This edition incorporates all amendments to Annex 14, Volume I, adopted by the Council prior to 6 March 1999 and supersedes on 4 November 1999 all previous editions of Annex 14, Volume I.

For information regarding the applicability of the Standards and Recommended Practices, see Chapter 1, 1.2 and Foreword.
AMENDMENTS

The issue of amendments is announced regularly in the ICAO Journal and in the monthly Supplement to the Catalogue of ICAO Publications and Audio-visual Training Aids, which holders of this publication should consult. The space below is provided to keep a record of such amendments.

RECORD OF AMENDMENTS AND CORRIGENDA

<table>
<thead>
<tr>
<th>AMENDMENTS</th>
<th>CORRIGENDA</th>
</tr>
</thead>
<tbody>
<tr>
<td>No.</td>
<td>Date applicable</td>
</tr>
<tr>
<td>1-3</td>
<td>Incorporated in this edition</td>
</tr>
<tr>
<td>4</td>
<td>1/11/01</td>
</tr>
</tbody>
</table>
## TABLE OF CONTENTS

<table>
<thead>
<tr>
<th>Chapter</th>
<th>Page</th>
<th>Page</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>General</td>
<td>1</td>
</tr>
<tr>
<td>1.1</td>
<td>Definitions</td>
<td>1</td>
</tr>
<tr>
<td>1.2</td>
<td>Applicability</td>
<td>5</td>
</tr>
<tr>
<td>1.3</td>
<td>Certification of aerodromes</td>
<td>6</td>
</tr>
<tr>
<td>1.4</td>
<td>Reference code</td>
<td>6</td>
</tr>
<tr>
<td>2</td>
<td>Aerodrome data</td>
<td>7</td>
</tr>
<tr>
<td>2.1</td>
<td>Aeronautical data</td>
<td>7</td>
</tr>
<tr>
<td>2.2</td>
<td>Aerodrome reference point</td>
<td>7</td>
</tr>
<tr>
<td>2.3</td>
<td>Aerodrome and runway elevations</td>
<td>8</td>
</tr>
<tr>
<td>2.4</td>
<td>Aerodrome reference temperature</td>
<td>8</td>
</tr>
<tr>
<td>2.5</td>
<td>Aerodrome dimensions and related information</td>
<td>8</td>
</tr>
<tr>
<td>2.6</td>
<td>Strength of pavements</td>
<td>8</td>
</tr>
<tr>
<td>2.7</td>
<td>Pre-flight altimeter check location</td>
<td>10</td>
</tr>
<tr>
<td>2.8</td>
<td>Declared distances</td>
<td>10</td>
</tr>
<tr>
<td>2.9</td>
<td>Condition of the movement area and related facilities</td>
<td>10</td>
</tr>
<tr>
<td>2.10</td>
<td>Disabled aircraft removal</td>
<td>11</td>
</tr>
<tr>
<td>2.11</td>
<td>Rescue and fire fighting</td>
<td>12</td>
</tr>
<tr>
<td>2.12</td>
<td>Visual approach slope indicator systems</td>
<td>12</td>
</tr>
<tr>
<td>2.13</td>
<td>Coordination between aeronautical information services and aerodrome authorities</td>
<td>12</td>
</tr>
<tr>
<td>3</td>
<td>Physical characteristics</td>
<td>14</td>
</tr>
<tr>
<td>3.1</td>
<td>Runways</td>
<td>14</td>
</tr>
<tr>
<td>3.2</td>
<td>Runway shoulders</td>
<td>17</td>
</tr>
<tr>
<td>3.3</td>
<td>Runway strips</td>
<td>18</td>
</tr>
<tr>
<td>3.4</td>
<td>Runway end safety areas</td>
<td>20</td>
</tr>
<tr>
<td>3.5</td>
<td>Clearways</td>
<td>20</td>
</tr>
<tr>
<td>3.6</td>
<td>Stopways</td>
<td>21</td>
</tr>
<tr>
<td>3.7</td>
<td>Radio altimeter operating area</td>
<td>22</td>
</tr>
<tr>
<td>3.8</td>
<td>Taxiways</td>
<td>22</td>
</tr>
<tr>
<td>3.9</td>
<td>Taxiway shoulders</td>
<td>26</td>
</tr>
<tr>
<td>3.10</td>
<td>Taxiway strips</td>
<td>26</td>
</tr>
<tr>
<td>3.11</td>
<td>Holding bays, runway-holding positions, intermediate holding positions and road-holding positions</td>
<td>27</td>
</tr>
<tr>
<td>3.12</td>
<td>Aprons</td>
<td>28</td>
</tr>
<tr>
<td>3.13</td>
<td>Isolated aircraft parking position</td>
<td>29</td>
</tr>
<tr>
<td>3.14</td>
<td>De-icing/anti-ice facilities</td>
<td>29</td>
</tr>
<tr>
<td>4</td>
<td>Obstacle restriction and removal</td>
<td>32</td>
</tr>
<tr>
<td>4.1</td>
<td>Obstacle limitation surfaces</td>
<td>32</td>
</tr>
<tr>
<td>4.2</td>
<td>Obstacle limitation requirements</td>
<td>36</td>
</tr>
<tr>
<td>4.3</td>
<td>Objects outside the obstacle limitation surfaces</td>
<td>40</td>
</tr>
<tr>
<td>4.4</td>
<td>Other objects</td>
<td>40</td>
</tr>
<tr>
<td>5</td>
<td>Visual aids for navigation</td>
<td>41</td>
</tr>
<tr>
<td>5.1</td>
<td>Indicators and signalling devices</td>
<td>41</td>
</tr>
<tr>
<td>5.1.1</td>
<td>Wind direction indicators</td>
<td>41</td>
</tr>
<tr>
<td>5.1.2</td>
<td>Landing direction indicator</td>
<td>41</td>
</tr>
<tr>
<td>5.1.3</td>
<td>Signalling lamp</td>
<td>41</td>
</tr>
<tr>
<td>5.1.4</td>
<td>Signal panels and signal area</td>
<td>42</td>
</tr>
<tr>
<td>5.2</td>
<td>Markings</td>
<td>42</td>
</tr>
<tr>
<td>5.2.1</td>
<td>General</td>
<td>42</td>
</tr>
<tr>
<td>5.2.2</td>
<td>Runway designation marking</td>
<td>43</td>
</tr>
<tr>
<td>5.2.3</td>
<td>Runway centre line marking</td>
<td>44</td>
</tr>
<tr>
<td>5.2.4</td>
<td>Threshold marking</td>
<td>45</td>
</tr>
<tr>
<td>5.2.5</td>
<td>Aiming point marking</td>
<td>46</td>
</tr>
<tr>
<td>5.2.6</td>
<td>Touchdown zone marking</td>
<td>47</td>
</tr>
<tr>
<td>5.2.7</td>
<td>Runway side stripe marking</td>
<td>49</td>
</tr>
<tr>
<td>5.2.8</td>
<td>Taxiway centre line marking</td>
<td>49</td>
</tr>
<tr>
<td>5.2.9</td>
<td>Runway-holding position marking</td>
<td>49</td>
</tr>
<tr>
<td>5.2.10</td>
<td>Intermediate holding position marking</td>
<td>51</td>
</tr>
<tr>
<td>5.2.11</td>
<td>VOR aerodrome check-point marking</td>
<td>52</td>
</tr>
<tr>
<td>5.2.12</td>
<td>Aircraft stand markings</td>
<td>52</td>
</tr>
<tr>
<td>5.2.13</td>
<td>Apron safety lines</td>
<td>53</td>
</tr>
<tr>
<td>5.2.14</td>
<td>Road-holding position marking</td>
<td>54</td>
</tr>
<tr>
<td>5.2.15</td>
<td>Mandatory instruction marking</td>
<td>54</td>
</tr>
<tr>
<td>5.2.16</td>
<td>Information marking</td>
<td>55</td>
</tr>
<tr>
<td>5.3</td>
<td>Lights</td>
<td>55</td>
</tr>
<tr>
<td>5.3.1</td>
<td>General</td>
<td>55</td>
</tr>
<tr>
<td>5.3.2</td>
<td>Emergency lighting</td>
<td>56</td>
</tr>
<tr>
<td>5.3.3</td>
<td>Aeronautical beacons</td>
<td>57</td>
</tr>
<tr>
<td>5.3.4</td>
<td>Approach lighting systems</td>
<td>58</td>
</tr>
<tr>
<td>5.3.5</td>
<td>Visual approach slope indicator systems</td>
<td>61</td>
</tr>
<tr>
<td>5.3.6</td>
<td>Circling guidance lights</td>
<td>73</td>
</tr>
<tr>
<td>5.3.7</td>
<td>Runway lead-in lighting systems</td>
<td>73</td>
</tr>
<tr>
<td>5.3.8</td>
<td>Runway threshold identification lights</td>
<td>74</td>
</tr>
<tr>
<td>5.3.9</td>
<td>Runway edge lights</td>
<td>74</td>
</tr>
<tr>
<td>5.3.10</td>
<td>Runway threshold and wing bar lights</td>
<td>74</td>
</tr>
<tr>
<td>5.3.11</td>
<td>Runway end lights</td>
<td>76</td>
</tr>
<tr>
<td>5.3.12</td>
<td>Runway centre line lights</td>
<td>77</td>
</tr>
<tr>
<td>5.3.13</td>
<td>Runway touchdown zone lights</td>
<td>77</td>
</tr>
<tr>
<td>5.3.14</td>
<td>Stopway lights</td>
<td>78</td>
</tr>
<tr>
<td>5.3.15</td>
<td>Taxiway centre line lights</td>
<td>78</td>
</tr>
<tr>
<td>5.3.16</td>
<td>Taxiway edge lights</td>
<td>82</td>
</tr>
<tr>
<td>5.3.17</td>
<td>Stop bars</td>
<td>83</td>
</tr>
<tr>
<td>5.3.18</td>
<td>Intermediate holding position lights</td>
<td>84</td>
</tr>
</tbody>
</table>
## Annex 14 — Aerodromes

<table>
<thead>
<tr>
<th>Page</th>
<th>Volume I</th>
</tr>
</thead>
<tbody>
<tr>
<td>5.3.19</td>
<td>De-icing/anti-icing facility exit lights</td>
</tr>
<tr>
<td>5.3.20</td>
<td>Runway guard lights</td>
</tr>
<tr>
<td>5.3.21</td>
<td>Apron floodlighting</td>
</tr>
<tr>
<td>5.3.22</td>
<td>Visual docking guidance system</td>
</tr>
<tr>
<td>5.3.23</td>
<td>Aircraft stand manoeuvring guidance lights</td>
</tr>
<tr>
<td>5.3.24</td>
<td>Road-holding position light</td>
</tr>
<tr>
<td>5.4</td>
<td>Signs</td>
</tr>
<tr>
<td>5.4.1</td>
<td>General</td>
</tr>
<tr>
<td>5.4.2</td>
<td>Mandatory instruction signs</td>
</tr>
<tr>
<td>5.4.3</td>
<td>Information signs</td>
</tr>
<tr>
<td>5.4.4</td>
<td>VOR aerodrome check-point sign</td>
</tr>
<tr>
<td>5.4.5</td>
<td>Aerodrome identification sign</td>
</tr>
<tr>
<td>5.4.6</td>
<td>Aircraft stand identification signs</td>
</tr>
<tr>
<td>5.4.7</td>
<td>Road-holding position sign</td>
</tr>
<tr>
<td>5.5</td>
<td>Markers</td>
</tr>
<tr>
<td>5.5.1</td>
<td>General</td>
</tr>
<tr>
<td>5.5.2</td>
<td>Unpaved runway edge markers</td>
</tr>
<tr>
<td>5.5.3</td>
<td>Stopway edge markers</td>
</tr>
<tr>
<td>5.5.4</td>
<td>Edge markers for snow-covered runways</td>
</tr>
<tr>
<td>5.5.5</td>
<td>Taxiway edge markers</td>
</tr>
<tr>
<td>5.5.6</td>
<td>Taxiway centre line markers</td>
</tr>
<tr>
<td>5.5.7</td>
<td>Unpaved taxiway edge markers</td>
</tr>
<tr>
<td>5.5.8</td>
<td>Boundary markers</td>
</tr>
<tr>
<td>6.1</td>
<td>Objects to be marked and/or lighted</td>
</tr>
<tr>
<td>6.2</td>
<td>Marking of objects</td>
</tr>
<tr>
<td>6.3</td>
<td>Lighting of objects</td>
</tr>
<tr>
<td>7.1</td>
<td>Closed runways and taxiways, or parts thereof</td>
</tr>
<tr>
<td>7.2</td>
<td>Non-load-bearing surfaces</td>
</tr>
<tr>
<td>7.3</td>
<td>Pre-threshold area</td>
</tr>
<tr>
<td>7.4</td>
<td>Unserviceable areas</td>
</tr>
<tr>
<td>8.1</td>
<td>Secondary power supply</td>
</tr>
<tr>
<td>8.2</td>
<td>Electrical systems</td>
</tr>
<tr>
<td>8.3</td>
<td>Monitoring</td>
</tr>
<tr>
<td>8.4</td>
<td>Fencing</td>
</tr>
<tr>
<td>8.5</td>
<td>Security lighting</td>
</tr>
<tr>
<td>8.6</td>
<td>Airport design</td>
</tr>
<tr>
<td>8.7</td>
<td>Siting and construction of equipment and installations on operational areas</td>
</tr>
<tr>
<td>8.8</td>
<td>Aerodrome vehicle operations</td>
</tr>
<tr>
<td>8.9</td>
<td>Surface movement guidance and control systems</td>
</tr>
<tr>
<td>9.1</td>
<td>Aerodrome emergency planning</td>
</tr>
<tr>
<td>9.2</td>
<td>Rescue and fire fighting</td>
</tr>
<tr>
<td>9.3</td>
<td>Disabled aircraft removal</td>
</tr>
<tr>
<td>APPENDIX 1</td>
<td>Colours for aeronautical ground lights, markings, signs and panels</td>
</tr>
<tr>
<td>1.</td>
<td>General</td>
</tr>
<tr>
<td>2.</td>
<td>Colours for aeronautical ground lights</td>
</tr>
<tr>
<td>3.</td>
<td>Colours for markings, signs and panels</td>
</tr>
<tr>
<td>APPENDIX 2</td>
<td>Aeronautical ground light characteristics</td>
</tr>
<tr>
<td>APPENDIX 3</td>
<td>Mandatory instruction markings and information markings</td>
</tr>
<tr>
<td>APPENDIX 4</td>
<td>Requirements concerning design of taxiing guidance signs</td>
</tr>
<tr>
<td>APPENDIX 5</td>
<td>Aeronautical data quality requirements</td>
</tr>
<tr>
<td>APPENDIX 6</td>
<td>Location of lights on obstacles</td>
</tr>
<tr>
<td>ATTACHMENT A</td>
<td>Guidance material supplementary to Annex 14, Volume I</td>
</tr>
<tr>
<td>1.</td>
<td>Number, siting and orientation of runways</td>
</tr>
<tr>
<td>2.</td>
<td>Clearways and stopways</td>
</tr>
<tr>
<td>3.</td>
<td>Calculation of declared distances</td>
</tr>
<tr>
<td>4.</td>
<td>Slopes on a runway</td>
</tr>
<tr>
<td>5.</td>
<td>Runway surface evenness</td>
</tr>
<tr>
<td>6.</td>
<td>Determining and expressing the friction characteristics of snow- and ice-covered paved surfaces</td>
</tr>
<tr>
<td>7.</td>
<td>Determination of friction characteristics of wet paved runways</td>
</tr>
<tr>
<td>8.</td>
<td>Strips</td>
</tr>
<tr>
<td>9.</td>
<td>Runway end safety areas</td>
</tr>
<tr>
<td>10.</td>
<td>Location of threshold</td>
</tr>
<tr>
<td>11.</td>
<td>Approach lighting systems</td>
</tr>
<tr>
<td>12.</td>
<td>Priority of installation of visual approach slope indicator systems</td>
</tr>
<tr>
<td>13.</td>
<td>Lighting of unserviceable areas</td>
</tr>
<tr>
<td>14.</td>
<td>Intensity control of approach and runway lights</td>
</tr>
<tr>
<td>15.</td>
<td>Signal area</td>
</tr>
<tr>
<td>16.</td>
<td>Rescue and fire fighting services</td>
</tr>
<tr>
<td>17.</td>
<td>Operators of vehicles</td>
</tr>
<tr>
<td>18.</td>
<td>The ACN-PCN method of reporting pavement strength</td>
</tr>
<tr>
<td>ATTACHMENT B</td>
<td>Obstacle limitation surfaces</td>
</tr>
</tbody>
</table>

LIMITED INDEX OF SIGNIFICANT SUBJECTS

INCLUDED IN ANNEX 14, VOLUME I | 207
### Abbreviations

<table>
<thead>
<tr>
<th>Abbreviation</th>
<th>Definition</th>
<th>Abbreviation</th>
<th>Definition</th>
</tr>
</thead>
<tbody>
<tr>
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<td>Aircraft classification number</td>
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<td>aprx</td>
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<td>NM</td>
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</tr>
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</tr>
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</tr>
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<td>Degree Celsius</td>
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</tr>
<tr>
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<td>Runway end safety area</td>
</tr>
<tr>
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</tr>
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</tr>
<tr>
<td>DME</td>
<td>Distance measuring equipment</td>
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<td>Visual meteorological conditions</td>
</tr>
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</tr>
<tr>
<td>ILS</td>
<td>Instrument landing system</td>
<td></td>
<td></td>
</tr>
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<td>IMC</td>
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<td></td>
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</tr>
<tr>
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<td>Degree Kelvin</td>
<td></td>
<td></td>
</tr>
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<td>Kilogram</td>
<td></td>
<td></td>
</tr>
<tr>
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<td>Kilometre</td>
<td></td>
<td></td>
</tr>
<tr>
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<td></td>
<td></td>
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<td>L</td>
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<td></td>
<td></td>
</tr>
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<td></td>
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<td>mmm</td>
<td>Minimum</td>
<td></td>
<td></td>
</tr>
<tr>
<td>MN</td>
<td>Meganewton</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

### Symbols

<table>
<thead>
<tr>
<th>Symbol</th>
<th>Definition</th>
</tr>
</thead>
<tbody>
<tr>
<td>°</td>
<td>Degree</td>
</tr>
<tr>
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</tr>
<tr>
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</tr>
<tr>
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</tr>
<tr>
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</tr>
<tr>
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</tr>
<tr>
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</tr>
<tr>
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<td>Plus or minus</td>
</tr>
</tbody>
</table>

### Manuals

(related to the specifications of this Annex)

- **Aerodrome Design Manual** (Doc 9157)
  - Part 1 — Runways
  - Part 2 — Taxiways, Aprons and Holding Bays
  - Part 3 — Pavements
  - Part 4 — Visual Aids
  - Part 5 — Electrical Systems
  - Part 6 — Frangibility (in preparation)

- **Airport Planning Manual** (Doc 9184)
  - Part 1 — Master Planning
  - Part 2 — Land Use and Environmental Control
  - Part 3 — Guidelines for Consultant/Construction Services

- **Airport Services Manual** (Doc 9137)
  - Part 1 — Rescue and Fire Fighting
  - Part 2 — Pavement Surface Conditions
  - Part 3 — Bird Control and Reduction
  - Part 4 — Fog Dispersal (withdrawn)
  - Part 5 — Removal of Disabled Aircraft

- **Part 6 — Control of Obstacles**
- **Part 7 — Airport Emergency Planning**
- **Part 8 — Airport Operational Services**
- **Part 9 — Airport Maintenance Practices**

- **Heliport Manual** (Doc 9261)
- **Stolport Manual** (Doc 9150)
- **Manual on the ICAO Bird Strike Information System (IBIS)** (Doc 9332)
- **Manual of Surface Movement Guidance and Control Systems (SMGCS)** (Doc 9476)
- **Manual on Certification of Aerodromes** (Doc 9774)
- **Human Factors Training Manual** (Doc 9683)
FOREWORD

Historical background

Standards and Recommended Practices for Aerodromes were first adopted by the Council on 29 May 1951 pursuant to the provisions of Article 37 of the Convention on International Civil Aviation (Chicago 1944) and designated as Annex 14 to the Convention. The Standards and Recommended Practices were based on recommendations of the Aerodromes, Air Routes and Ground Aids Division at its third session in September 1947 and at its fourth session in November 1949.

Table A shows the origin of subsequent amendments together with a list of the principal subjects involved and the dates on which the Annex and the amendments were adopted by the Council, when they became effective and when they became applicable.

Action by Contracting States

Notification of differences. The attention of Contracting States is drawn to the obligation imposed by Article 38 of the Convention by which Contracting States are required to notify the Organization of any differences between their national regulations and practices and the International Standards contained in this Annex and any amendments thereto. Contracting States are invited to extend such notification to any differences from the Recommended Practices contained in this Annex and any amendments thereto, when the notification of such differences is important for the safety of air navigation. Further, Contracting States are invited to keep the Organization currently informed of any differences which may subsequently occur, or of the withdrawal of any differences previously notified. A specified request for notification of differences will be sent to Contracting States immediately after the adoption of each amendment to this Annex.

The attention of States is also drawn to the provisions of Annex 15 related to the publication of differences between their national regulations and practices and the related ICAO Standards and Recommended Practices through the Aeronautical Information Service, in addition to the obligation of States under Article 38 of the Convention.

Promulgation of information. The establishment and withdrawal of and changes to facilities, services and procedures affecting aircraft operations provided in accordance with the Standards and Recommended Practices specified in this Annex should be notified and take effect in accordance with the provisions of Annex 15.

Status of Annex components

An Annex is made up of the following component parts, not all of which, however, are necessarily found in every Annex; they have the status indicated:

1. — Material comprising the Annex proper:

a) Standards and Recommended Practices adopted by the Council under the provisions of the Convention. They are defined as follows:

Standard: Any specification for physical characteristics, configuration, matériel, performance, personnel or procedure, the uniform application of which is recognized as necessary for the safety or regularity of international air navigation and to which Contracting States will conform in accordance with the Convention; in the event of impossibility of compliance, notification to the Council is compulsory under Article 38.

Recommended Practice: Any specification for physical characteristics, configuration, matériel, performance, personnel or procedure, the uniform application of which is recognized as desirable in the interest of safety, regularity or efficiency of international air navigation, and to which Contracting States will endeavour to conform in accordance with the Convention.

b) Appendices comprising material grouped separately for convenience but forming part of the Standards and Recommended Practices adopted by the Council.

c) Definitions of terms used in the Standards and Recommended Practices which are not self-explanatory in that they do not have accepted dictionary meanings. A definition does not have independent status but is an essential part of each Standard and Recommended Practice in which the term is used, since a change in the meaning of the term would affect the specification.

d) Tables and Figures which add to or illustrate a Standard or Recommended Practice and which are referred to therein, form part of the associated Standard or Recommended Practice and have the same status.
2.— Material approved by the Council for publication in association with the Standards and Recommended Practices:

a) Forewords comprising historical and explanatory material based on the action of the Council and including an explanation of the obligations of States with regard to the application of the Standards and Recommended Practices ensuing from the Convention and the Resolution of Adoption.

b) Introductions comprising explanatory material introduced at the beginning of parts, chapters or sections of the Annex to assist in the understanding of the application of the text.

c) Notes included in the text, where appropriate, to give factual information or references bearing on the Standards or Recommended Practices in question, but not constituting part of the Standards or Recommended Practices.

d) Attachments comprising material supplementary to the Standards and Recommended Practices, or included as a guide to their application.

Selection of language

This Annex has been adopted in five languages — English, Arabic, French, Russian and Spanish. Each Contracting State is requested to select one of those texts for the purpose of national implementation and for other effects provided for in the Convention, either through direct use or through translation into its own national language, and to notify the Organization accordingly.

Editorial practices

The following practice has been adhered to in order to indicate at a glance the status of each statement: Standards have been printed in light face roman; Recommended Practices have been printed in light face italics, the status being indicated by the prefix Recommendation; Notes have been printed in light face italics, the status being indicated by the prefix Note.

The following editorial practice has been followed in the writing of specifications: for Standards the operative verb “shall” is used, and for Recommended Practices the operative verb “should” is used.

The units of measurement used in this document are in accordance with the International System of Units (SI) as specified in Annex 5 to the Convention on International Civil Aviation. Where Annex 5 permits the use of non-SI alternative units these are shown in parentheses following the basic units. Where two sets of units are quoted it must not be assumed that the pairs of values are equal and interchangeable. It may, however, be inferred that an equivalent level of safety is achieved when either set of units is used exclusively.

Any reference to a portion of this document, which is identified by a number and/or title, includes all subdivisions of that portion.
Table A. Amendments to Annex 14, Volume I

<table>
<thead>
<tr>
<th>Amendment</th>
<th>Source(s)</th>
<th>Subject(s)</th>
<th>Adopted Effective Applicable</th>
</tr>
</thead>
<tbody>
<tr>
<td>1st Edition</td>
<td>Third and Fourth Sessions of the Aerodromes, Air Routes and Ground Aids Division</td>
<td>—</td>
<td>29 May 1951 1 November 1951 1 June 1952* 1 June 1954</td>
</tr>
<tr>
<td>1 to 6</td>
<td>Fifth Session of the Aerodromes, Air Routes and Ground Aids Division</td>
<td>Physical characteristics of runways, strips, clearways, stopways, taxiways and aprons; physical characteristics of channels, turning basins, taxi channels and mooring areas; approach areas; clearing and restriction of obstructions; obstruction marking; marking of unserviceable portions of the movement area; secondary power supply; aerodrome beacon; runway markings; stopway markers; approach, lead-in and runway lighting.</td>
<td>20 May 1953 1 September 1953 1 April 1954* 1 January 1955</td>
</tr>
<tr>
<td>7 to 13</td>
<td>Sixth Session of the Aerodromes, Air Routes and Ground Aids Division</td>
<td>Physical characteristics of runways, strips, taxiways and aprons; approach and take-off areas and surfaces; clearing and restriction of obstructions; obstruction markings; runway markings; stopway markers; taxiway markings; approach, runway and taxiway lighting; circling guidance lights; rescue and fire fighting services.</td>
<td>12 May 1958 1 September 1958 1 December 1958</td>
</tr>
<tr>
<td>14</td>
<td>Correspondence</td>
<td>Precision approach lighting system.</td>
<td>7 May 1959 1 October 1959 1 October 1959</td>
</tr>
<tr>
<td>15</td>
<td>Vertical Separation Panel</td>
<td>Pre-flight altimeter check-point.</td>
<td>15 May 1959 1 October 1959 1 October 1959</td>
</tr>
<tr>
<td>16</td>
<td>Correspondence</td>
<td>Extinguishing agents.</td>
<td>2 December 1960 2 December 1960 2 December 1960</td>
</tr>
<tr>
<td>17</td>
<td>Correspondence</td>
<td>Pre-flight altimeter check-point.</td>
<td>2 December 1960 2 December 1960 2 December 1960</td>
</tr>
<tr>
<td>18</td>
<td>First Meeting of the ANC Visual Aids Panel</td>
<td>VASIS</td>
<td>9 June 1961 1 October 1961 1 October 1961</td>
</tr>
<tr>
<td>19</td>
<td>Seventh Session of the Aerodromes, Air Routes and Ground Aids Division</td>
<td>Physical characteristics of runways, clearways, stopways, taxiways and aprons; take-off and approach areas; clearing and restriction of obstructions; obstruction markings; wind direction indicator; landing direction indicator; aerodrome beacon; runway markings; approach lighting system; runway alignment indicator; runway centre line; touchdown zone and taxiway lighting; rescue and fire fighting services.</td>
<td>23 March 1964 1 August 1964 1 November 1964</td>
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* Two applicability dates approved.
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<tr>
<th>Amendment</th>
<th>Source(s)</th>
<th>Subject(s)</th>
<th>Adopted</th>
<th>Effective</th>
<th>Applicable</th>
</tr>
</thead>
<tbody>
<tr>
<td>21</td>
<td>Fourth Air Navigation Conference and Fourth Meeting of the ANC Visual Aids Panel</td>
<td>Emergency lighting; threshold marking; fixed distance marking; approach light beacons; taxiway centre line lighting; secondary power supply; maintenance of aerodrome lighting and marking aids; monitoring of visual aids.</td>
<td>28 June 1967</td>
<td>28 October 1967</td>
<td>8 February 1968</td>
</tr>
<tr>
<td>22</td>
<td>Correspondence and ANC Visual Aids Panel</td>
<td>VOR aerodrome check-point marking and sign.</td>
<td>28 June 1968</td>
<td>28 October 1968</td>
<td>18 September 1969</td>
</tr>
<tr>
<td>23</td>
<td>Fifth Air Navigation Conference</td>
<td>Declared distances; strength of pavements; information on aerodrome conditions; reference code letters; runway length correction for slope; runway strips; taxiway clearances; holding bays; taxi-holding position markings; approach lighting systems; visual approach slope indicator systems; secondary power supply; rescue and fire fighting services; bird hazard reduction services.</td>
<td>23 January 1969</td>
<td>23 May 1969</td>
<td>18 September 1969</td>
</tr>
<tr>
<td>24</td>
<td>Fifth Meeting of the ANC Visual Aids Panel and First Meeting of the ANC Rescue and Fire Fighting Panel</td>
<td>Marking of unusable or unserviceable portions of the movement area; touchdown zone markings; category II holding position marking and sign; T-VASIS and AT-VASIS; runway edge lighting; exit taxiway centre line lighting; stop bars and clearance bars; emergency access roads; colour specifications for lights.</td>
<td>31 March 1971</td>
<td>6 September 1971</td>
<td>6 January 1972</td>
</tr>
<tr>
<td>26</td>
<td>Seventeenth Session of the Assembly and Middle East/South East Asia Regional Air Navigation Meeting</td>
<td>Aerodrome security; water rescue vehicles.</td>
<td>15 December 1971</td>
<td>15 April 1972</td>
<td>7 December 1971</td>
</tr>
<tr>
<td>27</td>
<td>ANC Visual Aids Panel and Middle East/South East Asia Regional Air Navigation Meeting</td>
<td>Runway centre line light colour coding; maintenance services.</td>
<td>20 March 1972</td>
<td>20 July 1972</td>
<td>7 December 1972</td>
</tr>
<tr>
<td>28</td>
<td>Secretariat and Sixth Meeting of the ANC Visual Aids Panel</td>
<td>Definition for snow on the ground; frangibility of light fixtures; runway centre line marking; taxiway centre line lighting; colour specifications for lights.</td>
<td>11 December 1972</td>
<td>11 April 1973</td>
<td>16 August 1973</td>
</tr>
<tr>
<td>29</td>
<td>Council action in pursuance of Assembly Resolutions A17-10 and A18-10</td>
<td>Aerodrome security.</td>
<td>7 December 1973</td>
<td>7 April 1974</td>
<td>23 May 1974</td>
</tr>
<tr>
<td>30</td>
<td>Eighth Air Navigation Conference and editorial revision of the Annex</td>
<td>Runway shoulders and strips; runway end safety areas; aerodrome reference temperature; clearways; holding bays; physical characteristics of taxiways; taxiway shoulders and strips; pavement strength; runway transverse slopes; runway braking action; obstacle limitation surfaces; category III runway lighting and marking; taxiway lighting; stop bars; rescue and fire fighting services; disabled aircraft removal.</td>
<td>3 February 1976</td>
<td>3 June 1976</td>
<td>30 December 1976</td>
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4/11/99 (x)
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<th>Amendment</th>
<th>Source(s)</th>
<th>Subject(s)</th>
<th>Adopted Effective Applicable</th>
</tr>
</thead>
<tbody>
<tr>
<td>31</td>
<td>Seventh Meeting of the ANC Visual Aids Panel and Fifth Meeting of the ANC Obstacle Clearance Panel</td>
<td>Obstacle limitation surfaces; light intensity control; inset light temperatures; taxiway centre line lights; apron floodlighting; visual docking guidance systems; signs; maintenance of visual aids.</td>
<td>13 December 1976 13 April 1977 6 October 1977</td>
</tr>
<tr>
<td>32</td>
<td>Correspondence and ANC Visual Aids Panel</td>
<td>Definition of frangibility; siting and construction of equipment and installations on operational areas; colour specifications for lights and markings.</td>
<td>14 December 1977 14 April 1978 10 August 1978</td>
</tr>
<tr>
<td>33</td>
<td>Correspondence and Secretariat</td>
<td>Reporting of information on visual approach slope indicator systems; runway, taxiway and taxi-holding position markings; approach lighting for displaced thresholds; runway edge and centre line lights; aerodrome emergency planning.</td>
<td>26 March 1979 26 July 1979 29 November 1979</td>
</tr>
<tr>
<td>34</td>
<td>Eighth Meeting of the ANC Visual Aids Panel</td>
<td>Apron markings; precision approach lighting systems; visual approach slope indicator systems; circling guidance lights; runway lead-in lighting systems; stop bars; visual docking guidance system; aircraft stand manoeuvring guidance lights; aircraft stand identification signs; marking and lighting of obstacles.</td>
<td>30 November 1979 30 March 1980 27 November 1980</td>
</tr>
<tr>
<td>35</td>
<td>Secretariat and the ANC Visual Aids Panel</td>
<td>Reporting of pavement strength; visual approach slope indicator systems; approach lighting systems; maintenance of lighting.</td>
<td>23 March 1981 23 July 1981 26 November 1981</td>
</tr>
<tr>
<td>36</td>
<td>Aerodromes, Air Routes and Ground Aids Divisional Meeting (1981), Ninth Meeting of the ANC Visual Aids Panel and Secretariat</td>
<td>Aerodrome reference code; runway friction characteristics; runway end safety areas; taxiway separation distances; rapid exit taxiways; taxiways on bridges; holding bays; obstacle limitation surfaces; PAPI; taxi-holding position marking and lights; runway centre line guidance; visual ground signals; rescue and fire fighting; apron management service; declared distances; ground servicing of aircraft; units of measure.</td>
<td>22 November 1982 23 March 1983 24 November 1983</td>
</tr>
<tr>
<td>37</td>
<td>Secretariat</td>
<td>Fuelling.</td>
<td>29 March 1983 29 July 1983 24 November 1983</td>
</tr>
<tr>
<td>38</td>
<td>Secretariat and the ANC Visual Aids Panel</td>
<td>Aerodrome data; APAPI; colour coding of exit taxiway centre line lights; stop bars; taxi-holding position lights; taxiway edge markers; markers for overhead wires; obstacle lighting of lighthouses; maintenance of taxiway centre line lights; surface marking colours.</td>
<td>17 March 1986 27 July 1986 20 November 1986</td>
</tr>
<tr>
<td></td>
<td></td>
<td>2. Take-off runways; aerodrome reference code; reporting of pavement strength; runway friction characteristics; conditions of movement area; separation of parallel runways; taxiway minimum separation distances; taxi-holding position marking; installation tolerances for PAPI; obstacle protection surface; stop bars; signs; taxiway centre line markers; aerodrome security; surface movement guidance and control; aerodrome emergency planning; rescue and fire fighting; maintenance; runway pavement overlay; bird hazard reduction; apron management service; colours for transilluminated signs and panels; aeronautical ground light characteristics.</td>
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</tr>
<tr>
<td>Amendment</td>
<td>Source(s)</td>
<td>Subject(s)</td>
<td>Adopted</td>
</tr>
<tr>
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<td>----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------</td>
<td>-----------</td>
</tr>
<tr>
<td>1</td>
<td>Twelfth Meeting of the ANC Visual Aids Panel and Secretariat</td>
<td>Definitions of frangible object, precision approach runways, road, road-holding position, runway guard lights, taxi-holding position; standard geodetic reference system; radio altimeter operating area, minimum distance between parallel runways; frangibility; runway and taxiway markings, aeronautical beacons, lighting aids for MLS operations, deletion of specifications on VASIS (AVASIS) and 3-BAR VASIS (3-BAR AVASIS), stop bars, runway guard lights, visual docking guidance system, taxiing guidance signs; obstacle lighting; visual aids for denoting restricted use areas; secondary power supply, electrical systems, monitoring, airport design, surface movement guidance and control systems; rescue and fire fighting, maintenance of visual aids; aeronautical ground light characteristics; form and proportions of information marking; design of taxiing guidance signs; friction characteristics of wet runways.</td>
<td>13 March 1995</td>
</tr>
<tr>
<td>3</td>
<td>Thirteenth Meeting of the ANC Visual Aids Panel and Secretariat</td>
<td>Definitions of aerodrome traffic density, de/anti-icing facility, de/anti-icing pad, holdover time, Human Factors principles, human performance, intermediate holding position, runway-holding position, signs, switch-over time; new aerodrome reference code letter F in Table1-1; runways, taxiways and taxiway minimum separation distances related to code letter F aeroplane operations, sight distance, runway strips, runway end safety areas, clearways, stopways, taxiways on bridges, holding bays, runway-holding positions, intermediate holding positions and road holding positions, de/anti-icing facilities; obstacle-free zone width for code letter F; runway-holding position marking, intermediate holding position marking, mandatory instruction marking, marking of de/anti-icing facilities, approach lighting systems, runway and taxiway centre line lights, stop bars, intermediate holding position lights, lighting of de/anti-icing facilities, runway guard lights, variable message signs, intersection take-off signs; visual aids for denoting obstacles; secondary power supply switch-over time, security measures in airport design, frangibility of non-visual aids on operational areas; Human Factors principles applied to aerodrome emergency planning, rescue and fire fighting, and maintenance, system of preventive maintenance for precision approach runways, categories II and III; colour measurement of aeronautical ground lights; isocandela diagrams for high-intensity taxiway centre line lights and runway guard lights; measurement of the average luminance of a sign, Table 4.1 of Appendix 4; Appendix 6.</td>
<td>5 March 1999</td>
</tr>
<tr>
<td>4</td>
<td>Secretariat and the Twelfth Meeting of the ANC Obstacle Clearance Panel</td>
<td>Definitions of aerodrome certificate, certified aerodrome, safety management system; certification of aerodromes; obstacle limitation surfaces; specifications concerning aerodrome emergency planning; rescue and fire fighting.</td>
<td>12 March 2001</td>
</tr>
</tbody>
</table>
INTernational Standards
And Recommended Practices

Chapter 1. General

Introductory Note.— This Annex contains Standards and Recommended Practices (specifications) that prescribe the physical characteristics and obstacle limitation surfaces to be provided for at aerodromes, and certain facilities and technical services normally provided at an aerodrome. It is not intended that these specifications limit or regulate the operation of an aircraft.

To a great extent, the specifications for individual facilities detailed in Annex 14, Volume I, have been interrelated by a reference code system, described in this chapter, and by the designation of the type of runway for which they are to be provided, as specified in the definitions. This not only simplifies the reading of Volume I of this Annex, but in most cases, provides for efficiently proportioned aerodromes when the specifications are followed.

This document sets forth the minimum aerodrome specifications for aircraft which have the characteristics of those which are currently operating or for similar aircraft that are planned for introduction. Accordingly, any additional safeguards that might be considered appropriate to provide for more demanding aircraft are not taken into account. Such matters are left to appropriate authorities to evaluate and take into account as necessary for each particular aerodrome. Guidance on some possible effects of future aircraft on these specifications is given in the Aerodrome Design Manual, Part 2.

It is to be noted that the specifications for precision approach runways categories II and III are only applicable to runways intended to be used by aeroplanes in code numbers 3 and 4.

Annex 14, Volume I, does not include specifications relating to the overall planning of aerodromes (such as separation between adjacent aerodromes or capacity of individual aerodromes) or to economic and other non-technical factors that need to be considered in the development of an aerodrome. Information on these subjects is included in the Airport Planning Manual, Part 1.

Aviation security is an integral part of aerodrome planning and operations. Annex 14, Volume I, contains several specifications aimed at enhancing the level of security at aerodromes. Specifications on other facilities related to security are given in Annex 17 and detailed guidance on the subject is contained in the ICAO Security Manual.

1.1 Definitions

When the following terms are used in this Annex they have the following meanings:

Accuracy. A degree of conformance between the estimated or measured value and the true value.

Note.— For measured positional data, the accuracy is normally expressed in terms of a distance from a stated position within which there is a defined confidence of the true position falling.

Aerodrome. A defined area on land or water (including any buildings, installations, and equipment) intended to be used either wholly or in part for the arrival, departure and surface movement of aircraft.

Aerodrome beacon. Aeronautical beacon used to indicate the location of an aerodrome from the air.

Aerodrome certificate. A certificate issued by the appropriate authority under applicable regulations for the operation of an aerodrome.

Aerodrome elevation. The elevation of the highest point of the landing area.

Aerodrome identification sign. A sign placed on an aerodrome to aid in identifying the aerodrome from the air.

Aerodrome reference point. The designated geographical location of an aerodrome.

Aerodrome traffic density.

a) Light. Where the number of movements in the mean busy hour is not greater than 15 per runway or typically less than 20 total aerodrome movements.

b) Medium. Where the number of movements in the mean busy hour is of the order of 16 to 25 per runway or typically between 20 to 35 total aerodrome movements.

c) Heavy. Where the number of movements in the mean busy hour is of the order of 26 or more per runway or typically more than 35 total aerodrome movements.
Note 1.— The number of movements in the mean busy hour is the arithmetic mean over the year of the number of movements in the daily busiest hour.

Note 2.— Either a take-off or a landing constitutes a movement.

Aeronautical beacon. An aeronautical ground light visible at all azimuths, either continuously or intermittently, to designate a particular point on the surface of the earth.

Aeronautical ground light. Any light specially provided as an aid to air navigation, other than a light displayed on an aircraft.

Aeroplane reference field length. The minimum field length required for take-off at maximum certificated take-off mass, sea level, standard atmospheric conditions, still air and zero runway slope, as shown in the appropriate aeroplane flight manual prescribed by the certificating authority or equivalent data from the aeroplane manufacturer. Field length means balanced field length for aeroplanes, if applicable, or take-off distance in other cases.

Note.— Attachment A, Section 2 provides information on the concept of balanced field length and the Airworthiness Manual (Doc 9760) contains detailed guidance on matters related to take-off distance.

Aircraft classification number (ACN). A number expressing the relative effect of an aircraft on a pavement for a specified standard subgrade category.

Note.— The aircraft classification number is calculated with respect to the center of gravity (CG) position which yields the critical loading on the critical gear. Normally the aftmost CG position appropriate to the maximum gross apron (ramp) mass is used to calculate the ACN. In exceptional cases the forwardmost CG position may result in the nose gear loading being more critical.

Aircraft stand. A designated area on an apron intended to be used for parking an aircraft.

Apron. A defined area, on a land aerodrome, intended to accommodate aircraft for purposes of loading or unloading passengers, mail or cargo, fuelling, parking or maintenance.

Apron management service. A service provided to regulate the activities and the movement of aircraft and vehicles on an apron.

Barrette. Three or more aeronautical ground lights closely spaced in a transverse line so that from a distance they appear as a short bar of light.

Capacitor discharge light. A lamp in which high-intensity flashes of extremely short duration are produced by the discharge of electricity at high voltage through a gas enclosed in a tube.

Certified aerodrome. An aerodrome whose operator has been granted an aerodrome certificate.

Clearway. A defined rectangular area on the ground or water under the control of the appropriate authority, selected or prepared as a suitable area over which an aeroplane may make a portion of its initial climb to a specified height.

Cyclic redundancy check (CRC). A mathematical algorithm applied to the digital expression of data that provides a level of assurance against loss or alteration of data.

Data quality. A degree or level of confidence that the data provided meet the requirements of the data user in terms of accuracy, resolution and integrity.

De-icing/anti-icing facility. A facility where frost, ice or snow is removed (de-icing) from the aeroplane to provide clean surfaces, and/or where clean surfaces of the aeroplane receive protection (anti-icing) against the formation of frost or ice and accumulation of snow or slush for a limited period of time.

Note.— Further guidance is given in the Manual of Aircraft Ground De-icing/Anti-icing Operations (Doc 9640).

De-icing/anti-icing pad. An area comprising an inner area for the parking of an aeroplane to receive de-icing/anti-icing treatment and an outer area for the manoeuvring of two or more mobile de-icing/anti-icing equipment.

Declared distances.

a) Take-off run available (TORA). The length of runway declared available and suitable for the ground run of an aeroplane taking off.

b) Take-off distance available (TODA). The length of the take-off run available plus the length of the clearway, if provided.

c) Accelerate-stop distance available (ASDA). The length of the take-off run available plus the length of the stopway, if provided.

d) Landing distance available (LDA). The length of runway which is declared available and suitable for the ground run of an aeroplane landing.

Dependent parallel approaches. Simultaneous approaches to parallel or near-parallel instrument runways where radar separation minima between aircraft on adjacent extended runway centre lines are prescribed.

Displaced threshold. A threshold not located at the extremity of a runway.
**Effective intensity.** The effective intensity of a flashing light is equal to the intensity of a fixed light of the same colour which will produce the same visual range under identical conditions of observation.

**Ellipsoid height (Geodetic height).** The height related to the reference ellipsoid, measured along the ellipsoidal outer normal through the point in question.

**Fixed light.** A light having constant luminous intensity when observed from a fixed point.

**Frangible object.** An object of low mass designed to break, distort or yield on impact so as to present the minimum hazard to aircraft.

*Note.— Guidance on design for frangibility is contained in the Aerodrome Design Manual, Part 6 (in preparation).*

**Geodetic datum.** A minimum set of parameters required to define location and orientation of the local reference system with respect to the global reference system/frame.

**Geoid.** The equipotential surface in the gravity field of the Earth which coincides with the undisturbed mean sea level (MSL) extended continuously through the continents.

*Note.— The geoid is irregular in shape because of local gravitational disturbances (wind tides, salinity, current, etc.) and the direction of gravity is perpendicular to the geoid at every point.*

**Geoid undulation.** The distance of the geoid above (positive) or below (negative) the mathematical reference ellipsoid.

*Note.— In respect to the World Geodetic System — 1984 (WGS-84) defined ellipsoid, the difference between the WGS-84 ellipsoidal height and orthometric height represents WGS-84 geoid undulation.*

**Hazard beacon.** An aeronautical beacon used to designate a danger to air navigation.

**Heliport.** An aerodrome or a defined area on a structure intended to be used wholly or in part for the arrival, departure and surface movement of helicopters.

**Holding bay.** A defined area where aircraft can be held, or bypassed, to facilitate efficient surface movement of aircraft.

**Holdover time.** The estimated time the anti-icing fluid (treatment) will prevent the formation of ice and frost and the accumulation of snow on the protected (treated) surfaces of an aeroplane.

**Human Factors principles.** Principles which apply to aeronautical design, certification, training, operations and maintenance and which seek safe interface between the human and other system components by proper consideration to human performance.

**Human performance.** Human capabilities and limitations which have an impact on the safety and efficiency of aeronautical operations.

**Identification beacon.** An aeronautical beacon emitting a coded signal by means of which a particular point of reference can be identified.

**Independent parallel approaches.** Simultaneous approaches to parallel or near-parallel instrument runways where radar separation minima between aircraft on adjacent extended runway centre lines are not prescribed.

**Independent parallel departures.** Simultaneous departures from parallel or near-parallel instrument runways.

**Instrument runway.** One of the following types of runways intended for the operation of aircraft using instrument approach procedures:

a) **Non-precision approach runway.** An instrument runway served by visual aids and a non-visual aid providing at least directional guidance adequate for a straight-in approach.

b) **Precision approach runway, category I.** An instrument runway served by ILS and/or MLS and visual aids intended for operations with a decision height not lower than 60 m (200 ft) and either a visibility not less than 800 m or a runway visual range not less than 550 m.

c) **Precision approach runway, category II.** An instrument runway served by ILS and/or MLS and visual aids intended for operations with a decision height lower than 60 m (200 ft) but not lower than 30 m (100 ft) and a runway visual range not less than 350 m.

d) **Precision approach runway, category III.** An instrument runway served by ILS and/or MLS to and along the surface of the runway and:

A — intended for operations with a decision height lower than 30 m (100 ft), or no decision height and a runway visual range not less than 200 m.

B — intended for operations with a decision height lower than 15 m (50 ft), or no decision height and a runway visual range less than 200 m but not less than 50 m.

C — intended for operations with no decision height and no runway visual range limitations.

*Note 1.— See Annex 10, Volume I, Part I, for related ILS and/or MLS specifications.*
Note 2.— Visual aids need not necessarily be matched to the scale of non-visual aids provided. The criterion for the selection of visual aids is the conditions in which operations are intended to be conducted.

Integrity (aeronautical data). A degree of assurance that an aeronautical data and its value has not been lost nor altered since the data origination or authorized amendment.

Intermediate holding position. A designated position intended for traffic control at which taxiing aircraft and vehicles shall stop and hold until further cleared to proceed, when so instructed by the aerodrome control tower.

Landing area. That part of a movement area intended for the landing or take-off of aircraft.

Landing direction indicator. A device to indicate visually the direction currently designated for landing and for take-off.

Lighting system reliability. The probability that the complete installation operates within the specified tolerances and that the system is operationally usable.

Manoeuvring area. That part of an aerodrome to be used for the take-off, landing and taxiing of aircraft, excluding aprons.

Marker. An object displayed above ground level in order to indicate an obstacle or delineate a boundary.

Marking. A symbol or group of symbols displayed on the surface of the movement area in order to convey aeronautical information.

Movement area. That part of an aerodrome to be used for the take-off, landing and taxiing of aircraft, consisting of the manoeuvring area and the apron(s).

Near-parallel runways. Non-intersecting runways whose extended centre lines have an angle of convergence/divergence of 15 degrees or less.

Non-instrument runway. A runway intended for the operation of aircraft using visual approach procedures.

Obstacle. All fixed (whether temporary or permanent) and mobile objects, or parts thereof, that are located on an area intended for the surface movement of aircraft or that extend above a defined surface intended to protect aircraft in flight.

Obstacle free zone (OFZ). The airspace above the inner approach surface, inner transitional surfaces, and balked landing surface and that portion of the strip bounded by these surfaces, which is not penetrated by any fixed obstacle other than a low-mass and frangibly mounted one required for air navigation purposes.

Orthometric height. Height of a point related to the geoid, generally presented as an MSL elevation.

Pavement classification number (PCN). A number expressing the bearing strength of a pavement for unrestricted operations.

Precision approach runway, see Instrument runway.

Primary runway(s). Runway(s) used in preference to others whenever conditions permit.

Road. An established surface route on the movement area meant for the exclusive use of vehicles.

Road-holding position. A designated position at which vehicles may be required to hold.

Runway. A defined rectangular area on a land aerodrome prepared for the landing and take-off of aircraft.

Runway end safety area (RESA). An area symmetrical about the extended runway centre line and adjacent to the end of the strip primarily intended to reduce the risk of damage to an aeroplane undershooting or overrunning the runway.

Runway guard lights. A light system intended to caution pilots or vehicle drivers that they are about to enter an active runway.

Runway-holding position. A designated position intended to protect a runway, an obstacle limitation surface, or an ILS/MLS critical/sensitive area at which taxiing aircraft and vehicles shall stop and hold, unless otherwise authorized by the aerodrome control tower.

Runway strip. A defined area including the runway and stopway, if provided, intended:

a) to reduce the risk of damage to aircraft running off a runway; and

b) to protect aircraft flying over it during take-off or landing operations.

Runway visual range (RVR). The range over which the pilot of an aircraft on the centre line of a runway can see the runway surface markings or the lights delineating the runway or identifying its centre line.

Safety management system. A system for the management of safety at aerodromes, including the organizational structure, responsibilities, procedures, processes and provisions for the implementation of aerodrome safety policies by an aerodrome operator, which provides for control of safety at, and the safe use of, the aerodrome.

Segregated parallel operations. Simultaneous operations on parallel or near-parallel instrument runways in which one
runway is used exclusively for approaches and the other runway is used exclusively for departures.

**Shoulder.** An area adjacent to the edge of a pavement so prepared as to provide a transition between the pavement and the adjacent surface.

**Sign.**

a) **Fixed message sign.** A sign presenting only one message.

b) **Variable message sign.** A sign capable of presenting several pre-determined messages or no message, as applicable.

**Signal area.** An area on an aerodrome used for the display of ground signals.

**Slush.** Water-saturated snow which with a heel-and-toe slap-down motion against the ground will be displaced with a splatter; specific gravity: 0.5 up to 0.8.

Note.— Combinations of ice, snow and/or standing water may, especially when rain, rain and snow, or snow is falling, produce substances with specific gravities in excess of 0.8. These substances, due to their high water/ice content, will have a transparent rather than a cloudy appearance and, at the higher specific gravities, will be readily distinguishable from slush.

**Snow (on the ground).**

a) **Dry snow.** Snow which can be blown if loose or, if compacted by hand, will fall apart again upon release; specific gravity: up to but not including 0.35.

b) **Wet snow.** Snow which, if compacted by hand, will stick together and tend to or form a snowball; specific gravity: 0.35 up to but not including 0.5.

c) **Compacted snow.** Snow which has been compressed into a solid mass that resists further compression and will hold together or break up into lumps if picked up; specific gravity: 0.5 and over.

**Station declination.** An alignment variation between the zero degree radial of a VOR and true north, determined at the time the VOR station is calibrated.

**Stopway.** A defined rectangular area on the ground at the end of take-off run available prepared as a suitable area in which an aircraft can be stopped in the case of an abandoned take-off.

**Switch-over time (light).** The time required for the actual intensity of a light measured in a given direction to fall from 50 per cent and recover to 50 per cent during a power supply changeover, when the light is being operated at intensities of 25 per cent or above.

**Take-off runway.** A runway intended for take-off only.

**Taxiway.** A defined path on a land aerodrome established for the taxiing of aircraft and intended to provide a link between one part of the aerodrome and another, including:

a) **Aircraft stand taxilane.** A portion of an apron designated as a taxiway and intended to provide access to aircraft stands only.

b) **Apron taxiway.** A portion of a taxiway system located on an apron and intended to provide a through taxi route across the apron.

c) **Rapid exit taxiway.** A taxiway connected to a runway at an acute angle and designed to allow landing aeroplanes to turn off at higher speeds than are achieved on other exit taxiways thereby minimizing runway occupancy times.

**Taxiway intersection.** A junction of two or more taxiways.

**Taxiway strip.** An area including a taxiway intended to protect an aircraft operating on the taxiway and to reduce the risk of damage to an aircraft accidentally running off the taxiway.

**Threshold.** The beginning of that portion of the runway usable for landing.

**Touchdown zone.** The portion of a runway, beyond the threshold, where it is intended landing aeroplanes first contact the runway.

**Usability factor.** The percentage of time during which the use of a runway or system of runways is not restricted because of the cross-wind component.

Note.— Cross-wind component means the surface wind component at right angles to the runway centre line.

**1.2 Applicability**

1.2.1 The interpretation of some of the specifications in the Annex expressly requires the exercising of discretion, the taking of a decision or the performance of a function by the appropriate authority. In other specifications, the expression appropriate authority does not actually appear although its inclusion is implied. In both cases, the responsibility for whatever determination or action is necessary shall rest with the State having jurisdiction over the aerodrome.

1.2.2 The specifications, unless otherwise indicated in a particular context, shall apply to all aerodromes open to public use in accordance with the requirements of Article 15 of the Convention. The specifications of Annex 14, Volume I, Chapter 3 shall apply only to land aerodromes. The specifications in this volume shall apply, where appropriate, to heliports but shall not apply to stolports.
Note.— Although there are at present no specifications relating to stolports, it is intended that specifications for these aerodromes will be included as they are developed. In the interim, guidance material on stolports is given in the Stolport Manual.

1.2.3 Wherever a colour is referred to in this Annex, the specifications for that colour given in Appendix 1 shall apply.

1.3 Certification of Aerodromes

Note.— The intent of these specifications is to ensure the establishment of a regulatory regime so that compliance with the specifications in this Annex can be effectively enforced. It is recognized that the methods of ownership, operation and surveillance of aerodromes differ among States. The most effective and transparent means of ensuring compliance with applicable specifications is the availability of a separate safety oversight entity and a well-defined safety oversight mechanism with support of appropriate legislation to be able to carry out the function of safety regulation of aerodromes.

1.3.1 As of 27 November 2003, States shall certify aerodromes used for international operations in accordance with the specifications contained in this Annex as well as other relevant ICAO specifications through an appropriate regulatory framework.

1.3.2 Recommendation.— States should certify aerodromes open to public use in accordance with these specifications as well as other relevant ICAO specifications through an appropriate regulatory framework.

1.3.3 The regulatory framework shall include the establishment of criteria for the certification of aerodromes.

Note.— Guidance on a regulatory framework is given in the Manual on Certification of Aerodromes.

1.3.4 Recommendation.— A certified aerodrome should have in operation a safety management system.

Note.— The intent of a safety management system is to have in place an organized and orderly approach in the management of aerodrome safety by the aerodrome operator. Guidance on an aerodrome safety management system is given in the Manual on Certification of Aerodromes.

1.3.5 Recommendation.— As part of the certification process, States should ensure that an aerodrome manual which will include all pertinent information on the aerodrome site, facilities, services, equipment, operating procedures, organization and management including a safety management system, is submitted by the applicant for approval/acceptance prior to granting the aerodrome certificate.

1.3.6 As of 24 November 2005, a certified aerodrome shall have in operation a safety management system.

1.4 Reference code

Introductory Note.— The intent of the reference code is to provide a simple method for interrelating the numerous specifications concerning the characteristics of aerodromes so as to provide a series of aerodrome facilities that are suitable for the aeroplanes that are intended to operate at the aerodrome. The code is not intended to be used for determining runway length or pavement strength requirements. The code is composed of two elements which are related to the aeroplane performance characteristics and dimensions. Element 1 is a number based on the aeroplane reference field length and element 2 is a letter based on the aeroplane wing span and outer main gear wheel span. A particular specification is related to the more appropriate of the two elements of the code or to an appropriate combination of the two code elements. The code letter or number within an element selected for design purposes is related to the critical aeroplane characteristics for which the facility is provided. When applying Annex 14, Volume I, the aeroplanes which the aerodrome is intended to serve are first identified and then the two elements of the code.

1.4.1 An aerodrome reference code — code number and letter — which is selected for aerodrome planning purposes shall be determined in accordance with the characteristics of the aeroplane for which an aerodrome facility is intended.

1.4.2 The aerodrome reference code numbers and letters shall have the meanings assigned to them in Table 1-1.

1.4.3 The code number for element 1 shall be determined from Table 1-1, column 1, selecting the code number corresponding to the highest value of the aeroplane reference field lengths of the aeroplanes for which the runway is intended.

Note.— The determination of the aeroplane reference field length is solely for the selection of a code number and is not intended to influence the actual runway length provided.

1.4.4 The code letter for element 2 shall be determined from Table 1-1, column 3, by selecting the code letter which corresponds to the greatest wing span, or the greatest outer main gear wheel span, whichever gives the more demanding code letter of the aeroplanes for which the facility is intended.

Note.— Guidance to assist the appropriate authority in determining the aerodrome reference code is given in the Aerodrome Design Manual, Parts 1 and 2.
Table 1-1. Aerodrome reference code
(see 1.4.2 to 1.4.4)

<table>
<thead>
<tr>
<th>Code number</th>
<th>Code element 1</th>
<th>Code letter</th>
<th>Wing span</th>
<th>Outer main gear wheel span a</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>Less than 800 m</td>
<td>A</td>
<td>Up to but not including 15 m</td>
<td>Up to but not including 4.5 m</td>
</tr>
<tr>
<td>2</td>
<td>800 m up to but not including 1 200 m</td>
<td>B</td>
<td>15 m up to but not including 24 m</td>
<td>4.5 m up to but not including 6 m</td>
</tr>
<tr>
<td>3</td>
<td>1 200 m up to but not including 1 800 m</td>
<td>C</td>
<td>24 m up to but not including 36 m</td>
<td>6 m up to but not including 9 m</td>
</tr>
<tr>
<td>4</td>
<td>1 800 m and over</td>
<td>D</td>
<td>36 m up to but not including 52 m</td>
<td>9 m up to but not including 14 m</td>
</tr>
<tr>
<td></td>
<td></td>
<td>E</td>
<td>52 m up to but not including 65 m</td>
<td>9 m up to but not including 14 m</td>
</tr>
<tr>
<td></td>
<td></td>
<td>F</td>
<td>65 m up to but not including 80 m</td>
<td>14 m up to but not including 16 m</td>
</tr>
</tbody>
</table>

a. Distance between the outside edges of the main gear wheels.

**Note.**— Guidance on planning for aeroplanes with wing spans greater than 80 m is given in the Aerodrome Design Manual, Parts 1 and 2.
CHAPTER 2. AERODROME DATA

2.1 Aeronautical data

2.1.1 Determination and reporting of aerodrome related aeronautical data shall be in accordance with the accuracy and integrity requirements set forth in Tables 1 to 5 contained in Appendix 5 while taking into account the established quality system procedures. Accuracy requirements for aeronautical data are based upon a 95 per cent confidence level and in that respect, three types of positional data shall be identified: surveyed points (e.g. runway threshold), calculated points (mathematical calculations from the known surveyed points of points in space, fixes) and declared points (e.g. flight information region boundary points).

Note.— Specifications governing the quality system are given in Annex 15, Chapter 3.

2.1.2 Contracting States shall ensure that integrity of aeronautical data is maintained throughout the data process from survey/origin to the next intended user. Aeronautical data integrity requirements shall be based upon the potential risk resulting from the corruption of data and upon the use to which the data item is put. Consequently, the following classification and data integrity level shall apply:

a) critical data, integrity level \(1 \times 10^{-8}\): there is a high probability when using corrupted critical data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe;

b) essential data, integrity level \(1 \times 10^{-5}\): there is a low probability when using corrupted essential data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe; and

c) routine data, integrity level \(1 \times 10^{-3}\): there is a very low probability when using corrupted routine data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe.

2.1.3 Protection of electronic aeronautical data while stored or in transit shall be totally monitored by the cyclic redundancy check (CRC). To achieve protection of the integrity level of critical and essential aeronautical data as classified in 2.1.2 above, a 32 or 24 bit CRC algorithm shall apply respectively.

2.1.4 Recommendation. — To achieve protection of the integrity level of routine aeronautical data as classified in 2.1.2 above, a 16 bit CRC algorithm should apply.

Note.— Guidance material on the aeronautical data quality requirements (accuracy, resolution, integrity, protection and traceability) is contained in the World Geodetic System — 1984 (WGS-84) Manual (Doc 9674). Supporting material in respect of the provisions of Appendix 5 related to accuracy and integrity of aeronautical data, is contained in RTCA Document DO-201A and European Organization for Civil Aviation Equipment (EUROCAE) Document ED-77, entitled Industry Requirements for Aeronautical Information.

2.1.5 Geographical coordinates indicating latitude and longitude shall be determined and reported to the aeronautical information services authority in terms of the World Geodetic System — 1984 (WGS-84) geodetic reference datum, identifying those geographical coordinates which have been transformed into WGS-84 coordinates by mathematical means and whose accuracy of original field work does not meet the requirements in Appendix 5, Table 1.

2.1.6 The order of accuracy of the field work shall be such that the resulting operational navigation data for the phases of flight will be within the maximum deviations, with respect to an appropriate reference frame, as indicated in tables contained in Appendix 5.

2.1.7 In addition to the elevation (referenced to mean sea level) of the specific surveyed ground positions at aerodromes, geoid undulation (referenced to the WGS-84 ellipsoid) for those positions as indicated in Appendix 5, shall be determined and reported to the aeronautical information services authority.

Note 1.— An appropriate reference frame is that which enables WGS-84 to be realized on a given aerodrome and with respect to which all coordinate data are related.

Note 2.— Specifications governing the publication of WGS-84 coordinates are given in Annex 4, Chapter 2 and Annex 15, Chapter 3.

2.2 Aerodrome reference point

2.2.1 An aerodrome reference point shall be established for an aerodrome.

2.2.2 The aerodrome reference point shall be located near the initial or planned geometric centre of the aerodrome and shall normally remain where first established.

2.2.3 The position of the aerodrome reference point shall be measured and reported to the aeronautical information services authority in degrees, minutes and seconds.
2.3 Aerodrome and runway elevations

2.3.1 The aerodrome elevation and geoid undulation at the aerodrome elevation position shall be measured to the accuracy of one-half metre or foot and reported to the aeronautical information services authority.

2.3.2 For an aerodrome used by international civil aviation for non-precision approaches, the elevation and geoid undulation of each threshold, the elevation of the runway end and any significant high and low intermediate points along the runway shall be measured to the accuracy of one-half metre or foot and reported to the aeronautical information services authority.

2.3.3 For precision approach runway, the elevation and geoid undulation of the threshold, the elevation of the runway end and the highest elevation of the touchdown zone shall be measured to the accuracy of one-quarter metre or foot and reported to the aeronautical information services authority.

Note.— Geoid undulation must be measured in accordance with the appropriate system of coordinates.

2.4 Aerodrome reference temperature

2.4.1 An aerodrome reference temperature shall be determined for an aerodrome in degrees Celsius.

2.4.2 Recommendation.— The aerodrome reference temperature should be the monthly mean of the daily maximum temperatures for the hottest month of the year (the hottest month being that which has the highest monthly mean temperature). This temperature should be averaged over a period of years.

2.5 Aerodrome dimensions and related information

2.5.1 The following data shall be measured or described, as appropriate, for each facility provided on an aerodrome:

a) runway — true bearing to one-hundredth of a degree, designation number, length, width, displaced threshold location to the nearest metre or foot, slope, surface type, type of runway and, for a precision approach runway category I, the existence of an obstacle free zone when provided;

b) strip runway end safety area stopway length, width to the nearest metre or foot, surface type;

c) taxiway — designation, width, surface type;

d) apron — surface type, aircraft stands;

e) the boundaries of the air traffic control service;

f) clearway — length to the nearest metre or foot, ground profile;

g) visual aids for approach procedures, marking and lighting of runways, taxiways and aprons, other visual guidance and control aids on taxiways and aprons, including taxi-holding positions and stopbars, and location and type of visual docking guidance systems;

h) location and radio frequency of any VOR aerodrome check-point;

i) location and designation of standard taxi-routes; and

j) distances to the nearest metre or foot of localizer and glide path elements comprising an instrument landing system (ILS) or azimuth and elevation antenna of microwave landing system (MLS) in relation to the associated runway extremities.

Note.— This information may best be shown in the form of charts such as those required for the preparation of aeronautical publications as specified in Annexes 4 and 15.

2.5.2 The geographical coordinates of each threshold shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds.

2.5.3 The geographical coordinates of appropriate taxiway centre line points shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds.

2.5.4 The geographical coordinates of each aircraft stand shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds.

2.5.5 The geographical coordinates of significant obstacles in the approach and take-off areas, in the circling area and in the vicinity of an aerodrome shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and tenths of seconds. In addition, the top elevation rounded up to the nearest metre or foot, type, marking and lighting (if any) of the significant obstacles shall be reported to the aeronautical information services authority.

Note.— This information may best be shown in the form of charts such as those required for the preparation of aeronautical publications as specified in Annexes 4 and 15.

2.6 Strength of pavements

2.6.1 The bearing strength of a pavement shall be determined.
2.6.2 The bearing strength of a pavement intended for aircraft of apron (ramp) mass greater than 5,700 kg shall be made available using the aircraft classification number — pavement classification number (ACN-PCN) method by reporting all of the following information:

a) the pavement classification number (PCN);

b) pavement type for ACN-PCN determination;

c) subgrade strength category;

d) maximum allowable tire pressure category or maximum allowable tire pressure value; and

e) evaluation method.

Note.— If necessary, PCNs may be published to an accuracy of one-tenth of a whole number.

2.6.3 The pavement classification number (PCN) reported shall indicate that an aircraft with an aircraft classification number (ACN) equal to or less than the reported PCN can operate on the pavement subject to any limitation on the tire pressure, or aircraft all-up mass for specified aircraft type(s).

Note.— Different PCNs may be reported if the strength of the pavement is subject to significant seasonal variation.

2.6.4 The ACN of an aircraft shall be determined in accordance with the standard procedures associated with the ACN-PCN method.

Note.— The standard procedures for determining the ACN of an aircraft are given in the Aerodrome Design Manual, Part 3. For convenience several aircraft types currently in use have been evaluated on rigid and flexible pavements founded on the four subgrade categories in 2.6.6 b) below and the results tabulated in that manual.

2.6.5 For the purposes of determining the ACN, the behaviour of a pavement shall be classified as equivalent to a rigid or flexible construction.

2.6.6 Information on pavement type for ACN-PCN determination, subgrade strength category, maximum allowable tire pressure category and evaluation method shall be reported using the following codes:

a) Pavement type for ACN-PCN determination:

<table>
<thead>
<tr>
<th>Code</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>R</td>
<td>Rigid pavement</td>
</tr>
<tr>
<td>F</td>
<td>Flexible pavement</td>
</tr>
</tbody>
</table>

Note.— If the actual construction is composite or non-standard, include a note to that effect (see example 2 below).

b) Subgrade strength category:

<table>
<thead>
<tr>
<th>Code</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>A</td>
<td>High strength: characterized by $K = 150 \text{ MN/m}^3$ and representing all $K$ values above $120 \text{ MN/m}^3$ for rigid pavements, and by $CBR = 15$ and representing all CBR values above 13 for flexible pavements.</td>
</tr>
<tr>
<td>B</td>
<td>Medium strength: characterized by $K = 80 \text{ MN/m}^3$ and representing a range in $K$ of 60 to 120 MN/m$^3$ for rigid pavements, and by $CBR = 10$ and representing a range in CBR of 8 to 13 for flexible pavements.</td>
</tr>
<tr>
<td>C</td>
<td>Low strength: characterized by $K = 40 \text{ MN/m}^3$ and representing a range in $K$ of 25 to 60 MN/m$^3$ for rigid pavements, and by $CBR = 6$ and representing a range in CBR of 4 to 8 for flexible pavements.</td>
</tr>
<tr>
<td>D</td>
<td>Ultra low strength: characterized by $K = 20 \text{ MN/m}^3$ and representing all $K$ values below 25 MN/m$^3$ for rigid pavements, and by $CBR = 3$ and representing all CBR values below 4 for flexible pavements.</td>
</tr>
</tbody>
</table>

c) Maximum allowable tire pressure category:

<table>
<thead>
<tr>
<th>Code</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>W</td>
<td>High: no pressure limit</td>
</tr>
<tr>
<td>X</td>
<td>Medium: pressure limited to 1.50 MPa</td>
</tr>
<tr>
<td>Y</td>
<td>Low: pressure limited to 1.00 MPa</td>
</tr>
<tr>
<td>Z</td>
<td>Very low: pressure limited to 0.50 MPa</td>
</tr>
</tbody>
</table>

d) Evaluation method:

<table>
<thead>
<tr>
<th>Code</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>T</td>
<td>Technical evaluation: representing a specific study of the pavement characteristics and application of pavement behaviour technology.</td>
</tr>
<tr>
<td>U</td>
<td>Using aircraft experience: representing a knowledge of the specific type and mass of aircraft satisfactorily being supported under regular use.</td>
</tr>
</tbody>
</table>

Note.— The following examples illustrate how pavement strength data are reported under the ACN-PCN method.

Example 1.— If the bearing strength of a rigid pavement, resting on a medium strength subgrade, has been assessed by technical evaluation to be PCN 80 and there is no tire pressure limitation, then the reported information would be:

PCN 80 / R / B / W / T
Example 2.— If the bearing strength of a composite pavement, behaving like a flexible pavement and resting on a high strength subgrade, has been assessed by using aircraft experience to be PCN 50 and the maximum tire pressure allowable is 1.00 MPa, then the reported information would be:

PCN 50 / F / A / Y / U

Note.— Composite construction.

Example 3.— If the bearing strength of a flexible pavement, resting on a medium strength subgrade, has been assessed by technical evaluation to be PCN 40 and the maximum allowable tire pressure is 0.80 MPa, then the reported information would include the following note.

Note.— The reported PCN is subject to a B747-400 all-up mass limitation of 390 000 kg.

2.6.7 Recommendation.— Criteria should be established to regulate the use of a pavement by an aircraft with an ACN higher than the PCN reported for that pavement in accordance with 2.6.2 and 2.6.3.

Note.— Attachment A, Section 18 details a simple method for regulating overload operations while the Aerodrome Design Manual, Part 3 includes the descriptions of more detailed procedures for evaluation of pavements and their suitability for restricted overload operations.

2.6.8 The bearing strength of a pavement intended for aircraft of apron (ramp) mass equal to or less than 5 700 kg shall be made available by reporting the following information:

a) maximum allowable aircraft mass; and
b) maximum allowable tire pressure.

Example: 4 000 kg/0.50 MPa.

2.7 Pre-flight altimeter check location

2.7.1 One or more pre-flight altimeter check locations shall be established for an aerodrome.

2.7.2 Recommendation.— A pre-flight check location should be located on an apron.

Note 1.— Locating a pre-flight altimeter check location on an apron enables an altimeter check to be made prior to obtaining taxi clearance and eliminates the need for stopping for that purpose after leaving the apron.

2.7.3 The elevation of a pre-flight altimeter check location shall be given as the average elevation, rounded to the nearest metre or foot, of the area on which it is located. The elevation of any portion of a pre-flight altimeter check location shall be within 3 m (10 ft) of the average elevation for that location.

2.8 Declared distances

The following distances shall be calculated to the nearest metre or foot for a runway intended for use by international commercial air transport:

a) take-off run available;
b) take-off distance available;
c) accelerate-stop distance available; and
d) landing distance available.

Note.— Guidance on calculation of declared distances is given in Attachment A, Section 3.

2.9 Condition of the movement area and related facilities

2.9.1 Information on the condition of the movement area and the operational status of related facilities shall be provided to the appropriate aeronautical information service units, and similar information of operational significance to the air traffic services units, to enable those units to provide the necessary information to arriving and departing aircraft. The information shall be kept up to date and changes in conditions reported without delay.

2.9.2 The condition of the movement area and the operational status of related facilities shall be monitored and reports on matters of operational significance or affecting aircraft performance given, particularly in respect of the following:

a) construction or maintenance work;
b) rough or broken surfaces on a runway, a taxiway or an apron;
c) snow, slush or ice on a runway, a taxiway or an apron;
d) water on a runway, a taxiway or an apron;
e) snow banks or drifts adjacent to a runway, a taxiway or an apron;
f) anti-icing or de-icing liquid chemicals on a runway or a taxiway;
g) other temporary hazards, including parked aircraft;
h) failure or irregular operation of part or all of the aerodrome visual aids; and
i) failure of the normal or secondary power supply.

2.9.3 Recommendation.— To facilitate compliance with 2.9.1 and 2.9.2 inspections of the movement area should be carried out each day at least once where the code number is 1 or 2 and at least twice where the code number is 3 or 4.

Note.— Guidance on carrying out daily inspections of the movement area is given in the Airport Services Manual, Part 8 and in the Manual of Surface Movement Guidance and Control Systems (SMGCS).

Water on a runway

2.9.4 Recommendation.— Whenever water is present on a runway, a description of the runway surface conditions on the centre half of the width of the runway, including the possible assessment of water depth, where applicable, should be made available using the following terms:

- DAMP — the surface shows a change of colour due to moisture.
- WET — the surface is soaked but there is no standing water.
- WATER PATCHES — significant patches of standing water are visible.
- FLOODED — extensive standing water is visible.

2.9.5 Information that a runway or portion thereof may be slippery when wet shall be made available.

2.9.6 A runway or portion thereof shall be determined as being slippery when wet when the measurements specified in 9.4.5 show that the runway surface friction characteristics as measured by a continuous friction measuring device are below the minimum friction level specified by the State.

Note.— Guidance on determining and expressing the minimum friction level is provided in Attachment A, Section 7.

2.9.7 Information on the minimum friction level specified by the State for reporting slippery runway conditions and the type of friction measuring device used shall be made available.

2.9.8 Recommendation.— When it is suspected that a runway may become slippery under unusual conditions, then additional measurements should be made when such conditions occur, and information on the runway surface friction characteristics made available when these additional measurements show that the runway or a portion thereof has become slippery.

Snow, slush or ice on a runway

Note 1.— The intent of these specifications is to satisfy the SNOWTAM and NOTAM promulgation requirements contained in Annex 15.

Note 2.— Runway surface condition sensors may be used to detect and continuously display current or predicted information on surface conditions such as the presence of moisture, or imminent formation of ice on pavements.

2.9.9 Recommendation.— Whenever a runway is affected by snow, slush or ice, and it has not been possible to clear the precipitant fully, the condition of the runway should be assessed, and the friction coefficient measured.

Note.— Guidance on determining and expressing the friction characteristics of snow- and ice-covered paved surfaces is provided in Attachment A, Section 6.

2.9.10 Recommendation.— The readings of the friction measuring device on snow-, slush-, or ice-covered surfaces should adequately correlate with the readings of one other such device.

Note.— The principal aim is to measure surface friction in a manner that is relevant to the friction experienced by an aircraft tire, thereby providing correlation between the friction measuring device and aircraft braking performance.

2.9.11 Recommendation.— Whenever dry snow, wet snow or slush is present on a runway, an assessment of the mean depth over each third of the runway should be made to an accuracy of approximately 2 cm for dry snow, 1 cm for wet snow and 0.3 cm for slush.

2.10 Disabled aircraft removal

Note.— See 9.3 for information on disabled aircraft removal services.

2.10.1 Recommendation.— The telephone/telex number(s) of the office of the aerodrome coordinator of operations for the removal of an aircraft disabled on or adjacent to the movement area should be made available, on request, to aircraft operators.

2.10.2 Recommendation.— Information concerning the capability to remove an aircraft disabled on or adjacent to the movement area should be made available.

Note.— The capability to remove a disabled aircraft may be expressed in terms of the largest type of aircraft which the aerodrome is equipped to remove.
2.11 Rescue and fire fighting

Note.— See 9.2 for information on rescue and fire fighting services.

2.11.1 Information concerning the level of protection provided at an aerodrome for aircraft rescue and fire fighting purposes shall be made available.

2.11.2 Recommendation.— The level of protection normally available at an aerodrome should be expressed in terms of the category of the rescue and fire fighting services as described in 9.2 and in accordance with the types and amounts of extinguishing agents normally available at the aerodrome.

2.11.3 Significant changes in the level of protection normally available at an aerodrome for rescue and fire fighting shall be notified to the appropriate air traffic services units and aeronautical information units to enable those units to provide the necessary information to arriving and departing aircraft. When such a change has been corrected, the above units shall be advised accordingly.

Note.— A significant change in the level of protection is considered to be a change in the category of the rescue and fire fighting service from the category normally available at the aerodrome, resulting from a change in availability of extinguishing agents, equipment to deliver the agents or personnel to operate the equipment, etc.

2.11.4 Recommendation.— A significant change should be expressed in terms of the new category of the rescue and fire fighting service available at the aerodrome.

2.12 Visual approach slope indicator systems

The following information concerning a visual approach slope indicator system installation shall be made available:

a) associated runway designation number;

b) type of system according to 5.3.5.2. For an AT-VASIS, PAPI or APAPI installation, the side of the runway on which the lights are installed, i.e. left or right, shall be given;

c) where the axis of the system is not parallel to the runway centre line, the angle of displacement and the direction of displacement, i.e. left or right shall be indicated;

d) nominal approach slope angle(s). For a T-VASIS or an AT-VASIS this shall be angle $\theta$ according to the formula in Figure 5-14 and for a PAPI and an APAPI this shall be angle $(B + C) / 2$ and $(A + B) / 2$, respectively as in Figure 5-16; and e) minimum eye height(s) over the threshold of the on-slope signal(s). For a T-VASIS or an AT-VASIS this shall be the lowest height at which only the wing bar(s) are visible; however, the additional heights at which the wing bar(s) plus one, two or three fly down light units come into view may also be reported if such information would be of benefit to aircraft using the approach. For a PAPI this shall be the setting angle of the third unit from the runway minus 2′, i.e. angle B minus 2′, and for an APAPI this shall be the setting angle of the unit farther from the runway minus 2′, i.e. angle A minus 2′.

2.13 Coordination between aeronautical information services and aerodrome authorities

2.13.1 To ensure that aeronautical information services units obtain information to enable them to provide up-to-date pre-flight information and to meet the need for in-flight information, arrangements shall be made between aeronautical information services and aerodrome authorities responsible for aerodrome services to report to the responsible aeronautical information services unit, with a minimum of delay:

a) information on aerodrome conditions (ref. 2.9, 2.10, 2.11 and 2.12 above);

b) the operational status of associated facilities, services and navigation aids within their area of responsibility;

c) any other information considered to be of operational significance.

2.13.2 Before introducing changes to the air navigation system, due account shall be taken by the services responsible for such changes of the time needed by the aeronautical information service for the preparation, production and issue of relevant material for promulgation. To ensure timely provision of the information to the aeronautical information service, close coordination between those services concerned is therefore required.

2.13.3 Of a particular importance are changes to aeronautical information that affect charts and/or computer-based navigation systems which qualify to be notified by the aeronautical information regulation and control (AIRAC) system, as specified in Annex 15, Chapter 6 and Appendix 4. The predetermined, internationally agreed AIRAC effective dates in addition to 14 days postage time shall be observed by the responsible aerodrome services when submitting the raw information/data to aeronautical information services.

2.13.4 The aerodrome services responsible for the provision of raw aeronautical information/data to the aeronautical information services shall do that while taking
into account accuracy and integrity requirements for aeronautical data as specified in Appendix 5 to this Annex.

Note 1.— Specifications for the issue of a NOTAM and SNOWTAM are contained in Annex 15, Chapter 5, Appendices 6 and 2 respectively.

Note 2.— AIRAC information is distributed by the AIS at least 42 days in advance of the AIRAC effective dates with the objective of reaching recipients at least 28 days in advance of the effective date.

Note 3.— The schedule of the predetermined internationally agreed AIRAC common effective dates at intervals of 28 days, including 6 November 1997 and guidance for the AIRAC use are contained in the Aeronautical Information Services Manual (Doc 8126, Chapter 3, 3.1.1 and Chapter 4, 4.4).
CHAPTER 3. PHYSICAL CHARACTERISTICS

3.1 Runways

Number and orientation of runways

Introductory Note.— Many factors affect the determination of the orientation, siting and number of runways.

One important factor is the usability factor, as determined by the wind distribution, which is specified hereunder. Another important factor is the alignment of the runway to facilitate the provision of approaches conforming to the approach surface specifications of Chapter 4. In Attachment A, Section 1, information is given concerning these and other factors.

When a new instrument runway is being located, particular attention needs to be given to areas over which aeroplanes will be required to fly when following instrument approach and missed approach procedures, so as to ensure that obstacles in these areas or other factors will not restrict the operation of the aeroplanes for which the runway is intended.

3.1.1 Recommendation.— The number and orientation of runways at an aerodrome should be such that the usability factor of the aerodrome is not less than 95 per cent for the aeroplanes that the aerodrome is intended to serve.

3.1.2 Choice of maximum permissible cross-wind components

Recommendation.— In the application of 3.1.1 it should be assumed that landing or take-off of aeroplanes is, in normal circumstances, precluded when the cross-wind component exceeds:

— 37 km/h (20 kt) in the case of aeroplanes whose reference field length is 1 500 m or over, except that when poor runway braking action owing to an insufficient longitudinal coefficient of friction is experienced with some frequency, a cross-wind component not exceeding 24 km/h (13 kt) should be assumed;

— 24 km/h (13 kt) in the case of aeroplanes whose reference field length is 1 200 m or up to but not including 1 500 m; and

— 19 km/h (10 kt) in the case of aeroplanes whose reference field length is less than 1 200 m.

Note.— In Attachment A, Section 1, guidance is given on factors affecting the calculation of the estimate of the usability factor and allowances which may have to be made to take account of the effect of unusual circumstances.

3.1.3 Data to be used

Recommendation.— The selection of data to be used for the calculation of the usability factor should be based on reliable wind distribution statistics that extend over as long a period as possible, preferably of not less than five years. The observations used should be made at least eight times daily and spaced at equal intervals of time.

Note.— These winds are mean winds. Reference to the need for some allowance for gusty conditions is made in Attachment A, Section 1.

Location of threshold

3.1.4 Recommendation.— A threshold should normally be located at the extremity of a runway unless operational considerations justify the choice of another location.

Note.— Guidance on the siting of the threshold is given in Attachment A, Section 10.

3.1.5 Recommendation.— When it is necessary to displace a threshold, either permanently or temporarily, from its normal location, account should be taken of the various factors which may have a bearing on the location of the threshold. Where this displacement is due to an unserviceable runway condition, a cleared and graded area of at least 60 m in length should be available between the unserviceable area and the displaced threshold. Additional distance should also be provided to meet the requirements of the runway end safety area as appropriate.

Note.— Guidance on factors which may be considered in the determination of the location of a displaced threshold is given in Attachment A, Section 10.

Actual length of runways

3.1.6 Primary runway

Recommendation.— Except as provided in 3.1.8, the actual runway length to be provided for a primary runway
should be adequate to meet the operational requirements of the aeroplanes for which the runway is intended and should be not less than the longest length determined by applying the corrections for local conditions to the operations and performance characteristics of the relevant aeroplanes.

Note 1.— This specification does not necessarily mean providing for operations by the critical aeroplane at its maximum mass.

Note 2.— Both take-off and landing requirements need to be considered when determining the length of runway to be provided and the need for operations to be conducted in both directions of the runway.

Note 3.— Local conditions that may need to be considered include elevation, temperature, runway slope, humidity and the runway surface characteristics.

Note 4.— When performance data on aeroplanes for which the runway is intended are not known, guidance on the determination of the actual length of a primary runway by application of general correction factors is given in the Aerodrome Design Manual, Part 1.

3.1.7 Secondary runway

Recommendation.— The length of a secondary runway should be determined similarly to primary runways except that it needs only to be adequate for those aeroplanes which require to use that secondary runway in addition to the other runway or runways in order to obtain a usability factor of at least 95 per cent.

3.1.8 Runways with stopways or clearways

Recommendation.— Where a runway is associated with a stopway or clearway, an actual runway length less than that resulting from application of 3.1.6 or 3.1.7, as appropriate, may be considered satisfactory, but in such a case any combination of runway, stopway and clearway provided should permit compliance with the operational requirements for take-off and landing of the aeroplanes the runway is intended to serve.

Note.— Guidance on use of stopways and clearways is given in Attachment A, Section 2.

Width of runways

3.1.9 Recommendation.— The width of a runway should be not less than the appropriate dimension specified in the following tabulation:

<table>
<thead>
<tr>
<th>Code letter</th>
<th>Code number</th>
<th>A</th>
<th>B</th>
<th>C</th>
<th>D</th>
<th>E</th>
<th>F</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>18 m</td>
<td>18 m</td>
<td>23 m</td>
<td>–</td>
<td>–</td>
<td>–</td>
<td></td>
</tr>
<tr>
<td>2</td>
<td>23 m</td>
<td>30 m</td>
<td>45 m</td>
<td>–</td>
<td>–</td>
<td>–</td>
<td></td>
</tr>
<tr>
<td>3</td>
<td>30 m</td>
<td>45 m</td>
<td>60 m</td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

a. The width of a precision approach runway should be not less than 30 m where the code number is 1 or 2.

Note 1.— The combinations of code numbers and letters for which widths are specified have been developed for typical aeroplane characteristics.

Note 2.— Factors affecting runway width are given in the Aerodrome Design Manual, Part 1.

Minimum distance between parallel runways

3.1.10 Recommendation.— Where parallel non-instrument runways are intended for simultaneous use, the minimum distance between their centre lines should be:

— 210 m where the higher code number is 3 or 4;
— 150 m where the higher code number is 2; and
— 120 m where the higher code number is 1.

Note.— Procedures for wake turbulence categorization of aircraft and wake turbulence separation minima are contained in the Procedures for Air Navigation Services — Air Traffic Management (PANS-ATM), Doc 4444, Part V, Section 16.

3.1.11 Recommendation.— Where parallel instrument runways are intended for simultaneous use subject to conditions specified in the PANS-ATM (Doc 4444) and the PANS-OPS (Doc 8168), Volume I, the minimum distance between their centre lines should be:

— 1 035 m for independent parallel approaches;
— 915 m for dependent parallel approaches;
— 760 m for independent parallel departures;
— 760 m for segregated parallel operations;

except that:

a) for segregated parallel operations the specified minimum distance:
1) may be decreased by 30 m for each 150 m that the arrival runway is staggered toward the arriving aircraft, to a minimum of 300 m; and

2) should be increased by 30 m for each 150 m that the arrival runway is staggered away from the arriving aircraft;

b) for independent parallel approaches, combinations of minimum distances and associated conditions other than those specified in the PANS-ATM (Doc 4444) may be applied when it is determined that such combinations would not adversely affect the safety of aircraft operations.

Note.— Procedures and facilities requirements for simultaneous operations on parallel or near-parallel instrument runways are contained in the PANS-ATM (Doc 4444), Part IV and the PANS-OPS (Doc 8168), Volume I, Part VII and Volume II, Parts II and III and relevant guidance is contained in the Manual of Simultaneous Operations on Parallel or Near-Parallel Instrument Runways (Doc 9643).

Slopes on runways

3.1.12 Longitudinal slopes

Recommendation.— The slope computed by dividing the difference between the maximum and minimum elevation along the runway centre line by the runway length should not exceed:

— 1 per cent where the code number is 3 or 4; and

— 2 per cent where the code number is 1 or 2.

3.1.13 Recommendation.— Along no portion of a runway should the longitudinal slope exceed:

— 1.25 per cent where the code number is 4, except that for the first and last quarter of the length of the runway the longitudinal slope should not exceed 0.8 per cent;

— 1.5 per cent where the code number is 3, except that for the first and last quarter of the length of a precision approach runway category II or III the longitudinal slope should not exceed 0.8 per cent; and

— 2 per cent where the code number is 1 or 2.

3.1.14 Longitudinal slope changes

Recommendation.— Where slope changes cannot be avoided, a slope change between two consecutive slopes should not exceed:

— 1.5 per cent where the code number is 3 or 4; and

— 2 per cent where the code number is 1 or 2.

Note.— Guidance on slope changes before a runway is given in Attachment A, Section 4.

3.1.15 Recommendation.— The transition from one slope to another should be accomplished by a curved surface with a rate of change not exceeding:

— 0.1 per cent per 30 m (minimum radius of curvature of 30 000 m) where the code number is 4;

— 0.2 per cent per 30 m (minimum radius of curvature of 15 000 m) where the code number is 3; and

— 0.4 per cent per 30 m (minimum radius of curvature of 7 500 m) where the code number is 1 or 2.

3.1.16 Sight distance

Recommendation.— Where slope changes cannot be avoided, they should be such that there will be an unobstructed line of sight from:

— any point 3 m above a runway to all other points 3 m above the runway within a distance of at least half the length of the runway where the code letter is C, D, E or F.

— any point 2 m above a runway to all other points 2 m above the runway within a distance of at least half the length of the runway where the code letter is B; and

— any point 1.5 m above a runway to all other points 1.5 m above the runway within a distance of at least half the length of the runway where the code letter is A.

Note.— Consideration will have to be given to providing an unobstructed line of sight over the entire length of a single runway where a full-length parallel taxiway is not available. Where an aerodrome has intersecting runways, additional criteria on the line of sight of the intersection area would need to be considered for operational safety. See the Aerodrome Design Manual, Part I.

3.1.17 Distance between slope changes

Recommendation.— Undulations or appreciable changes in slopes located close together along a runway should be avoided. The distance between the points of intersection of two successive curves should not be less than:

a) the sum of the absolute numerical values of the corresponding slope changes multiplied by the appropriate value as follows:
Chapter 3

— 30 000 m where the code number is 4;
— 15 000 m where the code number is 3; and
— 5 000 m where the code number is 1 or 2; or
b) 45 m;
whichever is greater.

Note.— Guidance on implementing this specification is given in Attachment A, Section 4.

3.1.18 Transverse slopes

**Recommendation.**— To promote the most rapid drainage of water, the runway surface should, if practicable, be cambered except where a single crossfall from high to low in the direction of the wind most frequently associated with rain would ensure rapid drainage. The transverse slope should ideally be:

— 1.5 per cent where the code letter is C, D, E or F; and
— 2 per cent where the code letter is A or B;
but in any event should not exceed 1.5 per cent or 2 per cent, as applicable, nor be less than 1 per cent except at runway or taxiway intersections where flatter slopes may be necessary.

For a cambered surface the transverse slope on each side of the centre line should be symmetrical.

Note.— On wet runways with cross-wind conditions the problem of aquaplaning from poor drainage is apt to be accentuated. In Attachment A, Section 7, information is given concerning this problem and other relevant factors.

3.1.19 **Recommendation.**— The transverse slope should be substantially the same throughout the length of a runway except at an intersection with another runway or a taxiway where an even transition should be provided taking account of the need for adequate drainage.

Note.— Guidance on transverse slope is given in the Aerodrome Design Manual, Part 3.

Strength of runways

3.1.20 **Recommendation.**— A runway should be capable of withstanding the traffic of aeroplanes the runway is intended to serve.

Surface of runways

3.1.21 The surface of a runway shall be constructed without irregularities that would result in loss in friction characteristics or otherwise adversely affect the take-off or landing of an aeroplane.

Note 1.— Surface irregularities may adversely affect the take-off or landing of an aeroplane by causing excessive bouncing, pitching, vibration, or other difficulties in the control of an aeroplane.

Note 2.— Guidance on design tolerances and other information is given in Attachment A, Section 5. Additional guidance is included in the Aerodrome Design Manual, Part 3.

3.1.22 The surface of a paved runway shall be so constructed as to provide good friction characteristics when the runway is wet.

3.1.23 **Recommendation.**— Measurements of the friction characteristics of a new or resurfaced runway should be made with a continuous friction measuring device using self-wetting features in order to assure that the design objectives with respect to its friction characteristics have been achieved.

Note.— Guidance on friction characteristics of new runway surfaces is given in Attachment A, Section 7. Additional guidance is included in the Airport Services Manual, Part 2.

3.1.24 **Recommendation.**— The average surface texture depth of a new surface should be not less than 1.0 mm.

Note 1.— This normally requires some form of special surface treatment.

Note 2.— Guidance on methods used to measure surface texture is given in the Airport Services Manual, Part 2.

3.1.25 **Recommendation.**— When the surface is grooved or scored, the grooves or scorings should be either perpendicular to the runway centre line or parallel to non-perpendicular transverse joints, where applicable.

Note.— Guidance on methods for improving the runway surface texture is given in the Aerodrome Design Manual, Part 3.

3.2 Runway shoulders

**General**

Note.— Guidance on characteristics and treatment of runway shoulders is given in Attachment A, Section 8, and in the Aerodrome Design Manual, Part 1.

3.2.1 **Recommendation.**— Runway shoulders should be provided for a runway where the code letter is D or E, and the runway width is less than 60 m.
3.2.2 **Recommendation.**— Runway shoulders should be provided for a runway where the code letter is F.

**Width of runway shoulders**

3.2.3 **Recommendation.**— The runway shoulders should extend symmetrically on each side of the runway so that the overall width of the runway and its shoulders is not less than:

- 60 m where the code letter is D or E; and
- 75 m where the code letter is F.

**Slopes on runway shoulders**

3.2.4 **Recommendation.**— The surface of the shoulder that abuts the runway should be flush with the surface of the runway and its transverse slope should not exceed 2.5 per cent.

**Strength of runway shoulders**

3.2.5 **Recommendation.**— A runway shoulder should be prepared or constructed so as to be capable, in the event of an aeroplane running off the runway, of supporting the aeroplane without inducing structural damage to the aeroplane and of supporting ground vehicles which may operate on the shoulder.


### 3.3 Runway strips

**General**

3.3.1 A runway and any associated stopways shall be included in a strip.

**Length of runway strips**

3.3.2 A strip shall extend before the threshold and beyond the end of the runway or stopway for a distance of at least:

- 60 m where the code number is 2, 3 or 4;
- 60 m where the code number is 1 and the runway is an instrument one; and
- 30 m where the code number is 1 and the runway is a non-instrument one.

**Width of runway strips**

3.3.3 A strip including a precision approach runway shall, wherever practicable, extend laterally to a distance of at least:

- 150 m where the code number is 3 or 4; and
- 75 m where the code number is 1 or 2;

on each side of the centre line of the runway and its extended centre line throughout the length of the strip.

3.3.4 **Recommendation.**— A strip including a non-precision approach runway should extend laterally to a distance of at least:

- 150 m where the code number is 3 or 4; and
- 75 m where the code number is 1 or 2;

on each side of the centre line of the runway and its extended centre line throughout the length of the strip.

3.3.5 **Recommendation.**— A strip including a non-instrument runway should extend on each side of the centre line of the runway and its extended centre line throughout the length of the strip, to a distance of at least:

- 75 m where the code number is 3 or 4;
- 40 m where the code number is 2; and
- 30 m where the code number is 1.

**Objects on runway strips**

*Note.*— See 8.7 for information regarding siting and construction of equipment and installations on runway strips.

3.3.6 **Recommendation.**— An object situated on a runway strip which may endanger aeroplanes should be regarded as an obstacle and should, as far as practicable, be removed.

3.3.7 No fixed object, other than visual aids required for air navigation purposes and satisfying the relevant frangibility requirement in Chapter 5, shall be permitted on a runway strip:

a) within 77.5 m of the runway centre line of a precision approach runway category I, II or III where the code number is 4 and the code letter is F; or

b) within 60 m of the runway centre line of a precision approach runway category I, II or III where the code number is 3 or 4; or

c) within 45 m of the runway centre line of a precision approach runway category I where the code number is 1 or 2.
No mobile object shall be permitted on this part of the runway strip during the use of the runway for landing or take-off.

**Grading of runway strips**

3.3.8  **Recommendation.**— That portion of a strip of an instrument runway within a distance of at least:

- 75 m where the code number is 3 or 4; and
- 40 m where the code number is 1 or 2;

from the centre line of the runway and its extended centre line should provide a graded area for aeroplanes which the runway is intended to serve in the event of an aeroplane running off the runway.

**Note.**— Guidance on grading of a greater area of a strip including a precision approach runway where the code number is 3 or 4 is given in Attachment A, Section 8.

3.3.9  **Recommendation.**— That portion of a strip of a non-instrument runway within a distance of at least:

- 75 m where the code number is 3 or 4;
- 40 m where the code number is 2; and
- 30 m where the code number is 1;

from the centre line of the runway and its extended centre line should provide a graded area for aeroplanes which the runway is intended to serve in the event of an aeroplane running off the runway.

3.3.10  The surface of that portion of a strip that abuts a runway, shoulder or stopway shall be flush with the surface of the runway, shoulder or stopway.

3.3.11  **Recommendation.**— That portion of a strip to at least 30 m before a threshold should be prepared against blast erosion in order to protect a landing aeroplane from the danger of an exposed edge.

**Slopes on runway strips**

3.3.12  **Longitudinal slopes**

**Recommendation.**— A longitudinal slope along that portion of a strip to be graded should not exceed:

- 1.5 per cent where the code number is 4;
- 1.75 per cent where the code number is 3; and
- 2 per cent where the code number is 1 or 2.

3.3.13  **Longitudinal slope changes**

**Recommendation.**— Slope changes on that portion of a strip to be graded should be as gradual as practicable and abrupt changes or sudden reversals of slopes avoided.

3.3.14  **Transverse slopes**

**Recommendation.**— Transverse slopes on that portion of a strip to be graded should be adequate to prevent the accumulation of water on the surface but should not exceed:

- 2.5 per cent where the code number is 3 or 4; and
- 3 per cent where the code number is 1 or 2;

except that to facilitate drainage the slope for the first 3 m outward from the runway, shoulder or stopway edge should be negative as measured in the direction away from the runway and may be as great as 5 per cent.

3.3.15  **Recommendation.**— The transverse slopes of any portion of a strip beyond that to be graded should not exceed an upward slope of 5 per cent as measured in the direction away from the runway.

**Strength of runway strips**

3.3.16  **Recommendation.**— That portion of a strip of an instrument runway within a distance of at least:

- 75 m where the code number is 3 or 4; and
- 40 m where the code number is 1 or 2;

from the centre line of the runway and its extended centre line should be so prepared or constructed as to minimize hazards arising from differences in load bearing capacity to aeroplanes which the runway is intended to serve in the event of an aeroplane running off the runway.

**Note.**— Guidance on preparation of runway strips is given in the Aerodrome Design Manual, Part 1.

3.3.17  **Recommendation.**— That portion of a strip containing a non-instrument runway within a distance of at least:

- 75 m where the code number is 3 or 4;
- 40 m where the code number is 1 or 2;

from the centre line of the runway and its extended centre line should be so prepared or constructed as to minimize hazards arising from differences in load bearing capacity to aeroplanes which the runway is intended to serve in the event of an aeroplane running off the runway.
3.4 Runway end safety areas

General

3.4.1 A runway end safety area shall be provided at each end of a runway strip where:

— the code number is 3 or 4; and
— the code number is 1 or 2 and the runway is an instrument one.

Note.— Guidance on runway end safety areas is given in Attachment A, Section 9.

Dimensions of runway end safety areas

3.4.2 A runway end safety area shall extend from the end of a runway strip to a distance of at least 90 m.

3.4.3 Recommendation.— A runway end safety area should, as far as practicable, extend from the end of a runway strip to a distance of at least:

— 240 m where the code number is 3 or 4; and
— 120 m where the code number is 1 or 2.

3.4.4 The width of a runway end safety area shall be at least twice that of the associated runway.

3.4.5 Recommendation.— The width of a runway end safety area should, wherever practicable, be equal to that of the graded portion of the associated runway strip.

Objects on runway end safety areas

Note.— See 8.7 for information regarding siting and construction of equipment and installations on runway end safety areas.

3.4.6 Recommendation.— An object situated on a runway end safety area which may endanger aeroplanes should be regarded as an obstacle and should, as far as practicable, be removed.

Clearing and grading of runway end safety areas

3.4.7 Recommendation.— A runway end safety area should provide a cleared and graded area for aeroplanes which the runway is intended to serve in the event of an aeroplane undershooting or overrunning the runway.

Note.— The surface of the ground in the runway end safety area does not need to be prepared to the same quality as the runway strip. See, however, 3.4.11.

Slopes on runway end safety areas

3.4.8 General

Recommendation.— The slopes of a runway end safety area should be such that no part of the runway end safety area penetrates the approach or take-off climb surface.

3.4.9 Longitudinal slopes

Recommendation.— The longitudinal slopes of a runway end safety area should not exceed a downward slope of 5 per cent. Longitudinal slope changes should be as gradual as practicable and abrupt changes or sudden reversals of slopes avoided.

3.4.10 Transverse slopes

Recommendation.— The transverse slopes of a runway end safety area should not exceed an upward or downward slope of 5 per cent. Transitions between differing slopes should be as gradual as practicable.

Strength of runway end safety areas

3.4.11 Recommendation.— A runway end safety area should be so prepared or constructed as to reduce the risk of damage to an aeroplane undershooting or overrunning the runway, enhance aeroplane deceleration and facilitate the movement of rescue and fire fighting vehicles as required in 9.2.26 to 9.2.28.

Note.— Guidance on strength of a runway end safety area is given in the Aerodrome Design Manual, Part 1.

3.5 Clearways

Note.— The inclusion of detailed specifications for clearways in this section is not intended to imply that a clearway has to be provided. Attachment A, Section 2 provides information on the use of clearways.

Location of clearways

3.5.1 Recommendation.— The origin of a clearway should be at the end of the take-off run available.
Length of clearways

3.5.2 Recommendation.— The length of a clearway should not exceed half the length of the take-off run available.

Width of clearways

3.5.3 Recommendation.— A clearway should extend laterally to a distance of at least 75 m on each side of the extended centre line of the runway.

Slopes on clearways

3.5.4 Recommendation.— The ground in a clearway should not project above a plane having an upward slope of 1.25 per cent, the lower limit of this plane being a horizontal line which:

a) is perpendicular to the vertical plane containing the runway centre line; and

b) passes through a point located on the runway centre line at the end of the take-off run available.

Note.— Because of transverse or longitudinal slopes on a runway, shoulder or strip, in certain cases the lower limit of the clearway plane specified above may be below the corresponding elevation of the runway, shoulder or strip. It is not intended that these surfaces be graded to conform with the lower limit of the clearway plane nor is it intended that terrain or objects which are above the clearway plane beyond the end of the strip but below the level of the strip be removed unless it is considered they may endanger aeroplanes.

3.5.5 Recommendation.— Abrupt upward changes in slope should be avoided when the slope on the ground in a clearway is relatively small or when the mean slope is upward. In such situations, in that portion of the clearway within a distance of 22.5 m or half the runway width whichever is greater on each side of the extended centre line, the slopes, slope changes and the transition from runway to clearway should generally conform with those of the runway with which the clearway is associated.

Objects on clearways

Note.— See 8.7 for information regarding siting and construction of equipment and installations on clearways.

3.5.6 Recommendation.— An object situated on a clearway which may endanger aeroplanes in the air should be regarded as an obstacle and should be removed.

3.6 Stopways

Note.— The inclusion of detailed specifications for stopways in this section is not intended to imply that a stopway has to be provided. Attachment A, Section 2 provides information on the use of stopways.

Width of stopways

3.6.1 A stopway shall have the same width as the runway with which it is associated.

Slopes on stopways

3.6.2 Recommendation.— Slopes and changes in slope on a stopway, and the transition from a runway to a stopway, should comply with the specifications of 3.1.12 to 3.1.18 for the runway with which the stopway is associated except that:

a) the limitation in 3.1.13 of a 0.8 per cent slope for the first and last quarter of the length of a runway need not be applied to the stopway; and

b) at the junction of the stopway and runway and along the stopway the maximum rate of slope change may be 0.3 per cent per 30 m (minimum radius of curvature of 10,000 m) for a runway where the code number is 3 or 4.

Strength of stopways

3.6.3 Recommendation.— A stopway should be prepared or constructed so as to be capable, in the event of an abandoned take-off, of supporting the aeroplane which the stopway is intended to serve without inducing structural damage to the aeroplane.

Note.— Attachment A, Section 2 presents guidance relative to the support capability of a stopway.

Surface of stopways

3.6.4 Recommendation.— The surface of a paved stopway should be so constructed as to provide a good coefficient of friction to be compatible with that of the associated runway when the stopway is wet.

3.6.5 Recommendation.— The friction characteristics of an unpaved stopway should not be substantially less than that of the runway with which the stopway is associated.
3.7 Radio altimeter operating area

General

3.7.1 Recommendation.— A radio altimeter operating area should be established in the pre-threshold area of a precision approach runway.

Length of the area

3.7.2 Recommendation.— A radio altimeter operating area should extend before the threshold for a distance of at least 300 m.

Width of the area

3.7.3 Recommendation.— A radio altimeter operating area should extend laterally, on each side of the extended centre line of the runway, to a distance of 60 m, except that, when special circumstances so warrant, the distance may be reduced to no less than 30 m if an aeronautical study indicates that such reduction would not affect the safety of operations of aircraft.

Longitudinal slope changes

3.7.4 Recommendation.— On a radio altimeter operating area, slope changes should be avoided or kept to a minimum. Where slope changes cannot be avoided, the slope changes should be as gradual as practicable and abrupt changes or sudden reversals of slopes avoided. The rate of change between two consecutive slopes should not exceed 2 per cent per 30 m.

Note.— Guidance on radio altimeter operating area is given in Attachment A, Section 4.3 and in the Manual of All-Weather Operations, (Doc 9365), Section 5.2. Guidance on the use of radio altimeter is given in the PANS-OPS, Volume II, Part III, Chapter 21.

3.8 Taxiways

Note.— Unless otherwise indicated the requirements in this section are applicable to all types of taxiways.

General

3.8.1 Recommendation.— Taxiways should be provided to permit the safe and expeditious surface movement of aircraft.

Note.— Guidance on layout of taxiways is given in the Aerodrome Design Manual, Part 2.

3.8.2 Recommendation.— Sufficient entrance and exit taxiways for a runway should be provided to expedite the movement of aeroplanes to and from the runway and provision of rapid exit taxiways considered when traffic volumes are high.

Note.— Where the end of a runway is not served by a taxiway, it may be necessary to provide additional pavement at the end of the runway for the turning of aeroplanes. Such areas may also be useful along the runway to reduce taxiing time and distance for some aeroplanes.

3.8.3 Recommendation.— The design of a taxiway should be such that, when the cockpit of the aeroplane for which the taxiway is intended remains over the taxiway centre line markings, the clearance distance between the outer main wheel of the aeroplane and the edge of the taxiway should be not less than that given by the following tabulation:

<table>
<thead>
<tr>
<th>Code letter</th>
<th>Clearance</th>
</tr>
</thead>
<tbody>
<tr>
<td>A</td>
<td>1.5 m</td>
</tr>
<tr>
<td>B</td>
<td>2.25 m</td>
</tr>
<tr>
<td>C</td>
<td>3 m if the taxiway is intended to be used by aeroplanes with a wheel base less than 18 m; 4.5 m if the taxiway is intended to be used by aeroplanes with a wheel base equal to or greater than 18 m.</td>
</tr>
<tr>
<td>D</td>
<td>4.5 m</td>
</tr>
<tr>
<td>E</td>
<td>4.5 m</td>
</tr>
<tr>
<td>F</td>
<td>4.5 m</td>
</tr>
</tbody>
</table>

Note 1.— Wheel base means the distance from the nose gear to the geometric centre of the main gear.

Note 2.— Where the code letter is F and the traffic density is high, a wheel-to-edge clearance greater than 4.5 m may be provided to permit higher taxiing speeds.

Width of taxiways

3.8.4 Recommendation.— A straight portion of a taxiway should have a width of not less than that given by the following tabulation:

<table>
<thead>
<tr>
<th>Code letter</th>
<th>Taxiway width</th>
</tr>
</thead>
<tbody>
<tr>
<td>A</td>
<td>7.5 m</td>
</tr>
<tr>
<td>B</td>
<td>10.5 m</td>
</tr>
</tbody>
</table>
Chapter 3

Annex 14 — Aerodromes

C 15 m if the taxiway is intended to be used by aeroplanes with a wheel base less than 18 m;
18 m if the taxiway is intended to be used by aeroplanes with a wheel base equal to or greater than 18 m.

D 18 m if the taxiway is intended to be used by aeroplanes with an outer main gear wheel span of less than 9 m;
23 m if the taxiway is intended to be used by aeroplanes with an outer main gear wheel span equal to or greater than 9 m.

E 23 m

F 25 m

Note.— Guidance on width of taxiways is given in the Aerodrome Design Manual, Part 2.

Taxiway curves

3.8.5 Recommendation.— Changes in direction of taxiways should be as few and small as possible. The radii of the curves should be compatible with the manoeuvring capability and normal taxiing speeds of the aeroplanes for which the taxiway is intended. The design of the curve should be such that, when the cockpit of the aeroplane remains over the taxiway centre line markings, the clearance distance between the outer main wheels of the aeroplane and the edge of the taxiway should not be less than those specified in 3.8.3.

Note 1.— An example of widening taxiways to achieve the wheel clearance specified is illustrated in Figure 3-1. Guidance on the values of suitable dimensions is given in the Aerodrome Design Manual, Part 2.

Note 2.— The location of taxiway centre line markings and lights is specified in 5.2.8.4 and 5.3.15.10.

Note 3.— Compound curves may reduce or eliminate the need for extra taxiway width.

The figure shows an example of taxiway widening to achieve the specified wheel clearances on taxiway curves (see 3.8.5). Guidance material on suitable dimensions is given in the Aerodrome Design Manual, Part 2.

Figure 3-1. Taxiway curve
3.8.6 **Recommendation.**— To facilitate the movement of aeroplanes, fillets should be provided at junctions and intersections of taxiways with runways, aprons and other taxiways. The design of the fillets should ensure that the minimum wheel clearances specified in 3.8.3 are maintained when aeroplanes are manoeuvring through the junctions or intersections.

Note.— Consideration will have to be given to the aeroplane datum length when designing fillets. Guidance on the design of fillets and the definition of the term aeroplane datum length are given in the Aerodrome Design Manual, Part 2.

### Taxiway minimum separation distances

3.8.7 **Recommendation.**— The separation distance between the centre line of a taxiway and the centre line of a runway, the centre line of a parallel taxiway or an object should not be less than the appropriate dimension specified in Table 3-1, except that it may be permissible to operate with lower separation distances at an existing aerodrome if an aeronautical study indicates that such lower separation distances would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

#### Slopes on taxiways

3.8.8 **Longitudinal slopes**

**Recommendation.**— The longitudinal slope of a taxiway should not exceed:
— 1.5 per cent where the code letter is C, D, E or F; and
— 3 per cent where the code letter is A or B.

3.8.9 Longitudinal slope changes

**Recommendation.**— Where slope changes on a taxiway cannot be avoided, the transition from one slope to another slope should be accomplished by a curved surface with a rate of change not exceeding:

— 1 per cent per 30 m (minimum radius of curvature of 3 000 m) where the code letter is C, D, E or F; and
— 1 per cent per 25 m (minimum radius of curvature of 2 500 m) where the code letter is A or B.

3.8.10 Sight distance

**Recommendation.**— Where a change in slope on a taxiway cannot be avoided, the change should be such that, from any point:

— 3 m above the taxiway, it will be possible to see the whole surface of the taxiway for a distance of at least 300 m from that point, where the code letter is C, D, E or F;

— 2 m above the taxiway, it will be possible to see the whole surface of the taxiway for a distance of at least 200 m from that point, where the code letter is B; and

— 1.5 m above the taxiway, it will be possible to see the whole surface of the taxiway for a distance of at least 150 m from that point, where the code letter is A.

3.8.11 Transverse slopes

**Recommendation.**— The transverse slopes of a taxiway should be sufficient to prevent the accumulation of water on the surface of the taxiway but should not exceed:

— 1.5 per cent where the code letter is C, D, E or F; and
— 2 per cent where the code letter is A or B.

Note.— See 3.12.4 regarding transverse slopes on an aircraft stand taxilane.

Strength of taxiways

3.8.12 **Recommendation.**— The strength of a taxiway should be at least equal to that of the runway it serves, due consideration being given to the fact that a taxiway will be subjected to a greater density of traffic and, as a result of slow moving and stationary aeroplanes, to higher stresses than the runway it serves.

Note.— Guidance on the relation of the strength of taxiways to the strength of runways is given in the Aerodrome Design Manual, Part 3.

Surface of taxiways

3.8.13 **Recommendation.**— The surface of a taxiway should not have irregularities that cause damage to aeroplane structures.

3.8.14 **Recommendation.**— The surface of a paved taxiway should be so constructed as to provide good friction characteristics when the taxiway is wet.

Rapid exit taxiways

Note.— The following specifications detail requirements particular to rapid exit taxiways. See Figure 3-2. General requirements for taxiways also apply to this type of taxiway. Guidance on the provision, location and design of rapid exit taxiways is included in the Aerodrome Design Manual, Part 2.

3.8.15 **Recommendation.**— A rapid exit taxiway should be designed with a radius of turn-off curve of at least:

— 550 m where the code number is 3 or 4; and
— 275 m where the code number is 1 or 2; and

to enable exit speeds under wet conditions of:

— 93 km/h where the code number is 3 or 4; and
— 65 km/h where the code number is 1 or 2.

Note.— The locations of rapid exit taxiways along a runway are based on several criteria described in the Aerodrome Design Manual, Part 2, in addition to different speed criteria.

3.8.16 **Recommendation.**— The radius of the fillet on the inside of the curve at a rapid exit taxiway should be sufficient to provide a widened taxiway throat in order to facilitate early recognition of the entrance and turn-off onto the taxiway.

3.8.17 **Recommendation.**— A rapid exit taxiway should include a straight distance after the turn-off curve sufficient for an exiting aircraft to come to a full stop clear of any intersecting taxiway.

3.8.18 **Recommendation.**— The intersection angle of a rapid exit taxiway with the runway should not be greater than 45° nor less than 25° and preferably should be 30°.
Annex 14 — Aerodromes

4/11/99 26

Taxis on bridges

3.8.19 The width of that portion of a taxiway bridge capable of supporting aeroplanes, as measured perpendicularly to the taxiway centre line, shall not be less than the width of the graded area of the strip provided for that taxiway, unless a proven method of lateral restraint is provided which shall not be hazardous for aeroplanes for which the taxiway is intended.

3.8.20 Recommendation.— Access should be provided to allow rescue and fire fighting vehicles to intervene in both directions within the specified response time to the largest aeroplane for which the taxiway bridge is intended.

Note.— If aeroplane engines overhang the bridge structure, protection of adjacent areas below the bridge from engine blast may be required.

3.8.21 Recommendation.— A bridge should be constructed on a straight section of the taxiway with a straight section on both ends of the bridge to facilitate the alignment of aeroplanes approaching the bridge.

3.9 Taxiway shoulders

Note.— Guidance on characteristics of taxiway shoulders and on shoulder treatment is given in the Aerodrome Design Manual, Part 2.

3.9.1 Recommendation.— Straight portions of a taxiway where the code letter is C, D, E or F should be provided with shoulders which extend symmetrically on each side of the taxiway so that the overall width of the taxiway and its shoulders on straight portions is not less than:

— 60 m where the code letter is F;
— 44 m where the code letter is E;
— 38 m where the code letter is D; and
— 25 m where the code letter is C.

On taxiway curves and on junctions or intersections where increased pavement is provided, the shoulder width should be not less than that on the adjacent straight portions of the taxiway.

3.9.2 Recommendation.— When a taxiway is intended to be used by turbine-engined aeroplanes, the surface of the taxiway shoulder should be so prepared as to resist erosion and the ingestion of the surface material by aeroplane engines.

3.10 Taxiway strips

Note.— Guidance on characteristics of taxiway strips is given in the Aerodrome Design Manual, Part 2.
Chapter 3

General

3.10.1 A taxiway, other than an aircraft stand taxilane, shall be included in a strip.

Width of taxiway strips

3.10.2 Recommendation.— A taxiway strip should extend symmetrically on each side of the centre line of the taxiway throughout the length of the taxiway to at least the distance from the centre line given in Table 3-1, column 11.

Objects on taxiway strips

Note.— See 8.7 for information regarding siting and construction of equipment and installations on taxiway strips.

3.10.3 Recommendation.— The taxiway strip should provide an area clear of objects which may endanger taxiing aeroplanes.

Note.— Consideration will have to be given to the location and design of drains on a taxiway strip to prevent damage to an aeroplane accidentally running off a taxiway. Suitably designed drain covers may be required.

Grading of taxiway strips

3.10.4 Recommendation.— The centre portion of a taxiway strip should provide a graded area to a distance from the centre line of the taxiway of at least:

— 11 m where the code letter is A;
— 12.5 m where the code letter is B or C;
— 19 m where the code letter is D;
— 22 m where the code letter is E; and
— 30 m where the code letter is F.

Slopes on taxiway strips

3.10.5 Recommendation.— The surface of the strip should be flush at the edge of the taxiway or shoulder, if provided, and the graded portion should not have an upward transverse slope exceeding:

— 2.5 per cent for strips where the code letter is C, D, E or F; and
— 3 per cent for strips of taxiways where the code letter is A or B;

the upward slope being measured with reference to the transverse slope of the adjacent taxiway surface and not the horizontal. The downward transverse slope should not exceed 5 per cent measured with reference to the horizontal.

3.10.6 Recommendation.— The transverse slopes on any portion of a taxiway strip beyond that to be graded should not exceed an upward or downward slope of 5 per cent as measured in the direction away from the taxiway.

3.11 Holding bays, runway-holding positions, intermediate holding positions and road-holding positions

General

3.11.1 Recommendation.— Holding bay(s) should be provided when the traffic density is medium or heavy.

3.11.2 A runway-holding position or positions shall be established:

a) on the taxiway, at the intersection of a taxiway and a runway; and
b) at an intersection of a runway with another runway when the former runway is part of a standard taxi-route.

3.11.3 A runway-holding position shall be established on a taxiway if the location or alignment of the taxiway is such that a taxiing aircraft or vehicle can infringe an obstacle limitation surface or interfere with the operation of radio navigation aids.

3.11.4 Recommendation.— An intermediate holding position should be established on a taxiway at any point other than a runway-holding position where it is desirable to define a specific holding limit.

3.11.5 A road-holding position shall be established at an intersection of a road with a runway.

Location

3.11.6 The distance between a holding bay, runway-holding position established at a taxiway/runway intersection or road-holding position and the centre line of a runway shall be in accordance with Table 3-2 and, in the case of a precision approach runway, such that a holding aircraft or vehicle will not interfere with the operation of radio navigation aids.

3.11.7 Recommendation.— At elevations greater than 700 m (2 300 ft) the distance of 90 m specified in Table 3-2 for a precision approach runway code number 4 should be increased as follows:
Annex 14 — Aerodromes

Table 3-2. Minimum distance from the runway centre line to a holding bay, runway-holding position or road-holding position

<table>
<thead>
<tr>
<th>Type of runway</th>
<th>1</th>
<th>2</th>
<th>3</th>
<th>4</th>
</tr>
</thead>
<tbody>
<tr>
<td>Non-instrument</td>
<td>30 m</td>
<td>40 m</td>
<td>75 m</td>
<td>75 m</td>
</tr>
<tr>
<td>Non-precision approach</td>
<td>40 m</td>
<td>40 m</td>
<td>75 m</td>
<td>75 m</td>
</tr>
<tr>
<td>Precision approach category I</td>
<td>60 m&lt;sup&gt;b&lt;/sup&gt;</td>
<td>60 m&lt;sup&gt;b&lt;/sup&gt;</td>
<td>90 m&lt;sup&gt;a,b&lt;/sup&gt;</td>
<td>90 m&lt;sup&gt;a,b,c&lt;/sup&gt;</td>
</tr>
<tr>
<td>Precision approach categories II and III</td>
<td>–</td>
<td>–</td>
<td>90 m&lt;sup&gt;a,b&lt;/sup&gt;</td>
<td>90 m&lt;sup&gt;a,b,c&lt;/sup&gt;</td>
</tr>
<tr>
<td>Take-off runway</td>
<td>30 m</td>
<td>40 m</td>
<td>75 m</td>
<td>75 m</td>
</tr>
</tbody>
</table>

a. If a holding bay, runway-holding position or road-holding position is at a lower elevation compared to the threshold, the distance may be decreased 5 m for every metre the bay or holding position is lower than the threshold, contingent upon not infringing the inner transitional surface.

b. This distance may need to be increased to avoid interference with radio navigation aids, particularly the glide path and localizer facilities. Information on critical and sensitive areas of ILS and MLS is contained in Annex 10, Volume I, Attachments C and G to Part I, respectively (see also 3.11.6).

Note 1.— The distance of 90 m for code number 3 or 4 is based on an aircraft with a tail height of 20 m, a distance from the nose to the highest part of the tail of 52.7 m and a nose height of 10 m holding at an angle of 45° or more with respect to the runway centre line, being clear of the obstacle free zone and not accountable for the calculation of OCA/H.

Note 2.— The distance of 60 m for code number 2 is based on an aircraft with a tail height of 8 m, a distance from the nose to the highest part of the tail of 24.6 m and a nose height of 5.2 m holding at an angle of 45° or more with respect to the runway centre line, being clear of the obstacle free zone.

c. Where the code letter is F, this distance should be 107.5 m.

Note.— The distance of 107.5 m for code number 4 where the code letter is F is based on an aircraft with a tail height of 24 m, a distance from the nose to the highest part of the tail of 62.2 m and a nose height of 10 m holding at an angle of 45° or more with respect to the runway centre line, being clear of the obstacle free zone.

3.12 Aprons

General

3.12.1 Recommendation.— Aprons should be provided where necessary to permit the on- and off-loading of passengers, cargo or mail as well as the servicing of aircraft without interfering with the aerodrome traffic.

Size of aprons

3.12.2 Recommendation.— The total apron area should be adequate to permit expeditious handling of the aerodrome traffic at its maximum anticipated density.
Chapter 3

**Strength of aprons**

3.12.3 **Recommendation.**— Each part of an apron should be capable of withstanding the traffic of the aircraft it is intended to serve, due consideration being given to the fact that some portions of the apron will be subjected to a higher density of traffic and, as a result of slow moving or stationary aircraft, to higher stresses than a runway.

**Slopes on aprons**

3.12.4 **Recommendation.**— Slopes on an apron, including those on an aircraft stand taxilane, should be sufficient to prevent accumulation of water on the surface of the apron but should be kept as level as drainage requirements permit.

3.12.5 **Recommendation.**— On an aircraft stand the maximum slope should not exceed 1 per cent.

**Clearance distances on aircraft stands**

3.12.6 **Recommendation.**— An aircraft stand should provide the following minimum clearances between an aircraft using the stand and any adjacent building, aircraft on another stand and other objects:

<table>
<thead>
<tr>
<th>Code letter</th>
<th>Clearance</th>
</tr>
</thead>
<tbody>
<tr>
<td>A</td>
<td>3 m</td>
</tr>
<tr>
<td>B</td>
<td>3 m</td>
</tr>
<tr>
<td>C</td>
<td>4.5 m</td>
</tr>
<tr>
<td>D</td>
<td>7.5 m</td>
</tr>
<tr>
<td>E</td>
<td>7.5 m</td>
</tr>
<tr>
<td>F</td>
<td>7.5 m</td>
</tr>
</tbody>
</table>

When special circumstances so warrant, these clearances may be reduced at a nose-in aircraft stand, where the code letter is D, E or F:

a) between the terminal, including any fixed passenger bridge, and the nose of an aircraft; and

b) over any portion of the stand provided with azimuth guidance by a visual docking guidance system.

**Note.**— On aprons, consideration also has to be given to the provision of service roads and to manoeuvring and storage area for ground equipment (see the Aerodrome Design Manual, Part 2, for guidance on storage of ground equipment).

**3.13 Isolated aircraft parking position**

3.13.1 An isolated aircraft parking position shall be designated or the aerodrome control tower shall be advised of an area or areas suitable for the parking of an aircraft which is known or believed to be the subject of unlawful interference, or which for other reasons needs isolation from normal aerodrome activities.

3.13.2 **Recommendation.**— The isolated aircraft parking position should be located at the maximum distance practicable and in any case never less than 100 m from other parking positions, buildings or public areas, etc. Care should be taken to ensure that the position is not located over underground utilities such as gas and aviation fuel and, to the extent feasible, electrical or communication cables.

**3.14 De-icing/anti-icing facilities**

**Note.**— Safe and efficient aeroplane operations are of primary importance in the development of an aeroplane de-icing/anti-icing facility. For further guidance, see the Manual on Aircraft Ground De-icing/Anti-icing Operations (Doc 9640).

**General**

3.14.1 **Recommendation.**— Aeroplane de-icing/anti-icing facilities should be provided at an aerodrome where icing conditions are expected to occur.

**Location**

3.14.2 **Recommendation.**— De-icing/anti-icing facilities should be provided either at aircraft stands or at specified remote areas along the taxiway leading to the runway meant for take-off, provided that adequate drainage arrangements for the collection and safe disposal of excess de-icing/anti-icing fluids are available to prevent ground water contamination. The effect of volume of traffic and departure flow rates should also be considered.

**Note 1.**— One of the primary factors influencing the location of a de-icing/anti-icing facility is to ensure that the holdover time of the anti-icing treatment is still in effect at the end of taxiing and when take-off clearance of the treated aeroplane is given.

**Note 2.**— Remote facilities compensate for changing weather conditions when icing conditions or blowing snow are expected to occur along the taxi route taken by the aeroplane to the runway meant for take-off.

3.14.3 **Recommendation.**— The remote de-icing/anti-icing facility should be located to be clear of the obstacle limitation surfaces specified in Chapter 4, not cause interference to the radio navigation aids and be clearly visible from the air traffic control tower for clearing the treated aeroplane.
3.14.4 Recommendation.— The remote de-icing/anti-icing facility should be so located as to provide for an expeditious traffic flow, perhaps with a bypass configuration, and not require unusual taxiing manoeuvre into and out of the pads.

Note.— The jet blast effects caused by a moving aeroplane on other aeroplanes receiving the anti-icing treatment or taxiing behind will have to be taken into account to prevent degradation of the treatment.

Size and number of de-icing/anti-icing pads

Note.— An aeroplane de-icing/anti-icing pad consists of:
   a) an inner area for parking of an aeroplane to be treated, and
   b) an outer area for movement of two or more mobile de-icing/anti-icing equipment.

3.14.5 Recommendation.— The size of a de-icing/anti-icing pad should be equal to the parking area required by the most demanding aeroplane in a given category with at least 3.8 m clear paved area all round the aeroplane for the movement of the de-icing/anti-icing vehicles.

Note.— Where more than one de-icing/anti-icing pad is provided, consideration will have to be given to providing de-icing/anti-icing vehicle movement areas of adjacent pads that do not overlap, but are exclusive for each pad. Consideration will also need to be given to bypassing of the area by other aeroplanes with the clearances specified in 3.14.9 and 3.14.10.

3.14.6 Recommendation.— The number of de-icing/anti-icing pads required should be determined based on the meteorological conditions, the type of aeroplanes to be treated, the method of application of de-icing/anti-icing fluid, the type and capacity of the dispensing equipment used, and the departure flow rates.

Note.— See the Aerodrome Design Manual, Part 2.

Slopes on de-icing/anti-icing pads

3.14.7 Recommendation.— The de-icing/anti-icing pads should be provided with suitable slopes to ensure satisfactory drainage of the area and to permit collection of all excess de-icing/anti-icing fluid running off an aeroplane. The maximum longitudinal slope should be as little as practicable and the transverse slope should not exceed 1 per cent.

Strength of de-icing/anti-icing pads

3.14.8 Recommendation.— The de-icing/anti-icing pad should be capable of withstanding the traffic of the aircraft it is intended to serve, due consideration being given to the fact that the de-icing/anti-icing pad (like an apron) will be subjected to a higher density of traffic and, as a result of slow-moving or stationary aircraft, to higher stresses than a runway.

Clearance distances on a de-icing/anti-icing pad

3.14.9 Recommendation.— A de-icing/anti-icing pad should provide the minimum clearances specified in 3.12.6 for aircraft stands. If the pad layout is such as to include bypass configuration, the minimum separation distances specified in Table 3-1, column 12, should be provided.

3.14.10 Recommendation.— Where the de-icing/anti-icing facility is located adjoining a regular taxiway, the taxiway minimum separation distance specified in Table 3-1, column 11, should be provided. (See Figure 3-3.)

Environmental considerations

Note.— The excess de-icing/anti-icing fluid running off an aeroplane poses the risk of contamination of ground water in addition to affecting the pavement surface friction characteristics.

3.14.11 Recommendation.— Where de-icing/anti-icing activities are carried out, the surface drainage should be planned to collect the run-off separately, preventing its mixing with the normal surface run-off so that it does not pollute the ground water.
Figure 3.3. Minimum separation distance on a de-icing/anti-icing facility
CHAPTER 4. OBSTACLE RESTRICTION AND REMOVAL

Note 1.— The objectives of the specifications in this chapter are to define the airspace around aerodromes to be maintained free from obstacles so as to permit the intended aeroplane operations at the aerodromes to be conducted safely and to prevent the aerodromes from becoming unusable by the growth of obstacles around the aerodromes. This is achieved by establishing a series of obstacle limitation surfaces that define the limits to which objects may project into the airspace.

Note 2.— Objects which penetrate the obstacle limitation surfaces contained in this chapter may in certain circumstances cause an increase in the obstacle clearance altitude/height for an instrument approach procedure or any associated visual circling procedure. Criteria for evaluating obstacles are contained in Procedures for Air Navigation Services — Aircraft Operations (PANS-OPS) (Doc 8168).

Note 3.— The establishment of, and requirements for, an obstacle protection surface for visual approach slope indicator systems are specified in 5.3.5.41 to 5.3.5.45.

4.1 Obstacle limitation surfaces

Note.— See Figure 4-1.

Inner horizontal surface

4.1.4 Description.— Inner horizontal surface. A surface located in a horizontal plane above an aerodrome and its environs.

4.1.5 Characteristics.— The radius or outer limits of the inner horizontal surface shall be measured from a reference point or points established for such purpose.

Note.— The shape of the inner horizontal surface need not necessarily be circular. Guidance on determining the extent of the inner horizontal surface is contained in the Airport Services Manual, Part 6.

4.1.6 The height of the inner horizontal surface shall be measured above an elevation datum established for such purpose.

Note.— Guidance on determining the elevation datum is contained in the Airport Services Manual, Part 6.

Approach surface

4.1.7 Description.— Approach surface. An inclined plane or combination of planes preceding the threshold.

4.1.8 Characteristics.— The limits of the approach surface shall comprise:

a) an inner edge of specified length, horizontal and perpendicular to the extended centre line of the runway and located at a specified distance before the threshold;

b) two sides originating at the ends of the inner edge and diverging uniformly at a specified rate from the extended centre line of the runway; and

c) an outer edge parallel to the inner edge.

The above surfaces shall be varied when lateral offset, offset or curved approaches are utilized, specifically, two sides originating at the ends of the inner edge and diverging uniformly at a specified rate from the extended centre line of the lateral offset, offset or curved ground track.

4.1.9 The elevation of the inner edge shall be equal to the elevation of the mid-point of the threshold.

4.1.10 The slope(s) of the approach surface shall be measured in the vertical plane containing the centre line of the runway and shall continue containing the centre line of any lateral offset or curved ground track.
See Figure 4-2 for inner transitional and balked landing obstacle limitation surfaces and Attachment B for a three-dimensional view.

Figure 4-1. Obstacle limitation surfaces
**Inner approach surface**

4.1.11 Description.— Inner approach surface. A rectangular portion of the approach surface immediately preceding the threshold.

4.1.12 Characteristics.— The limits of the inner approach surface shall comprise:

a) an inner edge coincident with the location of the inner edge of the approach surface but of its own specified length;

b) two sides originating at the ends of the inner edge and extending parallel to the vertical plane containing the centre line of the runway; and

c) an outer edge parallel to the inner edge.

**Transitional surface**

4.1.13 Description.— Transitional surface. A complex surface along the side of the strip and part of the side of the approach surface, that slopes upwards and outwards to the inner horizontal surface.

4.1.14 Characteristics.— The limits of a transitional surface shall comprise:

a) a lower edge beginning at the intersection of the side of the approach surface with the inner horizontal surface and
extending down the side of the approach surface to the inner edge of the approach surface and from there along the length of the strip parallel to the runway centre line; and

b) an upper edge located in the plane of the inner horizontal surface.

4.1.15 The elevation of a point on the lower edge shall be:

a) along the side of the approach surface — equal to the elevation of the approach surface at that point; and

b) along the strip — equal to the elevation of the nearest point on the centre line of the runway or its extension.

Note.— As a result of b) the transitional surface along the strip will be curved if the runway profile is curved, or a plane if the runway profile is a straight line. The intersection of the transitional surface with the inner horizontal surface will also be a curved or straight line depending on the runway profile.

4.1.16 The slope of the transitional surface shall be measured in a vertical plane at right angles to the centre line of the runway.

Inner transitional surface

Note.— It is intended that the inner transitional surface be the controlling obstacle limitation surface for navigation aids, aircraft and other vehicles that must be near the runway and which is not to be penetrated except for frangible objects. The transitional surface described in 4.1.13 is intended to remain as the controlling obstacle limitation surface for buildings, etc.

4.1.17 Description.— Inner transitional surface. A surface similar to the transitional surface but closer to the runway.

4.1.18 Characteristics.— The limits of an inner transitional surface shall comprise:

a) a lower edge beginning at the end of the inner approach surface and extending down the side of the inner approach surface to the inner edge of that surface, from there along the strip parallel to the runway centre line to the inner edge of the balked landing surface and from there up the side of the balked landing surface to the point where the side intersects the inner horizontal surface; and

b) an upper edge located in the plane of the inner horizontal surface.

4.1.19 The elevation of a point on the lower edge shall be:

a) along the side of the inner approach surface and balked landing surface — equal to the elevation of the particular surface at that point; and

b) along the strip — equal to the elevation of the nearest point on the centre line of the runway or its extension.

Note.— As a result of b) the inner transitional surface along the strip will be curved if the runway profile is curved or a plane if the runway profile is a straight line. The intersection of the inner transitional surface with the inner horizontal surface will also be a curved or straight line depending on the runway profile.

4.1.20 The slope of the inner transitional surface shall be measured in a vertical plane at right angles to the centre line of the runway.

Balked landing surface

4.1.21 Description.— Balked landing surface. An inclined plane located at a specified distance after the threshold, extending between the inner transitional surface.

4.1.22 Characteristics.— The limits of the balked landing surface shall comprise:

a) an inner edge horizontal and perpendicular to the centre line of the runway and located at a specified distance after the threshold;

b) two sides originating at the ends of the inner edge and diverging uniformly at a specified rate from the vertical plane containing the centre line of the runway; and

c) an outer edge parallel to the inner edge and located in the plane of the inner horizontal surface.

4.1.23 The elevation of the inner edge shall be equal to the elevation of the runway centre line at the location of the inner edge.

4.1.24 The slope of the balked landing surface shall be measured in the vertical plane containing the centre line of the runway.

Take-off climb surface

4.1.25 Description.— Take-off climb surface. An inclined plane or other specified surface beyond the end of a runway or clearway.

4.1.26 Characteristics.— The limits of the take-off climb surface shall comprise:

a) an inner edge horizontal and perpendicular to the centre line of the runway and located either at a specified distance beyond the end of the runway or at the end of the clearway when such is provided and its length exceeds the specified distance;

b) two sides originating at the ends of the inner edge, diverging uniformly at a specified rate from the take-off
track to a specified final width and continuing thereafter at that width for the remainder of the length of the take-off climb surface; and

c) an outer edge horizontal and perpendicular to the specified take-off track.

4.1.27 The elevation of the inner edge shall be equal to the highest point on the extended runway centre line between the end of the runway and the inner edge, except that when a clearway is provided the elevation shall be equal to the highest point on the ground on the centre line of the clearway.

4.1.28 In the case of a straight take-off flight path, the slope of the take-off climb surface shall be measured in the vertical plane containing the centre line of the runway.

4.1.29 In the case of a take-off flight path involving a turn, the take-off climb surface shall be a complex surface containing the horizontal normals to its centre line, and the slope of the centre line shall be the same as that for a straight take-off flight path.

4.2 Obstacle limitation requirements

Note.— The requirements for obstacle limitation surfaces are specified on the basis of the intended use of a runway, i.e. take-off or landing and type of approach, and are intended to be applied when such use is made of the runway. In cases where operations are conducted to or from both directions of a runway, then the function of certain surfaces may be nullified because of more stringent requirements of another lower surface.

Non-instrument runways

4.2.1 The following obstacle limitation surfaces shall be established for a non-instrument runway:

— conical surface;
— inner horizontal surface;
— approach surface; and
— transitional surfaces.

4.2.2 The heights and slopes of the surfaces shall not be greater than, and their other dimensions not less than, those specified in Table 4-1.

4.2.3 New objects or extensions of existing objects shall not be permitted above an approach or transitional surface except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object.

Note.— Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6.

4.2.4 Recommendation.— New objects or extensions of existing objects should not be permitted above the conical surface or inner horizontal surface except when, in the opinion of the appropriate authority, the object would be shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

4.2.5 Recommendation.— Existing objects above any of the surfaces required by 4.2.1 should as far as practicable be removed except when, in the opinion of the appropriate authority, the object is shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

Note.— Because of transverse or longitudinal slopes on a strip, in certain cases the inner edge or portions of the inner edge of the approach surface may be below the corresponding elevation of the strip. It is not intended that the strip be graded to conform with the inner edge of the approach surface, nor is it intended that terrain or objects which are above the approach surface beyond the end of the strip, but below the level of the strip, be removed unless it is considered they may endanger aeroplanes.

4.2.6 Recommendation.— In considering proposed construction, account should be taken of the possible future development of an instrument runway and consequent requirement for more stringent obstacle limitation surfaces.

Non-precision approach runways

4.2.7 The following obstacle limitation surfaces shall be established for a non-precision approach runway:

— conical surface;
— inner horizontal surface;
— approach surface; and
— transitional surfaces.

4.2.8 The heights and slopes of the surfaces shall not be greater than, and their other dimensions not less than, those specified in Table 4-1, except in the case of the horizontal section of the approach surface (see 4.2.9).

4.2.9 The approach surface shall be horizontal beyond the point at which the 2.5 per cent slope intersects:

a) a horizontal plane 150 m above the threshold elevation; or

b) the horizontal plane passing through the top of any object that governs the obstacle clearance altitude/height (OCA/H);

whichever is the higher.
Table 4-1. Dimensions and slopes of obstacle limitation surfaces — Approach runways

**APPROACH RUNWAYS**

<table>
<thead>
<tr>
<th>Surface and dimensions&lt;sup&gt;a&lt;/sup&gt;</th>
<th>Non-instrument</th>
<th>Non-precision approach</th>
<th>Precision approach category</th>
</tr>
</thead>
<tbody>
<tr>
<td>(1)</td>
<td>(2)</td>
<td>(3)</td>
<td>(4)</td>
</tr>
<tr>
<td>CONICAL</td>
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<td></td>
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<td></td>
</tr>
<tr>
<td>Height</td>
<td>5%</td>
<td>5%</td>
<td>5%</td>
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<tr>
<td>Radius</td>
<td>35 m</td>
<td>55 m</td>
<td>75 m</td>
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<td></td>
</tr>
<tr>
<td>Height</td>
<td>45 m</td>
<td>45 m</td>
<td>45 m</td>
</tr>
<tr>
<td>Radius</td>
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<td>2 500 m</td>
<td>4 000 m</td>
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<td></td>
</tr>
<tr>
<td>Width</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Distance from threshold</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Length</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Slope</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>APPROACH</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Length of inner edge</td>
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<td>80 m</td>
<td>150 m</td>
</tr>
<tr>
<td>Distance from threshold</td>
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<td>60 m</td>
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<tr>
<td>Divergence (each side)</td>
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<td>10%</td>
<td>10%</td>
</tr>
<tr>
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<td>2 500 m</td>
<td>3 000 m</td>
</tr>
<tr>
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<td>Second section</td>
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<td></td>
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<tr>
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<td></td>
</tr>
<tr>
<td>Slope</td>
<td></td>
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<td>Horizontal section</td>
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<td>Total length</td>
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<td>Slope</td>
<td>20%</td>
<td>20%</td>
<td>14.3%</td>
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<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Slope</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>BALKED LANDING SURFACE</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Length of inner edge</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Distance from threshold</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Divergence (each side)</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Slope</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

---

a. All dimensions are measured horizontally unless specified otherwise.
b. Variable length (see 4.2.9 or 4.2.17).
c. Distance to the end of strip.
d. Or end of runway whichever is less.
e. Where the code letter is F (Column (3) of Table 1-1), the width is increased to 155 m.
4.2.10 New objects or extensions of existing objects shall not be permitted above an approach surface within 3 000 m of the inner edge or above a transitional surface except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object.

Note.— Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6.

4.2.11 Recommendation.— New objects or extensions of existing objects should not be permitted above the approach surface beyond 3 000 m from the inner edge, the conical surface or inner horizontal surface except when, in the opinion of the appropriate authority, the object would be shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

4.2.12 Recommendation.— Existing objects above any of the surfaces required by 4.2.7 should as far as practicable be removed except when, in the opinion of the appropriate authority, the object is shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

Note.— Because of transverse or longitudinal slopes on a strip, in certain cases the inner edge or portions of the inner edge of the approach surface may be below the corresponding elevation of the strip. It is not intended that the strip be graded to conform with the inner edge of the approach surface, nor is it intended that terrain or objects which are above the approach surface beyond the end of the strip, be removed unless it is considered they may endanger aeroplanes.

Precision approach runways

Note 1.— See 8.7 for information regarding siting and construction of equipment and installations on operational areas.

Note 2.— Guidance on obstacle limitation surfaces for precision approach runways is given in the Airport Services Manual, Part 6.

4.2.13 The following obstacle limitation surfaces shall be established for a precision approach runway category I:

— conical surface;
— inner horizontal surface;
— approach surface; and
— transitional surfaces.

4.2.14 Recommendation.— The following obstacle limitation surfaces should be established for a precision approach runway category I:

— inner approach surface;
— inner transitional surfaces; and
— balked landing surface.

4.2.15 The following obstacle limitation surfaces shall be established for a precision approach runway category II or III:

— conical surface;
— inner horizontal surface;
— approach surface and inner approach surface;
— transitional surfaces;
— inner transitional surfaces; and
— balked landing surface.

4.2.16 The heights and slopes of the surfaces shall not be greater than, and their other dimensions not less than, those specified in Table 4-1, except in the case of the horizontal section of the approach surface (see 4.2.17).

4.2.17 The approach surface shall be horizontal beyond the point at which the 2.5 per cent slope intersects:

a) a horizontal plane 150 m above the threshold elevation; or

b) the horizontal plane passing through the top of any object that governs the obstacle clearance limit;

whichever is the higher.

4.2.18 Fixed objects shall not be permitted above the inner approach surface, the inner transitional surface or the balked landing surface, except for frangible objects which because of their function must be located on the strip. Mobile objects shall not be permitted above these surfaces during the use of the runway for landing.

4.2.19 New objects or extensions of existing objects shall not be permitted above an approach surface or a transitional surface except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object.

Note.— Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6.

4.2.20 Recommendation.— New objects or extensions of existing objects should not be permitted above the conical surface and the inner horizontal surface except when, in the opinion of the appropriate authority, an object would be shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

4.2.21 Recommendation.— Existing objects above an approach surface, a transitional surface, the conical surface and inner horizontal surface should as far as practicable be removed except when, in the opinion of the appropriate
authority, an object is shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

Note.— Because of transverse or longitudinal slopes on a strip, in certain cases the inner edge or portions of the inner edge of the approach surface may be below the corresponding elevation of the strip. It is not intended that the strip be graded to conform with the inner edge of the approach surface, nor is it intended that terrain or objects which are above the approach surface beyond the end of the strip, but below the level of the strip, be removed unless it is considered they may endanger aeroplanes.

Runways meant for take-off

4.2.22 The following obstacle limitation surface shall be established for a runway meant for take-off:

— take-off climb surface.

4.2.23 The dimensions of the surface shall be not less than the dimensions specified in Table 4-2, except that a lesser length may be adopted for the take-off climb surface where such lesser length would be consistent with procedural measures adopted to govern the outward flight of aeroplanes.

4.2.24 Recommendation.— The operational characteristics of aeroplanes for which the runway is intended should be examined to see if it is desirable to reduce the slope specified in Table 4-2 when critical operating conditions are to be catered to. If the specified slope is reduced, corresponding adjustment in the length of take-off climb surface should be made so as to provide protection to a height of 300 m.

Note.— When local conditions differ widely from sea level standard atmospheric conditions, it may be advisable for the slope specified in Table 4-2 to be reduced. The degree of this reduction depends on the divergence between local conditions and sea level standard atmospheric conditions, and on the performance characteristics and operational requirements of the aeroplanes for which the runway is intended.

4.2.25 New objects or extensions of existing objects shall not be permitted above a take-off climb surface except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object.

Note.— Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6.

4.2.26 Recommendation.— If no object reaches the 2 per cent (1:50) take-off climb surface, new objects should be limited to preserve the existing obstacle free surface or a surface down to a slope of 1.6 per cent (1:62.5).

Table 4-2. Dimensions and slopes of obstacle limitation surfaces

<table>
<thead>
<tr>
<th>RUNWAYS MEANT FOR TAKE-OFF</th>
<th>Code number</th>
</tr>
</thead>
<tbody>
<tr>
<td>Surface and dimensionsa</td>
<td>1</td>
</tr>
<tr>
<td>(1)</td>
<td>(2)</td>
</tr>
<tr>
<td>TAKE-OFF CLIMB</td>
<td></td>
</tr>
<tr>
<td>Length of inner edge</td>
<td>60 m</td>
</tr>
<tr>
<td>Distance from runway endb</td>
<td>30 m</td>
</tr>
<tr>
<td>Divergence (each side)</td>
<td>10%</td>
</tr>
<tr>
<td>Final width</td>
<td>380 m</td>
</tr>
<tr>
<td></td>
<td></td>
</tr>
<tr>
<td>Length</td>
<td>1 600 m</td>
</tr>
<tr>
<td>Slope</td>
<td>5%</td>
</tr>
</tbody>
</table>

a. All dimensions are measured horizontally unless specified otherwise.
b. The take-off climb surface starts at the end of the clearway if the clearway length exceeds the specified distance.
c. 1 800 m when the intended track includes changes of heading greater than 15° for operations conducted in IMC, VMC by night.
d. See 4.2.24 and 4.2.26.
4.2.27 **Recommendation.**— Existing objects that extend above a take-off climb surface should as far as practicable be removed except when, in the opinion of the appropriate authority, an object is shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of aeroplanes.

Note.— Because of transverse slopes on a strip or clearway, in certain cases portions of the inner edge of the take-off climb surface may be below the corresponding elevation of the strip or clearway. It is not intended that the strip or clearway be graded to conform with the inner edge of the take-off climb surface, nor is it intended that terrain or objects which are above the take-off climb surface beyond the end of the strip or clearway, but below the level of the strip or clearway, be removed unless it is considered they may endanger aeroplanes. Similar considerations apply at the junction of a clearway and strip where differences in transverse slopes exist.

4.3 **Objects outside the obstacle limitation surfaces**

4.3.1 **Recommendation.**— Arrangements should be made to enable the appropriate authority to be consulted concerning proposed construction beyond the limits of the obstacle limitation surfaces that extend above a height established by that authority, in order to permit an aeronautical study of the effect of such construction on the operation of aeroplanes.

4.3.2 **Recommendation.**— In areas beyond the limits of the obstacle limitation surfaces, at least those objects which extend to a height of 150 m or more above ground elevation should be regarded as obstacles, unless a special aeronautical study indicates that they do not constitute a hazard to aeroplanes.

Note.— This study may have regard to the nature of operations concerned and may distinguish between day and night operations.

4.4 **Other objects**

4.4.1 **Recommendation.**— Objects which do not project through the approach surface but which would nevertheless adversely affect the optimum siting or performance of visual or non-visual aids should, as far as practicable, be removed.

4.4.2 **Recommendation.**— Anything which may, in the opinion of the appropriate authority after aeronautical study, endanger aeroplanes on the movement area or in the air within the limits of the inner horizontal and conical surfaces should be regarded as an obstacle and should be removed in so far as practicable.

Note.— In certain circumstances, objects that do not project above any of the surfaces enumerated in 4.1 may constitute a hazard to aeroplanes as, for example, where there are one or more isolated objects in the vicinity of an aerodrome.
CHAPTER 5. VISUAL AIDS FOR NAVIGATION

5.1 Indicators and signalling devices

5.1.1 Wind direction indicators

Application

5.1.1.1 An aerodrome shall be equipped with at least one wind direction indicator.

Location

5.1.1.2 A wind direction indicator shall be located so as to be visible from aircraft in flight or on the movement area and in such a way as to be free from the effects of air disturbances caused by nearby objects.

Characteristics

5.1.1.3 Recommendation.— The wind direction indicator should be in the form of a truncated cone made of fabric and should have a length of not less than 3.6 m and a diameter, at the larger end, of not less than 0.9 m. It should be constructed so that it gives a clear indication of the direction of the surface wind and a general indication of the wind speed. The colour or colours should be so selected as to make the wind direction indicator clearly visible and understandable from a height of at least 300 m, having regard to background. Where practicable, a single colour, preferably white or orange, should be used. Where a combination of two colours is required to give adequate conspicuity against changing backgrounds, they should preferably be orange and white, red and white, or black and white, and should be arranged in five alternate bands, the first and last bands being the darker colour.

5.1.1.4 Recommendation.— The location of at least one wind direction indicator should be marked by a circular band 15 m in diameter and 1.2 m wide. The band should be centred about the wind direction indicator support and should be in a colour chosen to give adequate conspicuity, preferably white.

5.1.1.5 Recommendation.— Provision should be made for illuminating at least one wind indicator at an aerodrome intended for use at night.

5.1.2 Landing direction indicator

Location

5.1.2.1 Where provided, a landing direction indicator shall be located in a conspicuous place on the aerodrome.

Characteristics

5.1.2.2 Recommendation.— The landing direction indicator should be in the form of a “T”.

5.1.2.3 The shape and minimum dimensions of a landing “T” shall be as shown in Figure 5-1. The colour of the landing “T” shall be either white or orange, the choice being dependent on the colour that contrasts best with the background against which the indicator will be viewed. Where required for use at night the landing “T” shall either be illuminated or outlined by white lights.

5.1.3 Signalling lamp

Application

5.1.3.1 A signalling lamp shall be provided at a controlled aerodrome in the aerodrome control tower.

Characteristics

5.1.3.2 Recommendation.— A signalling lamp should be capable of producing red, green and white signals, and of:

![Figure 5-1. Landing direction indicator](image_url)
a) being aimed manually at any target as required;

b) giving a signal in any one colour followed by a signal in either of the two other colours; and

c) transmitting a message in any one of the three colours by Morse Code up to a speed of at least four words per minute.

When selecting the green light, use should be made of the restricted boundary of green as specified in Appendix 1, 2.1.2.

5.1.3.3 Recommendation.— The beam spread should be not less than 1° nor greater than 3°, with negligible light beyond 3°. When the signalling lamp is intended for use in the daytime the intensity of the coloured light should be not less than 6 000 cd.

5.1.4 Signal panels and signal area

Note.— The inclusion of detailed specifications for a signal area in this section is not intended to imply that one has to be provided. Attachment A, Section 15 provides guidance on the need to provide ground signals. Annex 2, Appendix 1 specifies the shape, colour and use of visual ground signals. The Aerodrome Design Manual, Part 4 provides guidance on their design.

Location of signal area

5.1.4.1 Recommendation.— The signal area should be located so as to be visible for all angles of azimuth above an angle of 10° above the horizontal when viewed from a height of 300 m.

Characteristics of signal area

5.1.4.2 The signal area shall be an even horizontal surface at least 9 m square.

5.1.4.3 Recommendation.— The colour of the signal area should be chosen to contrast with the colours of the signal panels used, and it should be surrounded by a white border not less than 0.3 m wide.

5.2 Markings

5.2.1 General

Interruption of runway markings

5.2.1.1 At an intersection of two (or more) runways the markings of the more important runway, except for the runway side stripe marking, shall be displayed and the markings of the other runway(s) shall be interrupted. The runway side stripe marking of the more important runway may be either continued across the intersection or interrupted.

5.2.1.2 Recommendation.— The order of importance of runways for the display of runway markings should be as follows:

1st — precision approach runway;

2nd — non-precision approach runway; and

3rd — non-instrument runway.

5.2.1.3 At an intersection of a runway and taxiway the markings of the runway shall be displayed and the markings of the taxiway interrupted, except that runway side stripe markings may be interrupted.

Note.— See 5.2.8.5 regarding the manner of connecting runway and taxiway centre line markings.

Colour and conspicuity

5.2.1.4 Runway markings shall be white.

Note 1.— It has been found that, on runway surfaces of light colour, the conspicuity of white markings can be improved by outlining them in black.

Note 2.— It is preferable that the risk of uneven friction characteristics on markings be reduced in so far as practicable by the use of a suitable kind of paint.

Note 3.— Markings may consist of solid areas or a series of longitudinal stripes providing an effect equivalent to the solid areas.

5.2.1.5 Taxiway markings and aircraft stand markings shall be yellow.

5.2.1.6 Apron safety lines shall be of a conspicuous colour which shall contrast with that used for aircraft stand markings.

5.2.1.7 Recommendation.— At aerodromes where operations take place at night, pavement markings should be made with reflective materials designed to enhance the visibility of the markings.

Unpaved taxiways

5.2.1.8 **Recommendation.**— An unpaved taxiway should be provided, so far as practicable, with the markings prescribed for paved taxiways.

5.2.2 Runway designation marking

**Application**

5.2.2.1 A runway designation marking shall be provided at the thresholds of a paved runway.

5.2.2.2 **Recommendation.**— A runway designation marking should be provided, so far as practicable, at the thresholds of an unpaved runway.

**Location**

5.2.2.3 A runway designation marking shall be located at a threshold as shown in Figure 5-2 as appropriate.

---

**Characteristics**

5.2.2.4 A runway designation marking shall consist of a two-digit number and on parallel runways shall be supplemented with a letter. On a single runway, dual parallel runways and triple parallel runways the two-digit number shall be the whole number nearest the one-tenth of the magnetic North when viewed from the direction of approach. On four or more parallel runways, one set of adjacent runways shall be numbered to the nearest one-tenth magnetic azimuth and the other set of adjacent runways numbered to the next nearest one-tenth of the magnetic azimuth. When the above rule would give a single digit number, it shall be preceded by a zero.

5.2.2.5 In the case of parallel runways, each runway designation number shall be supplemented by a letter as follows, in the order shown from left to right when viewed from the direction of approach:

---

**Note.**— If the runway threshold is displaced from the extremity of the runway, a sign showing the designation of the runway may be provided for aeroplanes taking off.
5.2.2.6 The numbers and letters shall be in the form and proportion shown in Figure 5-3. The dimensions shall be not less than those shown in Figure 5-3, but where the numbers are incorporated in the threshold marking, larger dimensions shall be used in order to fill adequately the gap between the stripes of the threshold marking.

5.2.3 Runway centre line marking

**Application**

5.2.3.1 A runway centre line marking shall be provided on a paved runway.
Chapter 5

Location

5.2.3.2 A runway centre line marking shall be located along the centre line of the runway between the runway designation markings as shown in Figure 5-2, except when interrupted in compliance with 5.2.1.1.

Characteristics

5.2.3.3 A runway centre line marking shall consist of a line of uniformly spaced stripes and gaps. The length of a stripe plus a gap shall be not less than 50 m or more than 75 m. The length of each stripe shall be at least equal to the length of the gap or 30 m, whichever is greater.

5.2.3.4 The width of the stripes shall be not less than:

— 0.90 m on precision approach category II and III runways;
— 0.45 m on non-precision approach runways where the code number is 3 or 4, and precision approach category I runways; and
— 0.30 m on non-precision approach runways where the code number is 1 or 2, and on non-instrument runways.

5.2.4 Threshold marking

Application

5.2.4.1 A threshold marking shall be provided at the threshold of a paved instrument runway, and of a paved non-instrument runway where the code number is 3 or 4 and the runway is intended for use by international commercial air transport.

5.2.4.2 Recommendation.— A threshold marking should be provided at the threshold of a paved non-instrument runway where the code number is 3 or 4 and the runway is intended for use by other than international commercial air transport.

5.2.4.3 Recommendation.— A threshold marking should be provided, so far as practicable, at the thresholds of an unpaved runway.

Note.— The Aerodrome Design Manual, Part 4, shows a form of marking which has been found satisfactory for the marking of downward slopes immediately before the threshold.

Location

5.2.4.4 The stripes of the threshold marking shall commence 6 m from the threshold.

Characteristics

5.2.4.5 A runway threshold marking shall consist of a pattern of longitudinal stripes of uniform dimensions disposed symmetrically about the centre line of a runway as shown in Figure 5-2 (A) and (B) for a runway width of 45 m. The number of stripes shall be in accordance with the runway width as follows:

<table>
<thead>
<tr>
<th>Runway width</th>
<th>Number of stripes</th>
</tr>
</thead>
<tbody>
<tr>
<td>18 m</td>
<td>4</td>
</tr>
<tr>
<td>23 m</td>
<td>6</td>
</tr>
<tr>
<td>30 m</td>
<td>8</td>
</tr>
<tr>
<td>45 m</td>
<td>12</td>
</tr>
<tr>
<td>60 m</td>
<td>16</td>
</tr>
</tbody>
</table>

except that on non-precision approach and non-instrument runways 45 m or greater in width, they may be as shown in Figure 5-2 (C).

5.2.4.6 The stripes shall extend laterally to within 3 m of the edge of a runway or to a distance of 27 m on either side of a runway centre line, whichever results in the smaller lateral distance. Where a runway designation marking is placed within a threshold marking there shall be a minimum of three stripes on each side of the centre line of the runway. Where a runway designation marking is placed above a threshold marking, the stripes shall be continued across the runway. The stripes shall be at least 30 m long and approximately 1.80 m wide with spacings of approximately 1.80 m between them except that, where the stripes are continued across a runway, a double spacing shall be used to separate the two stripes nearest the centre line of the runway, and in the case where the designation marking is included within the threshold marking this spacing shall be 22.5 m.

Transverse stripe

5.2.4.7 Recommendation.— Where a threshold is displaced from the extremity of a runway or where the extremity of a runway is not square with the runway centre line, a transverse stripe as shown in Figure 5-4 (B) should be added to the threshold marking.

5.2.4.8 A transverse stripe shall be not less than 1.80 m wide.

Arrows

5.2.4.9 Where a runway threshold is permanently displaced, arrows conforming to Figure 5-4 (B) shall be provided on the portion of the runway before the displaced threshold.
5.2.4.10 When a runway threshold is temporarily displaced from the normal position, it shall be marked as shown in Figure 5-4 (A) or 5-4 (B) and all markings prior to the displaced threshold shall be obscured except the runway centre line marking, which shall be converted to arrows.

Note 1.— In the case where a threshold is temporarily displaced for only a short period of time, it has been found satisfactory to use markers in the form and colour of a displaced threshold marking rather than attempting to paint this marking on the runway.

Note 2.— When the runway before a displaced threshold is unfit for the surface movement of aircraft, closed markings, as described in 7.1.4, are required to be provided.

5.2.5 Aiming point marking

Application

5.2.5.1 The provisions of Sections 5.2.5 and 5.2.6 shall not require the replacement of existing markings before 1 January 2005.

5.2.5.2 An aiming point marking shall be provided at each approach end of a paved instrument runway where the code number is 2, 3 or 4.

5.2.5.3 Recommendation.— An aiming point marking should be provided at each approach end of:

a) a paved non-instrument runway where the code number is 3 or 4,

b) a paved instrument runway where the code number is 1, when additional conspicuity of the aiming point is desirable.

Location

5.2.5.4 The aiming point marking shall commence no closer to the threshold than the distance indicated in the appropriate column of Table 5-1, except that, on a runway equipped with a visual approach slope indicator system, the beginning of the marking shall be coincident with the visual approach slope origin.
5.2.5.5 An aiming point marking shall consist of two conspicuous stripes. The dimensions of the stripes and the lateral spacing between their inner sides shall be in accordance with the provisions of the appropriate column of Table 5-1. Where a touchdown zone marking is provided, the lateral spacing between the markings shall be the same as that of the touchdown zone marking.

5.2.6 Touchdown zone marking

Application

5.2.6.1 A touchdown zone marking shall be provided in the touchdown zone of a paved precision approach runway where the code number is 2, 3 or 4.

5.2.6.2 Recommendation.—A touchdown zone marking should be provided in the touchdown zone of a paved non-precision approach or non-instrument runway where the code number is 3 or 4 and additional conspicuity of the touchdown zone is desirable.

Location and characteristics

5.2.6.3 A touchdown zone marking shall consist of pairs of rectangular markings symmetrically disposed about the runway centre line with the number of such pairs related to the landing distance available and, where the marking is to be displayed at both the approach directions of a runway, the distance between the thresholds, as follows:

<table>
<thead>
<tr>
<th>Location and dimensions</th>
<th>Landing distance available or the distance between thresholds</th>
<th>Pair(s) of markings</th>
</tr>
</thead>
<tbody>
<tr>
<td>Distance from threshold to beginning of marking</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Less than 800 m</td>
<td>150 m</td>
<td>250 m</td>
</tr>
<tr>
<td>800 m up to but not including 1 200 m</td>
<td></td>
<td></td>
</tr>
<tr>
<td>1 200 m up to but not including 2 400 m</td>
<td></td>
<td></td>
</tr>
<tr>
<td>2 400 m and above</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Length of stripe*</td>
<td>30-45 m</td>
<td>30-45 m</td>
</tr>
<tr>
<td>Width of stripe</td>
<td>4 m</td>
<td>6 m</td>
</tr>
<tr>
<td>Lateral spacing between inner sides of stripes</td>
<td>6 m*</td>
<td>9 m*</td>
</tr>
</tbody>
</table>

a. The greater dimensions of the specified ranges are intended to be used where increased conspicuity is required.
b. The lateral spacing may be varied within these limits to minimize the contamination of the marking by rubber deposits.
c. These figures were deduced by reference to the outer main gear wheel span which is element 2 of the aerodrome reference code at Chapter 1, Table 1-1.

5.2.6.4 A touchdown zone marking shall conform to either of the two patterns shown in Figure 5-5. For the pattern shown in Figure 5-5 (A), the markings shall be not less than 22.5 m long and 3 m wide. For the pattern shown in Figure 5-5 (B), each stripe of each marking shall be not less than 22.5 m long and 1.8 m wide with a spacing of 1.5 m between adjacent stripes. The lateral spacing between the inner sides of the rectangles shall be equal to that of the aiming point marking where provided. Where an aiming point marking is not provided, the lateral spacing between the inner sides of the rectangles shall correspond to the lateral spacing specified for the aiming point marking in Table 5-1 (columns 2, 3, 4 or 5, as appropriate). The pairs of markings shall be provided at longitudinal spacings of 150 m beginning from the threshold.
Figure 5-5. Aiming point and touchdown zone markings
(illustrated for a runway with a length of 2400 m or more)
Chapter 5

Annex 14 — Aerodromes

except that pairs of touchdown zone markings coincident with or located within 50 m of an aiming point marking shall be deleted from the pattern.

5.2.6.5 Recommendation.— On a non-precision approach runway where the code number is 2, an additional pair of touchdown zone marking stripes should be provided 150 m beyond the beginning of the aiming point marking.

5.2.7 Runway side stripe marking

Application

5.2.7.1 A runway side stripe marking shall be provided between the thresholds of a paved runway where there is a lack of contrast between the runway edges and the shoulders or the surrounding terrain.

5.2.7.2 Recommendation.— A runway side stripe marking should be provided on a precision approach runway irrespective of the contrast between the runway edges and the shoulders or the surrounding terrain.

Location

5.2.7.3 Recommendation.— A runway side stripe marking should consist of two stripes, one placed along each edge of the runway with the outer edge of each stripe approximately on the edge of the runway, except that, where the runway is greater than 60 m in width, the stripes should be located 30 m from the runway centre line.

Characteristics

5.2.7.4 Recommendation.— A runway side stripe should have an overall width of at least 0.9 m on runways 30 m or more in width and at least 0.45 m on narrower runways.

5.2.8 Taxiway centre line marking

Application

5.2.8.1 Taxiway centre line marking shall be provided on a paved taxiway, de-icing/anti-icing facility and apron where the code number is 3 or 4 in such a way as to provide continuous guidance between the runway centre line and aircraft stands.

5.2.8.2 Recommendation.— Taxiway centre line marking should be provided on a paved taxiway, de-icing/anti-icing facility and apron where the code number is 1 or 2 in such a way as to provide continuous guidance between the runway centre line and aircraft stands.

5.2.8.3 Taxiway centre line marking shall be provided on a paved runway when the runway is part of a standard taxi-route and:

a) there is no runway centre line marking; or

b) where the taxiway centre line is not coincident with the runway centre line.

Location

5.2.8.4 Recommendation.— On a straight section of a taxiway the taxiway centre line marking should be located along the taxiway centre line. On a taxiway curve the marking should continue from the straight portion of the taxiway at a constant distance from the outside edge of the curve.

Note.— See 3.8.5 and Figure 3-1.

5.2.8.5 Recommendation.— At an intersection of a taxiway with a runway where the taxiway serves as an exit from the runway, the taxiway centre line marking should be curved into the runway centre line marking as shown in Figures 5-6 and 5-21. The taxiway centre line marking should be extended parallel to the runway centre line marking for a distance of at least 60 m beyond the point of tangency where the code number is 3 or 4, and for a distance of at least 30 m where the code number is 1 or 2.

5.2.8.6 Recommendation.— Where taxiway centre line marking is provided on a runway in accordance with 5.2.8.3, the marking should be located on the centre line of the designated taxiway.

Characteristics

5.2.8.7 A taxiway centre line marking shall be at least 15 cm in width and continuous in length except where it intersects with a runway-holding position marking or an intermediate holding position marking as shown in Figure 5-6.

5.2.9 Runway-holding position marking

Application and location

5.2.9.1 A runway-holding position marking shall be displayed along a runway-holding position.

Note.— See 5.4.2 concerning the provision of signs at runway-holding positions.

Characteristics

5.2.9.2 At an intersection of a taxiway and a non-instrument, non-precision approach or take-off runway, the
RUNWAY-HOLDING POSITION MARKING

PATTERN A:
4 lines and
3 spaces at
0.15 m each

PATTERN B:
2 lines at
0.3 m each
1 space at
0.5 m

INTERMEDIATE HOLDING
POSITION MARKING

Figure 5-6. Taxiway markings
*(shown with basic runway markings)*
runway-holding position marking shall be as shown in Figure 5-6, pattern A.

5.2.9.3 Where a single runway-holding position is provided at an intersection of a taxiway and a precision approach category I, II or III runway, the runway-holding position marking shall be as shown in Figure 5-6, pattern A. Where two or three runway-holding positions are provided at such an intersection, the runway-holding position marking closer (closest) to the runway shall be as shown in Figure 5-6, pattern A and the markings farther from the runway shall be as shown in Figure 5-6, pattern B.

5.2.9.4 The runway-holding position marking displayed at a runway-holding position established in accordance with 3.11.3 shall be as shown in Figure 5-6, pattern A.

5.2.9.5 Recommendation.— Where increased conspicuity of the runway-holding position marking is required, the runway-holding position marking should be as shown in Figure 5-7, pattern A or pattern B, as appropriate.

5.2.9.6 Recommendation.— Where a pattern B runway-holding position marking is located on an area where it would exceed 60 m in length, the term “CAT II” or “CAT III” as appropriate should be marked on the surface at the ends of the runway-holding position marking and at equal intervals of 45 m maximum between successive marks. The letters should be not less than 1.8 m high and should be placed not more than 0.9 m beyond the holding position marking.

5.2.9.7 The runway-holding position marking displayed at a runway/runway intersection shall be perpendicular to the centre line of the runway forming part of the standard taxi-route. The pattern of the marking shall be as shown in Figure 5-7, pattern A.

5.2.10 Intermediate holding position marking

Application and location

5.2.10.1 Recommendation.— An intermediate holding position marking should be displayed along an intermediate holding position.

5.2.10.2 Recommendation.— An intermediate holding position marking should be displayed at the exit boundary of a remote de-icing/anti-icing facility adjoining a taxiway.
5.2.10.3 Where an intermediate holding position marking is displayed at an intersection of two paved taxiways, it shall be located across the taxiway at sufficient distance from the near edge of the intersecting taxiway to ensure safe clearance between taxiing aircraft. It shall be coincident with a stop bar or intermediate holding position lights, where provided.

5.2.10.4 The distance between an intermediate holding position marking at the exit boundary of a remote de-icing/anti-icing facility and the centre line of the adjoining taxiway shall not be less than the dimension specified in Table 3-1, column 11.

Characteristics

5.2.10.5 An intermediate holding position marking shall consist of a single broken line as shown in Figure 5-6.

5.2.11 VOR aerodrome check-point marking

Application

5.2.11.1 When a VOR aerodrome check-point is established, it shall be indicated by a VOR aerodrome check-point marking and sign.

Note.— See 5.4.4 for VOR aerodrome check-point sign.

5.2.11.2 Site selection

Note.— Guidance on the selection of sites for VOR aerodrome check-points is given in Annex 10, Volume I, Attachment E to Part I.

Location

5.2.11.3 A VOR aerodrome check-point marking shall be centred on the spot at which an aircraft is to be parked to receive the correct VOR signal.

Characteristics

5.2.11.4 A VOR aerodrome check-point marking shall consist of a circle 6 m in diameter and have a line width of 15 cm (see Figure 5-8 (A)).

5.2.11.5 Recommendation.— When it is preferable for an aircraft to be aligned in a specific direction, a line should be provided that passes through the centre of the circle on the desired azimuth. The line should extend 6 m outside the circle in the desired direction of heading and terminate in an arrowhead. The width of the line should be 15 cm (see Figure 5-8 (B)).

5.2.11.6 Recommendation.— A VOR aerodrome check-point marking should preferably be white in colour but should differ from the colour used for the taxiway markings.

Note.— To provide contrast, markings may be bordered with black.

5.2.12 Aircraft stand markings

Note.— Guidance on the layout of aircraft stand markings is contained in the Aerodrome Design Manual, Part 4.
Chapter 5

Application

5.2.12.1 Recommendation.— Aircraft stand markings should be provided for designated parking positions on a paved apron and on a de-icing/anti-icing facility.

Location

5.2.12.2 Recommendation.— Aircraft stand markings on a paved apron and on a de-icing/anti-icing facility should be located so as to provide the clearances specified in 3.12.6 and in 3.14.9 respectively, when the nose wheel follows the stand marking.

Characteristics

5.2.12.3 Recommendation.— Aircraft stand markings should include such elements as stand identification, lead-in line, turn bar, turning line, alignment bar, stop line and lead-out line, as are required by the parking configuration and to complement other parking aids.

5.2.12.4 Recommendation.— An aircraft stand identification (letter and/or number) should be included in the lead-in line a short distance after the beginning of the lead-in line. The height of the identification should be adequate to be readable from the cockpit of aircraft using the stand.

5.2.12.5 Recommendation.— Where two sets of aircraft stand markings are superimposed on each other in order to permit more flexible use of the apron and it is difficult to identify which stand marking should be followed, or safety would be impaired if the wrong marking was followed, then identification of the aircraft for which each set of markings is intended should be added to the stand identification.

Note.— Example: 2A-B747, 2B-F28.

5.2.12.6 Recommendation.— Lead-in, turning and lead-out lines should normally be continuous in length and have a width of not less than 15 cm. Where one or more sets of stand markings are superimposed on a stand marking, the lines should be continuous for the most demanding aircraft and broken for other aircraft.

5.2.12.7 Recommendation.— The curved portions of lead-in, turning and lead-out lines should have radii appropriate to the most demanding aircraft type for which the markings are intended.

5.2.12.8 Recommendation.— Where it is intended that an aircraft proceed in one direction only, arrows pointing in the direction to be followed should be added as part of the lead-in and lead-out lines.

5.2.12.9 Recommendation.— A turn bar should be located at right angles to the lead-in line, abeam the left pilot position at the point of initiation of any intended turn. It should have a length and width of not less than 6 m and 15 cm, respectively, and include an arrowhead to indicate the direction of turn.

Note.— The distances to be maintained between the turn bar and the lead-in line may vary according to different aircraft types, taking into account the pilot’s field of view.

5.2.12.10 Recommendation.— If more than one turn bar and/or stop line is required, they should be coded.

5.2.12.11 Recommendation.— An alignment bar should be placed so as to be coincident with the extended centre line of the aircraft in the specified parking position and visible to the pilot during the final part of the parking manoeuvre. It should have a width of not less than 15 cm.

5.2.12.12 Recommendation.— A stop line should be located at right angles to the alignment bar, abeam the left pilot position at the intended point of stop. It should have a length and width of not less than 6 m and 15 cm, respectively.

Note.— The distances to be maintained between the stop line and the lead-in line may vary according to different aircraft types, taking into account the pilot’s field of view.

5.2.13 Apron safety lines

Note.— Guidance on apron safety lines is contained in the Aerodrome Design Manual, Part 4.

Application

5.2.13.1 Recommendation.— Apron safety lines should be provided on a paved apron as required by the parking configurations and ground facilities.

Location

5.2.13.2 Apron safety lines shall be located so as to define the areas intended for use by ground vehicles and other aircraft servicing equipment, etc., to provide safe separation from aircraft.

Characteristics

5.2.13.3 Recommendation.— Apron safety lines should include such elements as wing tip clearance lines and service road boundary lines as required by the parking configurations and ground facilities.
5.2.13.4 **Recommendation.**— An apron safety line should be continuous in length and at least 10 cm in width.

5.2.14 Road-holding position marking

**Application**

5.2.14.1 A road-holding position marking shall be provided at all road entrances to a runway.

**Location**

5.2.14.2 The road-holding position marking shall be located across the road at the holding position.

**Characteristics**

5.2.14.3 The road-holding position marking shall be in accordance with the local road traffic regulations.

5.2.15 Mandatory instruction marking

**Note.**— *Guidance on mandatory instruction marking is given in the Aerodrome Design Manual, Part 4.*

**Application**

5.2.15.1 Where it is impracticable to install a mandatory instruction sign in accordance with 5.4.2.1, a mandatory instruction marking shall be provided on the surface of the pavement.

5.2.15.2 **Recommendation.**— Where operationally required, such as on taxiways exceeding 60 m in width, a mandatory instruction sign should be supplemented by a mandatory instruction marking.

**Location**

5.2.15.3 The mandatory instruction marking shall be located on the left-hand side of the taxiway centre line marking and on the holding side of the runway-holding position marking as shown in Figure 5-9. The distance between the nearest edge of the marking and the runway-holding position marking or the taxiway centre line marking shall be not less than 1 m.

5.2.15.4 **Recommendation.**— Except where operationally required, a mandatory instruction marking should not be located on a runway.

**Characteristics**

5.2.15.5 A mandatory instruction marking shall consist of an inscription in white on a red background. Except for a NO ENTRY marking, the inscription shall provide information identical to that of the associated mandatory instruction sign.

5.2.15.6 A NO ENTRY marking shall consist of an inscription in white reading NO ENTRY on a red background.

5.2.15.7 Where there is insufficient contrast between the marking and the pavement surface, the mandatory instruction marking shall include an appropriate border, preferably white or black.

---

**Figure 5-9.** Mandatory instruction marking
5.2.15.8 **Recommendation.**— The character height should be 4 m. The inscriptions should be in the form and proportions shown in Appendix 3.

5.2.15.9 **Recommendation.**— The background should be rectangular and extend a minimum of 0.5 m laterally and vertically beyond the extremities of the inscription.

5.2.16 Information marking

*Note.*— Guidance on information marking is contained in the Aerodrome Design Manual, Part 4.

**Application**

5.2.16.1 Where an information sign would normally be installed and it is physically impossible to install a sign, an information marking shall be displayed on the surface of the pavement.

5.2.16.2 **Recommendation.**— Where operationally required an information sign should be supplemented by an information marking.

**Location**

5.2.16.3 **Recommendation.**— The information marking should be displayed across the surface of the taxiway or apron where necessary and positioned so as to be legible from the cockpit of an approaching aircraft.

**Characteristics**

5.2.16.4 An information marking shall consist of:

a) an inscription in yellow, when it replaces or supplements a location sign; and

b) an inscription in black, when it replaces or supplements a direction or destination sign.

5.2.16.5 Where there is insufficient contrast between the marking and the pavement surface, the marking shall include:

a) a black background where the inscriptions are in yellow; and

b) a yellow background where the inscriptions are in black.

5.2.16.6 **Recommendation.**— The character height should be 4 m. The inscriptions should be in the form and proportions shown in Appendix 3.

5.3 Lights

**5.3.1 General**

**Lights which may endanger the safety of aircraft**

5.3.1.1 A non-aeronautical ground light near an aerodrome which might endanger the safety of aircraft shall be extinguished, screened or otherwise modified so as to eliminate the source of danger.

**Lights which may cause confusion**

5.3.1.2 **Recommendation.**— A non-aeronautical ground light which, by reason of its intensity, configuration or colour, might prevent, or cause confusion in, the clear interpretation of aeronautical ground lights should be extinguished, screened or otherwise modified so as to eliminate such a possibility. In particular, attention should be directed to a non-aeronautical ground light visible from the air within the areas described hereunder:

a) Instrument runway — code number 4:

within the areas before the threshold and beyond the end of the runway extending at least 4 500 m in length from the threshold and runway end and 750 m either side of the extended runway centre line in width.

b) Instrument runway — code number 2 or 3:

as in a), except that the length should be at least 3 000 m.

c) Instrument runway — code number 1;

and non-instrument runway:

within the approach area.

**Aeronautical ground lights which may cause confusion to mariners**

*Note.*— In the case of aeronautical ground lights near navigable waters, consideration needs to be given to ensuring that the lights do not cause confusion to mariners.

**Light fixtures and supporting structures**

*Note.*— See 8.7 for information regarding siting and construction of equipment and installations on operational areas, and the Aerodrome Design Manual, Part 6 (in preparation) for guidance on fragility of light fixtures and supporting structures.
Elevated approach lights

5.3.1.3 Elevated approach lights and their supporting structures shall be frangible except that, in that portion of the approach lighting system beyond 300 m from the threshold:

a) where the height of a supporting structure exceeds 12 m, the frangibility requirement shall apply to the top 12 m only; and

b) where a supporting structure is surrounded by non-frangible objects, only that part of the structure that extends above the surrounding objects shall be frangible.

5.3.1.4 The provisions of 5.3.1.3 shall not require the replacement of existing installations before 1 January 2005.

5.3.1.5 When an approach light fixture or supporting structure is not in itself sufficiently conspicuous, it shall be suitably marked.

Elevated lights

5.3.1.6 Elevated runway, stopway and taxiway lights shall be frangible. Their height shall be sufficiently low to preserve clearance for propellers and for the engine pods of jet aircraft.

Surface lights

5.3.1.7 Light fixtures inset in the surface of runways, stopways, taxiways and aprons shall be so designed and fitted as to withstand being run over by the wheels of an aircraft without damage either to the aircraft or to the lights themselves.

5.3.1.8 Recommendation.— The temperature produced by conduction or radiation at the interface between an installed inset light and an aircraft tire should not exceed 160°C during a 10-minute period of exposure.

Note.— Guidance on measuring the temperature of inset lights is given in the Aerodrome Design Manual, Part 4.

Light intensity and control

Note.— In dusk or poor visibility conditions by day, lighting can be more effective than marking. For lights to be effective in such conditions or in poor visibility by night, they must be of adequate intensity. To obtain the required intensity, it will usually be necessary to make the light directional, in which case the arcs over which the light shows will have to be adequate and so orientated as to meet the operational requirements. The runway lighting system will have to be considered as a whole, to ensure that the relative light intensities are suitably matched to the same end. (See Attachment A, Section 14, and the Aerodrome Design Manual, Part 4.)

5.3.1.9 The intensity of runway lighting shall be adequate for the minimum conditions of visibility and ambient light in which use of the runway is intended, and compatible with that of the nearest section of the approach lighting system when provided.

Note.— While the lights of an approach lighting system may be of higher intensity than the runway lighting, it is good practice to avoid abrupt changes in intensity as these could give a pilot a false impression that the visibility is changing during approach.

5.3.1.10 Where a high-intensity lighting system is provided, a suitable intensity control shall be incorporated to allow for adjustment of the light intensity to meet the prevailing conditions. Separate intensity controls or other suitable methods shall be provided to ensure that the following systems, when installed, can be operated at compatible intensities:

— approach lighting system;
— runway edge lights;
— runway threshold lights;
— runway end lights;
— runway centre line lights;
— runway touchdown zone lights; and
— taxiway centre line lights.

5.3.1.11 On the perimeter of and within the ellipse defining the main beam in Appendix 2, Figures 2.1 to 2.10, the maximum light intensity value shall not be greater than three times the minimum light intensity value measured in accordance with Appendix 2, collective notes for Figures 2.1 to 2.11, Note 2.

5.3.1.12 On the perimeter of and within the rectangle defining the main beam in Appendix 2, Figures 2.12 to 2.20, the maximum light intensity value shall not be greater than three times the minimum light intensity value measured in accordance with Appendix 2, collective notes for Figures 2.12 to 2.21, Note 2.

5.3.2 Emergency lighting

Application

5.3.2.1 Recommendation.— At an aerodrome provided with runway lighting and without a secondary power supply, sufficient emergency lights should be conveniently available for installation on at least the primary runway in the event of failure of the normal lighting system.
Chapter 5

Note.— Emergency lighting may also be useful to mark obstacles or delineate taxiways and apron areas.

Location

5.3.2.2 Recommendation.— When installed on a runway the emergency lights should, as a minimum, conform to the configuration required for a non-instrument runway.

Characteristics

5.3.2.3 Recommendation.— The colour of the emergency lights should conform to the colour requirements for runway lighting, except that, where the provision of coloured lights at the threshold and the runway end is not practicable, all lights may be variable white or as close to variable white as practicable.

5.3.3 Aeronautical beacons

Application

5.3.3.1 Where operationally necessary an aerodrome beacon or an identification beacon shall be provided at each aerodrome intended for use at night.

5.3.3.2 The operational requirement shall be determined having regard to the requirements of the air traffic using the aerodrome, the conspicuity of the aerodrome features in relation to its surroundings and the installation of other visual and non-visual aids useful in locating the aerodrome.

Aerodrome beacon

5.3.3.3 An aerodrome beacon shall be provided at an aerodrome intended for use at night if one or more of the following conditions exist:

a) aircraft navigate predominantly by visual means;

b) reduced visibilities are frequent; or

c) it is difficult to locate the aerodrome from the air due to surrounding lights or terrain.

Location

5.3.3.4 The aerodrome beacon shall be located on or adjacent to the aerodrome in an area of low ambient background lighting.

5.3.3.5 Recommendation.— The location of the beacon should be such that the beacon is not shielded by objects in significant directions and does not dazzle a pilot approaching to land.

Characteristics

5.3.3.6 The aerodrome beacon shall show either coloured flashes alternating with white flashes, or white flashes only. The frequency of total flashes shall be from 20 to 30 per minute. Where used, the coloured flashes emitted by beacons at land aerodromes shall be green and coloured flashes emitted by beacons at water aerodromes shall be yellow. In the case of a combined water and land aerodrome, coloured flashes, if used, shall have the colour characteristics of whichever section of the aerodrome is designated as the principal facility.

5.3.3.7 The light from the beacon shall show at all angles of azimuth. The vertical light distribution shall extend upwards from an elevation of not more than 1° to an elevation determined by the appropriate authority to be sufficient to provide guidance at the maximum elevation at which the beacon is intended to be used and the effective intensity of the flash shall be not less than 2 000 cd.

Note.— At locations where a high ambient background lighting level cannot be avoided, the effective intensity of the flash may be required to be increased by a factor up to a value of 10.

Identification beacon

Application

5.3.3.8 An identification beacon shall be provided at an aerodrome which is intended for use at night and cannot be easily identified from the air by other means.

Location

5.3.3.9 The identification beacon shall be located on the aerodrome in an area of low ambient background lighting.

5.3.3.10 Recommendation.— The location of the beacon should be such that the beacon is not shielded by objects in significant directions and does not dazzle a pilot approaching to land.

Characteristics

5.3.3.11 An identification beacon at a land aerodrome shall show at all angles of azimuth. The vertical light distribution shall extend upwards from an elevation of not more than 1° to an elevation determined by the appropriate authority to be sufficient to provide guidance at the maximum elevation at which the beacon is intended to be used and the effective intensity of the flash shall be not less than 2 000 cd.
Annex 14 — Aerodromes

Note.— At locations where a high ambient background lighting level cannot be avoided, the effective intensity of the flash may be required to be increased by a factor up to a value of 10.

5.3.3.12 An identification beacon shall show flashing-green at a land aerodrome and flashing-yellow at a water aerodrome.

5.3.3.13 The identification characters shall be transmitted in the International Morse Code.

5.3.3.14 Recommendation.— The speed of transmission should be between six and eight words per minute, the corresponding range of duration of the Morse dots being from 0.15 to 0.2 seconds per dot.

5.3.4 Approach lighting systems

Note.— It is intended that existing lighting systems not conforming to the specifications in 5.3.4.21, 5.3.4.39, 5.3.9.10, 5.3.10.10, 5.3.10.11, 5.3.11.5, 5.3.12.8, 5.3.13.6 and 5.3.15.8 be replaced not later than 1 January 2005.

Application

5.3.4.1 Application

A.— Non-instrument runway

Recommendation.— Where physically practicable, a simple approach lighting system as specified in 5.3.4.2 to 5.3.4.9 should be provided to serve a non-instrument runway where the code number is 3 or 4 and intended for use at night, except when the runway is used only in conditions of good visibility, and sufficient guidance is provided by other visual aids.

Note.— A simple approach lighting system can also provide visual guidance by day.

B.— Non-precision approach runway

Where physically practicable, a simple approach lighting system as specified in 5.3.4.2 to 5.3.4.9 shall be provided to serve a non-precision approach runway, except when the runway is used only in conditions of good visibility or sufficient guidance is provided by other visual aids.

Note.— It is advisable to give consideration to the installation of a precision approach category I lighting system or to the addition of a runway lead-in lighting system.

C.— Precision approach runway category I

Where physically practicable, a precision approach category I lighting system as specified in 5.3.4.10 to 5.3.4.21 shall be provided to serve a precision approach runway category I.

D.— Precision approach runway categories II and III

A precision approach category II and III lighting system as specified in 5.3.4.22 to 5.3.4.39 shall be provided to serve a precision approach runway category II or III.

Simple approach lighting system

Location

5.3.4.2 A simple approach lighting system shall consist of a row of lights on the extended centre line of the runway extending, whenever possible, over a distance of not less than 420 m from the threshold with a row of lights forming a crossbar 18 m or 30 m in length at a distance of 300 m from the threshold.

5.3.4.3 The lights forming the crossbar shall be as nearly as practicable in a horizontal straight line at right angles to, and bisected by, the line of the centre line lights. The lights of the crossbar shall be spaced so as to produce a linear effect, except that, when a crossbar of 30 m is used, gaps may be left on each side of the centre line. These gaps shall be kept to a minimum to meet local requirements and each shall not exceed 6 m.

Note 1.— Spacings for the crossbar lights between 1 m and 4 m are in use. Gaps on each side of the centre line may improve directional guidance when approaches are made with a lateral error, and facilitate the movement of rescue and fire fighting vehicles.

Note 2.— See Attachment A, Section 11 for guidance on installation tolerances.

5.3.4.4 The lights forming the centre line shall be placed at longitudinal intervals of 60 m, except that, when it is desired to improve the guidance, an interval of 30 m may be used. The innermost light shall be located either 60 m or 30 m from the threshold, depending on the longitudinal interval selected for the centre line lights.

5.3.4.5 Recommendation.— If it is not physically possible to provide a centre line extending for a distance of 420 m from the threshold, it should be extended to 300 m so as to include the crossbar. If this is not possible, the centre line lights should be extended as far as practicable, and each centre line light should then consist of a barrette at least 3 m in length. Subject to the approach system having a crossbar at 300 m from the threshold, an additional crossbar may be provided at 150 m from the threshold.

5.3.4.6 The system shall lie as nearly as practicable in the horizontal plane passing through the threshold, provided that:
a) no object other than an ILS or MLS azimuth antenna shall protrude through the plane of the approach lights within a distance of 60 m from the centre line of the system; and

b) no light other than a light located within the central part of a crossbar or a centre line barrette (not their extremities) shall be screened from an approaching aircraft.

Any ILS or MLS azimuth antenna protruding through the plane of the lights shall be treated as an obstacle and marked and lighted accordingly.

**Characteristics**

5.3.4.7 The lights of a simple approach lighting system shall be fixed lights and the colour of the lights shall be such as to ensure that the system is readily distinguishable from other aeronautical ground lights, and from extraneous lighting if present. Each centre line light shall consist of either:

a) a single source; or

b) a barrette at least 3 m in length.

**Note 1.**—When the barrette as in b) is composed of lights approximating to point sources, a spacing of 1.5 m between adjacent lights in the barrette has been found satisfactory.

**Note 2.**—It may be advisable to use barrettes 4 m in length if it is anticipated that the simple approach lighting system will be developed into a precision approach lighting system.

**Note 3.**—At locations where identification of the simple approach lighting system is difficult at night due to surrounding lights, sequence flashing lights installed in the outer portion of the system may resolve this problem.

5.3.4.8 **Recommendation.**—Where provided for a non-instrument runway, the lights should show at all angles in azimuth necessary to a pilot on base leg and final approach. The intensity of the lights should be adequate for all conditions of visibility and ambient light for which the system has been provided.

5.3.4.9 **Recommendation.**—Where provided for a non-precision approach runway, the lights should show at all angles in azimuth necessary to the pilot of an aircraft which on final approach does not deviate by an abnormal amount from the path defined by the non-visual aid. The lights should be designed to provide guidance during both day and night in the most adverse conditions of visibility and ambient light for which it is intended that the system should remain usable.

### Precision approach category I lighting system

**Location**

5.3.4.10 A precision approach category I lighting system shall consist of a row of lights on the extended centre line of the runway extending, wherever possible, over a distance of 900 m from the runway threshold with a row of lights forming a crossbar 30 m in length at a distance of 300 m from the runway threshold.

**Note.**—The installation of an approach lighting system of less than 900 m in length may result in operational limitations on the use of the runway. See Attachment A, Section 11.

5.3.4.11 The lights forming the crossbar shall be as nearly as practicable in a horizontal straight line at right angles to, and bisected by, the line of the centre line lights. The lights of the crossbar shall be spaced so as to produce a linear effect, except that gaps may be left on each side of the centre line. These gaps shall be kept to a minimum to meet local requirements and each shall not exceed 6 m.

**Note 1.**—Spacing of the crossbar lights between 1 m and 4 m are in use. Gaps on each side of the centre line may improve directional guidance when approaches are made with a lateral error, and facilitate the movement of rescue and fire fighting vehicles.

**Note 2.**—See Attachment A, Section 11 for guidance on installation tolerances.

5.3.4.12 The lights forming the centre line shall be placed at longitudinal intervals of 30 m with the innermost light located 30 m from the threshold.

5.3.4.13 The system shall lie as nearly as practicable in the horizontal plane passing through the threshold, provided that:

a) no object other than an ILS or MLS azimuth antenna shall protrude through the plane of the approach lights within a distance of 60 m from the centre line of the system; and

b) no light other than a light located within the central part of a crossbar or a centre line barrette (not their extremities) shall be screened from an approaching aircraft.

Any ILS or MLS azimuth antenna protruding through the plane of the lights shall be treated as an obstacle and marked and lighted accordingly.

**Characteristics**

5.3.4.14 The centre line and crossbar lights of a precision approach category I lighting system shall be fixed lights showing variable white. Each centre line light position shall consist of either:
a) a single light source in the innermost 300 m of the
centre line, two light sources in the central 300 m of the
centre line and three light sources in the outer 300 m of
the centre line to provide distance information; or

b) a barrette.

5.3.4.15 Where the serviceability level of the approach
lights specified as a maintenance objective in 9.4.29 can be
demonstrated, each centre line light position may consist of
either:

a) a single light source; or

b) a barrette.

5.3.4.16 The barrettes shall be at least 4 m in length.
When barrettes are composed of lights approximating to point
sources, the lights shall be uniformly spaced at intervals of not
more than 1.5 m.

5.3.4.17 **Recommendation.**— **If the centre line consists of barrettes as described in 5.3.4.14 b) or 5.3.4.15 b), each barrette should be supplemented by a capacitor discharge light, except where such lighting is considered unnecessary taking into account the characteristics of the system and the nature of the meteorological conditions.**

5.3.4.18 Each capacitor discharge light as described in
5.3.4.17 shall be flashed twice a second in sequence,
beginning with the outermost light and progressing toward the
threshold to the innermost light of the system. The design of
the electrical circuit shall be such that these lights can be
operated independently of the other lights of the approach
lighting system.

5.3.4.19 If the centre line consists of lights as described in
5.3.4.14 a) or 5.3.4.15 a), additional crossbars of lights to the
crossbar provided at 300 m from the threshold shall be
provided at 150 m, 450 m, 600 m and 750 m from the
threshold. The lights forming each crossbar shall be as nearly
as practicable in a horizontal straight line at right angles to,
and bisected by, the line of the centre line lights. The lights
shall be spaced so as to produce a linear effect, except that
gaps may be left on each side of the centre line. These gaps
shall be kept to a minimum to meet local requirements and
each shall not exceed 6 m.

**Note.— See Attachment A, Section 11 for detailed configuration.**

5.3.4.20 Where the additional crossbars described in
5.3.4.19 are incorporated in the system, the outer ends of the
crossbars shall lie on two straight lines that either are parallel
to the line of the centre line lights or converge to meet the
runway centre line 300 m from threshold.

5.3.4.21 The lights shall be in accordance with the speci-
fications of Appendix 2, Figure 2.1.

**Note.— The flight path envelopes used in the design of these lights are given in Attachment A, Figure A-4.**

**Precision approach category II and III lighting system**

**Location**

5.3.4.22 The approach lighting system shall consist of a row of lights on the extended centre line of the runway, extending, wherever possible, over a distance of 900 m from the runway threshold. In addition, the system shall have two side rows of lights, extending 270 m from the threshold, and two crossbars, one at 150 m and one at 300 m from the threshold, all as shown in Figure 5-10. Where the serviceability level of the approach lights specified as maintenance objectives in 9.4.26 can be demonstrated, the system may have two side rows of lights, extending 240 m from the threshold, and two crossbars, one at 150 m and one at 300 m from the threshold, all as shown in Figure 5-11.

**Note.— The length of 900 m is based on providing guidance for operations under category I, II and III conditions. Reduced lengths may support category II and III operations but may impose limitations on category I operations. See Attachment A, Section 11.**

5.3.4.23 The lights forming the centre line shall be placed at longitudinal intervals of 30 m with the innermost lights located 30 m from the threshold.

5.3.4.24 The lights forming the side rows shall be placed on each side of the centre line, at a longitudinal spacing equal to that of the centre line lights and with the first light located 30 m from the threshold. Where the serviceability level of the approach lights specified as maintenance objectives in 9.4.26 can be demonstrated, lights forming the side rows may be placed on each side of the centre line, at a longitudinal spacing of 60 m with the first light located 60 m from the threshold. The lateral spacing (or gauge) between the innermost lights of the side rows shall be not less than 18 m nor more than 22.5 m, and preferably 18 m, but in any event shall be equal to that of the touchdown zone lights.

5.3.4.25 The crossbar provided at 150 m from the
threshold shall fill in the gaps between the centre line and side row lights.

5.3.4.26 The crossbar provided at 300 m from the
threshold shall extend on both sides of the centre line lights to
a distance of 15 m from the centre line.

5.3.4.27 If the centre line beyond a distance of 300 m from the threshold consists of lights as described in 5.3.4.31 b) or 5.3.4.32 b), additional crossbars of lights shall be provided at 450 m, 600 m and 750 m from the threshold.
5.3.4.28 Where the additional crossbars described in
5.3.4.27 are incorporated in the system, the outer ends of these
crossbars shall lie on two straight lines that either are parallel
to the centre line or converge to meet the runway centre line
300 m from the threshold.

5.3.4.29 The system shall lie as nearly as practicable in the
horizontal plane passing through the threshold, provided that:

a) no object other than an ILS or MLS azimuth antenna shall
protrude through the plane of the approach lights within a
distance of 60 m from the centre line of the system; and

b) no light other than a light located within the central part of
a crossbar or a centre line barrette (not their extremities)
shall be screened from an approaching aircraft.

Any ILS or MLS azimuth antenna protruding through the
plane of the lights shall be treated as an obstacle and marked
and lighted accordingly.

Characteristics

5.3.4.30 The centre line of a precision approach
category II and III lighting system for the first 300 m from the
threshold shall consist of barrettes showing variable white,
except that, where the threshold is displaced 300 m or more,
the centre line may consist of single light sources showing
variable white. Where the serviceability level of the approach
lights specified as maintenance objectives in 9.4.26 can be
demonstrated, the centre line of a precision approach category
II and III lighting system for the first 300 m from the threshold
may consist of either:

a) barrettes, where the centre line beyond 300 m from the
threshold consists of barrettes as described in
5.3.4.32 a); or

b) alternate single light sources and barrettes, where the
centre line beyond 300 m from the threshold consists of
single light sources as described in 5.3.4.32 b), with the
innermost single light source located 30 m and the
innermost barrette located 60 m from the threshold; or

c) single light sources where the threshold is displaced
300 m or more;

all of which shall show variable white.

5.3.4.31 Beyond 300 m from the threshold each centre
line light position shall consist of either:

a) a barrette as used on the inner 300 m; or

b) two light sources in the central 300 m of the centre line
and three light sources in the outer 300 m of the centre
line;

all of which shall show variable white.

5.3.4.32 Where the serviceability level of the approach
lights specified as maintenance objectives in 9.4.26 can be
demonstrated, beyond 300 m from the threshold each centre
line light position may consist of either:

a) a barrette; or

b) a single light source;

all of which shall show variable white.

5.3.4.33 The barrettes shall be at least 4 m in length.
When barrettes are composed of lights approximating to point
sources, the lights shall be uniformly spaced at intervals of not
more than 1.5 m.

5.3.4.34 Recommendation.— If the centre line beyond
300 m from the threshold consists of barrettes as described in
5.3.4.31 a) or 5.3.4.32 a), each barrette beyond 300 m should
be supplemented by a capacitor discharge light, except where
such lighting is considered unnecessary taking into account
the characteristics of the system and the nature of the
meteorological conditions.

5.3.4.35 Each capacitor discharge light shall be flashed
twice a second in sequence, beginning with the outermost light
and progressing toward the threshold to the innermost light of
the system. The design of the electrical circuit shall be such
that these lights can be operated independently of the other
lights of the approach lighting system.

5.3.4.36 The side row shall consist of barrettes showing
red. The length of a side row barrette and the spacing of its
lights shall be equal to those of the touchdown zone light
barrettes.

5.3.4.37 The lights forming the crossbars shall be fixed
lights showing variable white. The lights shall be uniformly
spaced at intervals of not more than 2.7 m.

5.3.4.38 The intensity of the red lights shall be compat-
ible with the intensity of the white lights.

5.3.4.39 The lights shall be in accordance with the speci-
fications of Appendix 2, Figures 2.1 and 2.2.

Note.— The flight path envelopes used in the design of
these lights are given in Attachment A, Figure A-4.

5.3.5 Visual approach slope indicator systems

Application

5.3.5.1 A visual approach slope indicator system shall be
provided to serve the approach to a runway whether or not the
runway is served by other visual approach aids or by non-
visual aids, where one or more of the following conditions
exist:
Figure 5-10. Inner 300 m approach and runway lighting for precision approach runways categories II and III
Figure 5-11. Inner 300 m approach and runway lighting for precision approach runways categories II and III where the serviceability levels of the lights specified as maintenance objectives in Section 9.4 can be demonstrated.
a) the runway is used by turbojet or other aeroplanes with similar approach guidance requirements;

b) the pilot of any type of aeroplane may have difficulty in judging the approach due to:

1) inadequate visual guidance such as is experienced during an approach over water or featureless terrain by day or in the absence of sufficient extraneous lights in the approach area by night, or

2) misleading information such as is produced by deceptive surrounding terrain or runway slopes;

c) the presence of objects in the approach area may involve serious hazard if an aeroplane descends below the normal approach path, particularly if there are no non-visual or other visual aids to give warning of such objects;

d) physical conditions at either end of the runway present a serious hazard in the event of an aeroplane undershooting or overrunning the runway; and

e) terrain or prevalent meteorological conditions are such that the aeroplane may be subjected to unusual turbulence during approach.

Note.— Guidance on the priority of installation of visual approach slope indicator systems is contained in Attachment A, Section 12.

5.3.5.2 The standard visual approach slope indicator systems shall consist of the following:

a) T-VASIS and AT-VASIS conforming to the specifications contained in 5.3.5.6 to 5.3.5.22 inclusive;

b) PAPI and APAPI systems conforming to the specifications contained in 5.3.5.23 to 5.3.5.40 inclusive;
as shown in Figure 5-12.

5.3.5.3 PAPI, T-VASIS or AT-VASIS shall be provided where the code number is 3 or 4 when one or more of the conditions specified in 5.3.5.1 exist.

5.3.5.4 PAPI or APAPI shall be provided where the code number is 1 or 2 when one or more of the conditions specified in 5.3.5.1 exist.

5.3.5.5 Recommendation.— Where a runway threshold is temporarily displaced from the normal position and one or more of the conditions specified in 5.3.5.1 exist, a PAPI should be provided except that where the code number is 1 or 2 an APAPI may be provided.

T-VASIS and AT-VASIS

Description

5.3.5.6 The T-VASIS shall consist of twenty light units symmetrically disposed about the runway centre line in the form of two wing bars of four light units each, with bisecting longitudinal lines of six lights, as shown in Figure 5-13.
INSTALLATION TOLERANCES

The appropriate authority may:

a) vary the nominal eye height over the threshold of the on-slope signal between the limits of 12 m and 16 m, except in cases where a standard ILS glide path and/or MLS minimum glide path is available; the height over threshold should be varied to avoid any conflict between the visual approach slope indications and the usable portion of the ILS glide path and/or MLS minimum glide path indications;

b) vary the longitudinal distance between individual light units or the overall length of the system by not more than 10 per cent;

c) vary the lateral displacement of the system from the runway edge by not more than ± 3 m:

Note.— The system must be symmetrically displaced about the runway centre line.

d) where there is a longitudinal slope of the ground, adjust the longitudinal distance of a light unit to compensate for its difference in level from that of the threshold; and

e) where there is a transverse slope in the ground, adjust the longitudinal distance of two light units or two wing bars to compensate for the difference in level between them as necessary to meet the requirements of 5.3.5.16.

The distance between the wing bar and the threshold is based on an approach slope of 3° to a level runway with a nominal eye height over the threshold of 15 m. In practice, the threshold to wing bar distance is determined by:

a) the selected approach slope;

b) the longitudinal slope of the runway; and

c) the selected nominal eye height over the threshold.
5.3.5.7 The AT-VASIS shall consist of ten light units arranged on one side of the runway in the form of a single wing bar of four light units with a bisecting longitudinal line of six lights.

5.3.5.8 The light units shall be constructed and arranged in such a manner that the pilot of an aeroplane during an approach will:

a) when above the approach slope, see the wing bar(s) white, and one, two or three fly-down lights, the more fly-down lights being visible the higher the pilot is above the approach slope;

b) when on the approach slope, see the wing bar(s) white; and

c) when below the approach slope, see the wing bar(s) and one, two or three fly-up lights white, the more fly-up lights being visible the lower the pilot is below the approach slope; and when well below the approach slope, see the wing bar(s) and the three fly-up lights red.

When on or above the approach slope, no light shall be visible from the fly-up light units; when on or below the approach slope, no light shall be visible from the fly-down light units.

Siting

5.3.5.9 The light units shall be located as shown in Figure 5-13, subject to the installation tolerances given therein.

Note.— The siting of T-VASIS will provide, for a 3° slope and a nominal eye height over the threshold of 15 m (see 5.3.5.6 and 5.3.5.19), a pilot’s eye height over threshold of 13 m to 17 m when only the wing bar lights are visible. If increased eye height at the threshold is required (to provide adequate wheel clearance), then the approaches may be flown with one or more fly-down lights visible. The pilot’s eye height over the threshold is then of the following order:

- Wing bar lights and one fly-down light visible: 17 m to 22 m
- Wing bar lights and two fly-down lights visible: 22 m to 28 m
- Wing bar lights and three fly-down lights visible: 28 m to 54 m

Characteristics of the light units

5.3.5.10 The systems shall be suitable for both day and night operations.

5.3.5.11 The light distribution of the beam of each light unit shall be of fan shape showing over a wide arc in azimuth in the approach direction. The wing bar light units shall produce a beam of white light from 1°54’ vertical angle up to 6° vertical angle and a beam of red light from 0° to 1°54’ vertical angle. The fly-down light units shall produce a white beam extending from an elevation of 6° down to approximately the approach slope, where it shall have a sharp cut-off. The fly-up light units shall produce a white beam from approximately the approach slope down to 1°54’ vertical angle and a red beam below a 1°54’ vertical angle. The angle of the top of the red beam in the wing bar units and fly-up units may be increased to comply with 5.3.5.21.

5.3.5.12 The light intensity distribution of the fly-down, wing bar and fly-up light units shall be as shown in Appendix 2, Figure 2-22.

5.3.5.13 The colour transition from red to white in the vertical plane shall be such as to appear to an observer, at a distance of not less than 300 m, to occur over a vertical angle of not more than 15’.

5.3.5.14 At full intensity the red light shall have a Y coordinate not exceeding 0.320.

5.3.5.15 A suitable intensity control shall be provided to allow adjustments to meet the prevailing conditions and to avoid dazzling the pilot during approach and landing.

5.3.5.16 The light units forming the wing bars, or the light units forming a fly-down or a fly-up matched pair, shall be mounted so as to appear to the pilot of an approaching aeroplane to be substantially in a horizontal line. The light units shall be mounted as low as possible and shall be frangible.

5.3.5.17 The light units shall be so designed that deposits of condensation, dirt, etc., on optically transmitting or reflecting surfaces shall interfere to the least possible extent with the light signals and shall in no way affect the elevation of the beams or the contrast between the red and white signals. The construction of the light units shall be such as to minimize the probability of the slots being wholly or partially blocked by snow or ice where these conditions are likely to be encountered.

Approach slope and elevation setting of light beams

5.3.5.18 The approach slope shall be appropriate for use by the aeroplanes using the approach.

5.3.5.19 When the runway on which a T-VASIS is provided is equipped with an ILS and/or MLS, the siting and elevations of the light units shall be such that the visual approach slope conforms as closely as possible with the glide path of the ILS and/or the minimum glide path of the MLS, as appropriate.
5.3.5.20 The elevation of the beams of the wing bar light units on both sides of the runway shall be the same. The elevation of the top of the beam of the fly-up light unit nearest to each wing bar, and that of the bottom of the beam of the fly-down light unit nearest to each wing bar, shall be equal and shall correspond to the approach slope. The cut-off angle of the top of the beams of successive fly-up light units shall decrease by 5' of arc in angle of elevation at each successive unit away from the wing bar. The cut-in angle of the bottom of the beam of the fly-down light units shall increase by 7' of arc at each successive unit away from the wing bar (see Figure 5-14).

5.3.5.21 The elevation setting of the top of the red light beams of the wing bar and fly-up light units shall be such that, during an approach, the pilot of an aeroplane to whom the wing bar and three fly-up light units are visible would clear all objects in the approach area by a safe margin if any such light did not appear red.

5.3.5.22 The azimuth spread of the light beam shall be suitably restricted where an object located outside the obstacle protection surface of the system, but within the lateral limits of its light beam, is found to extend above the plane of the obstacle protection surface and an aeronautical study indicates that the object could adversely affect the safety of operations. The extent of the restriction shall be such that the object remains outside the confines of the light beam.

Note.— See 5.3.5.41 to 5.3.5.45 concerning the related obstacle protection surface.

PAPI and APAPI

Description

5.3.5.23 The PAPI system shall consist of a wing bar of 4 sharp transition multi-lamp (or paired single lamp) units equally spaced. The system shall be located on the left side of the runway unless it is physically impracticable to do so.

Note.— Where a runway is used by aircraft requiring visual roll guidance which is not provided by other external means, then a second wing bar may be provided on the opposite side of the runway.

5.3.5.24 The APAPI system shall consist of a wing bar of 2 sharp transition multi-lamp (or paired single lamp) units. The system shall be located on the left side of the runway unless it is physically impracticable to do so.

Note.— Where a runway is used by aircraft requiring visual roll guidance which is not provided by other external means, then a second wing bar may be provided on the opposite side of the runway.

5.3.5.25 The wing bar of a PAPI shall be constructed and arranged in such a manner that a pilot making an approach will:

a) when on or close to the approach slope, see the two units nearest the runway as red and the two units farthest from the runway as white;

b) when above the approach slope, see the one unit nearest the runway as red and the three units farthest from the runway as white; and when further above the approach slope, see all the units as white; and

c) when below the approach slope, see the three units nearest the runway as red and the unit farthest from the runway as white, and when further below the approach slope, see all the units as red.

5.3.5.26 The wing bar of an APAPI shall be constructed and arranged in such a manner that a pilot making an approach will:
Annex I4 - Aerodromes

5.3.5.27 The light units shall be located as in the basic configuration illustrated in Figure 5-15, subject to the installation tolerances given therein. The units forming a wing bar shall be mounted so as to appear to the pilot of an approaching aeroplane to be substantially in a horizontal line. The light units shall be mounted as low as possible and shall be frangible.

Characteristics of the light units

5.3.5.28 The system shall be suitable for both day and night operations.

5.3.5.29 The colour transition from red to white in the vertical plane shall be such as to appear to an observer, at a distance of not less than 300 m, to occur within a vertical angle of not more than 3°.

5.3.5.30 At full intensity the red light shall have a Y coordinate not exceeding 0.320.

5.3.5.31 The light intensity distribution of the light units shall be as shown in Appendix 2, Figure 2.23.

Note — See the Aerodrome Design Manual, Part 4 for additional guidance on the characteristics of light units.

5.3.5.32 Suitable intensity control shall be provided so as to allow adjustment to meet the prevailing conditions and to avoid dazzling the pilot during approach and landing.

5.3.5.33 Each light unit shall be capable of adjustment in elevation so that the lower limit of the white part of the beam may be fixed at any desired angle of elevation between 1°30' and at least 4°30' above the horizontal.

5.3.5.34 The light units shall be so designed that deposits of condensation, snow, ice, dirt, etc., on optically transmitting or reflecting surfaces shall interfere to the least possible extent with the light signals and shall not affect the contrast between the red and white signals and the elevation of the transition sector.

Approach slope and elevation setting of light units

5.3.5.35 The approach slope as defined in Figure 5-16 shall be appropriate for use by the aeroplanes using the approach.

5.3.5.36 When the runway is equipped with an ILS and/or MLS, the siting and the angle of elevation of the light units shall be such that the visual approach slope conforms as closely as possible with the glide path of the ILS and/or the minimum glide path of the MLS, as appropriate.

5.3.5.37 The angle of elevation settings of the light units in a PAPI wing bar shall be such that, during an approach, the pilot of an aeroplane observing a signal of one white and three reds will clear all objects in the approach area by a safe margin.

5.3.5.38 The angle of elevation settings of the light units in an APAPI wing bar shall be such that, during an approach, the pilot of an aeroplane observing the lowest onslope signal, i.e. one white and one red, will clear all objects in the approach area by a safe margin.

5.3.5.39 The azimuth spread of the light beam shall be suitably restricted where an object located outside the obstacle protection surface of the PAPI or APAPI system, but within the lateral limits of its light beam, is found to extend above the plane of the obstacle protection surface and an aeronautical study indicates that the object could adversely affect the safety of operations. The extent of the restriction shall be such that the object remains outside the confines of the light beam.

Note.— See 5.3.5.41 to 5.3.5.45 concerning the related obstacle protection surface.

5.3.5.40 Where wing bars are installed on each side of the runway to provide roll guidance, corresponding units shall be set at the same angle so that the signals of each wing bar change symmetrically at the same time.

Obstacle protection surface

Note.— The following specifications apply to T-VASIS, AT-VASIS, PAPI and APAPI.

5.3.5.41 An obstacle protection surface shall be established when it is intended to provide a visual approach slope indicator system.

5.3.5.42 The characteristics of the obstacle protection surface, i.e. origin, divergence, length and slope shall correspond to those specified in the relevant column of Table 5-3 and in Figure 5-17.
Installation Tolerances

a) Where a PAPI or APAPI is installed on a runway not equipped with an ILS or MLS, the distance $D_t$ shall be calculated to ensure that the lowest height at which a pilot will see a correct approach path indication (Figure 5-16, angle $B$ for a PAPI and angle $A$ for an APAPI) provides the wheel clearance over the threshold specified in Table 5-2 for the most demanding amongst aeroplanes regularly using the runway.

b) Where a PAPI or APAPI is installed on a runway equipped with an ILS and/or MLS, the distance $D_t$ shall be calculated to provide the optimum compatibility between the visual and non-visual aids for the range of eye-to-antenna heights of the aeroplanes regularly using the runway. The distance shall be equal to that between the threshold and the effective origin of the ILS glide path or MLS minimum glide path, as appropriate, plus a correction factor for the variation of eye-to-antenna heights of the aeroplanes concerned. The correction factor is obtained by multiplying the average eye-to-antenna height of those aeroplanes by the cotangent of the approach angle. However, the distance shall be such that in no case will the wheel clearance over the threshold be lower than that specified in column (3) of Table 5-2.

Note.— See Section 5.2.5 for specifications on aiming point marking. Guidance on the harmonization of PAPI, ILS and/or MLS signals is contained in the Aerodrome Design Manual, Part 4.

c) If a wheel clearance, greater than that specified in a) above is required for specific aircraft, this can be achieved by increasing $D_t$.

d) Distance $D_t$ shall be adjusted to compensate for differences in elevation between the lens centres of the light units and the threshold.

e) To ensure that units are mounted as low as possible and to allow for any transverse slope, small height adjustments of up to 5 cm between units are acceptable. A lateral gradient not greater than 1.25 per cent can be accepted provided it is uniformly applied across the units.

f) A spacing of 6 m ($\pm 1$ m) between PAPI units should be used on code numbers 1 and 2. In such an event, the inner PAPI unit shall be located not less than 10 m ($\pm 1$ m) from the runway edge.

Note.— Reducing the spacing between light units results in a reduction in usable range of the system.

g) The lateral spacing between APAPI units may be increased to 9 m ($\pm 1$ m) if greater range is required or later conversion to a full PAPI is anticipated. In the latter case, the inner APAPI unit shall be located 15 m ($\pm 1$ m) from the runway edge.
The height of the pilot's eye above the aircraft's ILS glide path/MLS antenna varies with the type of aeroplane and approach attitude. Harmonization of the PAPI signal and ILS glide path and/or MLS minimum glide path to a point closer to the threshold may be achieved by increasing the on-course sector from 20° to 30°. The setting angles for a 3° glide slope would then be 2°25', 2°45', 3°15' and 3°35'.

A — 3° PAPI ILLUSTRATED

B — 3° APAPI ILLUSTRATED

Figure 5-16. Light beams and angle of elevation setting of PAPI and APAPI
Table 5-2. Wheel clearance over threshold for PAPI and APAPI

<table>
<thead>
<tr>
<th>Eye-to-wheel height of aeroplane in the approach configuration(^a)</th>
<th>Desired wheel clearance (metres)(^{b,c})</th>
<th>Minimum wheel clearance (metres)(^d)</th>
</tr>
</thead>
<tbody>
<tr>
<td>up to but not including 3 m</td>
<td>6</td>
<td>3(^e)</td>
</tr>
<tr>
<td>3 m up to but not including 5 m</td>
<td>9</td>
<td>4</td>
</tr>
<tr>
<td>5 m up to but not including 8 m</td>
<td>9</td>
<td>5</td>
</tr>
<tr>
<td>8 m up to but not including 14 m</td>
<td>9</td>
<td>6</td>
</tr>
</tbody>
</table>

\(^a\) In selecting the eye-to-wheel height group, only aeroplanes meant to use the system on a regular basis shall be considered. The most demanding amongst such aeroplanes shall determine the eye-to-wheel height group.

\(^b\) Where practicable the desired wheel clearances shown in column (2) shall be provided.

\(^c\) The wheel clearances in column (2) may be reduced to no less than those in column (3) where an aeronautical study indicates that such reduced wheel clearances are acceptable.

\(^d\) When a reduced wheel clearance is provided at a displaced threshold it shall be ensured that the corresponding desired wheel clearance specified in column (2) will be available when an aeroplane at the top end of the eye-to-wheel height group chosen overflies the extremity of the runway.

\(^e\) This wheel clearance may be reduced to 1.5 m on runways used mainly by light-weight non-turbo-jet aeroplanes.

Table 5-3. Dimensions and slopes of the obstacle protection surface

<table>
<thead>
<tr>
<th>Runway type/code number</th>
<th>Non-instrument Code number</th>
<th>Instrument Code number</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>1</td>
<td>2</td>
</tr>
<tr>
<td><strong>Surface dimensions</strong></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Length of inner edge</td>
<td>60 m</td>
<td>80 m(^a)</td>
</tr>
<tr>
<td>Distance from threshold</td>
<td>30 m</td>
<td>60 m</td>
</tr>
<tr>
<td>Divergence (each side)</td>
<td>10%</td>
<td>10%</td>
</tr>
<tr>
<td>Total length</td>
<td>7 500 m</td>
<td>7 500 m(^b)</td>
</tr>
</tbody>
</table>

**Slope**

<table>
<thead>
<tr>
<th></th>
<th>Non-instrument Code number</th>
<th>Instrument Code number</th>
</tr>
</thead>
<tbody>
<tr>
<td>a) T-VASIS and AT-VASIS</td>
<td>–</td>
<td>c 1.9° 1.9° 1.9°</td>
</tr>
<tr>
<td>b) PAPI(^d)</td>
<td>–</td>
<td>A–0.57° A–0.57° A–0.57°</td>
</tr>
<tr>
<td>c) APAPI(^d)</td>
<td>A–0.9°</td>
<td>A–0.9°</td>
</tr>
</tbody>
</table>

\(^a\) This length is to be increased to 150 m for a T-VASIS or AT-VASIS.

\(^b\) This length is to be increased to 15 000 m for a T-VASIS or AT-VASIS.

\(^c\) No slope has been specified if a system is unlikely to be used on runway type/code number indicated.

\(^d\) Angles as indicated in Figure 5-16.
5.3.5.43 New objects or extensions of existing objects shall not be permitted above an obstacle protection surface except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object.

Note.—Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6.

5.3.5.44 Existing objects above an obstacle protection surface shall be removed except when, in the opinion of the appropriate authority, the object is shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety of operations of aeroplanes.

5.3.5.45 Where an aeronautical study indicates that an existing object extending above an obstacle protection surface could adversely affect the safety of operations of aeroplanes one or more of the following measures shall be taken:

a) suitably raise the approach slope of the system;

b) reduce the azimuth spread of the system so that the object is outside the confines of the beam;

c) displace the axis of the system and its associated obstacle protection surface by no more than 5°;

d) suitably displace the threshold; and
5.3.6 Circling guidance lights

**Application**

5.3.6.1 **Recommendation.**— Circling guidance lights should be provided when existing approach and runway lighting systems do not satisfactorily permit identification of the runway and/or approach area to a circling aircraft in the conditions for which it is intended the runway be used for circling approaches.

**Location**

5.3.6.2 **Recommendation.**— The location and number of circling guidance lights should be adequate to enable a pilot, as appropriate, to:

a) join the downwind leg or align and adjust the aircraft’s track to the runway at a required distance from it and to distinguish the threshold in passing; and

b) keep in sight the runway threshold and/or other features which will make it possible to judge the turn on to base leg and final approach, taking into account the guidance provided by other visual aids.

5.3.6.3 **Recommendation.**— Circling guidance lights should consist of:

a) lights indicating the extended centre line of the runway and/or parts of any approach lighting system; or

b) lights indicating the position of the runway threshold; or

c) lights indicating the direction or location of the runway; or a combination of such lights as is appropriate to the runway under consideration.


**Characteristics**

5.3.6.4 **Recommendation.**— Circling guidance lights should be fixed or flashing lights of an intensity and beam spread adequate for the conditions of visibility and ambient light in which it is intended to make visual circling approaches. The flashing lights should be white, and the steady lights either white or gaseous discharge lights.

5.3.6.5 **Recommendation.**— The lights should be designed and be installed in such a manner that they will not dazzle or confuse a pilot when approaching to land, taking off or taxiing.

5.3.7 Runway lead-in lighting systems

**Application**

5.3.7.1 **Recommendation.**— A runway lead-in lighting system should be provided where it is desired to provide visual guidance along a specific approach path, for reasons such as avoiding hazardous terrain or for purposes of noise abatement.

Note.— Guidance on providing lead-in lighting systems is given in the Aerodrome Design Manual, Part 4.

**Location**

5.3.7.2 **Recommendation.**— A runway lead-in lighting system should consist of groups of lights positioned so as to define the desired approach path and so that one group may be sighted from the preceding group. The interval between adjacent groups should not exceed approximately 1600 m.

Note.— Runway lead-in lighting systems may be curved, straight or a combination thereof.

5.3.7.3 **Recommendation.**— A runway lead-in lighting system should extend from a point as determined by the appropriate authority, up to a point where the approach lighting system, if provided, or the runway or the runway lighting system is in view.

**Characteristics**

5.3.7.4 **Recommendation.**— Each group of lights of a runway lead-in lighting system should consist of at least three flashing lights in a linear or cluster configuration. The system may be augmented by steady burning lights where such lights would assist in identifying the system.

5.3.7.5 **Recommendation.**— The flashing lights should be white, and the steady burning lights gaseous discharge lights.

5.3.7.6 **Recommendation.**— Where practicable, the flashing lights in each group should flash in sequence towards the runway.
5.3.8 Runway threshold identification lights

**Application**

5.3.8.1 Recommendation.— Runway threshold identification lights should be installed:

a) at the threshold of a non-precision approach runway when additional threshold conspicuity is necessary or where it is not practicable to provide other approach lighting aids; and

b) where a runway threshold is permanently displaced from the runway extremity or temporarily displaced from the normal position and additional threshold conspicuity is necessary.

**Location**

5.3.8.2 Runway threshold identification lights shall be located symmetrically about the runway centre line, in line with the threshold and approximately 10 m outside each line of runway edge lights.

**Characteristics**

5.3.8.3 Recommendation.— Runway threshold identification lights should be flashing white lights with a flash frequency between 60 and 120 per minute.

5.3.8.4 The lights shall be visible only in the direction of approach to the runway.

5.3.9 Runway edge lights

**Application**

5.3.9.1 Runway edge lights shall be provided for a runway intended for use at night or for a precision approach runway intended for use by day or night.

5.3.9.2 Recommendation.— Runway edge lights should be provided on a runway intended for take-off with an operating minimum below an RVR of the order of 800 m by day.

**Location**

5.3.9.3 Runway edge lights shall be placed along the full length of the runway and shall be in two parallel rows equi-distant from the centre line.

5.3.9.4 Runway edge lights shall be placed along the edges of the area declared for use as the runway or outside the edges of the area at a distance of not more than 3 m.

5.3.9.5 Recommendation.— Where the width of the area which could be declared as runway exceeds 60 m, the distance between the rows of lights should be determined taking into account the nature of the operations, the light distribution characteristics of the runway edge lights, and other visual aids serving the runway.

5.3.9.6 The lights shall be uniformly spaced in rows at intervals of not more than 60 m for an instrument runway, and at intervals of not more than 100 m for a non-instrument runway. The lights on opposite sides of the runway axis shall be on lines at right angles to that axis. At intersections of runways, lights may be spaced irregularly or omitted, provided that adequate guidance remains available to the pilot.

**Characteristics**

5.3.9.7 Runway edge lights shall be fixed lights showing variable white, except that:

a) in the case of a displaced threshold, the lights between the beginning of the runway and the displaced threshold shall show red in the approach direction; and

b) a section of the lights 600 m or one-third of the runway length, whichever is the less, at the remote end of the runway from the end at which the take-off run is started, may show yellow.

5.3.9.8 The runway edge lights shall show at all angles in azimuth necessary to provide guidance to a pilot landing or taking off in either direction. When the runway edge lights are intended to provide circling guidance, they shall show at all angles in azimuth (see 5.3.6.1).

5.3.9.9 In all angles of azimuth required in 5.3.9.8, runway edge lights shall show at angles up to 15° above the horizontal with an intensity adequate for the conditions of visibility and ambient light in which use of the runway for take-off or landing is intended. In any case, the intensity shall be at least 50 cd except that at an aerodrome without extraneous lighting the intensity of the lights may be reduced to not less than 25 cd to avoid dazzling the pilot.

5.3.9.10 Runway edge lights on a precision approach runway shall be in accordance with the specifications of Appendix 2, Figure 2-9 or 2-10.

5.3.10 Runway threshold and wing bar lights

(see Figure 5-18)

**Application of runway threshold lights**

5.3.10.1 Runway threshold lights shall be provided for a runway equipped with runway edge lights except on a non-instrument or non-precision approach runway where the threshold is displaced and wing bar lights are provided.
Figure 5-10. Arrangement of runway threshold and runway end lights.
Location of runway threshold lights

5.3.10.2 When a threshold is at the extremity of a runway, the threshold lights shall be placed in a row at right angles to the runway axis as near to the extremity of the runway as possible and, in any case, not more than 3 m outside the extremity.

5.3.10.3 When a threshold is displaced from the extremity of a runway, threshold lights shall be placed in a row at right angles to the runway axis at the displaced threshold.

5.3.10.4 Threshold lighting shall consist of:

a) on a non-instrument or non-precision approach runway, at least six lights;

b) on a precision approach runway category I, at least the number of lights that would be required if the lights were uniformly spaced at intervals of 3 m between the rows of runway edge lights; and

c) on a precision approach runway category II or III, lights uniformly spaced between the rows of runway edge lights at intervals of not more than 3 m.

5.3.10.5 Recommendation.— The lights prescribed in 5.3.10.4 a) and b) should be either:

a) equally spaced between the rows of runway edge lights, or

b) symmetrically disposed about the runway centre line in two groups, with the lights uniformly spaced in each group and with a gap between the groups equal to the gauge of the touchdown zone marking or lighting, where such is provided, or otherwise not more than half the distance between the rows of runway edge lights.

Application of wing bar lights

5.3.10.6 Recommendation.— Wing bar lights should be provided on a precision approach runway when additional conspicuity is considered desirable.

5.3.10.7 Wing bar lights shall be provided on a non-instrument or non-precision approach runway where the threshold is displaced and runway threshold lights are required, but are not provided.

Location of wing bar lights

5.3.10.8 Wing bar lights shall be symmetrically disposed about the runway centre line at the threshold in two groups, i.e. wing bars. Each wing bar shall be formed by at least five lights extending at least 10 m outward from, and at right angles to, the line of the runway edge lights, with the innermost light of each wing bar in the line of the runway edge lights.

Characteristics of runway threshold and wing bar lights

5.3.10.9 Runway threshold and wing bar lights shall be fixed unidirectional lights showing green in the direction of approach to the runway. The intensity and beam spread of the lights shall be adequate for the conditions of visibility and ambient light in which use of the runway is intended.

5.3.10.10 Runway threshold lights on a precision approach runway shall be in accordance with the specifications of Appendix 2, Figure 2.3.

5.3.10.11 Threshold wing bar lights on a precision approach runway shall be in accordance with the specifications of Appendix 2, Figure 2.4.

5.3.11 Runway end lights

Application

5.3.11.1 Runway end lights shall be provided for a runway equipped with runway edge lights.

Note.— When the threshold is at the runway extremity, fittings serving as threshold lights may be used as runway end lights.

Location

5.3.11.2 Runway end lights shall be placed on a line at right angles to the runway axis as near to the end of the runway as possible and, in any case, not more than 3 m outside the end.

5.3.11.3 Recommendation.— Runway end lighting should consist of at least six lights. The lights should be either:

a) equally spaced between the rows of runway edge lights, or

b) symmetrically disposed about the runway centre line in two groups with the lights uniformly spaced in each group and with a gap between the groups of not more than half the distance between the rows of runway edge lights.

For a precision approach runway category III, the spacing between runway end lights, except between the two innermost lights if a gap is used, should not exceed 6 m.

Characteristics

5.3.11.4 Runway end lights shall be fixed unidirectional lights showing red in the direction of the runway. The intensity
and beam spread of the lights shall be adequate for the conditions of visibility and ambient light in which use of the runway is intended.

5.3.11.5 Runway end lights on a precision approach runway shall be in accordance with the specifications of Appendix 2, Figure 2-8.

5.3.12 Runway centre line lights

Application

5.3.12.1 Runway centre line lights shall be provided on a precision approach runway category II or III.

5.3.12.2 Recommendation.— Runway centre line lights should be provided on a precision approach runway category I, particularly when the runway is used by aircraft with high landing speeds or where the width between the runway edge lights is greater than 50 m.

5.3.12.3 Runway centre line lights shall be provided on a runway intended to be used for take-off with an operating minimum below an RVR of the order of 400 m.

5.3.12.4 Recommendation.— Runway centre line lights should be provided on a runway intended to be used for take-off with an operating minimum of an RVR of the order of 400 m or higher when used by aeroplanes with a very high take-off speed, particularly where the width between the runway edge lights is greater than 50 m.

Location

5.3.12.5 Runway centre line lights shall be located along the centre line of the runway, except that the lights may be uniformly offset to the same side of the runway centre line by not more than 60 cm where it is not practicable to locate them along the centre line. The lights shall be located from the threshold to the end at longitudinal spacing of approximately 15 m. Where the serviceability level of the runway centre line lights specified as maintenance objectives in 9.4.26 or 9.4.30, as appropriate, can be demonstrated and the runway is intended for use in runway visual range conditions of 350 m or greater, the longitudinal spacing may be approximately 30 m.

Note.— Existing centre line lighting where lights are spaced at 7.5 m need not be replaced.

5.3.12.6 Recommendation.— Centre line guidance for take-off from the beginning of a runway to a displaced threshold should be provided by:

- an approach lighting system if its characteristics and intensity settings afford the guidance required during take-off and it does not dazzle the pilot of an aircraft taking off; or
- runway centre line lights; or
- barrettes of at least 3 m length and spaced at uniform intervals of 30 m, as shown in Figure 5-19, designed so that their photometric characteristics and intensity setting afford the guidance required during take-off without dazzling the pilot of an aircraft taking off.

Where necessary, provision should be made to extinguish those centre line lights specified in b) or reset the intensity of the approach lighting system or barrettes when the runway is being used for landing. In no case should only the single source runway centre line lights show from the beginning of the runway to a displaced threshold when the runway is being used for landing.

Characteristics

5.3.12.7 Runway centre line lights shall be fixed lights showing variable white from the threshold to the point 900 m from the runway end; alternate red and variable white from 900 m to 300 m from the runway end; and red from 300 m to the runway end, except that for runways less than 1 800 m in length, the alternate red and variable white lights shall extend from the mid-point of the runway usable for landing to 300 m from the runway end.

Note.— Care is required in the design of the electrical system to ensure that failure of part of the electrical system will not result in a false indication of the runway distance remaining.

5.3.12.8 Runway centre line lights shall be in accordance with the specifications of Appendix 2, Figure 2.6 or 2.7.

5.3.13 Runway touchdown zone lights

Application

5.3.13.1 Touchdown zone lights shall be provided in the touchdown zone of a precision approach runway category II or III.

Location

5.3.13.2 Touchdown zone lights shall extend from the threshold for a longitudinal distance of 900 m, except that, on runways less than 1 800 m in length, the system shall be shortened so that it does not extend beyond the midpoint of the runway. The pattern shall be formed by pairs of barrettes symmetrically located about the runway centre line. The
lateral spacing between the innermost lights of a pair of barrettes shall be equal to the lateral spacing selected for the touchdown zone marking. The longitudinal spacing between pairs of barrettes shall be either 30 m or 60 m.

Note.— To allow for operations at lower visibility minima, it may be advisable to use a 30 m longitudinal spacing between barrettes.

**Characteristics**

5.3.13.3 A barrette shall be composed of at least three lights with a spacing between the lights of not more than 1.5 m.

5.3.13.4 Recommendation.— A barrette should be not less than 3 m nor more than 4.5 m in length.

5.3.13.5 Touchdown zone lights shall be fixed unidirectional lights showing variable white.

5.3.13.6 Touchdown zone lights shall be in accordance with the specifications of Appendix 2, Figure 2.5.

5.3.14 Stopway lights

**Application**

5.3.14.1 Stopway lights shall be provided for a stopway intended for use at night.

**Location**

5.3.14.2 Stopway lights shall be placed along the full length of the stopway and shall be in two parallel rows that are equidistant from the centre line and coincident with the rows of the runway edge lights. Stopway lights shall also be provided across the end of a stopway on a line at right angles to the stopway axis as near to the end of the stopway as possible and, in any case, not more than 3 m outside the end.

**Characteristics**

5.3.14.3 Stopway lights shall be fixed unidirectional lights showing red in the direction of the runway.

5.3.15 Taxiway centre line lights

**Application**

5.3.15.1 Taxiway centre line lights shall be provided on an exit taxiway, taxiway, de-icing/anti-icing facility and apron intended for use in runway visual range conditions less than a value of 350 m in such a manner as to provide continuous guidance between the runway centre line and aircraft stands, except that these lights need not be provided where the traffic density is light and taxiway edge lights and centre line marking provide adequate guidance.

5.3.15.2 Recommendation.— Taxiway centre line lights should be provided on a taxiway intended for use at night in runway visual range conditions of 350 m or greater, and particularly on complex taxiway intersections and exit taxiways, except that these lights need not be provided where the traffic density is light and taxiway edge lights and centre line marking provide adequate guidance.

Note.— Where there may be a need to delineate the edges of a taxiway, e.g. on a rapid exit taxiway, narrow taxiway or in snow conditions, this may be done with taxiway edge lights or markers.

5.3.15.3 Recommendation.— Taxiway centre line lights should be provided on an exit taxiway, taxiway, de-icing/anti-icing facility and apron in all visibility conditions where specified as components of an advanced surface movement guidance and control system in such a manner as to provide continuous guidance between the runway centre line and aircraft stands.

5.3.15.4 Taxiway centre line lights shall be provided on a runway forming part of a standard taxi-route and intended for taxiing in runway visual range conditions less than a value of 350 m, except that these lights need not be provided where the traffic density is light and taxiway edge lights and centre line marking provide adequate guidance.

Note.— See 8.2.3 for provisions concerning the interlocking of runway and taxiway lighting systems.

5.3.15.5 Recommendation.— Taxiway centre line lights should be provided in all visibility conditions on a runway forming part of a standard taxi-route where specified as components of an advanced surface movement guidance and control system.

**Characteristics**

5.3.15.6 Taxiway centre line lights on a taxiway other than an exit taxiway and on a runway forming part of a standard taxi-route shall be fixed lights showing green with beam dimensions such that the light is visible only from aeroplanes on or in the vicinity of the taxiway.

5.3.15.7 Taxiway centre line lights on an exit taxiway shall be fixed lights. Alternate taxiway centre line lights shall show green and yellow from their beginning near the runway centre line to the perimeter of the ILS/MLS critical/sensitive area or the lower edge of the inner transitional surface, whichever is farthest from the runway; and thereafter all lights shall show green (Figure 5-20). The light nearest to the perimeter shall always show yellow. Where aircraft may
Figure 5.19. Example of approach and runway lighting for runway with displaced thresholds
Figure 5-20. Taxiway lighting
follow the same centre line in both directions, all the centre line lights shall show green to aircraft approaching the runway.

Note 1.— Care is necessary to limit the light distribution of green lights on or near a runway so as to avoid possible confusion with threshold lights.

Note 2.— For yellow filter characteristics see Appendix 1, 2.2.

Note 3.— The size of the ILS/MLS critical/sensitive area depends on the characteristics of the associated ILS/MLS and other factors. Guidance is provided in Annex 10, Volume I, Attachments C and G to Part I.

Note 4.— See 5.4.3 for specifications on runway vacated signs.

5.3.15.8 Taxiway centre line lights shall be in accordance with the specifications of:

a) Appendix 2, Figure 2-12, 2-13, or 2-14 for taxiways intended for use in runway visual range conditions of less than a value of 350 m; and

b) Appendix 2, Figure 2-15 or 2-16 for other taxiways.

5.3.15.9 Recommendation.— Where taxiway centre line lights are specified as components of an advanced surface movement guidance and control system and where, from an operational point of view, higher intensities are required to maintain ground movements at a certain speed in very low visibilities or in bright daytime conditions, taxiway centre line lights should be in accordance with the specifications of Appendix 2, Figure 2-17, 2-18 or 2-19.

Note.— High-intensity centre line lights should only be used in case of an absolute necessity and following a specific study.

Location

5.3.15.10 Recommendation.— Taxiway centre line lights should normally be located on the taxiway centre line marking, except that they may be offset by not more than 30 cm where it is not practicable to locate them on the marking.

Taxiway centre line lights on taxiways

Location

5.3.15.11 Recommendation.— Taxiway centre line lights on a straight section of a taxiway should be spaced at longitudinal intervals of not more than 30 m, except that:

a) larger intervals not exceeding 60 m may be used where, because of the prevailing meteorological conditions, adequate guidance is provided by such spacing;

b) intervals less than 30 m should be provided on short straight sections; and

c) on a taxiway intended for use in RVR conditions of less than a value of 350 m, the longitudinal spacing should not exceed 15 m.

5.3.15.12 Recommendation.— Taxiway centre line lights on a taxiway curve should continue from the straight portion of the taxiway at a constant distance from the outside edge of the taxiway curve. The lights should be spaced at intervals such that a clear indication of the curve is provided.

5.3.15.13 Recommendation.— On a taxiway intended for use in RVR conditions of less than a value of 350 m, the lights on a curve should not exceed a spacing of 15 m and on a curve of less than 400 m radius the lights should be spaced at intervals of not greater than 7.5 m. This spacing should extend for 60 m before and after the curve.

Note 1.— Spacings on curves that have been found suitable for a taxiway intended for use in RVR conditions of 350 m or greater are:

<table>
<thead>
<tr>
<th>Curve radius</th>
<th>Light spacing</th>
</tr>
</thead>
<tbody>
<tr>
<td>up to 400 m</td>
<td>7.5 m</td>
</tr>
<tr>
<td>401 m to 899 m</td>
<td>15 m</td>
</tr>
<tr>
<td>900 m or greater</td>
<td>30 m</td>
</tr>
</tbody>
</table>

Note 2.— See 3.8.5 and Figure 3-1.

Taxiway centre line lights on rapid exit taxiways

Location

5.3.15.14 Recommendation.— Taxiway centre line lights on a rapid exit taxiway should commence at a point at least 60 m before the beginning of the taxiway centre line curve and continue beyond the end of the curve to a point on the centre line of the taxiway where an aeroplane can be expected to reach normal taxiing speed. The lights on that portion parallel to the runway centre line should always be at least 60 cm from any row of runway centre line lights, as shown in Figure 5-21.

5.3.15.15 Recommendation.— The lights should be spaced at longitudinal intervals of not more than 15 m, except that, where runway centre line lights are not provided, a greater interval not exceeding 30 m may be used.

Taxiway centre line lights on other exit taxiways

Location

5.3.15.16 Recommendation.— Taxiway centre line lights on exit taxiways other than rapid exit taxiways should commence at the point where the taxiway centre line marking
begins to curve from the runway centre line, and follow the curved taxiway centre line marking at least to the point where the marking leaves the runway. The first light should be at least 60 cm from any row of runway centre line lights, as shown in Figure 5-21.

5.3.15.17 Recommendation.— The lights should be spaced at longitudinal intervals of not more than 7.5 m.

Taxiway centre line lights on runways

Application

5.3.16 Taxiway edge lights

5.3.16.1 Taxiway edge lights shall be provided at the edges of a holding bay, de-icing/anti-icing facility, apron, etc. intended for use at night and on a taxiway not provided with taxiway centre line lights and intended for use at night, except that taxiway edge lights need not be provided where, considering the nature of the operations, adequate guidance can be achieved by surface illumination or other means.

Note.— See 5.5.5 for taxiway edge markers.

5.3.16.2 Taxiway edge lights shall be provided on a runway forming part of a standard taxi-route and intended for taxiing at night where the runway is not provided with taxiway centre line lights.

Note.— See 8.2.3 for provisions concerning the inter-locking of runway and taxiway lighting systems.
Location

5.3.16.3 Recommendation.— Taxiway edge lights on a straight section of a taxiway and on a runway forming part of a standard taxi-route should be spaced at uniform longitudinal intervals of not more than 60 m. The lights on a curve should be spaced at intervals less than 60 m so that a clear indication of the curve is provided.

5.3.16.4 Recommendation.— Taxiway edge lights on a holding bay, de-icing/anti-icing facility, apron, etc. should be spaced at uniform longitudinal intervals of not more than 60 m.

5.3.16.5 Recommendation.— The lights should be located as near as practicable to the edges of the taxiway, holding bay, de-icing/anti-icing facility, apron or runway, etc. or outside the edges at a distance of not more than 3 m.

Characteristics

5.3.16.6 Taxiway edge lights shall be fixed lights showing blue. The lights shall show up to at least 30° above the horizontal and at all angles in azimuth necessary to provide guidance to a pilot taxiing in either direction. At an intersection, exit or curve the lights shall be shielded as far as practicable so that they cannot be seen in angles of azimuth in which they may be confused with other lights.

5.3.17 Stop bars

Application

Note.— The provision of stop bars requires their control either manually or automatically by air traffic services.

5.3.17.1 A stop bar shall be provided at every runway-holding position serving a runway when it is intended that the runway will be used in runway visual range conditions less than a value of 350 m, except where:

a) appropriate aids and procedures are available to assist in preventing inadvertent incursions of aircraft and vehicles onto the runway; or

b) operational procedures exist to limit, in runway visual range conditions less than a value of 550 m, the number of:

1) aircraft on the manoeuvring area to one at a time; and

2) vehicles on the manoeuvring area to the essential minimum.

5.3.17.2 A stop bar shall be provided at every runway-holding position, serving a runway when it is intended that the runway will be used in runway visual range conditions of values between 350 m and 550 m, except where:

a) appropriate aids and procedures are available to assist in preventing inadvertent incursions of aircraft and vehicles onto the runway; or

b) operational procedures exist to limit, in runway visual range conditions less than a value of 550 m, the number of:

1) aircraft on the manoeuvring area to one at a time; and

2) vehicles on the manoeuvring area to the essential minimum.

5.3.17.3 Recommendation.— A stop bar should be provided at an intermediate holding position when it is desired to supplement markings with lights and to provide traffic control by visual means.

5.3.17.4 Recommendation.— Where the normal stop bar lights might be obscured (from a pilot’s view), for example, by snow or rain, or where a pilot may be required to stop the aircraft in a position so close to the lights that they are blocked from view by the structure of the aircraft, then a pair of elevated lights should be added to each end of the stop bar.

Location

5.3.17.5 Stop bars shall be located across the taxiway at the point where it is desired that traffic stop. Where the additional lights specified in 5.3.17.4 are provided, these lights shall be located not less than 3 m from the taxiway edge.

Characteristics

5.3.17.6 Stop bars shall consist of lights spaced at intervals of 3 m across the taxiway, showing red in the intended direction(s) of approach to the intersection or runway-holding position.

5.3.17.7 Stop bars installed at a runway-holding position shall be unidirectional and shall show red in the direction of approach to the runway.

5.3.17.8 Where the additional lights specified in 5.3.17.4 are provided, these lights shall have the same characteristics as the lights in the stop bar, but shall be visible to approaching aircraft up to the stop bar position.
5.3.17.9 Selectively switchable stop bars shall be installed in conjunction with at least three taxiway centre line lights (extending for a distance of at least 90 m from the stop bar) in the direction that it is intended for an aircraft to proceed from the stop bar.

Note.— See 5.3.15.11 for provisions concerning the spacing of taxiway centre line lights.

5.3.17.10 The intensity in red light and beam spreads of stop bar lights shall be in accordance with the specifications in Appendix 2, Figures 2-12 through 2-16, as appropriate.

5.3.17.11 Recommendation.— Where stop bars are specified as components of an advanced surface movement guidance and control system and where, from an operational point of view, higher intensities are required to maintain ground movements at a certain speed in very low visibilities or in bright daytime conditions, the intensity in red light and beam spreads of stop bar lights should be in accordance with the specifications of Appendix 2, Figure 2-17, 2-18 or 2-19.

Note.— High-intensity stop bars should only be used in case of an absolute necessity and following a specific study.

5.3.17.12 Recommendation.— Where a wide beam fixture is required, the intensity in red light and beam spreads of stop bar lights should be in accordance with the specifications of Appendix 2, Figure 2-17 or 2-19.

5.3.17.13 The lighting circuit shall be designed so that:

a) stop bars located across entrance taxiways are selectively switchable;

b) stop bars located across taxiways intended to be used only as exit taxiways are switchable selectively or in groups;

c) when a stop bar is illuminated, any taxiway centre line lights installed beyond the stop bar shall be extinguished for a distance of at least 90 m; and

d) stop bars shall be interlocked with the taxiway centre line lights so that when the centre line lights beyond the stop bar are illuminated the stop bar is extinguished and vice versa.

Note 1.— A stop bar is switched on to indicate that traffic stop and switched off to indicate that traffic proceed.

Note 2.— Care is required in the design of the electrical system to ensure that all of the lights of a stop bar will not fail at the same time. Guidance on this issue is given in the Aerodrome Design Manual, Part 5.

5.3.18 Intermediate holding position lights

Application

5.3.18.1 Except where a stop bar has been installed, intermediate holding position lights shall be provided at an intermediate holding position intended for use in runway visual range conditions less than a value of 350 m.

5.3.18.2 Recommendation.— Intermediate holding position lights should be provided at an intermediate holding position where there is no need for stop-and-go signals as provided by a stop bar.

Location

5.3.18.3 Intermediate holding position lights shall be located along the intermediate holding position marking at a distance of 0.3 m prior to the marking.

Characteristics

5.3.18.4 Intermediate holding position lights shall consist of three fixed unidirectional lights showing yellow in the direction of approach to the intermediate holding position with a light distribution similar to taxiway centre line lights if provided. The lights shall be disposed symmetrically about and at right angle to the taxiway centre line, with individual lights spaced 1.5 m apart.

5.3.19 De-icing/anti-icing facility exit lights

Application

5.3.19.1 Recommendation.— De-icing/anti-icing facility exit lights should be provided at the exit boundary of a remote de-icing/anti-icing facility adjoining a taxiway.

Location

5.3.19.2 De-icing/anti-icing facility exit lights shall be located 0.3 m inward of the intermediate holding position marking displayed at the exit boundary of a remote de-icing/anti-icing facility.

Characteristics

5.3.19.3 De-icing/anti-icing facility exit lights shall consist of in-pavement fixed unidirectional lights spaced at intervals of 6 m showing yellow in the direction of the approach to the exit boundary with a light distribution similar to taxiway centre line lights (see Figure 5-22).
5.3.20 Runway guard lights

Note.— There are two standard configurations of runway guard lights as illustrated in Figure 5-23.

Application

5.3.20.1 Runway guard lights, Configuration A, shall be provided at each taxiway/runway intersection associated with a runway intended for use in:

a) runway visual range conditions less than a value of 550 m where a stop bar is not installed; and

b) runway visual range conditions of values between 550 m and 1 200 m where the traffic density is heavy.

5.3.20.2 Recommendation.— Runway guard lights, Configuration A, should be provided at each taxiway/runway intersection associated with a runway intended for use in:

a) runway visual range conditions of values less than a value of 550 m where a stop bar is installed; and

b) runway visual range conditions of values between 550 m and 1 200 m where the traffic density is medium or light.

5.3.20.3 Recommendation.— Runway guard lights, Configuration A or Configuration B or both, should be provided at each taxiway/runway intersection where enhanced conspicuity of the taxiway/runway intersection is needed, such as on a wide-throat taxiway, except that Configuration B should not be collocated with a stop bar.

Location

5.3.20.4 Runway guard lights, Configuration A, shall be located at each side of the taxiway at a distance from the runway centre line not less than that specified for a take-off runway in Table 3-2.

5.3.20.5 Runway guard lights, Configuration B, shall be located across the taxiway at a distance from the runway centre line not less than that specified for a take-off runway in Table 3-2.

Characteristics

5.3.20.6 Runway guard lights, Configuration A, shall consist of two pairs of yellow lights.

5.3.20.7 Recommendation.— Where there is a need to enhance the contrast between the on and off state of runway guard lights, Configuration A, intended for use during the day, a visor of sufficient size to prevent sunlight from entering the lens without interfering with the function of the fixture should be located above each lamp.

Note.— Some other device or design, e.g. specially designed optics, may be used in lieu of the visor.

5.3.20.8 Runway guard lights, Configuration B, shall consist of yellow lights spaced at intervals of 3 m across the taxiway.

5.3.20.9 The light beam shall be unidirectional and aligned so as to be visible to the pilot of an aeroplane taxiing to the holding position.
Annex 14 — Aerodromes

5.3.20.10 **Recommendation.**— The intensity in yellow light and beam spreads of lights of Configuration A should be in accordance with the specifications in Appendix 2, Figure 2-24.

5.3.20.11 **Recommendation.**— Where runway guard lights are intended for use during the day, the intensity in yellow light and beam spreads of lights of Configuration A should be in accordance with the specifications in Appendix 2, Figure 2-25.

5.3.20.12 **Recommendation.**— Where runway guard lights are specified as components of an advanced surface movement guidance and control system where higher light intensities are required, the intensity in yellow light and beam spreads of lights of Configuration A should be in accordance with the specifications in Appendix 2, Figure 2-25.

Note.— Higher light intensities may be required to maintain ground movement at a certain speed in low visibilities.

5.3.20.13 **Recommendation.**— The intensity in yellow light and beam spreads of lights of Configuration B should be in accordance with the specifications in Appendix 2, Figure 2-12.

5.3.20.14 **Recommendation.**— Where runway guard lights are intended for use during the day, the intensity in yellow light and beam spreads of lights of Configuration B should be in accordance with the specifications in Appendix 2, Figure 2-20.

5.3.20.15 **Recommendation.**— Where runway guard lights are specified as components of an advanced surface movement guidance and control system where higher light intensities are required, the intensity in yellow light and beam spreads of lights of Configuration B should be in accordance with the specifications in Appendix 2, Figure 2-20.

5.3.20.16 The lights in each unit of Configuration A shall be illuminated alternately.

5.3.20.17 For Configuration B, adjacent lights shall be alternately illuminated and alternative lights shall be illuminated in unison.

5.3.20.18 The lights shall be illuminated between 30 and 60 cycles per minute and the light suppression and illumination periods shall be equal and opposite in each light.

Note — The optimum flash rate is dependent on the rise and fall times of the lamps used. Runway guard lights, Configuration A, installed on 6.6 ampere series circuits have been found to look best when operated at 45 to 50 flashes per minute per lamp. Runway guard lights, Configuration B, installed on 6.6 ampere series circuits have been found to look best when operated at 30 to 32 flashes per minute per lamp.

See 5.3.20.4 and 5.3.20.5

A pair of unidirectional, flashing yellow lights

Unidirectional flashing yellow lights spaced at intervals of 3 m

Figure 5-23. Runway guard lights
Chapter 5

5.3.21 Apron floodlighting
(see also 5.3.15.1 and 5.3.16.1)

Application

5.3.21.1 Recommendation.— Apron floodlighting should be provided on an apron, on a de-icing/anti-icing facility and on a designated isolated aircraft parking position intended to be used at night.

Note 1. — Where a de-icing/anti-icing facility is located in close proximity to the runway and permanent floodlighting could be confusing to pilots, other means of illumination of the facility may be required.

Note 2. — The designation of an isolated aircraft parking position is specified in 3.13.


Location

5.3.21.2 Recommendation.— Apron floodlights should be located so as to provide adequate illumination on all apron service areas, with a minimum of glare to pilots of aircraft in flight and on the ground, aerodrome and apron controllers, and personnel on the apron. The arrangement and aiming of floodlights should be such that an aircraft stand receives light from two or more directions to minimize shadows.

Characteristics

5.3.21.3 The spectral distribution of apron floodlights shall be such that the colours used for aircraft marking connected with routine servicing, and for surface and obstacle marking, can be correctly identified.

5.3.21.4 Recommendation.— The average illuminance should be at least the following:

Aircraft stand:
— horizontal illuminance — 20 lux with a uniformity ratio (average to minimum) of not more than 4 to 1; and
— vertical illuminance — 20 lux at a height of 2 m above the apron in relevant directions.

Other apron areas:
— horizontal illuminance — 50 per cent of the average illuminance on the aircraft stands with a uniformity ratio (average to minimum) of not more than 4 to 1.

5.3.22 Visual docking guidance system

Application

5.3.22.1 A visual docking guidance system shall be provided when it is intended to indicate, by a visual aid, the precise positioning of an aircraft on an aircraft stand and other alternative means, such as marshallsers, are not practicable.

Note.— The factors to be considered in evaluating the need for a visual docking guidance system are in particular: the number and type(s) of aircraft using the aircraft stand, weather conditions, space available on the apron and the precision required for manoeuvring into the parking position due to aircraft servicing installation, passenger loading bridges, etc. See the Aerodrome Design Manual, Part 4 — Visual Aids for guidance on the selection of suitable systems.

5.3.22.2 The provisions of 5.3.22.3 to 5.3.22.7, 5.3.22.9, 5.3.22.10, 5.3.22.12 to 5.3.22.15, 5.3.22.17, 5.3.22.18 and 5.3.22.20 shall not require the replacement of existing installations before 1 January 2005.

Characteristics

5.3.22.3 The system shall provide both azimuth and stopping guidance.

5.3.22.4 The azimuth guidance unit and the stopping position indicator shall be adequate for use in all weather, visibility, background lighting and pavement conditions for which the system is intended both by day and night, but shall not dazzle the pilot.

Note.— Care is required in both the design and on-site installation of the system to ensure that reflection of sunlight, or other light in the vicinity, does not degrade the clarity and conspicuity of the visual cues provided by the system.

5.3.22.5 The azimuth guidance unit and the stopping position indicator shall be of a design such that:

a) a clear indication of malfunction of either or both is available to the pilot; and

b) they can be turned off.

5.3.22.6 The azimuth guidance unit and the stopping position indicator shall be located in such a way that there is continuity of guidance between the aircraft stand markings, the aircraft stand manoeuvring guidance lights, if present, and the visual docking guidance system.

5.3.22.7 The accuracy of the system shall be adequate for the type of loading bridge and fixed aircraft servicing installations with which it is to be used.
5.3.22.8 **Recommendation.**— *The system should be usable by all types of aircraft for which the aircraft stand is intended, preferably without selective operation.*

5.3.22.9 If selective operation is required to prepare the system for use by a particular type of aircraft, then the system shall provide an identification of the selected aircraft type to both the pilot and the system operator as a means of ensuring that the system has been set properly.

**Azimuth guidance unit**

**Location**

5.3.22.10 The azimuth guidance unit shall be located on or close to the extension of the stand centre line ahead of the aircraft so that its signals are visible from the cockpit of an aircraft throughout the docking manoeuvre and aligned for use at least by the pilot occupying the left seat.

5.3.22.11 **Recommendation.**— *The azimuth guidance unit should be aligned for use by the pilots occupying both the left and right seats.*

**Characteristics**

5.3.22.12 The azimuth guidance unit shall provide unambiguous left/right guidance which enables the pilot to acquire and maintain the lead-in line without overcontrolling.

5.3.22.13 When azimuth guidance is indicated by colour change, green shall be used to identify the centre line and red for deviations from the centre line.

**Stopping position indicator**

**Location**

5.3.22.14 The stopping position indicator shall be located in conjunction with, or sufficiently close to, the azimuth guidance unit so that a pilot can observe both the azimuth and stop signals without turning the head.

5.3.22.15 The stopping position indicator shall be usable at least by the pilot occupying the left seat.

5.3.22.16 **Recommendation.**— *The stopping position indicator should be usable by the pilots occupying both the left and right seats.*

**Characteristics**

5.3.22.17 The stopping position information provided by the indicator for a particular aircraft type shall account for the anticipated range of variations in pilot eye height and/or viewing angle.

5.3.22.18 The stopping position indicator shall show the stopping position for the aircraft for which guidance is being provided, and shall provide closing rate information to enable the pilot to gradually decelerate the aircraft to a full stop at the intended stopping position.

5.3.22.19 **Recommendation.**— *The stopping position indicator should provide closing rate information over a distance of at least 10 m.*

5.3.22.20 When stopping guidance is indicated by colour change, green shall be used to show that the aircraft can proceed and red to show that the stop point has been reached except that for a short distance prior to the stop point a third colour may be used to warn that the stopping point is close.

5.3.23 **Aircraft stand manoeuvring guidance lights**

**Application**

5.3.23.1 **Recommendation.**— *Aircraft stand manoeuvring guidance lights should be provided to facilitate the positioning of an aircraft on an aircraft stand on a paved apron or on a de-icing/anti-icing facility intended for use in poor visibility conditions, unless adequate guidance is provided by other means.*

**Location**

5.3.23.2 Aircraft stand manoeuvring guidance lights shall be collocated with the aircraft stand markings.

**Characteristics**

5.3.23.3 Aircraft stand manoeuvring guidance lights, other than those indicating a stop position, shall be fixed yellow lights, visible throughout the segments within which they are intended to provide guidance.

5.3.23.4 **Recommendation.**— *The lights used to delinate lead-in, turning and lead-out lines should be spaced at intervals of not more than 7.5 m on curves and 15 m on straight sections.*

5.3.23.5 The lights indicating a stop position shall be fixed, unidirectional lights, showing red.

5.3.23.6 **Recommendation.**— *The intensity of the lights should be adequate for the condition of visibility and ambient light in which the use of the aircraft stand is intended.*

5.3.23.7 **Recommendation.**— *The lighting circuit should be designed so that the lights may be switched on to indicate that an aircraft stand is to be used and switched off to indicate that it is not to be used.*
5.3.24 Road-holding position light

Application

5.3.24.1 A road-holding position light shall be provided at each road-holding position serving a runway when it is intended that the runway will be used in runway visual range conditions less than a value of 350 m.

5.3.24.2 Recommendation.— A road-holding position light should be provided at each road-holding position serving a runway when it is intended that the runway will be used in runway visual range conditions of values between 350 m and 550 m.

Location

5.3.24.3 A road-holding position light shall be located adjacent to the holding position marking 1.5 m (± 0.5 m) from one edge of the road, i.e. left or right as appropriate to the local traffic regulations.

Note.— See 8.7 for the mass and height limitations and fragibility requirements of navigation aids located on runway strips.

Characteristics

5.3.24.4 The road-holding position light shall comprise:

a) a controllable red (stop)/green (go) traffic light; or

b) a flashing-red light.

Note.— It is intended that the lights specified in subparagraph a) be controlled by the air traffic services.

5.3.24.5 The road-holding position light beam shall be unidirectional and aligned so as to be visible to the driver of a vehicle approaching the holding position.

5.3.24.6 The intensity of the light beam shall be adequate for the conditions of visibility and ambient light in which the use of the holding position is intended, but shall not dazzle the driver.

Note.— The commonly used traffic lights are likely to meet the requirements in 5.3.24.5 and 5.3.24.6.

5.3.24.7 The flash frequency of the flashing-red light shall be between 30 and 60 per minute.

5.4 Signs

Application

5.4.1 General

Note.— Signs shall be either fixed message signs or variable message signs. Guidance on signs is contained in the Aerodrome Design Manual, Part 4.

Characteristics

5.4.1.3 Signs shall be frangible. Those located near a runway or taxiway shall be sufficiently low to preserve clearance for propellers and the engine pods of jet aircraft. The installed height of the sign shall not exceed the dimension shown in the appropriate column of Table 5-4.

5.4.1.4 Signs shall be rectangular, as shown in Figures 5-24 and 5-25 with the longer side horizontal.

5.4.1.5 The only signs on the movement area utilizing red shall be mandatory instruction signs.

5.4.1.6 The inscriptions on a sign shall be in accordance with the provisions of Appendix 4.

5.4.1.7 Signs shall be illuminated in accordance with the provisions of Appendix 4 when intended for use:

a) in runway visual range conditions less than a value of 800 m; or

b) at night in association with instrument runways; or

c) at night in association with non-instrument runways where the code number is 3 or 4.
Annex 14 — Aerodromes

5.4.1.8 Signs shall be retroreflective and/or illuminated in accordance with the provisions of Appendix 4 when intended for use at night in association with non-instrument runways where the code number is 1 or 2.

5.4.1.9 A variable message sign shall show a blank face when not in use.

5.4.1.10 In case of failure, a variable message sign shall not provide information that could lead to unsafe action from a pilot or a vehicle driver.

5.4.1.11 Recommendation.— The time interval to change from one message to another on a variable message sign should be as short as practicable and should not exceed 5 seconds.

5.4.2 Mandatory instruction signs

Note.— See Figure 5-24 for pictorial representation of mandatory instruction signs and Figure 5-26 for examples of locating signs at taxiway/runway intersections.

Application

5.4.2.1 A mandatory instruction sign shall be provided to identify a location beyond which an aircraft taxiing or vehicle shall not proceed unless authorized by the aerodrome control tower.

5.4.2.2 Mandatory instruction signs shall include runway designation signs, category I, II or III holding position signs,

![Diagram of mandatory instruction signs](image)
Figure 5-25. Information signs.
Annex 14 — Aerodromes

NON-INSTRUMENT, NON-PRECISION, TAKE-OFF RUNWAYS

Precision Approach Runways

<table>
<thead>
<tr>
<th>CATEGORY I</th>
<th>CATEGORY II</th>
<th>CATEGORY III</th>
</tr>
</thead>
<tbody>
<tr>
<td>A 27 27 A</td>
<td>A 27 27 CAT I</td>
<td>A 27 27 CAT III</td>
</tr>
<tr>
<td></td>
<td>A 27 27 A</td>
<td>A 27 27 CAT II</td>
</tr>
</tbody>
</table>

Note: Distance X is established in accordance with Table 3-2. Distance Y is established at the edge of the ILS/MLS critical/sensitive area.

Figure 5-26. Examples of sign positions at taxiway/runway intersections

4/11/99
runway-holding position signs, road-holding position signs and NO ENTRY signs.

Note.— See 5.4.7 for specifications on road-holding position signs.

5.4.2.3 A pattern “A” runway-holding position marking shall be supplemented at a taxiway/runway intersection or a runway/runway intersection with a runway designation sign.

5.4.2.4 A pattern “B” runway-holding position marking shall be supplemented with a category I, II or III holding position sign.

5.4.2.5 A pattern “A” runway-holding position marking at a runway-holding position established in accordance with 3.11.3 shall be supplemented with a runway-holding position sign.

Note.— See 5.2.9 for specifications on runway-holding position marking.

5.4.2.6 Recommendation.— A runway designation sign at a taxiway/runway intersection should be supplemented with a location sign in the outboard (farthest from the taxiway) position, as appropriate.

Note.— See 5.4.3 for characteristics of location signs.

5.4.2.7 A NO ENTRY sign shall be provided when entry into an area is prohibited.

Location

5.4.2.8 A runway designation sign at a taxiway/runway intersection or a runway/runway intersection shall be located on each side of the runway-holding position marking facing the direction of approach to the runway.

5.4.2.9 A category I, II or III holding position sign shall be located on each side of the runway-holding position marking facing the direction of the approach to the critical area.

5.4.2.10 A NO ENTRY sign shall be located at the beginning of the area to which entrance is prohibited on each side of the taxiway as viewed by the pilot.

5.4.2.11 A runway-holding position sign shall be located on each side of the runway-holding position established in accordance with 3.11.3, facing the approach to the obstacle limitation surface or ILS/MLS critical/sensitive area, as appropriate.

Characteristics

5.4.2.12 A mandatory instruction sign shall consist of an inscription in white on a red background.

5.4.2.13 The inscription on a runway designation sign shall consist of the runway designations of the intersecting runway properly oriented with respect to the viewing position of the sign, except that a runway designation sign installed in the vicinity of a runway extremity may show the runway designation of the concerned runway extremity only.

5.4.2.14 The inscription on a category I, II, III or joint II/III holding position sign shall consist of the runway designator followed by CAT I, CAT II, CAT III or CAT II/III, as appropriate.

5.4.2.15 The inscription on a NO ENTRY sign shall be in accordance with Figure 5-24.

5.4.2.16 The inscription on a runway-holding position sign at a runway-holding position established in accordance with 3.11.3 shall consist of the taxiway designation and a number.
5.4.2.17 Where appropriate, the following inscriptions/symbol shall be used:

<table>
<thead>
<tr>
<th>Inscription/symbol</th>
<th>Use</th>
</tr>
</thead>
<tbody>
<tr>
<td>Runway designation of a runway extremity</td>
<td>To indicate a runway-holding position at a runway extremity</td>
</tr>
<tr>
<td>OR Runway designation of both extremities of a runway</td>
<td>To indicate a runway-holding position located at other taxiway/runway intersections or runway/runway intersections</td>
</tr>
<tr>
<td>25 CAT I (Example)</td>
<td>To indicate a category I runway-holding position at the threshold of runway 25</td>
</tr>
<tr>
<td>25 CAT II (Example)</td>
<td>To indicate a category II runway-holding position at the threshold of runway 25</td>
</tr>
<tr>
<td>25 CAT III (Example)</td>
<td>To indicate a category III runway-holding position at the threshold of runway 25</td>
</tr>
<tr>
<td>25 CAT II/III (Example)</td>
<td>To indicate a joint category II/III runway-holding position at the threshold of runway 25</td>
</tr>
<tr>
<td>NO ENTRY symbol</td>
<td>To indicate that entry to an area is prohibited</td>
</tr>
<tr>
<td>B2 (Example)</td>
<td>To indicate a runway-holding position established in accordance with 3.11.3</td>
</tr>
</tbody>
</table>

5.4.3 Information signs

*Note.— See Figure 5-25 for pictorial representations of information signs.*

**Application**

5.4.3.1 An information sign shall be provided where there is an operational need to identify by a sign, a specific location, or routing (direction or destination) information.

5.4.3.2 Information signs shall include: direction signs, location signs, destination signs, runway exit signs, runway vacated signs and intersection take-off signs.

5.4.3.3 A runway exit sign shall be provided where there is an operational need to identify a runway exit.

5.4.3.4 A runway vacated sign shall be provided where the exit taxiway is not provided with taxiway centre line lights and there is a need to indicate to a pilot leaving a runway the perimeter of the ILS/MLS critical/sensitive area or the lower edge of the inner transitional surface whichever is farther from the runway centre line.

*Note.— See 5.3.15 for specifications on colour coding taxiway centre line lights.*

5.4.3.5 **Recommendation.—** An intersection take-off sign should be provided when there is an operational need to indicate the remaining take-off run available (TORA) for intersection take-offs.

5.4.3.6 **Recommendation.—** Where necessary, a destination sign should be provided to indicate the direction to a specific destination on the aerodrome, such as cargo area, general aviation, etc.

5.4.3.7 A combined location and direction sign shall be provided when it is intended to indicate routing information prior to a taxiway intersection.

5.4.3.8 A direction sign shall be provided when there is an operational need to identify the designation and direction of taxiways at an intersection.

5.4.3.9 **Recommendation.—** A location sign should be provided at an intermediate holding position.

5.4.3.10 A location sign shall be provided in conjunction with a runway designation sign except at a runway/runway intersection.

5.4.3.11 A location sign shall be provided in conjunction with a direction sign, except that it may be omitted where an aeronautical study indicates that it is not needed.

5.4.3.12 **Recommendation.—** Where necessary, a location sign should be provided to identify taxiways exiting an apron or taxiways beyond an intersection.

5.4.3.13 **Recommendation.—** Where a taxiway ends at an intersection such as a “T” and it is necessary to identify this, a barricade, direction sign and/or other appropriate visual aid should be used.

**Location**

5.4.3.14 Except as specified in 5.4.3.16 and 5.4.3.24 information signs shall, wherever practicable, be located on the left-hand side of the taxiway in accordance with Table 5-4.

5.4.3.15 At a taxiway intersection, information signs shall be located prior to the intersection and in line with the
taxiway intersection marking. Where there is no taxiway intersection marking, the signs shall be installed at least 60 m from the centre line of the intersecting taxiway where the code number is 3 or 4 and at least 40 m where the code number is 1 or 2.

Note.— A location sign installed beyond a taxiway intersection may be installed on either side of a taxiway.

5.4.3.16 A runway exit sign shall be located on the same side of the runway as the exit is located (i.e. left or right) and positioned in accordance with Table 5-4.

5.4.3.17 A runway exit sign shall be located prior to the runway exit point in line with a position at least 60 m prior to the point of tangency where the code number is 3 or 4, and at least 30 m where the code number is 1 or 2.

5.4.3.18 A runway vacated sign shall be located at least on one side of the taxiway. The distance between the sign and the centre line of a runway shall be not less than the greater of the following:

a) the distance between the centre line of the runway and the perimeter of the ILS/MLS critical/sensitive area; or

b) the distance between the centre line of the runway and the lower edge of the inner transitional surface.

5.4.3.19 Where provided in conjunction with a runway vacated sign, the taxiway location sign shall be positioned outboard of the runway vacated sign.

5.4.3.20 An intersection take-off sign shall be located at the left-hand side of the entry taxiway. The distance between the sign and the centre line of the runway shall be not less than 60 m where the code number is 3 or 4 and not less than 45 m where the code number is 1 or 2.

5.4.3.21 A taxiway location sign installed in conjunction with a runway designation sign shall be positioned outboard of the runway designation sign.

5.4.3.22 Recommendation.— A destination sign should not normally be collocated with a location or direction sign.

5.4.3.23 An information sign other than a location sign shall not be collocated with a mandatory instruction sign.

5.4.3.24 Recommendation.— A direction sign, barricade and/or other appropriate visual aid used to identify a “T” intersection should be located on the opposite side of the intersection facing the taxiway.

Characteristics

5.4.3.25 An information sign other than a location sign shall consist of an inscription in black on a yellow background.

5.4.3.26 A location sign shall consist of an inscription in yellow on a black background and where it is a stand-alone sign shall have a yellow border.

5.4.3.27 The inscription on a runway exit sign shall consist of the designator of the exit taxiway and an arrow indicating the direction to follow.

5.4.3.28 The inscription on a runway vacated sign shall depict the pattern A runway-holding position marking as shown in Figure 5-25.

5.4.3.29 The inscription on an intersection take-off sign shall consist of a numerical message indicating the remaining take-off run available in metres plus an arrow, appropriately located and oriented, indicating the direction of the take-off as shown in Figure 5-25.

5.4.3.30 The inscription on a destination sign shall comprise an alpha, alphanumerical or numerical message identifying the destination plus an arrow indicating the direction to proceed as shown in Figure 5-25.

5.4.3.31 The inscription on a direction sign shall comprise an alpha or alphanumerical message identifying the taxiway(s) plus an arrow or arrows appropriately oriented as shown in Figure 5-25.

5.4.3.32 The inscription on a location sign shall comprise the designation of the location taxiway, runway or other pavement the aircraft is on or is entering and shall not contain arrows.

5.4.3.33 Recommendation.— Where it is necessary to identify each of a series of intermediate holding positions on the same taxiway, the location sign should consist of the taxiway designation and a number.

5.4.3.34 Where a location sign and direction signs are used in combination:

a) all direction signs related to left turns shall be placed on the left side of the location sign and all direction signs related to right turns shall be placed on the right side of the location sign, except that where the junction consists of one intersecting taxiway, the location sign may alternatively be placed on the left hand side;

b) the direction signs shall be placed such that the direction of the arrows departs increasingly from the vertical with increasing deviation of the corresponding taxiway;

c) an appropriate direction sign shall be placed next to the location sign where the direction of the location taxiway changes significantly beyond the intersection; and

d) adjacent direction signs shall be delineated by a vertical black line as shown in Figure 5-25.
5.4.3.35 A taxiway shall be identified by a designator comprising a letter, letters or a combination of a letter or letters followed by a number.

5.4.3.36 **Recommendation.**— When designating taxiways, the use of the letters I, O or X and the use of words such as inner and outer should be avoided wherever possible to avoid confusion with the numerals 1, 0 and closed marking.

5.4.3.37 The use of numbers alone on the manoeuvring area shall be reserved for the designation of runways.

5.4.4 VOR aerodrome check-point sign

**Application**

5.4.4.1 When a VOR aerodrome check-point is established, it shall be indicated by a VOR aerodrome check-point marking and sign.

*Note.—* See 5.2.11 for VOR aerodrome check-point marking.

**Location**

5.4.4.2 A VOR aerodrome check-point sign shall be located as near as possible to the check-point and so that the inscriptions are visible from the cockpit of an aircraft properly positioned on the VOR aerodrome check-point marking.

**Characteristics**

5.4.4.3 A VOR aerodrome check-point sign shall consist of an inscription in black on a yellow background.

5.4.4.4 **Recommendation.**— The inscriptions on a VOR check-point sign should be in accordance with one of the alternatives shown in Figure 5-27 in which:

- **VOR** is an abbreviation identifying this as a VOR check-point;
- **116.3** is an example of the radio frequency of the VOR concerned;
- **147°** is an example of the VOR bearing, to the nearest degree, which should be indicated at the VOR check-point; and
- **4.3 NM** is an example of the distance in nautical miles to a DME collocated with the VOR concerned.

*Note.—* Tolerances for the bearing value shown on the sign are given in Annex 10, Volume I, Attachment E to Part I. It will be noted that a check-point can only be used operationally when periodic checks show it to be consistently within ± 2 degrees of the stated bearing.

![Figure 5-27. VOR aerodrome check-point sign](image_url)
5.4.5 Aerodrome identification sign

Application

5.4.5.1 Recommendation.— An aerodrome identification sign should be provided at an aerodrome where there is insufficient alternative means of visual identification.

Location

5.4.5.2 Recommendation.— The aerodrome identification sign should be placed on the aerodrome so as to be legible, in so far as is practicable, at all angles above the horizontal.

Characteristics

5.4.5.3 The aerodrome identification sign shall consist of the name of the aerodrome.

5.4.5.4 Recommendation.— The colour selected for the sign should give adequate conspicuity when viewed against its background.

5.4.5.5 Recommendation.— The characters should have a height of not less than 3 m.

5.4.6 Aircraft stand identification signs

Application

5.4.6.1 Recommendation.— An aircraft stand identification marking should be supplemented with an aircraft stand identification sign where feasible.

Location

5.4.6.2 Recommendation.— An aircraft stand identification sign should be located so as to be clearly visible from the cockpit of an aircraft prior to entering the aircraft stand.

Characteristics

5.4.6.3 Recommendation.— An aircraft stand identification sign should consist of an inscription in black on a yellow background.

5.4.7 Road-holding position sign

5.4.7.1 A road-holding position sign shall be provided at all road entrances to a runway.

Location

5.4.7.2 The road-holding position sign shall be located 1.5 m from one edge of the road (left or right as appropriate to the local traffic regulations) at the holding position.

Characteristics

5.4.7.3 A road-holding position sign shall consist of an inscription in white on a red background.

5.4.7.4 The inscription on a road-holding position sign shall be in the national language, be in conformity with the local traffic regulations and include the following:

a) a requirement to stop; and

b) where appropriate:

1) a requirement to obtain ATC clearance; and

2) location designator.

Note.— Examples of road-holding position signs are contained in the Aerodrome Design Manual, Part 4.

5.4.7.5 A road-holding position sign intended for night use shall be retroreflective or illuminated.

5.5 Markers

5.5.1 General

Markers shall be frangible. Those located near a runway or taxiway shall be sufficiently low to preserve clearance for propellers and for the engine pods of jet aircraft.

Note 1.— Anchors or chains, to prevent markers which have broken from their mounting from blowing away, are sometimes used.

Note 2.— Guidance on frangibility of markers is given in the Aerodrome Design Manual, Part 6 (in preparation).

5.5.2 Unpaved runway edge markers

Application

5.5.2.1 Recommendation.— Markers should be provided when the extent of an unpaved runway is not clearly indicated by the appearance of its surface compared with that of the surrounding ground.
Location

5.5.2.2 Recommendation.— Where runway lights are provided, the markers should be incorporated in the light fixtures. Where there are no lights, markers of flat rectangular or conical shape should be placed so as to delimit the runway clearly.

Characteristics

5.5.2.3 Recommendation.— The flat rectangular markers should have a minimum size of 1 m by 3 m and should be placed with their long dimension parallel to the runway centre line. The conical markers should have a height not exceeding 50 cm.

5.5.3 Stopway edge markers

Application

5.5.3.1 Recommendation.— Stopway edge markers should be provided when the extent of a stopway is not clearly indicated by its appearance compared with that of the surrounding ground.

Characteristics

5.5.3.2 The stopway edge markers shall be sufficiently different from any runway edge markers used to ensure that the two types of markers cannot be confused.

Note.— Markers consisting of small vertical boards camouflaged on the reverse side, as viewed from the runway, have proved operationally acceptable.

5.5.4 Edge markers for snow-covered runways

Application

5.5.4.1 Recommendation.— Edge markers for snow-covered runways should be used to indicate the usable limits of a snow-covered runway when the limits are not otherwise indicated.

Note.— Runway lights could be used to indicate the limits.

Location

5.5.4.2 Recommendation.— Edge markers for snow-covered runways should be placed along the sides of the runway at intervals of not more than 100 m, and should be located symmetrically about the runway centre line at such a distance from the centre line that there is adequate clearance for wing tips and power plants. Sufficient markers should be placed across the threshold and end of the runway.

Characteristics

5.5.4.3 Recommendation.— Edge markers for snow-covered runways should consist of conspicuous objects such as evergreen trees about 1.5 m high, or light-weight markers.

5.5.5 Taxiway edge markers

Application

5.5.5.1 Recommendation.— Taxiway edge markers should be provided on a taxiway where the code number is 1 or 2 and taxiway centre line or edge lights or taxiway centre line markers are not provided.

Location

5.5.5.2 Recommendation.— Taxiway edge markers should be installed at least at the same locations as would the taxiway edge lights had they been used.

Characteristics

5.5.5.3 A taxiway edge marker shall be retroreflective blue.

5.5.5.4 Recommendation.— The marked surface as viewed by the pilot should be a rectangle and should have a minimum viewing area of 150 cm².

5.5.5.5 Taxiway edge markers shall be frangible. Their height shall be sufficiently low to preserve clearance for propellers and for the engine pods of jet aircraft.

5.5.6 Taxiway centre line markers

Application

5.5.6.1 Recommendation.— Taxiway centre line markers should be provided on a taxiway where the code number is 1 or 2 and taxiway centre line or edge lights or taxiway edge markers are not provided.

5.5.6.2 Recommendation.— Taxiway centre line markers should be provided on a taxiway where the code number is 3 or 4 and taxiway centre line lights are not provided if there is a need to improve the guidance provided by the taxiway centre line marking.

Location

5.5.6.3 Recommendation.— Taxiway centre line markers should be installed at least at the same location as would taxiway centre line lights had they been used.
Chapter 5

Note.— See 5.3.15.11 for the spacing of taxiway centre line lights.

5.5.6.4 Recommendation.— Taxiway centre line markers should normally be located on the taxiway centre line marking except that they may be offset by not more than 30 cm where it is not practicable to locate them on the marking.

Characteristics

5.5.6.5 A taxiway centre line marker shall be retro-reflective green.

5.5.6.6 Recommendation.— The marked surface as viewed by the pilot should be a rectangle and should have a minimum viewing area of 20 cm².

5.5.6.7 Taxiway centre line markers shall be so designed and fitted as to withstand being run over by the wheels of an aircraft without damage either to the aircraft or to the markers themselves.

5.5.7 Unpaved taxiway edge markers

Application

5.5.7.1 Recommendation.— Where the extent of an unpaved taxiway is not clearly indicated by its appearance compared with that of the surrounding ground, markers should be provided.

Location

5.5.7.2 Recommendation.— Where taxiway lights are provided, the markers should be incorporated in the light fixtures. Where there are no lights, markers of conical shape should be placed so as to delimit the taxiway clearly.

5.5.8 Boundary markers

Application

5.5.8.1 Boundary markers shall be provided at an aerodrome where the landing area has no runway.

Location

5.5.8.2 Boundary markers shall be spaced along the boundary of the landing area at intervals of not more than 200 m, if the type shown in Figure 5-28 is used, or approximately 90 m, if the conical type is used with a marker at any corner.

Characteristics

5.5.8.3 Recommendation.— Boundary markers should be of a form similar to that shown in Figure 5-28, or in the form of a cone not less than 50 cm high and not less than 75 cm in diameter at the base. The markers should be coloured to contrast with the background against which they will be seen. A single colour, orange or red, or two contrasting colours, orange and white or alternatively red and white, should be used, except where such colours merge with the background.

Figure 5-28. Boundary markers
CHAPTER 6. VISUAL AIDS FOR DENOTING OBSTACLES

6.1 Objects to be marked and/or lighted

Note.— The marking and/or lighting of obstacles is intended to reduce hazards to aircraft by indicating the presence of the obstacles. It does not necessarily reduce operating limitations which may be imposed by an obstacle.

6.1.1 Recommendation.— A fixed obstacle that extends above a take-off climb surface within 3 000 m of the inner edge of the take-off climb surface should be marked and, if the runway is used at night, lighted, except that:

a) such marking and lighting may be omitted when the obstacle is shielded by another fixed obstacle;

b) the marking may be omitted when the obstacle is lighted by medium-intensity obstacle lights, Type A, by day and its height above the level of the surrounding ground does not exceed 150 m;

c) the marking may be omitted when the obstacle is lighted by high-intensity obstacle lights by day; and

d) the lighting may be omitted where the obstacle is a lighthouse and an aeronautical study indicates the lighthouse light to be sufficient.

6.1.2 Recommendation.— A fixed object, other than an obstacle, adjacent to a take-off climb surface should be marked and, if the runway is used at night, lighted if such marking and lighting is considered necessary to ensure its avoidance, except that the marking may be omitted when:

a) the object is lighted by medium-intensity obstacle lights, Type A, by day and its height above the level of the surrounding ground does not exceed 150 m; or

b) the object is lighted by high-intensity obstacle lights by day.

6.1.3 A fixed obstacle that extends above an approach or transitional surface within 3 000 m of the inner edge of the approach surface shall be marked and, if the runway is used at night, lighted, except that:

a) such marking and lighting may be omitted when the obstacle is shielded by another fixed obstacle;

b) the marking may be omitted when the obstacle is lighted by medium-intensity obstacle lights, Type A, by day and its height above the level of the surrounding ground does not exceed 150 m;

c) the marking may be omitted when the obstacle is lighted by high-intensity obstacle lights by day; and

d) the lighting may be omitted where the obstacle is a lighthouse and an aeronautical study indicates the lighthouse light to be sufficient.

6.1.4 Recommendation.— A fixed obstacle above a horizontal surface should be marked and, if the aerodrome is used at night, lighted except that:

a) such marking and lighting may be omitted when:

1) the obstacle is shielded by another fixed obstacle; or

2) for a circuit extensively obstructed by immovable objects or terrain, procedures have been established to ensure safe vertical clearance below prescribed flight paths; or

3) an aeronautical study shows the obstacle not to be of operational significance;

b) the marking may be omitted when the obstacle is lighted by medium-intensity obstacle lights, Type A, by day and its height above the level of the surrounding ground does not exceed 150 m;

c) the marking may be omitted when the obstacle is lighted by high-intensity obstacle lights by day; and

d) the lighting may be omitted where the obstacle is a lighthouse and an aeronautical study indicates the lighthouse light to be sufficient.

6.1.5 A fixed object that extends above an obstacle protection surface shall be marked and, if the runway is used at night, lighted.

Note.— See 5.3.5 for information on the obstacle protection surface.
6.1.6 Vehicles and other mobile objects, excluding aircraft, on the movement area of an aerodrome are obstacles and shall be marked and, if the vehicles and aerodrome are used at night or in conditions of low visibility, lighted, except that aircraft servicing equipment and vehicles used only on aprons may be exempt.

6.1.7 Elevated aeronautical ground lights within the movement area shall be marked so as to be conspicuous by day. Obstacle lights shall not be installed on elevated ground lights or signs in the movement area.

6.1.8 All obstacles within the distance specified in Table 3-1, column 11 or 12, from the centre line of a taxiway, an apron taxiway or aircraft stand taxilane shall be marked and, if the taxiway, apron taxiway or aircraft stand taxilane is used at night, lighted.

6.1.9 Recommendation.—Obstacles in accordance with 4.3.2 should be marked and lighted, except that the marking may be omitted when the obstacle is lighted by high-intensity obstacle lights by day.

6.1.10 Recommendation.—Overhead wires, cables, etc., crossing a river, valley or highway should be marked and their supporting towers marked and lighted if an aeronautical study indicates that the wires or cables could constitute a hazard to aircraft, except that the marking of the supporting towers may be omitted when they are lighted by high-intensity obstacle lights by day.

6.1.11 Recommendation.—When it has been determined that an overhead wire, cable, etc., needs to be marked but it is not practicable to install markers on the wire, cable, etc., then high-intensity obstacle lights, Type B, should be provided on their supporting towers.

6.2 Marking of objects

General

6.2.1 All fixed objects to be marked shall, whenever practicable, be coloured, but if this is not practicable, markers or flags shall be displayed on or above them, except that objects that are sufficiently conspicuous by their shape, size or colour need not be otherwise marked.

6.2.2 All mobile objects to be marked shall be coloured or display flags.

Use of colours

6.2.3 Recommendation.—An object should be coloured to show a chequered pattern if it has essentially unbroken surfaces and its projection on any vertical plane equals or exceeds 4.5 m in both dimensions. The pattern should consist of rectangles of not less than 1.5 m and not more than 3 m on a side, the corners being of the darker colour. The colours of the pattern should contrast each with the other and with the background against which they will be seen. Orange and white or alternatively red and white should be used, except where such colours merge with the background. (See Figure 6-1.)

6.2.4 Recommendation.—An object should be coloured to show alternating contrasting bands if:

a) it has essentially unbroken surfaces and has one dimension, horizontal or vertical, greater than 1.5 m, and the other dimension, horizontal or vertical, less than 4.5 m; or
b) it is of skeletal type with either a vertical or a horizontal dimension greater than 1.5 m.

The bands should be perpendicular to the longest dimension and have a width approximately 1/7 of the longest dimension or 30 m, whichever is less. The colours of the bands should contrast with the background against which they will be seen. Orange and white should be used, except where such colours are not conspicuous when viewed against the background. The bands on the extremities of the object should be of the darker colour. (See Figures 6-1 and 6-2.)

Note.— Table 6-1 shows a formula for determining band widths and for having an odd number of bands, thus permitting both the top and bottom bands to be of the darker colour.

6.2.5 Recommendation.— An object should be coloured in a single conspicuous colour if its projection on any vertical plane has both dimensions less than 1.5 m. Orange or red should be used, except where such colours merge with the background.

Note.— Against some backgrounds it may be found necessary to use a different colour from orange or red to obtain sufficient contrast.

6.2.6 Recommendation.— When mobile objects are marked by colour, a single conspicuous colour, preferably red or yellowish green for emergency vehicles and yellow for service vehicles should be used.

<table>
<thead>
<tr>
<th>Longest dimension</th>
<th>Not exceeding</th>
<th>Band width</th>
</tr>
</thead>
<tbody>
<tr>
<td>Greater than</td>
<td></td>
<td></td>
</tr>
<tr>
<td>1.5 m</td>
<td>210 m</td>
<td>1/7 of longest dimension</td>
</tr>
<tr>
<td>210 m</td>
<td>270 m</td>
<td>1/9</td>
</tr>
<tr>
<td>270 m</td>
<td>330 m</td>
<td>1/11</td>
</tr>
<tr>
<td>330 m</td>
<td>390 m</td>
<td>1/13</td>
</tr>
<tr>
<td>390 m</td>
<td>450 m</td>
<td>1/15</td>
</tr>
<tr>
<td>450 m</td>
<td>510 m</td>
<td>1/17</td>
</tr>
<tr>
<td>510 m</td>
<td>570 m</td>
<td>1/19</td>
</tr>
<tr>
<td>570 m</td>
<td>630 m</td>
<td>1/21</td>
</tr>
</tbody>
</table>

Use of markers

6.2.7 Markers displayed on or adjacent to objects shall be located in conspicuous positions so as to retain the general definition of the object and shall be recognizable in clear weather from a distance of at least 1 000 m for an object to be viewed from the air and 300 m for an object to be viewed from the ground in all directions in which an aircraft is likely to approach the object. The shape of markers shall be distinctive to the extent necessary to ensure that they are not mistaken for markers employed to convey other information, and they shall be such that the hazard presented by the object they mark is not increased.

6.2.8 Recommendation.— A marker displayed on an overhead wire, cable, etc., should be spherical and have a diameter of not less than 60 cm.

6.2.9 Recommendation.— The spacing between two consecutive markers or between a marker and a supporting tower should be appropriate to the diameter of the marker, but in no case should the spacing exceed:

a) 30 m where the marker diameter is 60 cm progressively increasing with the diameter of the marker to

b) 35 m where the marker diameter is 80 cm and further progressively increasing to a maximum of

c) 40 m where the marker diameter is of at least 130 cm.

Where multiple wires, cables, etc. are involved, a marker should be located not lower than the level of the highest wire at the point marked.

6.2.10 Recommendation.— A marker should be of one colour. When installed, white and red, or white and orange markers should be displayed alternately. The colour selected should contrast with the background against which it will be seen.

Use of flags

6.2.11 Flags used to mark objects shall be displayed around, on top of, or around the highest edge of, the object. When flags are used to mark extensive objects or groups of closely spaced objects, they shall be displayed at least every 15 m. Flags shall not increase the hazard presented by the object they mark.

6.2.12 Flags used to mark fixed objects shall not be less than 0.6 m square and flags used to mark mobile objects, not less than 0.9 m square.

6.2.13 Recommendation.— Flags used to mark fixed objects should be orange in colour or a combination of two triangular sections, one orange and the other white, or one red and the other white, except that where such colours merge with the background, other conspicuous colours should be used.

6.2.14 Flags used to mark mobile objects shall consist of a chequered pattern, each square having sides of not less than 0.3 m. The colours of the pattern shall contrast each with the other and with the background against which they will be seen. Orange and white or alternatively red and white shall be used, except where such colours merge with the background.
Chapter 6

Annex 14—Aerodromes

Figure 6-2. Examples of marking and lighting of tall structures

Note.—H is less the 45 m for the examples shown above. For greater heights intermediate lights must be added as shown below.

Number of lights \( N = \frac{Y \text{ (metres)}}{45} \)

Light spacing \( X = \frac{Y}{N} \leq 45 \text{ m} \)
6.3 Lighting of objects

Use of obstacle lights

6.3.1 The presence of objects which must be lighted, as specified in 6.1, shall be indicated by low-, medium- or high-intensity obstacle lights, or a combination of such lights.

Note.— High-intensity obstacle lights are intended for day use as well as night use. Care is needed to ensure that these lights do not create disconcerting dazzle. Guidance on the design, location and operation of high-intensity obstacle lights is given in the Aerodrome Design Manual, Part 4.

6.3.2 Recommendation.— Low-intensity obstacle lights, Type A or B, should be used where the object is a less extensive one and its height above the surrounding ground is less than 45 m.

6.3.3 Recommendation.— Where the use of low-intensity obstacle lights, Type A or B, would be inadequate or an early special warning is required, then medium- or high-intensity obstacle lights should be used.

6.3.4 Low-intensity obstacle lights, Type C, shall be displayed on vehicles and other mobile objects excluding aircraft.

6.3.5 Low-intensity obstacle lights, Type D, shall be displayed on follow-me vehicles.

6.3.6 Recommendation.— Low-intensity obstacle lights, Type B, should be used either alone or in combination with medium-intensity obstacle lights, Type B, in accordance with 6.3.7.

6.3.7 Recommendation.— Medium-intensity obstacle lights, Type A, B or C, should be used where the object is an extensive one or its height above the level of the surrounding ground is greater than 45 m. Medium-intensity obstacle lights, Types A and C, should be used alone, whereas medium-intensity obstacle lights, Type B, should be used either alone or in combination with low-intensity obstacle lights, Type B.

Note.— A group of trees or buildings is regarded as an extensive object.

6.3.8 Recommendation.— High-intensity obstacle lights, Type A, should be used to indicate the presence of an object if its height above the level of the surrounding ground exceeds 150 m and an aeronautical study indicates such lights to be essential for the recognition of the object by day.

6.3.9 Recommendation.— High-intensity obstacle lights, Type B, should be used to indicate the presence of a tower supporting overhead wires, cables, etc., where:

   a) an aeronautical study indicates such lights to be essential for the recognition of the presence of wires, cables, etc.; or

   b) it has not been found practicable to install markers on the wires, cables, etc.

6.3.10 Recommendation.— Where, in the opinion of the appropriate authority, the use of high-intensity obstacle lights, Type A or B, or medium-intensity obstacle lights, Type A, at night may dazzle pilots in the vicinity of an aerodrome (within approximately 10 000 m radius) or cause significant environmental concerns, a dual obstacle lighting system should be provided. This system should be composed of high-intensity obstacle lights, Type A or B, or medium-intensity obstacle lights, Type A, as appropriate, for daytime and twilight use and medium-intensity obstacle lights, Type B or C, for night-time use.

Location of obstacle lights

Note.— Recommendations on how a combination of low-, medium-, and/or high-intensity lights on obstacles should be displayed are given in Appendix 6.

6.3.11 One or more low-, medium- or high-intensity obstacle lights shall be located as close as practicable to the top of the object. The top lights shall be so arranged as to at least indicate the points or edges of the object highest in relation to the obstacle limitation surface.

6.3.12 Recommendation.— In the case of chimney or other structure of like function, the top lights should be placed sufficiently below the top so as to minimize contamination by smoke etc. (see Figures 6-2 and 6-3).

6.3.13 In the case of a tower or antenna structure indicated by high-intensity obstacle lights by day with an appurtenance, such as a rod or an antenna, greater than 12 m where it is not practicable to locate a high-intensity obstacle light on the top of the appurtenance, such a light shall be located at the highest practicable point and, if practicable, a medium-intensity obstacle light, Type A, mounted on the top.

6.3.14 In the case of an extensive object or of a group of closely spaced objects, top lights shall be displayed at least on the points or edges of the objects highest in relation to the obstacle limitation surface, so as to indicate the general definition and the extent of the objects. If two or more edges are of the same height, the edge nearest the landing area shall be marked. Where low-intensity lights are used, they shall be spaced at longitudinal intervals not exceeding 45 m. Where medium-intensity lights are used, they shall be spaced at longitudinal intervals not exceeding 900 m.

6.3.15 Recommendation.— When the obstacle limitation surface concerned is sloping and the highest point above the
obstacle limitation surface is not the highest point of the object, additional obstacle lights should be placed on the highest point of the object.

6.3.16 Where an object is indicated by medium-intensity obstacle lights, Type A, and the top of the object is more than 105 m above the level of the surrounding ground or the elevation of tops of nearby buildings (when the object to be marked is surrounded by buildings), additional lights shall be provided at intermediate levels. These additional intermediate lights shall be spaced as equally as practicable, between the top lights and ground level or the level of tops of nearby buildings, as appropriate, with the spacing not exceeding 105 m (see 6.3.7).

6.3.17 Where an object is indicated by medium-intensity obstacle lights, Type B, and the top of the object is more than 45 m above the level of the surrounding ground or the elevation of tops of nearby buildings (when the object to be marked is surrounded by buildings), additional lights shall be provided at intermediate levels. These additional intermediate lights shall be alternately low-intensity obstacle lights, Type B, and medium-intensity obstacle lights, Type B, and shall be spaced as equally as practicable between the top lights and ground level or the level of tops of nearby buildings, as appropriate, with the spacing not exceeding 52 m.

6.3.18 Where an object is indicated by medium-intensity obstacle lights, Type C, and the top of the object is more than 45 m above the level of the surrounding ground or the elevation of tops of nearby buildings (when the object to be marked is surrounded by buildings), additional lights shall be provided at intermediate levels. These additional intermediate lights shall be spaced as equally as practicable, between the top lights and ground level or the level of tops of nearby buildings, as appropriate, with the spacing not exceeding 52 m.

6.3.19 Where high-intensity obstacle lights, Type A, are used, they shall be spaced at uniform intervals not exceeding 105 m between the ground level and the top light(s) specified in 6.3.11 except that where an object to be marked is surrounded by buildings, the elevation of the tops of the buildings may be used as the equivalent of the ground level when determining the number of light levels.

6.3.20 Where high-intensity obstacle lights, Type B, are used, they shall be located at three levels:

— at the top of the tower;
— at the lowest level of the catenary of the wires or cables;
and
— at approximately midway between these two levels.

Note.— In some cases, this may require locating the lights off the tower.

6.3.21 Recommendation.— The installation setting angles for high-intensity obstacle lights, Types A and B, should be in accordance with Table 6-2.

6.3.22 The number and arrangement of low-, medium- or high-intensity obstacle lights at each level to be marked shall be such that the object is indicated from every angle in azimuth. Where a light is shielded in any direction by another part of the object, or by an adjacent object, additional lights shall be provided on that object in such a way as to retain the lights.
general definition of the object to be lighted. If the shielded light does not contribute to the definition of the object to be lighted, it may be omitted.

Table 6-2. Installation setting angles for high-intensity obstacle lights

<table>
<thead>
<tr>
<th>Height of light unit above terrain</th>
<th>Angle of the peak of the beam above the horizontal</th>
</tr>
</thead>
<tbody>
<tr>
<td>greater than 151 m AGL</td>
<td>0°</td>
</tr>
<tr>
<td>122 m to 151 m AGL</td>
<td>1°</td>
</tr>
<tr>
<td>92 m to 122 m AGL</td>
<td>2°</td>
</tr>
<tr>
<td>less than 92 m AGL</td>
<td>3°</td>
</tr>
</tbody>
</table>

Low-intensity obstacle light — Characteristics

6.3.23 Low-intensity obstacle lights on fixed objects, Types A and B, shall be fixed-red lights.

6.3.24 Low-intensity obstacle lights, Types A and B, shall be in accordance with the specifications in Table 6-3.

6.3.25 Low-intensity obstacle lights, Type C, displayed on vehicles associated with emergency or security shall be flashing-blue and those displayed on other vehicles shall be flashing-yellow.

6.3.26 Low-intensity obstacle lights, Type D, displayed on follow-me vehicles shall be flashing-yellow.

6.3.27 Low-intensity obstacle lights, Types C and D, shall be in accordance with the specifications in Table 6-3.

6.3.28 Low-intensity obstacle lights on objects with limited mobility such as aerobridges shall be fixed-red. The intensity of the lights shall be sufficient to ensure conspicuity considering the intensity of the adjacent lights and the general levels of illumination against which they would normally be viewed.

Note.— See Annex 2 for lights to be displayed by aircraft.

6.3.29 Low-intensity obstacle lights on objects with limited mobility shall as a minimum be in accordance with the specifications for low-intensity obstacle lights, Type A, in Table 6-3.

Medium-intensity obstacle light — Characteristics

6.3.30 Medium-intensity obstacle lights, Type A, shall be flashing-white lights, Type B shall be flashing-red lights and Type C shall be fixed-red lights.

6.3.31 Medium-intensity obstacle lights, Types A, B and C, shall be in accordance with the specifications in Table 6-3.

6.3.32 Medium-intensity obstacle lights, Types A and B, located on an object shall flash simultaneously.

High-intensity obstacle light — Characteristics

6.3.33 High-intensity obstacle lights, Types A and B, shall be flashing-white lights.

6.3.34 High-intensity obstacle lights, Types A and B, shall be in accordance with the specifications in Table 6-3.

6.3.35 High-intensity obstacle lights, Type A, located on an object shall flash simultaneously.

6.3.36 Recommendation.— High-intensity obstacle lights, Type B, indicating the presence of a tower supporting overhead wires, cables, etc., should flash sequentially; first the middle light, second the top light and last, the bottom light. The intervals between flashes of the lights should approximate the following ratios:

<table>
<thead>
<tr>
<th>Flash interval between</th>
<th>Ratio of cycle time</th>
</tr>
</thead>
<tbody>
<tr>
<td>middle and top light</td>
<td>1/13</td>
</tr>
<tr>
<td>top and bottom light</td>
<td>2/13</td>
</tr>
<tr>
<td>bottom and middle light</td>
<td>10/13</td>
</tr>
</tbody>
</table>
Table 6-3. Characteristics of obstacle lights

<table>
<thead>
<tr>
<th>Light Type</th>
<th>Colour</th>
<th>Signal type/ (flash rate)</th>
<th>Peak intensity (cd) at given Background Luminance</th>
<th>Vertical Beam Spread (c)</th>
<th>Intensity (cd) at given Elevation Angles when the light unit is levelled (d)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td></td>
<td></td>
<td>Above 500 cd/m²</td>
<td>50-500 cd/m²</td>
<td>Below 50 cd/m²</td>
</tr>
<tr>
<td>Low-intensity, Type A (fixed obstacle)</td>
<td>Red</td>
<td>Fixed</td>
<td>N/A</td>
<td>10 mm</td>
<td>10 mm</td>
</tr>
<tr>
<td>Low-intensity, Type B (fixed obstacle)</td>
<td>Red</td>
<td>Fixed</td>
<td>N/A</td>
<td>32 mm</td>
<td>32 mm</td>
</tr>
<tr>
<td>Low-intensity, Type C (mobile obstacle)</td>
<td>Yellow/Blue</td>
<td>Flashing (60-90 fpm)</td>
<td>N/A</td>
<td>40 mm (b)</td>
<td>40 mm (b)</td>
</tr>
<tr>
<td>Low-intensity, Type D Follow-me Vehicle</td>
<td>Yellow</td>
<td>Flashing (60-90 fpm)</td>
<td>N/A</td>
<td>200 mm (b)</td>
<td>200 mm (b)</td>
</tr>
<tr>
<td>Medium-intensity, Type A</td>
<td>White</td>
<td>Flashing (20-60 fpm)</td>
<td>20 000 (b)</td>
<td>20 000 (b)</td>
<td>2 000 (b)</td>
</tr>
<tr>
<td>Medium-intensity, Type B</td>
<td>Red</td>
<td>Flashing (20-60 fpm)</td>
<td>N/A</td>
<td>N/A</td>
<td>N/A</td>
</tr>
<tr>
<td>Medium-intensity, Type C</td>
<td>Red</td>
<td>Fixed</td>
<td>N/A</td>
<td>N/A</td>
<td>N/A</td>
</tr>
<tr>
<td>High-intensity, Type A</td>
<td>White</td>
<td>Flashing (40-60 fpm)</td>
<td>200 000 (b)</td>
<td>20 000 (b)</td>
<td>2 000 (b)</td>
</tr>
<tr>
<td>High-intensity, Type B</td>
<td>White</td>
<td>Flashing (40-60 fpm)</td>
<td>100 000 (b)</td>
<td>20 000 (b)</td>
<td>2 000 (b)</td>
</tr>
</tbody>
</table>

Note.— This table does not include recommended horizontal beam spreads. 6.3.22 requires 360° coverage around an obstacle. Therefore, the number of lights needed to meet this requirement will depend on the horizontal beam spreads of each light as well as the shape of the obstacle. Thus, with narrower beam spreads, more lights will be required.

a) See 6.3.25
b) Effective intensity, as determined in accordance with the Aerodrome Design Manual, Part 4.
c) Beam spread is defined as the angle between two directions in a plane for which the intensity is equal to 50% of the lower tolerance value of the intensity shown in columns 4, 5 and 6. The beam pattern is not necessarily symmetrical about the elevation angle at which the peak intensity occurs.
d) Elevation (vertical) angles are referenced to the horizontal.
e) Intensity at any specified horizontal radial as a percentage of the actual peak intensity at the same radial when operated at each of the intensities shown in columns 4, 5 and 6.
f) Intensity at any specified horizontal radial as a percentage of the lower tolerance value of the intensity shown in columns 4, 5 and 6.
g) In addition to specified values, lights shall have sufficient intensity to ensure conspicuity at elevation angles between ± 0° and 50°.
h) Peak intensity should be located at approximately 2.5° vertical.
i) Peak intensity should be located at approximately 17° vertical.

fpm — flashes per minute; N/A — not applicable
CHAPTER 7. VISUAL AIDS FOR DENOTING
RESTRICTED USE AREAS

7.1 Closed runways and taxiways,
or parts thereof

Application

7.1.1 A closed marking shall be displayed on a runway or
taxiway, or portion thereof, which is permanently closed to the
use of all aircraft.

7.1.2 Recommendation.— A closed marking should be
displayed on a temporarily closed runway or taxiway or
portion thereof, except that such marking may be omitted when
the closing is of short duration and adequate warning by air
traffic services is provided.

Location

7.1.3 On a runway a closed marking shall be placed at
each end of the runway, or portion thereof, declared closed,
and additional markings shall be so placed that the maximum
interval between markings does not exceed 300 m. On a
taxiway a closed marking shall be placed at least at each end
of the taxiway or portion thereof closed.

Characteristics

7.1.4 The closed marking shall be of the form and pro-
portions as detailed in Figure 7-1, Illustration a), when displayed
on a runway, and shall be of the form and proportions as
detailed in Figure 7-1, Illustration b), when displayed on a
taxiway. The marking shall be white when displayed on a
runway and shall be yellow when displayed on a taxiway.

Note.— When an area is temporarily closed, frangible
barriers or markings utilizing materials other than paint or
other suitable means may be used to identify the closed area.

7.1.5 When a runway or taxiway or portion thereof is
permanently closed, all normal runway and taxiway markings
shall be obliterated.

7.1.6 Lighting on a closed runway or taxiway or portion
thereof shall not be operated, except as required for mainten-
ance purposes.

7.1.7 In addition to closed markings, when the runway or
taxiway or portion thereof closed is intercepted by a usable
runway or taxiway which is used at night, unserviceability
lights shall be placed across the entrance to the closed area at
intervals not exceeding 3 m (see 7.4.4).

7.2 Non-load-bearing surfaces

Application

7.2.1 Shoulders for taxiways, holding bays and aprons
and other non-load-bearing surfaces which cannot readily be
distinguished from load-bearing surfaces and which, if used by
aircraft, might result in damage to the aircraft shall have the
boundary between such areas and the load-bearing surface
marked by a taxi side stripe marking.

Note.— The marking of runway sides is specified in 5.2.7.

Location

7.2.2 Recommendation.— A taxi side stripe marking
should be placed along the edge of the load-bearing pavement,
with the outer edge of the marking approximately on the edge
of the load-bearing pavement.

Characteristics

7.2.3 Recommendation.— A taxi side stripe marking
should consist of a pair of solid lines, each 15 cm wide and
spaced 15 cm apart and the same colour as the taxiway centre
line marking.

Note.— Guidance on providing additional transverse
stripes at an intersection or a small area on the apron is given

7.3 Pre-threshold area

Application

7.3.1 Recommendation.— When the surface before a
threshold is paved and exceeds 60 m in length and is not
suitable for normal use by aircraft, the entire length before the
threshold should be marked with a chevron marking.

Location

7.3.2 Recommendation.— A chevron marking should
point in the direction of the runway and be placed as shown
in Figure 7-2.
Chapter 7

Annex 14 — Aerodromes

Figure 7-1. Closed runway and taxiway markings

Figure 7-2. Pre-threshold marking
Annex 14 — Aerodromes

Characteristics

7.3.3 Recommendation.— A chevron marking should be of conspicuous colour and contrast with the colour used for the runway markings; it should preferably be yellow. It should have an over-all width of at least 0.9 m.

7.4 Unserviceable areas

Application

7.4.1 Unserviceability markers shall be displayed wherever any portion of a taxiway, apron or holding bay is unfit for the movement of aircraft but it is still possible for aircraft to bypass the area safely. On a movement area used at night, unserviceability lights shall be used.

Note.— Unserviceability markers and lights are used for such purposes as warning pilots of a hole in a taxiway or apron pavement or outlining a portion of pavement, such as on an apron, that is under repair. They are not suitable for use when a portion of a runway becomes unserviceable, nor on a taxiway when a major portion of the width becomes unserviceable. In such instances, the runway or taxiway is normally closed.

Location

7.4.2 Unserviceability markers and lights shall be placed at intervals sufficiently close so as to delineate the unserviceable area.

Note.— Guidance on the location of unserviceability lights is given in Attachment A, Section 13.

Characteristics of unserviceability markers

7.4.3 Unserviceability markers shall consist of conspicuous upstanding devices such as flags, cones or marker boards.

Characteristics of unserviceability lights

7.4.4 An unserviceability light shall consist of a red fixed light. The light shall have an intensity sufficient to ensure conspicuity considering the intensity of the adjacent lights and the general level of illumination against which it would normally be viewed. In no case shall the intensity be less than 10 cd of red light.

Characteristics of unserviceability cones

7.4.5 Recommendation.— An unserviceability cone should be at least 0.5 m in height and red, orange or yellow or any one of these colours in combination with white.

Characteristics of unserviceability flags

7.4.6 Recommendation.— An unserviceability flag should be at least 0.5 m square and red, orange or yellow or any one of these colours in combination with white.

Characteristics of unserviceability marker boards

7.4.7 Recommendation.— An unserviceability marker board should be at least 0.5 m in height and 0.9 m in length, with alternate red and white or orange and white vertical stripes.
CHAPTER 8. EQUIPMENT AND INSTALLATIONS

8.1 Secondary power supply

General

Application

8.1.1 Recommendation.— A secondary power supply should be provided, capable of supplying the power requirements of at least the aerodrome facilities listed below:

a) the signalling lamp and the minimum lighting necessary to enable air traffic services personnel to carry out their duties;

Note.— The requirement for minimum lighting may be met by other than electrical means.

b) all obstacle lights which, in the opinion of the appropriate authority, are essential to ensure the safe operation of aircraft;

c) approach, runway and taxiway lighting as specified in 8.1.6 to 8.1.9;

d) meteorological equipment;

e) essential security lighting, if provided in accordance with 8.5;

f) essential equipment and facilities for the aerodrome responding emergency agencies; and

g) floodlighting on a designated isolated aircraft parking position if provided in accordance with 5.3.21.1.

Note.— Specifications for secondary power supply for radio navigation aids and ground elements of communications systems are given in Annex 10, Volume I, Part I, Chapter 2.

Characteristics

8.1.2 Recommendation.— Electric power supply connections to those facilities for which secondary power is required should be so arranged that the facilities are automatically connected to the secondary power supply on failure of the normal source of power.

8.1.3 Recommendation.— The time interval between failure of the normal source of power and the complete restoration of the services required by 8.1.1 should be as short as practicable and should not exceed two minutes, except that for visual aids associated with non-precision, precision approach or take-off runways the requirements of Table 8-1 for maximum switch-over times should apply.

Note 1.— In certain cases, less than thirty seconds has been found to be attainable.

Note 2.— A definition of switch-over time is given in Chapter 1.

8.1.4 The provision of a definition of switch-over time shall not require the replacement of an existing secondary power supply before 1 January 2010. However, for a secondary power supply installed after 4 November 1999, the electric power supply connections to those facilities for which secondary power is required shall be so arranged that the facilities are capable of meeting the requirements of Table 8-1 for maximum switch-over times as defined in Chapter 1.

8.1.5 Recommendation.— Requirements for a secondary power supply should be met by either of the following:

— independent public power, which is a source of power supplying the aerodrome service from a substation other than the normal substation through a transmission line following a route different from the normal power supply route and such that the possibility of a simultaneous failure of the normal and independent public power supplies is extremely remote; or

— standby power unit(s), which are engine generators, batteries, etc., from which electric power can be obtained.

Note.— Guidance on secondary power supply is given in the Aerodrome Design Manual, Part 5.

Visual aids

Application

8.1.6 Recommendation.— At an aerodrome where the primary runway is a non-instrument runway, a secondary power supply capable of meeting the requirements of 8.1.3 should be provided, except that a secondary power supply for visual aids need not be provided when an emergency lighting system in accordance with the specification of 5.3.2 is provided and capable of being deployed in 15 minutes.
Table 8-1. Secondary power supply requirements
(see 8.1.3)

<table>
<thead>
<tr>
<th>Runway</th>
<th>Lighting aids requiring power</th>
<th>Maximum switch-over time</th>
</tr>
</thead>
<tbody>
<tr>
<td>Non-instrument</td>
<td>Visual approach slope indicators&lt;sup&gt;a&lt;/sup&gt;</td>
<td>See</td>
</tr>
<tr>
<td></td>
<td>Runway edge&lt;sup&gt;b&lt;/sup&gt;</td>
<td>8.1.3 and 8.1.6</td>
</tr>
<tr>
<td></td>
<td>Runway threshold&lt;sup&gt;b&lt;/sup&gt;</td>
<td></td>
</tr>
<tr>
<td></td>
<td>Runway end&lt;sup&gt;b&lt;/sup&gt;</td>
<td></td>
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<tr>
<td></td>
<td>Obstacle&lt;sup&gt;a&lt;/sup&gt;</td>
<td></td>
</tr>
<tr>
<td>Non-precision approach</td>
<td>Approach lighting system</td>
<td>15 seconds</td>
</tr>
<tr>
<td></td>
<td>Visual approach slope indicators&lt;sup&gt;a, d&lt;/sup&gt;</td>
<td></td>
</tr>
<tr>
<td></td>
<td>Runway edge&lt;sup&gt;d&lt;/sup&gt;</td>
<td>15 seconds</td>
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<tr>
<td></td>
<td>Runway threshold&lt;sup&gt;d&lt;/sup&gt;</td>
<td>15 seconds</td>
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<tr>
<td></td>
<td>Runway end</td>
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<tr>
<td></td>
<td>Obstacle&lt;sup&gt;a&lt;/sup&gt;</td>
<td>15 seconds</td>
</tr>
<tr>
<td>Precision approach category I</td>
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<td>15 seconds</td>
</tr>
<tr>
<td></td>
<td>Visual approach slope indicators&lt;sup&gt;a, d&lt;/sup&gt;</td>
<td>15 seconds</td>
</tr>
<tr>
<td></td>
<td>Runway threshold&lt;sup&gt;d&lt;/sup&gt;</td>
<td>15 seconds</td>
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<tr>
<td></td>
<td>Runway end</td>
<td>15 seconds</td>
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<tr>
<td></td>
<td>Essential taxiway&lt;sup&gt;a&lt;/sup&gt;</td>
<td>15 seconds</td>
</tr>
<tr>
<td></td>
<td>Obstacle&lt;sup&gt;a&lt;/sup&gt;</td>
<td>15 seconds</td>
</tr>
<tr>
<td>Precision approach category II/III</td>
<td>Approach lighting system</td>
<td>15 seconds</td>
</tr>
<tr>
<td></td>
<td>Supplementary approach lighting barrettes</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>Obstacle&lt;sup&gt;a&lt;/sup&gt;</td>
<td>15 seconds</td>
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<tr>
<td></td>
<td>Runway edge</td>
<td>15 seconds</td>
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<td></td>
<td>Runway threshold</td>
<td>1 second</td>
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<td></td>
<td>Runway end</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>Runway centre line</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>Runway touchdown zone</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>All stop bars</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>Essential taxiway</td>
<td>15 seconds</td>
</tr>
<tr>
<td></td>
<td>Essential taxiway</td>
<td>15 seconds</td>
</tr>
<tr>
<td>Runway meant for take-off in runway visual</td>
<td>Runway edge</td>
<td>15 seconds&lt;sup&gt;c&lt;/sup&gt;</td>
</tr>
<tr>
<td>range conditions less than a value of 800 m.</td>
<td>Runway end</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>Runway centre line</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>All stop bars</td>
<td>1 second</td>
</tr>
<tr>
<td></td>
<td>Essential taxiway&lt;sup&gt;a&lt;/sup&gt;</td>
<td>15 seconds</td>
</tr>
<tr>
<td></td>
<td>Obstacle&lt;sup&gt;a&lt;/sup&gt;</td>
<td>15 seconds</td>
</tr>
</tbody>
</table>

<sup>a</sup> Supplied with secondary power when their operation is essential to the safety of flight operation.

<sup>b</sup> See Chapter 5, 5.3.2 regarding the use of emergency lighting.

<sup>c</sup> One second where no runway centre line lights are provided.

<sup>d</sup> One second where approaches are over hazardous or precipitous terrain.
Chapter 8

8.1.7 Recommendation.— At an aerodrome where the primary runway is a non-precision approach runway, a secondary power supply capable of meeting the requirements of Table 8-1 should be provided except that a secondary power supply for visual aids need not be provided for more than one non-precision approach runway.

8.1.8 For a precision approach runway, a secondary power supply capable of meeting the requirements of Table 8-1 for the appropriate category of precision approach runway shall be provided. Electric power supply connections to those facilities for which secondary power is required shall be so arranged that the facilities are automatically connected to the secondary power supply on failure of the normal source of power.

8.1.9 For a runway meant for take-off in runway visual range conditions less than a value of 800 m, a secondary power supply capable of meeting the relevant requirements of Table 8-1 shall be provided.

Note.— Guidance on electrical systems is included in the Aerodrome Design Manual, Part 5.

8.2 Electrical systems

8.2.1 For a runway meant for use in runway visual range conditions less than a value of 550 m, the electrical systems for the power supply, lighting and control of the lighting systems included in Table 8-1 shall be so designed that an equipment failure will not leave the pilot with inadequate visual guidance or misleading information.

Note.— Guidance on means of providing this protection is given in the Aerodrome Design Manual, Part 5.

8.2.2 Where the secondary power supply of an aerodrome is provided by the use of duplicate feeders, such supplies shall be physically and electrically separate so as to ensure the required level of availability and independence.

Note.— Guidance on acceptable power source arrangements for the use of duplicate feeders for a secondary power supply is given in the Aerodrome Design Manual, Part 5.

8.2.3 Where a runway forming part of a standard taxi-route is provided with runway lighting and taxiway lighting, the lighting systems shall be interlocked to preclude the possibility of simultaneous operation of both forms of lighting.

8.3 Monitoring

8.3.1 Recommendation.— A system of monitoring visual aids should be employed to ensure lighting system reliability.

8.3.2 Where lighting systems are used for aircraft control purposes, such systems shall be monitored automatically so as to provide an immediate indication of any fault which may affect the control functions. This information shall be automatically relayed to the air traffic service unit.

8.3.3 Recommendation.— For a runway meant for use in runway visual range conditions less than a value of 550 m, the lighting systems detailed in Table 8-1 should be monitored so as to provide an immediate indication when the serviceability level of any element falls below the minimum serviceability level specified in 9.4.26 to 9.4.30, as appropriate. This information should be immediately relayed to the maintenance crew.

8.3.4 Recommendation.— For a runway meant for use in runway visual range conditions less than a value of 550 m, the lighting systems detailed in Table 8-1 should be monitored automatically to provide an immediate indication when the serviceability level of any element falls below the minimum level specified by the appropriate authority below which operations should not continue. This information should be automatically relayed to the air traffic services unit and displayed in a prominent position.

Note.— Guidance on air traffic control interface and visual aids monitoring is included in the Aerodrome Design Manual, Part 5.

8.4 Fencing

Application

8.4.1 Recommendation.— A fence or other suitable barrier should be provided on an aerodrome to prevent the entrance to the movement area of animals large enough to be a hazard to aircraft.

8.4.2 Recommendation.— A fence or other suitable barrier should be provided on an aerodrome to deter the inadvertent or premeditated access of an unauthorized person onto a non-public area of the aerodrome.

Note 1.— This is intended to include the barring of sewers, ducts, tunnels, etc., where necessary to prevent access.

Note 2.— Special measures may be required to prevent the access of an unauthorized person to runways or taxiways which overpass public roads.
8.4.3 **Recommendation.** Suitable means of protection should be provided to deter the inadvertent or premeditated access of unauthorized persons into ground installations and facilities essential for the safety of civil aviation located off the aerodrome.

**Location**

8.4.4 **Recommendation.** The fence or barrier should be located so as to separate the movement area and other facilities or zones on the aerodrome vital to the safe operation of aircraft from areas open to public access.

8.4.5 **Recommendation.** When greater security is thought necessary, a cleared area should be provided on both sides of the fence or barrier to facilitate the work of patrols and to make trespassing more difficult. Consideration should be given to the provision of a perimeter road inside the aerodrome fencing for the use of both maintenance personnel and security patrols.

8.5 **Security lighting**

**Recommendation.** At an aerodrome where it is deemed desirable for security reasons, a fence or other barrier provided for the protection of international civil aviation and its facilities should be illuminated at a minimum essential level. Consideration should be given to locating lights so that the ground area on both sides of the fence or barrier, particularly at access points, is illuminated.

8.6 **Airport design**

8.6.1 Architectural and infrastructure-related requirements for the optimum implementation of international civil aviation security measures shall be integrated into the design and construction of new facilities and alterations to existing facilities at an aerodrome.

*Note.* Guidance on all aspects of the planning of aerodromes including security considerations is contained in the Aerodrome Planning Manual, Part 1.

8.7 **Siting and construction of equipment and installations on operational areas**

*Note 1.* Requirements for obstacle limitation surfaces are specified in 4.2.

*Note 2.* The design of light fixtures and their supporting structures, light units of visual approach slope indicators, signs, and markers, is specified in 5.3.1, 5.3.5, 5.4.1 and 5.5.1, respectively. Guidance on the frangible design of visual and non-visual aids for navigation is given in the Aerodrome Design Manual, Part 6 (in preparation).

8.7.1 Unless its function requires it to be there for air navigation purposes, no equipment or installation shall be:

a) on a runway strip, a runway end safety area, a taxiway strip or within the distances specified in Table 3-1, column 11, if it would endanger an aircraft; or

b) on a clearway if it would endanger an aircraft in the air.

8.7.2 Any equipment or installation required for air navigation purposes which must be located:

a) on that portion of a runway strip within:

1) 75 m of the runway centre line where the code number is 3 or 4; or

2) 45 m of the runway centre line where the code number is 1 or 2; or

b) on a runway end safety area, a taxiway strip or within the distances specified in Table 3-1; or

c) on a clearway and which would endanger an aircraft in the air;

shall be frangible and mounted as low as possible.

8.7.3 Existing non-visual aids need not meet the requirement of 8.7.2 until 1 January 2010.

8.7.4 **Recommendation.** Any equipment or installation required for air navigation purposes which must be located on the non-graded portion of a runway strip should be regarded as an obstacle and should be frangible and mounted as low as possible.


8.7.5 Unless its function requires it to be there for air navigation purposes, no equipment or installation shall be located within 240 m from the end of the strip and within:

a) 60 m of the extended centre line where the code number is 3 or 4; or

b) 45 m of the extended centre line where the code number is 1 or 2;

of a precision approach runway category I, II or III.

8.7.6 Any equipment or installation required for air navigation purposes which must be located on or near a strip of a precision approach runway category I, II or III and which:
8.7.7 Existing non-visual aids need not meet the requirement of 8.7.6 b) until 1 January 2010.

Note.— See 5.3.1.4 for the protection date for existing elevated approach lights.

8.7.8 Recommendation.— Any equipment or installation required for air navigation purposes which is an obstacle of operational significance in accordance with 4.2.4, 4.2.11, 4.2.20 or 4.2.27 should be frangible and mounted as low as possible.

8.8 Aerodrome vehicle operations

Note 1.— Guidance on aerodrome vehicle operations is contained in Attachment A, Section 17 and on traffic rules and regulations for vehicles is contained in the Manual of Surface Movement Guidance and Control Systems (SMGCS).

Note 2.— It is intended that roads located on the movement area be restricted to the exclusive use of aerodrome personnel and other authorized persons, and that access to the public buildings by an unauthorized person will not require use of such roads.

8.8.1 A vehicle shall be operated:

a) on a manoeuvring area only as authorized by the aerodrome control tower; and

b) on an apron only as authorized by the appropriate designated authority.

8.8.2 The driver of a vehicle on the movement area shall comply with all mandatory instructions conveyed by markings and signs unless otherwise authorized by:

a) the aerodrome control tower when on the manoeuvring area; or

b) the appropriate designated authority when on the apron.

8.8.3 The driver of a vehicle on the movement area shall comply with all mandatory instructions conveyed by lights.

8.8.4 The driver of a vehicle on the movement area shall be appropriately trained for the tasks to be performed and shall comply with the instructions issued by:

a) the aerodrome control tower, when on the manoeuvring area; and

b) the appropriate designated authority, when on the apron.

8.8.5 The driver of a radio-equipped vehicle shall establish satisfactory two-way radio communication with the aerodrome control tower before entering the manoeuvring area and with the appropriate designated authority before entering the apron. The driver shall maintain a continuous listening watch on the assigned frequency when on the movement area.

8.9 Surface movement guidance and control systems

Application

8.9.1 A surface movement guidance and control system shall be provided at an aerodrome.


Characteristics

8.9.2 Recommendation.— The design of a surface movement guidance and control system should take into account:

a) the density of air traffic;

b) the visibility conditions under which operations are intended;

c) the need for pilot orientation;

d) the complexity of the aerodrome layout; and

e) movements of vehicles.

8.9.3 Recommendation.— The visual aid components of a surface movement guidance and control system, i.e. markings, lights and signs should be designed to conform with the relevant specifications in 5.2, 5.3 and 5.4, respectively.
8.9.4 **Recommendation.**— A surface movement guidance and control system should be designed to assist in the prevention of inadvertent incursions of aircraft and vehicles onto an active runway.

8.9.5 **Recommendation.**— The system should be designed to assist in the prevention of collisions between aircraft, and between aircraft and vehicles or objects, on any part of the movement area.

**Note.**— Guidance on control of stop bars through induction loops and on a visual taxiing guidance and control system is contained in the *Aerodrome Design Manual, Part 4.*

8.9.6 Where a surface movement guidance and control system is provided by selective switching of stop bars and taxiway centre line lights, the following requirements shall be met:

a) taxiway routes which are indicated by illuminated taxiway centre line lights shall be capable of being terminated by an illuminated stop bar;

b) the control circuits shall be so arranged that when a stop bar located ahead of an aircraft is illuminated the appropriate section of taxiway centre line lights beyond it is suppressed; and

c) the taxiway centre line lights are activated ahead of an aircraft when the stop bar is suppressed.

**Note 1.**— See Sections 5.3.15 and 5.3.17 for specifications on taxiway centre line lights and stop bars, respectively.

**Note 2.**— Guidance on installation of stop bars and taxiway centre line lights in surface movement guidance and control systems is given in the *Aerodrome Design Manual, Part 4.*

8.9.7 **Recommendation.**— Surface movement radar for the manoeuvring area should be provided at an aerodrome intended for use in runway visual range conditions less than a value of 350 m.

8.9.8 **Recommendation.**— Surface movement radar for the manoeuvring area should be provided at an aerodrome other than that in 8.9.7 when traffic density and operating conditions are such that regularity of traffic flow cannot be maintained by alternative procedures and facilities.

**Note.**— Guidance on the use of surface movement radar is given in the *Manual of Surface Movement Guidance and Control Systems (SMGCS)* and in the *Air Traffic Services Planning Manual (Doc 9426).*
CHAPTER 9. EMERGENCY AND OTHER SERVICES

9.1 Aerodrome emergency planning

General

Introductory Note.— Aerodrome emergency planning is the process of preparing an aerodrome to cope with an emergency occurring at the aerodrome or in its vicinity. The objective of aerodrome emergency planning is to minimize the effects of an emergency, particularly in respect of saving lives and maintaining aircraft operations. The aerodrome emergency plan sets forth the procedures for coordinating the response of different aerodrome agencies (or services) and of those agencies in the surrounding community that could be of assistance in responding to the emergency. Guidance material to assist the appropriate authority in establishing aerodrome emergency planning is given in the Airport Services Manual, Part 7.

9.1.1 An aerodrome emergency plan shall be established at an aerodrome, commensurate with the aircraft operations and other activities conducted at the aerodrome.

9.1.2 The aerodrome emergency plan shall provide for the coordination of the actions to be taken in an emergency occurring at an aerodrome or in its vicinity.

Note.— Examples of emergencies are: aircraft emergencies, sabotage including bomb threats, unlawfully seized aircraft, dangerous goods occurrences, building fires and natural disasters.

9.1.3 The plan shall coordinate the response or participation of all existing agencies which, in the opinion of the appropriate authority, could be of assistance in responding to an emergency.

Note.— Examples of agencies are:

— on the aerodrome: air traffic control unit, rescue and fire fighting services, aerodrome administration, medical and ambulance services, aircraft operators, security services, and police;

— off the aerodrome: fire departments, police, medical and ambulance services, hospitals, military, and harbour patrol or coast guard.

9.1.4 Recommendation.— The plan should provide for cooperation and coordination with the rescue coordination centre, as necessary.

9.1.5 Recommendation.— The aerodrome emergency plan document should include at least the following:

a) types of emergencies planned for;

b) agencies involved in the plan;

c) responsibility and role of each agency, the emergency operations centre and the command post, for each type of emergency;

d) information on names and telephone numbers of offices or people to be contacted in the case of a particular emergency; and

e) a grid map of the aerodrome and its immediate vicinity.

9.1.6 The plan shall observe Human Factors principles to ensure optimum response by all existing agencies participating in emergency operations.

Note.— Guidance material on Human Factors principles can be found in the Human Factors Training Manual.

Emergency operations centre and command post

9.1.7 Recommendation.— A fixed emergency operations centre and a mobile command post should be available for use during an emergency.

9.1.8 Recommendation.— The emergency operations centre should be a part of the aerodrome facilities and should be responsible for the overall coordination and general direction of the response to an emergency.

9.1.9 Recommendation.— The command post should be a facility capable of being moved rapidly to the site of an emergency, when required, and should undertake the local coordination of those agencies responding to the emergency.

9.1.10 Recommendation.— A person should be assigned to assume control of the emergency operations centre and, when appropriate, another person the command post.

Communication system

9.1.11 Recommendation.— Adequate communication systems linking the command post and the emergency operations centre with each other and with the participating agencies should be provided in accordance with the plan and consistent with the particular requirements of the aerodrome.
Annex 14 — Aerodromes

Aerodrome emergency exercise

9.1.12 The plan shall contain procedures for periodic testing of the adequacy of the plan and for reviewing the results in order to improve its effectiveness.

Note.— The plan includes all participating agencies and associated equipment.

9.1.13 The plan shall be tested by conducting:

a) a full-scale aerodrome emergency exercise at intervals not exceeding two years; and

b) partial emergency exercises in the intervening year to ensure that any deficiencies found during the full-scale aerodrome emergency exercise have been corrected; and reviewed thereafter, or after an actual emergency, so as to correct any deficiency found during such exercises or actual emergency.

Note.— The purpose of a full-scale exercise is to ensure the adequacy of the plan to cope with different types of emergencies. The purpose of a partial exercise is to ensure the adequacy of the response to individual participating agencies and components of the plan, such as the communications system.

Emergencies in difficult environments

9.1.14 The plan shall include the ready availability of and coordination with appropriate specialist rescue services to be able to respond to emergencies where an aerodrome is located close to water and/or swampy areas and where a significant portion of approach or departure operations takes place over these areas.

9.1.15 Recommendation.— At those aerodromes located close to water and/or swampy areas, or difficult terrain, the aerodrome emergency plan should include the establishment, testing and assessment at regular intervals of a pre-determined response for the specialist rescue services.

9.2 Rescue and fire fighting

General

Introductory Note.— The principal objective of a rescue and fire fighting service is to save lives. For this reason, the provision of means of dealing with an aircraft accident or incident occurring at, or in the immediate vicinity of, an aerodrome assumes primary importance because it is within this area that there are the greatest opportunities of saving lives. This must assume at all times the possibility of, and need for, extinguishing a fire which may occur either immediately following an aircraft accident or incident, or at any time during rescue operations.

The most important factors bearing on effective rescue in a survivable aircraft accident are: the training received, the effectiveness of the equipment and the speed with which personnel and equipment designated for rescue and fire fighting purposes can be put into use.

Requirements to combat building and fuel farm fires, or to deal with foaming of runways, are not taken into account.

Application

9.2.1 Rescue and fire fighting equipment and services shall be provided at an aerodrome.

Note.— Public or private organizations, suitably located and equipped, may be designated to provide the rescue and fire fighting service. It is intended that the fire station housing these organizations be normally located on the aerodrome, although an off-aerodrome location is not precluded provided the response time can be met.

9.2.2 Where an aerodrome is located close to water/swampy areas, or difficult terrain, and where a significant portion of approach or departure operations takes place over these areas, specialist rescue services and fire fighting equipment appropriate to the hazard and risk shall be available.

Note 1.— Special fire fighting equipment need not be provided for water areas; this does not prevent the provision of such equipment if it would be of practical use, such as when the areas concerned include reefs or islands.

Note 2.— The objective is to plan and deploy the necessary life-saving flotation equipment as expeditiously as possible in a number commensurate with the largest aeroplane normally using the aerodrome.

Note 3.— Additional guidance is available in Chapter 13 of the Airport Services Manual, Part 1.

Level of protection to be provided

9.2.3 The level of protection provided at an aerodrome for rescue and fire fighting shall be appropriate to the aerodrome category determined using the principles in 9.2.5 and 9.2.6, except that, where the number of movements of the aeroplanes in the highest category normally using the aerodrome is less than 700 in the busiest consecutive three months, the level of protection provided shall be not less than one category below the determined category.

Note.— Either a take-off or a landing constitutes a movement.
9.2.4 **Recommendation.**— From 1 January 2005, the level of protection provided at an aerodrome for rescue and fire fighting should be equal to the aerodrome category determined using the principles in 9.2.5 and 9.2.6.

9.2.5 The aerodrome category shall be determined from Table 9-1 and shall be based on the longest aeroplanes normally using the aerodrome and their fuselage width.

**Note.**— To categorize the aeroplanes using the aerodrome, first evaluate their overall length and second, their fuselage width.

9.2.6 If, after selecting the category appropriate to the longest aeroplane’s overall length, that aeroplane’s fuselage width is greater than the maximum width in Table 9-1, column 3 for that category, then the category for that aeroplane shall actually be one category higher.

**Note.**— Guidance on categorizing aerodromes for rescue and fire fighting purposes and on providing rescue and fire fighting equipment and services is given in Attachment A, Section 16 and in the Airport Services Manual, Part 1.

Table 9-1. Aerodrome category for rescue and fire fighting

<table>
<thead>
<tr>
<th>Aerodrome category</th>
<th>Aeroplane overall length</th>
<th>Maximum fuselage width</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>0 m up to but not including 9 m</td>
<td>2 m</td>
</tr>
<tr>
<td>2</td>
<td>9 m up to but not including 12 m</td>
<td>2 m</td>
</tr>
<tr>
<td>3</td>
<td>12 m up to but not including 18 m</td>
<td>3 m</td>
</tr>
<tr>
<td>4</td>
<td>18 m up to but not including 24 m</td>
<td>4 m</td>
</tr>
<tr>
<td>5</td>
<td>24 m up to but not including 28 m</td>
<td>4 m</td>
</tr>
<tr>
<td>6</td>
<td>28 m up to but not including 39 m</td>
<td>5 m</td>
</tr>
<tr>
<td>7</td>
<td>39 m up to but not including 49 m</td>
<td>5 m</td>
</tr>
<tr>
<td>8</td>
<td>49 m up to but not including 61 m</td>
<td>7 m</td>
</tr>
<tr>
<td>9</td>
<td>61 m up to but not including 76 m</td>
<td>7 m</td>
</tr>
<tr>
<td>10</td>
<td>76 m up to but not including 90 m</td>
<td>8 m</td>
</tr>
</tbody>
</table>

9.2.7 During anticipated periods of reduced activity, the level of protection available shall be no less than that needed for the highest category of aeroplane planned to use the aerodrome during that time irrespective of the number of movements.

9.2.8 **Recommendation.**— Both principal and complementary agents should normally be provided at an aerodrome.

**Note.**— Descriptions of the agents may be found in the Airport Services Manual, Part 1.

9.2.9 **Recommendation.**— The principal extinguishing agent should be:

a) a foam meeting the minimum performance level A; or

b) a foam meeting the minimum performance level B; or

c) a combination of these agents;

except that the principal extinguishing agent for aerodromes in categories 1 to 3 should preferably meet the minimum performance level B.

**Note.**— Information on the required physical properties and fire extinguishing performance criteria needed for a foam to achieve an acceptable performance level A or B rating is given in the Airport Services Manual, Part 1.

9.2.10 **Recommendation.**— The complementary extinguishing agent should be a dry chemical powder suitable for extinguishing hydrocarbon fires.

**Note 1.**— When selecting dry chemical powders for use with foam, care must be exercised to ensure compatibility.

**Note 2.**— Alternate complementary agents having equivalent fire fighting capability may be utilized. Additional information on extinguishing agents is given in the Airport Services Manual, Part 1.

9.2.11 The amounts of water for foam production and the complementary agents to be provided on the rescue and fire fighting vehicles shall be in accordance with the aerodrome category determined under 9.2.3, 9.2.4, 9.2.5, 9.2.6 and Table 9-2, except that these amounts may be modified as follows:

a) for aerodrome categories 1 and 2 up to 100 per cent of the water may be replaced by complementary agent; or

b) for aerodrome categories 3 to 10 when a foam meeting performance level A is used, up to 30 per cent of the water may be replaced by complementary agent.

For the purpose of agent substitution, the following equivalents shall be used:
Note 1.— The amounts of water specified for foam production are predicated on an application rate of 8.2 L/min/m² for a foam meeting performance level A, and 5.5 L/min/m² for a foam meeting performance level B.

Note 2.— When any other complementary agent is used, the substitution ratios need to be checked.

9.2.12 The quantity of foam concentrates separately provided on vehicles for foam production shall be in proportion to the quantity of water provided and the foam concentrate selected.

9.2.13 Recommendation.— The amount of foam concentrate provided on a vehicle should be sufficient to produce at least two loads of foam solution.

9.2.14 Recommendation.— Supplementary water supplies, for the expeditious replenishment of rescue and fire fighting vehicles at the scene of an aircraft accident, should be provided.

9.2.15 Recommendation.— When both a foam meeting performance level A and a foam meeting performance level B are to be used, the total amount of water to be provided for foam production should first be based on the quantity which would be required if only a foam meeting performance level A were used, and then reduced by 3 L for each 2 L of water provided for the foam meeting performance level B.

9.2.16 The discharge rate of the foam solution shall not be less than the rates shown in Table 9-2.

9.2.17 Recommendation.— The complementary agents should comply with the appropriate specifications of the International Organization for Standardization (ISO).*

9.2.18 Recommendation.— The discharge rate of complementary agents should be selected for optimum effectiveness of the agent.

9.2.19 Recommendation.— A reserve supply of foam concentrate and complementary agent, equivalent to 200 per cent of the quantities of these agents to be provided in the rescue and fire fighting vehicles, should be maintained on the aerodrome for vehicle replenishment purposes. Where a major delay in the replenishment of this supply is anticipated, the amount of reserve supply should be increased.

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Table 9-2. Minimum usable amounts of extinguishing agents

<table>
<thead>
<tr>
<th>Aerodrome category</th>
<th>Foam meeting performance level A</th>
<th>Foam meeting performance level B</th>
<th>Complementary agents</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Water¹ (L)</td>
<td>Discharge rate foam solution/ minute</td>
<td>Water¹ (L)</td>
</tr>
<tr>
<td>1</td>
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<td>350</td>
<td>230</td>
</tr>
<tr>
<td>2</td>
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<td>800</td>
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</tr>
<tr>
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<tr>
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<td>2600</td>
<td>2400</td>
</tr>
<tr>
<td>5</td>
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<tr>
<td>10</td>
<td>48200</td>
<td>16600</td>
<td>32300</td>
</tr>
</tbody>
</table>

Note 1.— The quantities of water shown in columns 2 and 4 are based on the average overall length of aeroplanes in a given category. Where operations of an aeroplane larger than the average size are expected, the quantities of water would need to be recalculated. See the Airport Services Manual, Part I for additional guidance.

Note 2.— Any other complementary agent having equivalent fire fighting capability may be used.

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1 kg complementary agent = 1.0 L water for production of a foam meeting performance level A

1 kg complementary agent = 0.66 L water for production of a foam meeting performance level B

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* See ISO Publications 5923 (Carbon Dioxide), 7201 (Halogenated Hydrocarbons) and 7202 (Powder).
Rescue equipment

9.2.20 **Recommendation.**— Rescue equipment commensurate with the level of aircraft operations should be provided on the rescue and fire fighting vehicle(s).

Note.— Guidance on the rescue equipment to be provided at an aerodrome is given in the Airport Services Manual, Part 1.

Response time

9.2.21 The operational objective of the rescue and fire fighting service shall be to achieve a response time not exceeding three minutes to any point of each operational runway, in optimum visibility and surface conditions.

9.2.22 **Recommendation.**— The operational objective of the rescue and fire fighting service should be to achieve a response time not exceeding two minutes to any point of each operational runway, in optimum visibility and surface conditions.

9.2.23 **Recommendation.**— The operational objective of the rescue and fire fighting service should be to achieve a response time not exceeding three minutes to any other part of the movement area in optimum visibility and surface conditions.

Note 1.— Response time is considered to be the time between the initial call to the rescue and fire fighting service, and the time when the first responding vehicle(s) is (are) in position to apply foam at a rate of at least 50 per cent of the discharge rate specified in Table 9-2.

Note 2.— To meet the operational objective as nearly as possible in less than optimum conditions of visibility, it may be necessary to provide suitable guidance and/or procedures for rescue and fire fighting vehicles.

Note 3.— Optimum visibility and surface conditions are defined as daytime, good visibility, no precipitation with normal response route free of surface contamination e.g. water, ice or snow.

9.2.24 **Recommendation.**— Any other vehicles required to deliver the amounts of extinguishing agents specified in Table 9-2 should arrive no more than one minute after the first responding vehicle(s) so as to provide continuous agent application.

9.2.25 **Recommendation.**— A system of preventive maintenance of rescue and fire fighting vehicles should be employed to ensure effectiveness of the equipment and compliance with the specified response time throughout the life of the vehicle.

Emergency access roads

9.2.26 **Recommendation.**— Emergency access roads should be provided on an aerodrome where terrain conditions permit their construction, so as to facilitate achieving minimum response times. Particular attention should be given to the provision of ready access to approach areas up to 1 000 m from the threshold, or at least within the aerodrome boundary. Where a fence is provided, the need for convenient access to outside areas should be taken into account.

Note.— Aerodrome service roads may serve as emergency access roads when they are suitably located and constructed.

9.2.27 **Recommendation.**— Emergency access roads should be capable of supporting the heaviest vehicles which will use them, and be usable in all weather conditions. Roads within 90 m of a runway should be surfaced to prevent surface erosion and the transfer of debris to the runway. Sufficient vertical clearance should be provided from overhead obstructions for the largest vehicles.

9.2.28 **Recommendation.**— When the surface of the road is indistinguishable from the surrounding area, or in areas where snow may obscure the location of the roads, edge markers should be placed at intervals of about 10 m.

Fire stations

9.2.29 **Recommendation.**— All rescue and fire fighting vehicles should normally be housed in a fire station. Satellite fire stations should be provided whenever the response time cannot be achieved from a single fire station.

9.2.30 **Recommendation.**— The fire station should be located so that the access for rescue and fire fighting vehicles into the runway area is direct and clear, requiring a minimum number of turns.

Communication and alerting systems

9.2.31 **Recommendation.**— A discrete communication system should be provided linking a fire station with the control tower, any other fire station on the aerodrome and the rescue and fire fighting vehicles.

9.2.32 **Recommendation.**— An alerting system for rescue and fire fighting personnel, capable of being operated from that station, should be provided at a fire station, any other fire station on the aerodrome and the aerodrome control tower.

Number of rescue and fire fighting vehicles

9.2.33 **Recommendation.**— The minimum number of rescue and fire fighting vehicles provided at an aerodrome should be in accordance with the following tabulation:
Annex 14 — Aerodromes

<table>
<thead>
<tr>
<th>Aerodrome category</th>
<th>Rescue and fire fighting vehicles</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
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</tr>
<tr>
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<td>1</td>
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<tr>
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<td>9</td>
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<tr>
<td>10</td>
<td>3</td>
</tr>
</tbody>
</table>

Note.— Guidance on minimum characteristics of rescue and fire fighting vehicles is given in the Airport Services Manual, Part 1.

Personnel

9.2.34 All rescue and fire fighting personnel shall be properly trained to perform their duties in an efficient manner and shall participate in live fire drills commensurate with the types of aircraft and type of rescue and fire fighting equipment in use at the aerodrome, including pressure-fed fuel fires.

Note 1.— Guidance to assist the appropriate authority in providing proper training is given in Attachment A, Section 16 of this volume of Annex 14; Airport Services Manual, Part 1; and Training Manual, Part E-2.

9.2.35 The rescue and fire fighting personnel training programme shall include training in human performance, including team coordination.

Note.— Guidance material to design training programmes on human performance and team coordination can be found in the Human Factors Training Manual.

9.2.36 Recommendation.— During flight operations, sufficient trained personnel should be detailed and be readily available to ride the rescue and fire fighting vehicles and to operate the equipment at maximum capacity. These trained personnel should be deployed in a way that ensures that minimum response times can be achieved and that continuous agent application at the appropriate rate can be fully maintained. Consideration should also be given for personnel to use hand lines, ladders and other rescue and fire fighting equipment normally associated with aircraft rescue and fire fighting operations.

9.2.37 Recommendation.— In determining the number of personnel required to provide for rescue, consideration should be given to the types of aircraft using the aerodrome.

9.2.38 All responding rescue and fire fighting personnel shall be provided with protective clothing and respiratory equipment to enable them to perform their duties in an effective manner.

9.3 Disabled aircraft removal

Note.— Guidance on removal of a disabled aircraft, including recovery equipment, is given in the Airport Services Manual, Part 5. See also Annex 13 concerning protection of evidence, custody and removal of aircraft.

9.3.1 Recommendation.— A plan for the removal of an aircraft disabled on, or adjacent to, the movement area should be established for an aerodrome, and a coordinator designated to implement the plan, when necessary.

9.3.2 Recommendation.— The disabled aircraft removal plan should be based on the characteristics of the aircraft that may normally be expected to operate at the aerodrome, and include among other things:

a) a list of equipment and personnel on, or in the vicinity of, the aerodrome which would be available for such purpose; and

b) arrangements for the rapid receipt of aircraft recovery equipment kits available from other aerodromes.

9.4 Maintenance

General

9.4.1 Recommendation.— A maintenance programme, including preventive maintenance where appropriate, should be established at an aerodrome to maintain facilities in a condition which does not impair the safety, regularity or efficiency of air navigation.

Note 1.— Preventive maintenance is programmed maintenance work done in order to prevent a failure or degradation of facilities.

Note 2.— “Facilities” are intended to include such items as pavements, visual aids, fencing, drainage systems and buildings.

9.4.2 Recommendation.— The design and application of the maintenance programme should observe Human Factors principles.

Note.— Guidance material on Human Factors principles can be found in the Human Factors Training Manual.
Chapter 9

Pavements

9.4.3 Recommendation.— The surface of pavements (runways, taxiways, aprons, etc.) should be kept clear of any loose stones or other objects that might cause damage to aircraft structures or engines, or impair the operation of aircraft systems.

Note.— Guidance on precautions to be taken in regard to the surface of shoulders is given in Attachment A, Section 8, and the Aerodrome Design Manual, Part 2.

9.4.4 Recommendation.— The surface of a runway should be maintained in a condition such as to preclude formation of harmful irregularities.

Note.— See Attachment A, Section 5.

9.4.5 Measurements of the friction characteristics of a runway surface shall be made periodically with a continuous friction measuring device using self-wetting features.

Note.— Guidance on evaluating the friction characteristics of a runway is provided in Attachment A, Section 7. Additional guidance is included in the Airport Services Manual, Part 2.

9.4.6 Corrective maintenance action shall be taken when the friction characteristics for either the entire runway or a portion thereof are below a minimum friction level specified by the State.

Note.— A portion of runway in the order of 100 m long may be considered significant for maintenance or reporting action.

9.4.7 Recommendation.— Corrective maintenance action should be considered when the friction characteristics for either the entire runway or a portion thereof are below a maintenance planning level specified by the State.

9.4.8 Recommendation.— When there is reason to believe that the drainage characteristics of a runway, or portions thereof, are poor due to slopes or depressions, then the runway friction characteristics should be assessed under natural or simulated conditions that are representative of local rain and corrective maintenance action should be taken as necessary.

9.4.9 Recommendation.— When a taxiway is used by turbine-engined aeroplanes, the surface of the taxiway shoulders should be maintained so as to be free of any loose stones or other objects that could be ingested by the aeroplane engines.

Note.— Guidance on this subject is given in the Aerodrome Design Manual, Part 2.

9.4.10 The surface of a paved runway shall be maintained in a condition so as to provide good friction characteristics and low rolling resistance. Snow, slush, ice, standing water, mud, dust, sand, oil, rubber deposits and other contaminants shall be removed as rapidly and completely as possible to minimize accumulation.

Note.— Guidance on determining and expressing the friction characteristics when conditions of snow or ice cannot be avoided is given in Attachment A, Section 6. The Airport Services Manual, Part 2, contains further information on this subject, on improving friction characteristics and on clearing of runways.

9.4.11 Recommendation.— A taxiway should be kept clear of snow, slush, ice, etc., to the extent necessary to enable aircraft to be taxied to and from an operational runway.

9.4.12 Recommendation.— Aprons should be kept clear of snow, slush, ice, etc., to the extent necessary to enable aircraft to manoeuvre safely or, where appropriate, to be towed or pushed.

9.4.13 Recommendation.— Whenever the clearance of snow, slush, ice, etc., from the various parts of the movement area cannot be carried out simultaneously, the order of priority should be as follows but may be altered following, as necessary, consultation with the aerodrome users:

1st — runway(s) in use;
2nd — taxiways serving runway(s) in use;
3rd — apron(s);
4th — holding bays; and
5th — other areas.

9.4.14 Recommendation.— Chemicals to remove or to prevent the formation of ice and frost on aerodrome pavements should be used when conditions indicate their use could be effective. Caution should be exercised in the application of the chemicals so as not to create a more slippery condition.

Note.— Guidance on the use of chemicals for aerodrome pavements is given in the Airport Services Manual, Part 2.

9.4.15 Chemicals which may have harmful effects on aircraft or pavements, or chemicals which may have toxic effects on the aerodrome environment, shall not be used.

Runway pavement overlays

Note.— The following specifications are intended for runway pavement overlay projects when the runway is to be returned to an operational status before overlay of the entire runway is complete thus normally necessitating a temporary
ramp between the new and old runway surfaces. Guidance on overlaying pavements and assessing their operational status is given in the Aerodrome Design Manual, Part 3.

9.4.16 The longitudinal slope of the temporary ramp, measured with reference to the existing runway surface or previous overlay course, shall be:

a) 0.5 to 1.0 per cent for overlays up to and including 5 cm in thickness; and

b) not more than 0.5 per cent for overlays more than 5 cm in thickness.

9.4.17 **Recommendation.**— Overlaying should proceed from one end of the runway toward the other end so that based on runway utilization most aircraft operations will experience a down ramp.

9.4.18 **Recommendation.**— The entire width of the runway should be overlaid during each work session.

9.4.19 Before a runway being overlaid is returned to a temporary operational status, a runway centre line marking conforming to the specifications in Section 5.2.3 shall be provided. Additionally, the location of any temporary threshold shall be identified by a 3.6 m wide transverse stripe.

**Visual aids**

*Note.— These specifications are intended to define the maintenance performance level objectives. They are not intended to define whether the lighting system is operationally out of service.*

9.4.20 A light shall be deemed to be unserviceable when the main beam average intensity is less than 50 per cent of the value specified in the appropriate figure in Appendix 2. For light units where the designed main beam average intensity is above the value shown in Appendix 2, the 50 per cent value shall be related to that design value.

9.4.21 A system of preventive maintenance of visual aids shall be employed to ensure lighting and marking system reliability.

*Note.— Guidance on preventive maintenance of visual aids is given in the Airport Services Manual, Part 9.*

9.4.22 **Recommendation.**— The system of preventive maintenance employed for a precision approach runway category II or III should include at least the following checks:

a) visual inspection and in-field measurement of the intensity, beam spread and orientation of lights included in the approach and runway lighting systems;

b) control and measurement of the electrical characteristics of each circuitry included in the approach and runway lighting systems; and

c) control of the correct functioning of light intensity settings used by air traffic control.

9.4.23 **Recommendation.**— In-field measurement of intensity, beam spread and orientation of lights included in approach and runway lighting systems for a precision approach runway category II or III should be undertaken by measuring all lights, as far as practicable, to ensure conformance with the applicable specification of Appendix 2.

9.4.24 **Recommendation.**— Measurement of intensity, beam spread and orientation of lights included in approach and runway lighting systems for a precision approach runway category II or III should be undertaken using a mobile measuring unit of sufficient accuracy to analyze the characteristics of the individual lights.

9.4.25 **Recommendation.**— The frequency of measurement of lights for a precision approach runway category II or III should be based on traffic density, the local pollution level, the reliability of the installed lighting equipment and the continuous assessment of the results of the in-field measurements but in any event should not be less than twice a year for in-pavement lights and not less than once a year for other lights.

9.4.26 The system of preventive maintenance employed for a precision approach runway category II or III shall have as its objective that, during any period of category II or III operations, all approach and runway lights are serviceable, and that in any event at least:

a) 95 per cent of the lights are serviceable in each of the following particular significant elements:

1) precision approach category II and III lighting system, the inner 450 m;

2) runway centre line lights;

3) runway threshold lights; and

4) runway edge lights;

b) 90 per cent of the lights are serviceable in the touchdown zone lights;

c) 85 per cent of the lights are serviceable in the approach lighting system beyond 450 m; and

d) 75 per cent of the lights are serviceable in the runway end lights.

In order to provide continuity of guidance, the allowable percentage of unserviceable lights shall not be permitted in
such a way as to alter the basic pattern of the lighting system. Additionally, an unserviceable light shall not be permitted adjacent to another unserviceable light, except in a barrette or a crossbar where two adjacent unserviceable lights may be permitted.

Note.— With respect to barrettes, crossbars and runway edge lights, lights are considered to be adjacent if located consecutively and:

— laterally: in the same barrette or crossbar; or

— longitudinally: in the same row of edge lights or barrettes.

9.4.27 The system of preventive maintenance employed for a stop bar provided at a runway-holding position used in conjunction with a runway intended for operations in runway visual range conditions less than a value of 350 m shall have the following objectives:

a) no more than two lights will remain unserviceable; and

b) two adjacent lights will not remain unserviceable unless the light spacing is significantly less than that specified.

9.4.28 The system of preventive maintenance employed for a taxiway intended for use in runway visual range conditions less than a value of 350 m shall have as its objective that no two adjacent taxiway centre line lights be unserviceable.

9.4.29 The system of preventive maintenance employed for a precision approach runway category I shall have as its objective that, during any period of category I operations, all approach and runway lights are serviceable, and that in any event at least 85 per cent of the lights are serviceable in each of the following:

a) precision approach category I lighting system;

b) runway threshold lights;

c) runway edge lights; and

d) runway end lights.

In order to provide continuity of guidance an unserviceable light shall not be permitted adjacent to another unserviceable light unless the light spacing is significantly less than that specified.

Note.— In barrettes and crossbars, guidance is not lost by having two adjacent unserviceable lights.

9.4.30 The system of preventive maintenance employed for a runway meant for take-off in runway visual range conditions less than a value of 550 m shall have as its objective that, during any period of operations, all runway lights are serviceable and that in any event:

a) at least 95 per cent of the lights are serviceable in the runway centre line lights (where provided) and in the runway edge lights; and

b) at least 75 per cent of the lights are serviceable in the runway end lights.

In order to provide continuity of guidance, an unserviceable light shall not be permitted adjacent to another unserviceable light.

9.4.31 The system of preventive maintenance employed for a runway meant for take-off in runway visual range conditions of a value of 550 m or greater shall have as its objective that, during any period of operations, all runway lights are serviceable and that, in any event, at least 85 per cent of the lights are serviceable in the runway edge lights and runway end lights. In order to provide continuity of guidance, an unserviceable light shall not be permitted adjacent to another unserviceable light.

9.4.32 Recommendation.— During low visibility procedures the appropriate authority should restrict construction or maintenance activities in the proximity of aerodrome electrical systems.

9.5 Bird hazard reduction

9.5.1 Recommendation.— The bird strike hazard on, or in the vicinity of, an aerodrome should be assessed through:

a) the establishment of a national procedure for recording and reporting bird strikes to aircraft; and

b) the collection of information from aircraft operators, airport personnel, etc. on the presence of birds on or around the aerodrome.

Note.— The ICAO Bird Strike Information System (IBIS) is designed to collect and disseminate information on bird strikes to aircraft. Information on the system is included in the Manual on the ICAO Bird Strike Information System (IBIS).

9.5.2 Recommendation.— When a bird strike hazard is identified at an aerodrome, the appropriate authority should take action to decrease the number of birds constituting a potential hazard to aircraft operations by adopting measures for discouraging their presence on, or in the vicinity of, an aerodrome.

Note.— Guidance on effective measures for establishing whether or not birds, on or near an aerodrome, constitute a potential hazard to aircraft operations, and on methods for discouraging their presence, is given in the Airport Services Manual, Part 3.
9.5.3 **Recommendation.** — Garbage disposal dumps or any such other source attracting bird activity on, or in the vicinity of, an aerodrome should be eliminated or their establishment prevented, unless an appropriate study indicates that they are unlikely to create conditions conducive to a bird hazard problem.

9.6 **Apron management service**

9.6.1 **Recommendation.** — When warranted by the volume of traffic and operating conditions, an appropriate apron management service should be provided on an apron by an aerodrome ATS unit, by another aerodrome operating authority, or by a cooperative combination of these, in order to:

a) regulate movement with the objective of preventing collisions between aircraft, and between aircraft and obstacles;

b) regulate entry of aircraft into, and coordinate exit of aircraft from, the apron with the aerodrome control tower; and

c) ensure safe and expeditious movement of vehicles and appropriate regulation of other activities.

9.6.2 **Recommendation.** — When the aerodrome control tower does not participate in the apron management service, procedures should be established to facilitate the orderly transition of aircraft between the apron management unit and the aerodrome control tower.

*Note.— Guidance on an apron management service is given in the Airport Services Manual, Part 8 and in the Manual of Surface Movement Guidance and Control Systems (SMGCS).*

9.6.3 An apron management service shall be provided with radiotelephony communications facilities.

9.6.4 Where low visibility procedures are in effect, persons and vehicles operating on an apron shall be restricted to the essential minimum.

*Note.— Guidance on related special procedures is given in the Manual of Surface Movement Guidance and Control Systems (SMGCS).*

9.6.5 An emergency vehicle responding to an emergency shall be given priority over all other surface movement traffic.

9.6.6 A vehicle operating on an apron shall:

a) give way to an emergency vehicle; an aircraft taxiing, about to taxi, or being pushed or towed; and

b) give way to other vehicles in accordance with local regulations.

9.6.7 An aircraft stand shall be visually monitored to ensure that the recommended clearance distances are provided to an aircraft using the stand.

9.7 **Ground servicing of aircraft**

9.7.1 Fire extinguishing equipment suitable for at least initial intervention in the event of a fuel fire and personnel trained in its use shall be readily available during the ground servicing of an aircraft, and there shall be a means of quickly summoning the rescue and fire fighting service in the event of a fire or major fuel spill.

9.7.2 When aircraft refuelling operations take place while passengers are embarking, on board or disembarking, ground equipment shall be positioned so as to allow:

a) the use of a sufficient number of exits for expeditious evacuation; and

b) a ready escape route from each of the exits to be used in an emergency.
APPENDIX 1. COLOURS FOR AERONAUTICAL GROUND LIGHTS, MARKINGS, SIGNS AND PANELS

1. General

Introductory Note.— The following specifications define the chromaticity limits of colours to be used for aeronautical ground lights, markings, signs and panels. The specifications are in accord with the 1983 specifications of the International Commission on Illumination (CIE).

It is not possible to establish specifications for colours such that there is no possibility of confusion. For reasonably certain recognition, it is important that the eye illumination be well above the threshold of perception, that the colour not be greatly modified by selective atmospheric attenuations and that the observer’s colour vision be adequate. There is also a risk of confusion of colour at an extremely high level of eye illumination such as may be obtained from a high-intensity source at very close range. Experience indicates that satisfactory recognition can be achieved if due attention is given to these factors.

The chromaticities are expressed in terms of the standard observer and coordinate system adopted by the International Commission on Illumination (CIE) at its Eighth Session at Cambridge, England, in 1931.*

2. Colours for aeronautical ground lights

2.1 Chromaticities

2.1.1 The chromaticities of aeronautical ground lights shall be within the following boundaries:

CIE Equations (see Figure 1.1):

a) Red
   Purple boundary \( y = 0.980 - x \)
   Yellow boundary \( y = 0.335 \)

b) Yellow
   Red boundary \( y = 0.382 \)
   White boundary \( y = 0.790 - 0.667x \)
   Green boundary \( y = x - 0.120 \)

c) Green
   Yellow boundary \( x = 0.360 - 0.080y \)
   White boundary \( x = 0.650y \)
   Blue boundary \( y = 0.390 - 0.171x \)

d) Blue
   Green boundary \( y = 0.805x + 0.065 \)
   White boundary \( y = 0.400 - x \)
   Purple boundary \( x = 0.600y + 0.133 \)

e) White
   Yellow boundary \( x = 0.500 \)
   Blue boundary \( x = 0.285 \)
   Green boundary \( y = 0.440 \)
   and \( y = 0.150 + 0.640x \)
   Purple boundary \( y = 0.050 + 0.750x \)
   and \( y = 0.382 \)

f) Variable white
   Yellow boundary \( x = 0.255 + 0.750y \)
   and \( x = 1.185 - 1.500 y \)
   Blue boundary \( x = 0.285 \)
   Green boundary \( y = 0.440 \)
   and \( y = 0.150 + 0.640x \)
   Purple boundary \( y = 0.050 + 0.750x \)
   and \( y = 0.382 \)

Note.— Guidance on chromaticity changes resulting from the effect of temperature on filtering elements is given in the Aerodrome Design Manual, Part 4.

2.1.2 Recommendation.— Where dimming is not required, or where observers with defective colour vision must be able to determine the colour of the light, green signals should be within the following boundaries:

Yellow boundary \( y = 0.726 - 0.726x \)
White boundary \( x = 0.650y \)
Blue boundary \( y = 0.390 - 0.171x \)

2.1.3 Recommendation.— Where increased certainty of recognition is more important than maximum visual range, green signals should be within the following boundaries:

Yellow boundary \( y = 0.726 - 0.726x \)
White boundary \( x = 0.625y - 0.041 \)
Blue boundary \( y = 0.390 - 0.171x \)

* See CIE Publication No. 15, Colorimetry (1971).
2.2 Discrimination between lights

2.2.1 **Recommendation.**— If there is a requirement to discriminate yellow and white from each other, they should be displayed in close proximity of time or space as, for example, by being flashed successively from the same beacon.

2.2.2 **Recommendation.**— If there is a requirement to discriminate yellow from green and/or white, as for example on exit taxiway centre line lights, the y coordinates of the yellow light should not exceed a value of 0.40.

*Note.— The limits of white have been based on the assumption that they will be used in situations in which the characteristics (colour temperature) of the light source will be substantially constant.*

2.2.3 **Recommendation.**— The colour variable white is intended to be used only for lights that are to be varied in intensity, e.g. to avoid dazzling. If this colour is to be discriminated from yellow, the lights should be so designed and operated that:

a) the x coordinate of the yellow is at least 0.050 greater than the x coordinate of the white; and

b) the disposition of the lights will be such that the yellow lights are displayed simultaneously and in close proximity to the white lights.

2.2.4 The colour of aeronautical ground lights shall be verified as being within the boundaries specified in Figure 1.1 by measurement at five points within the area limited by the innermost isocandela curve (isocandela diagrams in Appendix 2 refer), with operation at rated current or voltage. In the case of elliptical or circular isocandela curves, the colour measurements shall be taken at the centre and at the horizontal and vertical limits. In the case of rectangular isocandela curves, the colour measurements shall be taken at the centre and the limits of the diagonals (corners). In addition, the colour of the light shall be checked at the outermost isocandela curve to ensure that there is no colour shift that might cause signal confusion to the pilot.

*Note 1.— For the outermost isocandela curve, a measurement of colour coordinates should be made and recorded for review and judgement of acceptability by the appropriate authority.*

*Note 2.— Certain light units may have application so that they may be viewed and used by pilots from directions beyond that of the outermost isocandela curve (e.g. stop bar lights at significantly wide runway-holding positions). In such instances, the appropriate authority should assess the actual application and if necessary require a check of colour shift at angular ranges beyond the outermost curve.*

2.2.5 In the case of visual approach slope indicators and other light units having a colour transition sector, the colour shall be measured at points in accordance with 2.2.4, except that the colour areas shall be treated separately and no point shall be within 0.5 degrees of the transition sector.

3. Colours for markings, signs and panels

*Note 1.— The specifications of surface colours given below apply only to freshly coloured surfaces. Colours used for markings, signs and panels usually change with time and therefore require renewal.*


*Note 3.— The specifications recommended in 3.4 below for transilluminated panels are interim in nature and are based on the CIE specifications for transilluminated signs. It is intended that these specifications will be reviewed and updated as and when CIE develops specifications for transilluminated panels.*

3.1 The chromaticities and luminance factors of ordinary colours, colours of retro-reflective materials and colours of transilluminated (internally illuminated) signs and panels shall be determined under the following standard conditions:

a) angle of illumination: 45°;

b) direction of view: perpendicular to surface; and

c) illuminant: CIE standard illuminant D_65.

3.2 **Recommendation.**— The chromaticity and luminance factors of ordinary colours for markings and externally illuminated signs and panels should be within the following boundaries when determined under standard conditions.

**CIE Equations (see Figure 1.2):**

a) Red

Purple boundary \( y = 0.345 - 0.051x \)
White boundary \( y = 0.910 - x \)
Orange boundary \( y = 0.314 + 0.047x \)
Luminance factor \( \beta = 0.07 \text{ (mm)} \)

b) Orange

Red boundary \( y = 0.265 + 0.205x \)
White boundary \( y = 0.910 - x \)
Yellow boundary \( y = 0.207 + 0.390x \)
Luminance factor \( \beta = 0.20 \text{ (mm)} \)
c) Yellow
Orange boundary  \( y = 0.108 + 0.707x \)
White boundary  \( y = 0.910 - x \)
Green boundary  \( y = 1.35x - 0.093 \)
Luminance factor  \( \beta = 0.45 \) (mmn)

d) White
Purple boundary  \( y = 0.010 + x \)
Blue boundary  \( y = 0.610 - x \)
Green boundary  \( y = 0.030 + x \)
Yellow boundary  \( y = 0.710 - x \)
Luminance factor  \( \beta = 0.75 \) (mmn)

e) Black
Purple boundary  \( y = x - 0.030 \)
Blue boundary  \( y = 0.570 - x \)
Green boundary  \( y = 0.040 + x \)
Yellow boundary  \( y = 0.740 - x \)
Luminance factor  \( \beta = 0.03 \) (max)

f) Yellowish green
Green boundary  \( y = 1.317x + 0.4 \)
White boundary  \( y = 0.910 - x \)
Yellow boundary  \( y = 0.867x + 0.4 \)
Luminance factor  \( \beta = 0.03 \) (max)

Note.— The small separation between surface red and surface orange is not sufficient to ensure the distinction of these colours when seen separately.

3.3 Recommendation.— The chromaticity and luminance factors of colours of retro-reflective materials for markings, signs and panels should be within the following boundaries when determined under standard conditions.

CIE Equations (see Figure 1.3):

a) Red
Purple boundary  \( y = 0.345 - 0.051x \)
White boundary  \( y = 0.910 - x \)
Orange boundary  \( y = 0.314 + 0.047x \)
Luminance factor  \( \beta = 0.03 \) (mmn)

b) Orange
Red boundary  \( y = 0.265 + 0.205x \)
White boundary  \( y = 0.910 - x \)
Yellow boundary  \( y = 0.207 + 0.390x \)
Luminance factor  \( \beta = 0.14 \) (mmn)

c) Yellow
Orange boundary  \( y = 0.160 + 0.540x \)
White boundary  \( y = 0.910 - x \)
Green boundary  \( y = 1.35x - 0.093 \)
Luminance factor  \( \beta = 0.16 \) (mmn)

d) White
Purple boundary  \( y = x \)
Blue boundary  \( y = 0.610 - x \)
Green boundary  \( y = 0.040 + x \)
Yellow boundary  \( y = 0.710 - x \)
Luminance factor  \( \beta = 0.27 \) (mmn)

e) Blue
Green boundary  \( y = 0.118 + 0.675x \)
White boundary  \( y = 0.370 - x \)
Purple boundary  \( y = 1.65x - 0.187 \)
Luminance factor  \( \beta = 0.01 \) (mmn)

f) Green
Yellow boundary  \( y = 0.711 - 1.22x \)
White boundary  \( y = 0.243 + 0.670x \)
Blue boundary  \( y = 0.405 - 0.243x \)
Luminance factor  \( \beta = 0.03 \) (mmn)

3.4 Recommendation.— The chromaticity and luminance factors of colours of transilluminated (internally illuminated) signs and panels should be within the following boundaries when determined under standard conditions.

CIE Equations (see Figure 1.4):

a) Red
Purple boundary  \( y = 0.345 - 0.051x \)
White boundary  \( y = 0.910 - x \)
Orange boundary  \( y = 0.314 + 0.047x \)
Luminance factor  \( \beta = 0.07 \) (mnm)
Relative luminance to white (night condition) 5% (mnm)
Relative luminance to white (night condition) 20% (max)

b) Yellow
Orange boundary  \( y = 0.108 + 0.707x \)
White boundary  \( y = 0.910 - x \)
Green boundary  \( y = 1.35x - 0.093 \)
Luminance factor  \( \beta = 0.45 \) (mmn)
Relative luminance to white (night condition) 30% (mmn)
Relative luminance to white (night condition) 80% (max)

c) White
Purple boundary  \( y = 0.010 + x \)
Blue boundary  \( y = 0.610 - x \)
Green boundary  \( y = 0.030 + x \)
Yellow boundary  \( y = 0.710 - x \)
Luminance factor  \( \beta = 0.75 \) (mmn)
Relative luminance to white (night condition) 100%

d) Black
Purple boundary  \( y = x - 0.030 \)
Blue boundary  \( y = 0.570 - x \)
Green boundary  \( y = 0.050 + x \)
Yellow boundary  \( y = 0.740 - x \)
Luminance factor  \( \beta = 0.03 \) (max)
Relative luminance to white (night condition) 0% (mnm)
Relative luminance to white (night condition) 2% (max)
Figure 1.1 Colours for aeronautical ground lights
Figure 1.2 Ordinary colours for markings and externally illuminated signs and panels
Figure 1.3  Colours of retro-reflective materials for markings, signs and panels
Figure 1.4 Colours of transilluminated (internally illuminated) signs and panels
APPENDIX 2. AERONAUTICAL GROUND LIGHT CHARACTERISTICS

Notes:

1. Curves calculated on formula \( \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \)

<table>
<thead>
<tr>
<th>a</th>
<th>10</th>
<th>14</th>
<th>15</th>
</tr>
</thead>
<tbody>
<tr>
<td>b</td>
<td>5.5</td>
<td>6.5</td>
<td>8.5</td>
</tr>
</tbody>
</table>

2. Vertical setting angles of the lights shall be such that the following vertical coverage of the main beam will be met:

   distance from threshold | vertical main beam coverage
   ------------------------|---------------------------
   threshold to 315 m      | 0° — 11°
   316 m to 475 m         | 0.5° — 11.5°
   476 m to 640 m         | 1.2° — 12.5°
   641 m and beyond       | 2.5° — 13.5° (as illustrated above)

3. Lights in crossbars beyond 22.5 m from the centre line shall be toed-in 2 degrees. All other lights shall be aligned parallel to the centre line of the runway.

4. See collective notes for Figures 2.1 to 2.11.

Figure 2.1 Isocandela diagram for approach centre line light and crossbars (white light)
Appendix 2

Annex 14 — Aerodromes

Notes:

1. Curves calculated on formula \[ \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \]

2. Toe-in 2 degrees

3. Vertical setting angles of the lights shall be such that the following vertical coverage of the main beam will be met:

<table>
<thead>
<tr>
<th>distance from threshold</th>
<th>vertical main beam coverage</th>
</tr>
</thead>
<tbody>
<tr>
<td>threshold to 115 m</td>
<td>0.5° — 10.5°</td>
</tr>
<tr>
<td>116 m to 215 m</td>
<td>1° — 11°</td>
</tr>
<tr>
<td>216 m and beyond</td>
<td>1.5° — 11.5° (as illustrated above)</td>
</tr>
</tbody>
</table>

4. See collective notes for Figures 2.1 to 2.11.

Figure 2.2 Isocandela diagram for approach side row light (red light)
Notes:

1. Curves calculated on formula \( \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \)

2. Toe-in 3.5 degrees

3. See collective notes for Figures 2.1 to 2.11.

Figure 2.3 Isocandela diagram for threshold light (green light)
Appendix 2

Notes:

1. Curves calculated on formula $\frac{x^2}{a^2} + \frac{y^2}{b^2} = 1$

2. Toe-in 2 degrees

3. See collective notes for Figures 2.1 to 2.11.

Figure 2.4   Isocandela diagram for threshold wing bar light (green light)
Notes:

1. Curves calculated on formula \( \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \)

2. Toe-in 4 degrees

3. See collective notes for Figures 2.1 to 2.11.

Figure 2.5  Isocandela diagram for touchdown zone light (white light)
Notes:

1. Curves calculated on formula \[ \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \]

<table>
<thead>
<tr>
<th></th>
<th>6.0</th>
<th>7.0</th>
<th>8.5</th>
</tr>
</thead>
<tbody>
<tr>
<td>a</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>b</td>
<td>3.5</td>
<td>6.0</td>
<td>8.5</td>
</tr>
</tbody>
</table>

2. For red light multiply values by 0.15

3. See collective notes for Figures 2.1 to 2.11.

Figure 2.6 Isocandela diagram for runway centre line light with 30 m longitudinal spacing (white light)
Annex 14 — Aerodromes

Notes:

1. Curves calculated on formula \( \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \)

2. For red light multiply values by 0.15

3. See collective notes for Figures 2.1 to 2.11.

Figure 2.7 Isocandela diagram for runway centre line light with 15 m longitudinal spacing (white light)
Notes:

1. Curves calculated on formula \[ \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \]

2. See collective notes for Figures 2.1 to 2.11.

Figure 2.8 Isocandela diagram for runway end light (red light)
Annex 14 — Aerodromes

Notes:

1. Curves calculated on formula \[ \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \]
2. Toe-in 3.5 degrees
3. For yellow light multiply values by 0.4
4. See collective notes for Figures 2.1 to 2.11.

Figure 2.9 Isocandela diagram for runway edge light
where width of runway is 45 m (white light)
Appendix 2

Annex 14 — Aerodromes

Notes:

1. Curves calculated on formula \( \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \)

2. Toe-in 4.5 degrees

3. For yellow light multiply values by 0.4

4. See collective notes for Figures 2.1 to 2.11.

Figure 2.10 Isocandela diagram for runway edge light where width of runway is 60 m (white light)
Figure 2.11  Grid points to be used for the calculation of average intensity of approach and runway lights
Collective notes to Figures 2.1 to 2.11

1. The ellipses in each figure are symmetrical about the common vertical and horizontal axes.

2. Figures 2.1 to 2.10 show the minimum allowable light intensities. The average intensity of the main beam is calculated by establishing grid points as shown in Figure 2.11 and using the intensity values measured at all grid points located within and on the perimeter of the ellipse representing the main beam. The average value is the arithmetic average of light intensities measured at all considered grid points.

3. No deviations are acceptable in the main beam pattern when the lighting fixture is properly aimed.

4. Average intensity ratio. The ratio between the average intensity within the ellipse defining the main beam of a typical new light and the average light intensity of the main beam of a new runway edge light shall be as follows:

| Figure 2.1 | Approach centre line and crossbars | 1.5 to 2.0 (white light) |
| Figure 2.2 | Approach side row | 0.5 to 1.0 (red light) |
| Figure 2.3 | Threshold | 1.0 to 1.5 (green light) |
| Figure 2.4 | Threshold wing bar | 1.0 to 1.5 (green light) |
| Figure 2.5 | Touchdown zone | 0.5 to 1.0 (white light) |
| Figure 2.6 | Runway centre line (longitudinal spacing 30 m) | 0.5 to 1.0 (white light) |
| Figure 2.7 | Runway centre line (longitudinal spacing 15 m) | 0.5 to 1.0 for CAT III (white light) 0.25 to 0.5 for CAT I, II (white light) |
| Figure 2.8 | Runway end | 0.25 to 0.5 (red light) |
| Figure 2.9 | Runway edge (45 m runway width) | 1.0 (white light) |
| Figure 2.10 | Runway edge (60 m runway width) | 1.0 (white light) |

5. The beam coverages in the figures provide the necessary guidance for approaches down to an RVR of the order of 150 m and take-offs down to an RVR of the order of 100 m.

6. Horizontal angles are measured with respect to the vertical plane through the runway centre line. For lights other than centre line lights, the direction towards the runway centre line is considered positive. Vertical angles are measured with respect to the horizontal plane.

7. Where, for approach centre line lights and crossbars and for approach side row lights, inset lights are used in lieu of elevated lights, e.g. on a runway with a displaced threshold, the intensity requirements can be met by installing two or three fittings (lower intensity) at each position.

8. The importance of adequate maintenance cannot be over-emphasized. The average intensity should never fall to a value less than 50 per cent of the value shown in the figures and it should be the aim of airport authorities to maintain a level of light output close to the specified minimum average intensity.

9. The light unit shall be installed so that the main beam is aligned within one-half degree of the specified requirement.
Annex 14 — Aerodromes

Notes:

1. These beam coverages allow for displacement of the cockpit from the centre line up to distances of the order of 12 m and are intended for use before and after curves.

2. See collective notes for Figures 2.12 to 2.21.

Figure 2.12. Isocandela diagram for taxiway centre line (15 m spacing) and stop bar lights in straight sections intended for use in runway visual range conditions of less than a value of 350 m where large offsets can occur and for low-intensity runway guard lights, Configuration B

Notes:

1. These beam coverages are generally satisfactory and cater for a normal displacement of the cockpit from the centre line of approximately 3 m.

2. See collective notes for Figures 2.12 to 2.21.

Figure 2.13. Isocandela diagram for taxiway centre line (15 m spacing) and stop bar lights in straight sections intended for use in runway visual range conditions of less than a value of 350 m.
Appendix 2

Figure 2.14 Isocandela diagram for taxiway centre line (7.5 m spacing) and stop bar lights in curved sections intended for use in runway visual range conditions of less than a value of 350 m

Notes:
1. Lights on curves to be toed-in 15.75 degrees with respect to the tangent of the curve.
2. See collective notes for Figures 2.12 to 2.21.

Figure 2.15 Isocandela diagram for taxiway centre line (30 m, 60 m spacing) and stop bar lights in straight sections intended for use in runway visual range conditions of 350 m or greater

Notes:
1. At locations where high background luminance is usual and where deterioration of light output resulting from dust, snow and local contamination is a significant factor, the cd-values should be multiplied by 2.5.
2. Where omnidirectional lights are used they shall comply with the vertical beam requirements in this figure.
3. See collective notes for Figures 2.12 to 2.21.
Annex 14 — Aerodromes

Notes:

1. Lights on curves to be toed-in 15.75 degrees with respect to the tangent of the curve.

2. At locations where high background luminance is usual and where deterioration of light output resulting from dust, snow and local contamination is a significant factor, the cd-values should be multiplied by 2.5.

3. These beam coverages allow for displacement of the cockpit from the centre line up to distances of the order of 12 m as could occur at the end of curves.

4. See collective notes for Figures 2.12 to 2.21.

Figure 2.16 Isocandela diagram for taxiway centre line (7.5 m, 15 m, 30 m spacing) and stop bar lights in curved sections intended for use in runway visual range conditions of 350 m or greater.
Notes:

1. These beam coverages allow for displacement of the cockpit from the centre line up to distances of the order of 12 m and are intended for use before and after curves.

2. See collective notes for Figures 2.12 to 2.21.

Figure 2.17. Isocandela diagram for high-intensity taxiway centre line (15 m spacing) and stop bar lights in straight sections intended for use in an advanced surface movement guidance and control system where higher light intensities are required and where large offsets can occur.
Notes:

1. These beam coverages are generally satisfactory and cater for a normal displacement of the cockpit corresponding to the outer main gear wheel on the taxiway edge.

2. See collective notes for Figures 2.12 to 2.21.

Figure 2.18. Isocandela diagram for high-intensity taxiway centre line (15 m spacing) and stop bar lights in straight sections intended for use in an advanced surface movement guidance and control system where higher light intensities are required.
Notes:

1. Lights on curves to be toed-in 17 degrees with respect to the tangent of the curve.

2. See collective notes for Figures 2.12 to 2.21.

Figure 2.19. Isocandela diagram for high-intensity taxiway centre line (7.5 m spacing) and stop bar lights in curved sections intended for use in an advanced surface movement guidance and control system where higher light intensities are required.
Annex 14 — Aerodromes

Notes:

1. Although the lights flash in normal operation, the light intensity is specified as if the lights were fixed for incandescent lamps.

2. See collective notes for Figures 2.12 to 2.21.

Figure 2.20. Isocandela diagram for high-intensity runway guard lights, Configuration B

Figure 2.21. Grid points to be used for calculation of average intensity of taxiway centre line and stop bar lights
Collective notes to Figures 2.12 to 2.21

1. The intensities specified in Figures 2.12 to 2.20 are in green and yellow light for taxiway centre line lights, yellow light for runway guard lights and red light for stop bar lights.

2. Figures 2.12 to 2.20 show the minimum allowable light intensities. The average intensity of the main beam is calculated by establishing grid points as shown in Figure 2.21 and using the intensity values measured at all grid points located within and on the perimeter of the rectangle representing the main beam. The average value is the arithmetic average of the light intensities measured at all considered grid points.

3. No deviations are acceptable in the main beam or in the innermost beam, as applicable, when the lighting fixture is properly aimed.

4. Horizontal angles are measured with respect to the vertical plane through the taxiway centre line except on curves where they are measured with respect to the tangent to the curve.

5. Vertical angles are measured from the longitudinal slope of the taxiway surface.

6. The importance of adequate maintenance cannot be over-emphasized. The intensity, either average where applicable or as specified on the corresponding isocandela curves, should never fall to a value less than 50 per cent of the value shown in the figures and it should be the aim of airport authorities to maintain a level of light output close to the specified minimum average intensity.

7. The light unit shall be installed so that the main beam or the innermost beam, as applicable, is aligned within one-half degree of the specified requirement.
Figure 2.2: Light intensity distribution of T-VASIS and R-VASIS

Annex 14: ICAO

Note 1: These curves are by minimum intensities in white light.

Note 2: Both VASIS and R-VASIS are minimum at operating altitude.
Appendix 2

Note 1.— These curves are for minimum intensities in red light.

Note 2.— The intensity value in the white sector of the beam is no less than 2 and may be as high as 6.5 times the corresponding intensity in the red sector.

Note 3.— The intensity values shown in brackets are for APAPI.

Figure 2.23 Light intensity distribution of PAPI and APAPI
Notes:

1. Although the lights flash in normal operation, the light intensity is specified as if the lights were fixed for incandescent lamps.

2. The intensities specified are in yellow light.

Figure 2.24  Isocandela diagram for each light in low-intensity runway guard lights, Configuration A
Notes:

1. Although the lights flash in normal operation, the light intensity is specified as if the lights were fixed for incandescent lamps.

2. The intensities specified are in yellow light.

Figure 2.25. Isocandela diagram for each light in high-intensity runway guard lights, Configuration A
Note 1.— See Chapter 5, Sections 5.2.15 and 5.2.16 for specifications on the application, location and characteristics of mandatory instruction markings and information markings.

Note 2.— This appendix details the form and proportions of the letters, numbers and symbols of mandatory instruction markings and information markings on a 20 cm grid.
APPENDIX 4. REQUIREMENTS CONCERNING DESIGN OF TAXIING GUIDANCE SIGNS

Note.— See Chapter 5, Section 5.4 for specifications on the application, location and characteristics of signs.

1. Inscription heights shall conform to the following tabulation.

<table>
<thead>
<tr>
<th>Runway code number</th>
<th>Minimum character height</th>
</tr>
</thead>
<tbody>
<tr>
<td>Mand. instruction</td>
<td>Information sign</td>
</tr>
<tr>
<td>sign</td>
<td>Runway exit and runway</td>
</tr>
<tr>
<td></td>
<td>vacated signs</td>
</tr>
<tr>
<td>1 or 2</td>
<td>300 mm 300 mm 200 mm</td>
</tr>
<tr>
<td>3 or 4</td>
<td>400 mm 400 mm 300 mm</td>
</tr>
</tbody>
</table>

Note.— Where a taxiway location sign is installed in conjunction with a runway designation sign (see 5.4.3.22), the character size shall be that specified for mandatory instruction signs.

2. Arrow dimensions shall be as follows:

<table>
<thead>
<tr>
<th>Legend height</th>
<th>Stroke</th>
</tr>
</thead>
<tbody>
<tr>
<td>200 mm</td>
<td>32 mm</td>
</tr>
<tr>
<td>300 mm</td>
<td>48 mm</td>
</tr>
<tr>
<td>400 mm</td>
<td>64 mm</td>
</tr>
</tbody>
</table>

3. Stroke width for single letter shall be as follows:

<table>
<thead>
<tr>
<th>Legend height</th>
<th>Stroke</th>
</tr>
</thead>
<tbody>
<tr>
<td>200 mm</td>
<td>32 mm</td>
</tr>
<tr>
<td>300 mm</td>
<td>48 mm</td>
</tr>
<tr>
<td>400 mm</td>
<td>64 mm</td>
</tr>
</tbody>
</table>

4. Sign luminance shall be as follows:

a) Where operations are conducted in runway visual range conditions less than a value of 800 m, average sign luminance shall be at least:

<table>
<thead>
<tr>
<th>Color</th>
<th>Luminance (cd/m²)</th>
</tr>
</thead>
<tbody>
<tr>
<td>Red</td>
<td>30 cd/m²</td>
</tr>
<tr>
<td>Yellow</td>
<td>150 cd/m²</td>
</tr>
<tr>
<td>White</td>
<td>300 cd/m²</td>
</tr>
</tbody>
</table>

b) Where operations are conducted in accordance with 5.4.1.7 b) and c) and 5.4.1.8, average sign luminance shall be at least:

<table>
<thead>
<tr>
<th>Color</th>
<th>Luminance (cd/m²)</th>
</tr>
</thead>
<tbody>
<tr>
<td>Red</td>
<td>10 cd/m²</td>
</tr>
<tr>
<td>Yellow</td>
<td>50 cd/m²</td>
</tr>
<tr>
<td>White</td>
<td>100 cd/m²</td>
</tr>
</tbody>
</table>

Note.— In runway visual range conditions less than a value of 400 m, there will be some degradation in the performance of signs.

5. The luminance ratio between red and white elements of a mandatory sign shall be between 1:5 and 1:10.

6. The average luminance of the sign is calculated by establishing grid points as shown in Figure 4.1 and using the luminance values measured at all grid points located within the rectangle representing the sign.

7. The average value is the arithmetic average of the luminance values measured at all considered grid points.

Note.— Guidance on measuring the average luminance of a sign is contained in the Aerodrome Design Manual, Part 4.

8. The ratio between luminance values of adjacent grid points shall not exceed 1.5:1. For areas on the sign face where the grid spacing is 7.5 cm, the ratio between luminance values of adjacent grid points shall not exceed 1.25:1. The ratio between the maximum and minimum luminance value over the whole sign face shall not exceed 5:1.

9. The forms of characters, i.e. letters, numbers, arrows and symbols, shall conform to those shown in Figure 4.2. The width of characters and the space between individual characters shall be determined as indicated in Table 4.1.

10. The face height of signs shall be as follows:

<table>
<thead>
<tr>
<th>Legend height</th>
<th>Face height (min)</th>
</tr>
</thead>
<tbody>
<tr>
<td>200 mm</td>
<td>400 mm</td>
</tr>
<tr>
<td>300 mm</td>
<td>600 mm</td>
</tr>
<tr>
<td>400 mm</td>
<td>800 mm</td>
</tr>
</tbody>
</table>
11. The face width of signs shall be determined using Figure 4.3 except that, where a mandatory instruction sign is provided on one side of a taxiway only, the face width shall not be less than:

   a) 1.94 m where the code number is 3 or 4; and
   b) 1.46 m where the code number is 1 or 2.

   Note.— Additional guidance on determining the face width of a sign is contained in the Aerodrome Design Manual, Part 4.

12. Borders

   a) The black vertical delineator between adjacent direction signs should have a width of approximately 0.7 of the stroke width.
   b) The yellow border on a stand-alone location sign should be approximately 0.5 stroke width.

13. The colours of signs shall be in accordance with the appropriate specifications in Appendix 1.
Note 1.— The average luminance of a sign is calculated by establishing grid points on a sign face showing typical inscriptions and a background of the appropriate colour (red for mandatory instruction signs and yellow for direction and destination signs) as follows:

a) Starting at the top left corner of the sign face, establish a reference grid point at 7.5 cm from the left edge and the top of the sign face.

b) Create a grid of 15 cm spacing horizontally and vertically from the reference grid point. Grid points within 7.5 cm of the edge of the sign face shall be excluded.

c) Where the last point in a row/column of grid points is located between 22.5 cm and 15 cm from the edge of the sign face (but not inclusive), an additional point shall be added 7.5 cm from this point.

d) Where a grid point falls on the boundary of a character and the background, the grid point shall be slightly shifted to be completely outside the character.

Note 2.— Additional grid points may be required to ensure that each character includes at least five evenly spaced grid points.

Note 3.— Where one unit includes two types of signs, a separate grid shall be established for each type.

Figure 4.1 Grid points for calculating average luminance of a sign
Figure 4.2  Forms of characters
Figure 4.2 (cont.)
Figure 4.2  (cont.)
Figure 4.2 (cont.)
Note 1.— The arrow stroke width, diameter of the dot, and both width and length of the dash shall be proportioned to the character stroke widths.

Note 2.— The dimensions of the arrow shall remain constant for a particular sign size, regardless of orientation.
Table 4-1. Letter and numeral widths and space between letters or numerals

<table>
<thead>
<tr>
<th>a) Letter to letter code number</th>
<th>Following Letter</th>
<th>Preceding Letter</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>B, D, E, F,</td>
<td>A, J, T, V, W, Y</td>
</tr>
<tr>
<td></td>
<td>H, I, K, L,</td>
<td>Q, S, X, Z</td>
</tr>
<tr>
<td></td>
<td>M, N, P, R, U</td>
<td></td>
</tr>
<tr>
<td>Code number</td>
<td></td>
<td></td>
</tr>
<tr>
<td>A</td>
<td>2</td>
<td>2</td>
</tr>
<tr>
<td>B</td>
<td>1</td>
<td>2</td>
</tr>
<tr>
<td>C</td>
<td>2</td>
<td>2</td>
</tr>
<tr>
<td>D</td>
<td>1</td>
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<th>Following number</th>
<th>Preceding Numeral</th>
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<th>400</th>
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<th>Letter height (mm)</th>
<th>Width (mm)</th>
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<table>
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<th>Letter</th>
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<th>Width (mm)</th>
</tr>
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</tr>
<tr>
<td>0</td>
<td>143</td>
<td>214</td>
<td>286</td>
</tr>
</tbody>
</table>

INSTRUCTIONS

1. To determine the proper SPACE between letters or numerals, obtain the code number from table a or b and enter table c for that code number to the desired letter or numeral height.

2. The space between words or groups of characters forming an abbreviation or symbol should be equal to 0.5 to 0.75 of the height of the characters used except that where an arrow is located with a single character such as 'A →', the space may be reduced to not less than one quarter of the character of the height in order to provide a good visual balance.

3. Where the numeral follows a letter or vice versa use Code 1.

4. Where a hyphen, dot, or diagonal stroke follows a character or vice versa use Code 1.
Figure 4.3 Sign dimensions
## APPENDIX 5. AERONAUTICAL DATA QUALITY REQUIREMENTS

Table 1. Latitude and longitude

<table>
<thead>
<tr>
<th>Latitude and longitude</th>
<th>Accuracy</th>
<th>Data type</th>
<th>Classification</th>
<th>Integrity</th>
</tr>
</thead>
<tbody>
<tr>
<td>Aerodrome reference point</td>
<td>30 m</td>
<td>surveyed/calculated</td>
<td>routine</td>
<td>1 × 10⁻³</td>
</tr>
<tr>
<td>NAVAIDS located at the aerodrome</td>
<td>3 m</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10⁻⁵</td>
</tr>
<tr>
<td>Obstacles in the circling area and at the aerodrome</td>
<td>3 m</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10⁻⁵</td>
</tr>
<tr>
<td>Significant obstacles in the approach and take-off area</td>
<td>3 m</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10⁻⁵</td>
</tr>
<tr>
<td>Runway threshold</td>
<td>1 m</td>
<td>surveyed</td>
<td>critical</td>
<td>1 × 10⁻⁸</td>
</tr>
<tr>
<td>Runway end (flight path alignment point)</td>
<td>1 m</td>
<td>surveyed</td>
<td>critical</td>
<td>1 × 10⁻⁸</td>
</tr>
<tr>
<td>Runway centre line points</td>
<td>1 m</td>
<td>surveyed</td>
<td>critical</td>
<td>1 × 10⁻⁸</td>
</tr>
<tr>
<td>Taxiway centre line points</td>
<td>0.5 m</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10⁻⁵</td>
</tr>
<tr>
<td>Aircraft stand-points/INS check-points</td>
<td>0.5 m</td>
<td>surveyed</td>
<td>routine</td>
<td>1 × 10⁻³</td>
</tr>
</tbody>
</table>
### Table 2. Elevation/Altitude/Height

<table>
<thead>
<tr>
<th>Elevation/altitude/height</th>
<th>Accuracy</th>
<th>Data type</th>
<th>Classification</th>
<th>Integrity</th>
</tr>
</thead>
<tbody>
<tr>
<td>Aerodrome elevation</td>
<td>0.5 m or 1 ft</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>WGS-84 geoid undulation at aerodrome elevation position</td>
<td>0.5 m or 1 ft</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>Runway threshold, non-precision approaches</td>
<td>0.5 m or 1 ft</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>WGS-84 geoid undulation at runway threshold, non-precision approaches</td>
<td>0.5 m or 1 ft</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>Runway threshold, precision approaches</td>
<td>0.25 m or 1 ft</td>
<td>surveyed</td>
<td>critical</td>
<td>1 × 10^-8</td>
</tr>
<tr>
<td>WGS-84 geoid undulation at runway threshold, precision approaches</td>
<td>0.25 m or 1 ft</td>
<td>surveyed</td>
<td>critical</td>
<td>1 × 10^-8</td>
</tr>
<tr>
<td>Obstacles in the approach and take-off areas</td>
<td>1 m or 1 ft</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>Obstacles in the circling areas and at the aerodrome</td>
<td>1 m or 1 ft</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>Distance measuring equipment/precision (DME/P)</td>
<td>3 m (10 ft)</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
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</tbody>
</table>

### Table 3. Declination and Magnetic Variation

<table>
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<th>Declination/variation</th>
<th>Accuracy</th>
<th>Data type</th>
<th>Classification</th>
<th>Integrity</th>
</tr>
</thead>
<tbody>
<tr>
<td>Aerodrome magnetic variation</td>
<td>1 degree</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>ILS localizer antenna magnetic variation</td>
<td>1 degree</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
<tr>
<td>MLS azimuth antenna magnetic variation</td>
<td>1 degree</td>
<td>surveyed</td>
<td>essential</td>
<td>1 × 10^-5</td>
</tr>
</tbody>
</table>
### Table 4. Bearing

<table>
<thead>
<tr>
<th>Bearing</th>
<th>Accuracy Data type</th>
<th>Classification Integrity</th>
</tr>
</thead>
<tbody>
<tr>
<td>ILS localizer alignment</td>
<td>1/100 degree</td>
<td>essential</td>
</tr>
<tr>
<td></td>
<td>surveyed</td>
<td>$1 \times 10^{-5}$</td>
</tr>
<tr>
<td>MLS zero azimuth alignment</td>
<td>1/100 degree</td>
<td>essential</td>
</tr>
<tr>
<td></td>
<td>surveyed</td>
<td>$1 \times 10^{-5}$</td>
</tr>
<tr>
<td>Runway bearing</td>
<td>1/100 degree</td>
<td>routine</td>
</tr>
<tr>
<td></td>
<td>surveyed</td>
<td>$1 \times 10^{-3}$</td>
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### Table 5. Length/Distance/Dimension

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<th>Accuracy Data type</th>
<th>Classification Integrity</th>
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<td>Runway length</td>
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<td>critical</td>
</tr>
<tr>
<td></td>
<td>surveyed</td>
<td>$1 \times 10^{-8}$</td>
</tr>
<tr>
<td>Stopway length</td>
<td>1 m or 1 ft</td>
<td>critical</td>
</tr>
<tr>
<td></td>
<td>surveyed</td>
<td>$1 \times 10^{-8}$</td>
</tr>
<tr>
<td>Landing distance available</td>
<td>1 m or 1 ft</td>
<td>critical</td>
</tr>
<tr>
<td></td>
<td>surveyed</td>
<td>$1 \times 10^{-8}$</td>
</tr>
<tr>
<td>ILS localizer antenna-runway end, distance</td>
<td>3 m (10 ft)</td>
<td>routine</td>
</tr>
<tr>
<td></td>
<td>calculated</td>
<td>$1 \times 10^{-3}$</td>
</tr>
<tr>
<td>ILS glide slope antenna-threshold, distance along centre line</td>
<td>3 m (10 ft)</td>
<td>routine</td>
</tr>
<tr>
<td></td>
<td>calculated</td>
<td>$1 \times 10^{-3}$</td>
</tr>
<tr>
<td>ILS markers-threshold distance</td>
<td>3 m (10 ft)</td>
<td>essential</td>
</tr>
<tr>
<td></td>
<td>calculated</td>
<td>$1 \times 10^{-5}$</td>
</tr>
<tr>
<td>ILS DME antenna-threshold, distance along centre line</td>
<td>3 m (10 ft)</td>
<td>essential</td>
</tr>
<tr>
<td></td>
<td>calculated</td>
<td>$1 \times 10^{-5}$</td>
</tr>
<tr>
<td>MLS azimuth antenna-runway end, distance</td>
<td>3 m (10 ft)</td>
<td>routine</td>
</tr>
<tr>
<td></td>
<td>calculated</td>
<td>$1 \times 10^{-3}$</td>
</tr>
<tr>
<td>MLS elevation antenna-threshold, distance along centre line</td>
<td>3 m (10 ft)</td>
<td>routine</td>
</tr>
<tr>
<td></td>
<td>calculated</td>
<td>$1 \times 10^{-3}$</td>
</tr>
<tr>
<td>MLS DME/P antenna-threshold, distance along centre line</td>
<td>3 m (10 ft)</td>
<td>essential</td>
</tr>
<tr>
<td></td>
<td>calculated</td>
<td>$1 \times 10^{-5}$</td>
</tr>
</tbody>
</table>
Note.— High-intensity obstacle lighting is recommended on structures with a height of more than 150 m above ground level. If medium-intensity lighting is used, marking will also be required.

Figure 6.1 Medium-intensity flashing-white obstacle lighting system, Type A
Appendix 6  

Annex 14 — Aerodromes

Note.— For night-time use only.

Figure 6.2 Medium-intensity flashing-red obstacle lighting system, Type B
Note.— For night-time use only.

Figure 6.3 Medium-intensity fixed-red obstacle lighting system, Type C
Note.— High-intensity obstacle lighting is recommended on structures with a height of more than 150 m above ground level. If medium-intensity lighting is used, marking will also be required.

Figure 6.4 Medium-intensity dual obstacle lighting system, Type A/Type B
Note.— High-intensity obstacle lighting is recommended on structures with a height of more than 150 m above ground level. If medium-intensity lighting is used, marking will also be required.

Figure 6.5 Medium-intensity dual obstacle lighting system, Type A/Type C
Figure 6.6  High-intensity flashing-white obstacle lighting system, Type A
Figure 6.7  High-/medium-intensity dual obstacle lighting system, Type A/Type B
Figure 6.8  High-/medium-intensity dual obstacle lighting system, Type A/Type C
ATTACHMENT A. GUIDANCE MATERIAL
SUPPLEMENTARY TO ANNEX 14, VOLUME I

1. Number, siting and orientation of runways

Siting and orientation of runways

1.1 Many factors should be taken into account in the determination of the siting and orientation of runways. Without attempting to provide an exhaustive list of these factors nor an analysis of their effects, it appears useful to indicate those which most frequently require study. These factors may be classified under four headings:

1.1.1 Type of operation. Attention should be paid in particular to whether the aerodrome is to be used in all meteorological conditions or only in visual meteorological conditions, and whether it is intended for use by day and night, or only by day.

1.1.2 Climatological conditions. A study of the wind distribution should be made to determine the usability factor. In this regard, the following comments should be taken into account:

a) Wind statistics used for the calculation of the usability factor are normally available in ranges of speed and direction, and the accuracy of the results obtained depends, to a large extent, on the assumed distribution of observations within these ranges. In the absence of any sure information as to the true distribution, it is usual to assume a uniform distribution since, in relation to the most favourable runway orientations, this generally results in a slightly conservative figure for the usability factor.

b) The maximum mean cross-wind components given in Chapter 3, 3.1.2 refer to normal circumstances. There are some factors which may require that a reduction of those maximum values be taken into account at a particular aerodrome. These include:

1) the wide variations which may exist, in handling characteristics and maximum permissible cross-wind components, among diverse types of aeroplanes (including future types) within each of the three groups given in 3.1.2;

2) prevalence and nature of gusts;

3) prevalence and nature of turbulence;

4) the availability of a secondary runway;

5) the width of runways;

6) the runway surface conditions — water, snow and ice on the runway materially reduce the allowable cross-wind component; and

7) the strength of the wind associated with the limiting cross-wind component.

A study should also be made of the occurrence of poor visibility and/or low cloud base. Account should be taken of their frequency as well as the accompanying wind direction and speed.

1.1.3 Topography of the aerodrome site, its approaches, and surroundings, particularly:

a) compliance with the obstacle limitation surfaces;

b) current and future land use. The orientation and layout should be selected so as to protect as far as possible the particularly sensitive areas such as residential, school and hospital zones from the discomfort caused by aircraft noise;

c) current and future runway lengths to be provided;

d) construction costs; and

e) possibility of installing suitable non-visual and visual aids for approach-to-land.

1.1.4 Air traffic in the vicinity of the aerodrome, particularly:

a) proximity of other aerodromes or ATS routes;

b) traffic density; and

c) air traffic control and missed approach procedures.

Number of runways in each direction

1.2 The number of runways to be provided in each direction depends on the number of aircraft movements to be catered to.

2. Clearways and stopways

2.1 The decision to provide a stopway and/or a clearway as an alternative to an increased length of runway will depend on the physical characteristics of the area beyond the runway...
end, and on the operating performance requirements of the prospective aeroplanes. The runway, stopway and clearway lengths to be provided are determined by the aeroplane take-off performance, but a check should also be made of the landing distance required by the aeroplanes using the runway to ensure that adequate runway length is provided for landing. The length of a clearway, however, cannot exceed half the length of take-off run available.

2.2 The aeroplane performance operating limitations require a length which is enough to ensure that the aeroplane can, after starting a take-off, either be brought safely to a stop or complete the take-off safely. For the purpose of discussion it is supposed that the runway, stopway and clearway lengths provided at the aerodrome are only just adequate for the aeroplane requiring the longest take-off and accelerate-stop distances, taking into account its take-off mass, runway characteristics and ambient atmospheric conditions. Under these circumstances there is, for each take-off, a speed, called the decision speed; below this speed, the take-off must be abandoned if an engine fails, while above it the take-off must be completed. A very long take-off run and take-off distance would be required to complete a take-off when an engine fails before the decision speed is reached, because of the insufficient speed and the reduced power available. There would be no difficulty in stopping in the remaining accelerate-stop distance available provided action is taken immediately. In these circumstances the correct course of action would be to abandon the take-off.

On the other hand, if an engine fails after the decision speed is reached, the aeroplane will have sufficient speed and power available to complete the take-off safely in the remaining take-off distance available. However, because of the high speed, there would be difficulty in stopping the aeroplane in the remaining accelerate-stop distance available.

2.3 The decision speed is not a fixed speed for any aeroplane, but can be selected by the pilot within limits to suit the accelerate-stop and take-off distance available, aeroplane take-off mass, runway characteristics, and ambient atmospheric conditions at the aerodrome. Normally, a higher decision speed is selected as the accelerate-stop distance available increases.

2.4 A variety of combinations of accelerate-stop distances required and take-off distances required can be obtained to accommodate a particular aeroplane, taking into account the aeroplane take-off mass, runway characteristics, and ambient atmospheric conditions. Each combination requires its particular length of take-off run.

2.5 The most familiar case is where the decision speed is such that the take-off distance required is equal to the accelerate-stop distance required; this value is known as the balanced field length. Where stopway and clearway are not provided, these distances are both equal to the runway length. However, if landing distance is for the moment ignored, runway is not essential for the whole of the balanced field length, as the take-off run required is, of course, less than the balanced field length. The balanced field length can, therefore, be provided by a runway supplemented by an equal length of clearway and stopway, instead of wholly as a runway. If the runway is used for take-off in both directions, an equal length of clearway and stopway has to be provided at each runway end. The saving in runway length is, therefore, bought at the cost of a greater over-all length.

2.6 In case economic considerations preclude the provision of stopway and, as a result, only runway and clearway are to be provided, the runway length (neglecting landing requirements) should be equal to the accelerate-stop distance required or the take-off run required, whichever is the greater. The take-off distance available will be the length of the runway plus the length of clearway.

2.7 The minimum runway length and the maximum stopway or clearway length to be provided may be determined as follows, from the data in the aeroplane flight manual for the aeroplane considered to be critical from the viewpoint of runway length requirements:

a) if a stopway is economically possible, the lengths to be provided are those for the balanced field length. The runway length is the take-off run required or the landing distance required, whichever is the greater. If the accelerate-stop distance required is greater than the runway length so determined, the excess may be provided as stopway, usually at each end of the runway. In addition, a clearway of the same length as the stopway must also be provided;

b) if a stopway is not to be provided, the runway length is the landing distance required, or if it is greater, the accelerate-stop distance required, which corresponds to the lowest practical value of the decision speed. The excess of the take-off distance required over the runway length may be provided as clearway, usually at each end of the runway.

2.8 In addition to the above consideration, the concept of clearways in certain circumstances can be applied to a situation where the take-off distance required for all engines operating exceeds that required for the engine failure case.

2.9 The economy of a stopway can be entirely lost if, after each usage, it must be regraded and compacted. Therefore, it should be designed to withstand at least a certain number of loadings of the aeroplane which the stopway is intended to serve without inducing structural damage to the aeroplane.

3. Calculation of declared distances

3.1 The declared distances to be calculated for each runway direction comprise: the take-off run available (TORA), take-off distance available (TODA), accelerate-stop distance available (ASDA), and landing distance available (LDA).
3.2 Where a runway is not provided with a stopway or clearway and the threshold is located at the extremity of the runway, the four declared distances should normally be equal to the length of the runway, as shown in Figure A-1 (A).

3.3 Where a runway is provided with a clearway (CWY), then the TODA will include the length of clearway, as shown in Figure A-1 (B).

3.4 Where a runway is provided with a stopway (SWY), then the ASDA will include the length of stopway, as shown in Figure A-1 (C).

3.5 Where a runway has a displaced threshold, then the LDA will be reduced by the distance the threshold is displaced, as shown in Figure A-1 (D). A displaced threshold affects only the LDA for approaches made to that threshold; all declared distances for operations in the reciprocal direction are unaffected.

3.6 Figures A-1 (B) through A-1 (D) illustrate a runway provided with a clearway or a stopway or having a displaced threshold. Where more than one of these features exist, then more than one of the declared distances will be modified — but the modification will follow the same principle illustrated. An example showing a situation where all these features exist is shown in Figure A-1 (E).

3.7 A suggested format for providing information on declared distances is given in Figure A-1 (F). If a runway direction cannot be used for take-off or landing, or both, because it is operationally forbidden, then this should be declared and the words “not usable” or the abbreviation “NU” entered.

4. Slopes on a runway

4.1 Distance between slope changes

The following example illustrates how the distance between slope changes is to be determined (see Figure A-2):

D for a runway where the code number is 3 should be at least:

\[15 \times (|x - y| + |y - z|) \text{ m}\]

\[|x - y|\] being the absolute numerical value of \(x - y\)

\[|y - z|\] being the absolute numerical value of \(y - z\)

Assuming \(x = +0.01\)

\(y = -0.005\)

\(z = +0.005\)

then \(|x - y| = 0.015\)

\(|y - z| = 0.01\)

To comply with the specifications, D should be not less than:

\[15 \times 0.015 + 0.01 = 375 \text{ m}\]

5. Runway surface evenness

5.1 In adopting tolerances for runway surface irregularities, the following standard of construction is achievable for short distances of 3 m and conforms to good engineering practice:

Except across the crown of a camber or across drainage channels, the finished surface of the wearing course is to be of such regularity that, when tested with a 3 m straight-edge placed anywhere in any direction on the surface, there is no deviation greater than 3 mm between the bottom of the straight-edge and the surface of the pavement anywhere along the straight edge.

5.2 Caution should also be exercised when inserting runway lights or drainage grilles in runway surfaces to ensure that adequate smoothness of the surface is maintained.

5.3 The operation of aircraft and differential settlement of surface foundations will eventually lead to increases in surface irregularities. Small deviations in the above tolerances will not seriously hamper aircraft operations. In general, isolated irregularities of the order of 2.5 cm to 3 cm over a 45 m distance are tolerable. Exact information of the maximum acceptable deviation cannot be given, as it varies with the type and speed of an aircraft.
Annex I4 - Aerodromes

Note.— All declared distances are illustrated for operations from left to right.

Figure A-1. Illustration of declared distances.
5.4 Deformation of the runway with time may also increase the possibility of the formation of water pools. Pools as shallow as approximately 3 mm in depth, particularly if they are located where they are likely to be encountered at high speed by landing aeroplanes, can induce aquaplaning, which can then be sustained on a wet runway by a much shallower depth of water. Improved guidance regarding the significant length and depth of pools relative to aquaplaning is the subject of further research. It is, of course, especially necessary to prevent pools from forming whenever there is a possibility that they might become frozen.

6. Determining and expressing the friction characteristics of snow- and ice-covered paved surfaces

6.1 There is an operational need for reliable and uniform information concerning the friction characteristics of ice- and snow-covered runways. Accurate and reliable indications of surface friction characteristics can be obtained by friction measuring devices; however, further experience is required to correlate the results obtained by such equipment with aircraft performance, owing to the many variables involved, such as: aircraft mass, speed, braking mechanism, tire and undercarriage characteristics.

6.2 The friction coefficient should be measured if a runway is covered wholly or partly by snow or ice and repeated as conditions change. Friction measurements and/or braking action assessments on surfaces other than runways should be made when an unsatisfactory friction condition can be expected on such surfaces.

6.3 The measurement of the friction coefficient provides the best basis for determining surface friction conditions. The value of surface friction should be the maximum value which occurs when a wheel is slipping but still rolling. Various friction measuring devices may be used. As there is an operational need for uniformity in the method of assessing and reporting runway friction conditions, the measurements should preferably be made with equipment which provides continuous measuring of the maximum friction along the entire runway. Measuring techniques and information on limitations of the various friction measuring devices and precautions to be observed are given in the Airport Services Manual, Part 2.

6.4 A chart, based on results of tests conducted on selected ice- or snow-covered surfaces, showing the correlation between certain friction measuring devices on ice- or snow-covered surfaces is presented in the Airport Services Manual, Part 2.

6.5 The friction conditions of a runway should be expressed as “braking action information” in terms of the measured friction coefficient μ or estimated braking action. Specific numerical μ values are necessarily related to the design and construction of each friction measuring device as well as to the surface being measured and the speed employed.

6.6 The table below with associated descriptive terms was developed from friction data collected only in compacted snow and ice and should not therefore be taken to be absolute values applicable in all conditions. If the surface is affected by snow or ice and the braking action is reported as “good”, pilots should not expect to find conditions as good as on a clean dry runway (where the available friction may well be greater than that needed in any case). The value “good” is a comparative value and is intended to mean that aeroplanes should not experience directional control or braking difficulties, especially when landing.

<table>
<thead>
<tr>
<th>Measured coefficient</th>
<th>Estimated braking action</th>
<th>Code</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.40 and above</td>
<td>Good</td>
<td>5</td>
</tr>
<tr>
<td>0.39 to 0.36</td>
<td>Medium to good</td>
<td>4</td>
</tr>
<tr>
<td>0.35 to 0.30</td>
<td>Medium</td>
<td>3</td>
</tr>
<tr>
<td>0.29 to 0.26</td>
<td>Medium to poor</td>
<td>2</td>
</tr>
<tr>
<td>0.25 and below</td>
<td>Poor</td>
<td>1</td>
</tr>
</tbody>
</table>

6.7 It has been found necessary to provide surface friction information for each third of a runway. The thirds are called...
A, B and C. For the purpose of reporting information to aeronautical service units, section A is always the section associated with the lower runway designation number. When giving landing information to a pilot before landing, the sections are however referred to as first, second or third part of the runway. The first part always means the first third of the runway as seen in the direction of landing. Friction measurements are made along two lines parallel to the runway, i.e. along a line on each side of the centre line approximately 3 m or that distance from the centre line at which most operations take place. The objective of the tests is to determine the mean friction value for sections A, B and C. In cases where a continuous friction measuring device is used, the mean values are obtained from the friction values recorded for each section. The distance between each test point should be approximately 10 per cent of the usable length of the runway. If it is decided that a single test line on one side of the runway centre line gives adequate coverage of the runway, then it follows that each third of the runway should have three tests carried out on it. Test results and calculated mean friction values are entered in a special form (see Airport Services Manual, Part 2).

Note.—Where applicable, figures for stopway friction value should also be made available on request.

6.8 A continuous friction measuring device (e.g. Skiddometer, Surface Friction Tester, Mu-meter, Runway Friction Tester or Grip Tester), can be used for measuring the friction values for compacted snow- and ice-covered runways. A decelometer (e.g. Tapley Meter or Brakemeter — Dynometer) may be used on certain surface conditions, e.g. compacted snow, ice and very thin layers of dry snow. Other friction measuring devices can be used, provided they have been correlated with at least one of the types mentioned above. A decelometer should not be used in loose snow or slush, as it can give misleading friction values. Other friction measuring devices can also give misleading friction values under certain combinations of contaminants and air/pavement temperature.

6.9 The Airport Services Manual, Part 2 provides guidance on the uniform use of test equipment to achieve compatible test results and other information on removal of surface contamination and improvement of friction conditions.

7. **Determination of friction characteristics of wet paved runways**

7.1 The friction of a wet paved runway should be measured to:

a) verify the friction characteristics of new or resurfaced paved runways when wet (Chapter 3, 3.1.23);

b) assess periodically the slipperiness of paved runways when wet (Chapter 9, 9.4.5);

c) determine the effect on friction when drainage characteristics are poor (Chapter 9, 9.4.8); and

d) determine the friction of paved runways that become slippery under unusual conditions (Chapter 2, 2.9.8).

7.2 Runways should be evaluated when first constructed or after resurfacing to determine the wet runway surface friction characteristics. Although it is recognized that friction reduces with use, this value will represent the friction of the relatively long central portion of the runway that is uncontaminated by rubber deposits from aircraft operations and is therefore of operational value. Evaluation tests should be made on clean surfaces. If it is not possible to clean a surface before testing, then for purposes of preparing an initial report a test could be made on a portion of clean surface in the central part of the runway.

7.3 Friction tests of existing surface conditions should be taken periodically in order to identify runways with low friction when wet. A State should define what minimum friction level it considers acceptable before a runway is classified as slippery when wet and publish this value in the State’s aeronautical information publication (AIP). When the friction of a runway is found to be below this reported value, such information should be promulgated by NOTAM. The State should also establish a maintenance planning level, below which, appropriate corrective maintenance action should be initiated to improve the friction. However, when the friction characteristics for either the entire runway or a portion thereof are below the minimum friction level, corrective maintenance action must be taken without delay. Friction measurements should be taken at intervals that will ensure identification of runways in need of maintenance or special surface treatment before the condition becomes serious. The time interval between measurements will depend on factors such as: aircraft type and frequency of usage, climatic conditions, pavement type, and pavement service and maintenance requirements.

7.4 For uniformity and to permit comparison with other runways, friction tests of existing, new or resurfaced runways should be made with a continuous friction measuring device provided with a smooth tread tire. The device should have a capability of using self-wetting features to enable measurements of the friction characteristics of the surface to be made at a water depth of at least 1 mm.

7.5 When it is suspected that the friction characteristics of a runway may be reduced because of poor drainage, owing to inadequate slopes or depressions, then an additional test should be made, but this time under natural conditions representative of a local rain. This test differs from the previous one in that water depths in the poorly cleared areas are normally greater in a local rain condition. The test results are thus more apt to identify problem areas having low friction values that could induce aquaplaning than the previous test. If circumstances do not permit tests to be conducted during natural conditions representative of a rain, then this condition may be simulated.
7.6 Even when the friction has been found to be above the level set by the State to define a slippery runway, it may be known that under unusual conditions, such as after a long dry period, the runway may have become slippery. When such a condition is known to exist, then a friction measurement should be made as soon as it is suspected that the runway may have become slippery.

7.7 When the results of any of the measurements identified in 7.3 through 7.6 indicate that only a particular portion of a runway surface is slippery, then action to promulgate this information and, if appropriate, take corrective action is equally important.

7.8 When conducting friction tests on wet runways, it is important to note that, unlike compacted snow and ice conditions, in which there is very limited variation of the friction coefficient with speed, a wet runway produces a drop in friction with an increase in speed. However, as the speed increases, the rate at which the friction is reduced becomes less. Among the factors affecting the friction coefficient between the tire and the runway surface, texture is particularly important. If the runway has a good macro-texture allowing the water to escape beneath the tire, then the friction value will be less affected by speed. Conversely, a low macro-texture surface will produce a larger drop in friction with increase in speed. Accordingly, when testing runways to determine their friction characteristics and whether maintenance action is necessary to improve it, a speed high enough to reveal these friction/speed variations should be used.

7.9 Annex 14, Volume I requires States to specify two friction levels as follows:

- a) a maintenance friction level below which corrective maintenance action should be initiated; and
- b) a minimum friction level below which information that a runway may be slippery when wet should be made available.

Furthermore, States should establish criteria for the friction characteristics of new or resurfaced runway surfaces. Table A-1 provides guidance on establishing the design objective for new runway surfaces and minimum friction levels for runway surfaces in use.

7.10 The friction values given above are absolute values and are intended to be applied without any tolerance. These values were developed from a research study conducted in a State. The two friction measuring tires mounted on the Mu-meter were smooth tread and had a special rubber formulation, i.e. Type A. The tires were tested at a 15 degree included angle of alignment along the longitudinal axis of the trailer. The single friction measuring tires mounted on the Skiddometer, Surface Friction Tester, Runway Friction Tester and TATRA were smooth tread and used the same rubber formulation, i.e. Type B. The GRIPTESTER was tested with a single smooth tread tire having the same rubber formulation as Type B but the size was smaller, i.e. Type C. The specifications of these tires (i.e. Types A, B and C) are contained in the Airport Services Manual, Part 2. Friction measuring devices using rubber formulation, tire tread/groove patterns, water depth, tire pressures, or test speeds different from those used in the programme described above, cannot be directly equated with the friction values given in the table. The values in columns (5), (6) and (7) are averaged values representative of

---

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<tr>
<th>Test equipment</th>
<th>Type</th>
<th>Pressure (kPa)</th>
<th>Test speed (km/h)</th>
<th>Test water depth (mm)</th>
<th>Design objective for new surface</th>
<th>Maintenance planning level</th>
<th>Minimum friction level</th>
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<tr>
<td>Mu-meter Trailer</td>
<td>A</td>
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<tr>
<td></td>
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<td>0.60</td>
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<td>0.36</td>
<td>0.24</td>
</tr>
</tbody>
</table>
the runway or significant portion thereof. It is considered desirable to test the friction characteristics of a paved runway at more than one speed.

7.11 Other friction measuring devices can be used, provided they have been correlated with at least one test equipment mentioned above. The *Airport Services Manual*, Part 2 provides guidance on the methodology for determining the friction values corresponding to the design objective, maintenance planning level and minimum friction level for a friction tester not identified in the above table.

8. Strips

8.1 Shoulders

8.1.1 The shoulder of a runway or stopway should be prepared or constructed so as to minimize any hazard to an aeroplane running off the runway or stopway. Some guidance is given in the following paragraphs on certain special problems which may arise, and on the further question of measures to avoid the ingestion of loose stones or other objects by turbine engines.

8.1.2 In some cases, the bearing strength of the natural ground in the strip may be sufficient, without special preparation, to meet the requirements for shoulders. Where special preparation is necessary, the method used will depend on local soil conditions and the mass of the aeroplanes the runway is intended to serve. Soil tests will help in determining the best method of improvement (e.g. drainage, stabilization, surfacing, light paving).

8.1.3 Attention should also be paid when designing shoulders to prevent the ingestion of stones or other objects by turbine engines. Similar considerations apply here to those which are discussed for the margins of taxiways in the *Aerodrome Design Manual*, Part 2, both as to the special measures which may be necessary and as to the distance over which such special measures, if required, should be taken.

8.1.4 Where shoulders have been treated specially, either to provide the required bearing strength or to prevent the presence of stones or debris, difficulties may arise because of a lack of visual contrast between the runway surface and that of the adjacent strip. This difficulty can be overcome either by providing a good visual contrast in the surfacing of the runway or strip, or by providing a runway side stripe marking.

8.2 Objects on strips

Within the general area of the strip adjacent to the runway, measures should be taken to prevent an aeroplane’s wheel, when sinking into the ground, from striking a hard vertical face. Special problems may arise for runway light fittings or other objects mounted in the strip or at the intersection with a taxiway or another runway. In the case of construction, such as runways or taxiways, where the surface must also be flush with the strip surface, a vertical face can be eliminated by chamfering from the top of the construction to not less than 30 cm below the strip surface level. Other objects, the functions of which do not require them to be at surface level, should be buried to a depth of not less than 30 cm.

8.3 Grading of a strip for precision approach runways

Chapter 3, 3.3.8 recommends that the portion of a strip of an instrument runway within at least 75 m from the centre line should be graded where the code number is 3 or 4. For a precision approach runway, it may be desirable to adopt a greater width where the code number is 3 or 4. Figure A-3 shows the shape and dimensions of a wider strip that may be considered for such a runway. This strip has been designed using information on aircraft running off runways. The portion to be graded extends to a distance of 105 m from the centre line, except that the distance is gradually reduced to 75 m from the centre line at both ends of the strip, for a length of 150 m from the runway end.

9. Runway end safety areas

9.1 Where a runway end safety area is provided in accordance with Chapter 3, consideration should be given to providing an area long enough to contain overruns and undershoots resulting from a reasonably probable combination of adverse operational factors. On a precision approach runway, the ILS localizer is normally the first upstanding obstacle, and the runway end safety area should extend up to this facility. In other circumstances and on a non-precision approach or non-instrument runway, the first upstanding obstacle may be a road, a railroad or other constructed or natural feature. In such circumstances, the runway end safety area should extend as far as the obstacle.

9.2 Where provision of a runway end safety area may involve encroachment in areas where it would be particularly prohibitive to implement, and the appropriate authority considers a runway end safety area essential, consideration may have to be given to reducing some of the declared distances.

10. Location of threshold

10.1 General

10.1.1 The threshold is normally located at the extremity of a runway, if there are no obstacles penetrating above the
10.1.2 In determining that no obstacle penetrate above the approach surface, account should be taken of mobile objects (vehicles on roads, trains, etc.) at least within that portion of the approach area within 1 200 m longitudinally from the threshold and of an over-all width of not less than 150 m.

10.2 Displaced threshold

10.2.1 If an object extends above the approach surface and the object cannot be removed, consideration should be given to displacing the threshold permanently.

10.2.2 To meet the obstacle limitation objectives of Chapter 4, the threshold should ideally be displaced down the runway for the distance necessary to provide that the approach surface is cleared of obstacles.

10.2.3 However, displacement of the threshold from the runway extremity will inevitably cause reduction of the landing distance available, and this may be of greater operational significance than penetration of the approach surface by marked and lighted obstacles. A decision to displace the threshold, and the extent of such displacement, should therefore have regard to an optimum balance between the considerations of clear approach surfaces and adequate landing distance. In deciding this question, account will need to be taken of the types of aeroplanes which the runway is intended to serve, the limiting visibility and cloud base conditions under which the runway will be used, the position of the obstacles in relation to the threshold and extended centre line and, in the case of a precision approach runway, the significance of the obstacles to the determination of the obstacle clearance limit.

10.2.4 Notwithstanding the consideration of landing distance available, the selected position for the threshold should not be such that the obstacle-free surface to the threshold is steeper than 3.3 per cent where the code number is 4 or steeper than 5 per cent where the code number is 3.

10.2.5 In the event of a threshold being located according to the criteria for obstacle-free surfaces in the preceding paragraph, the obstacle marking requirements of Chapter 6 should continue to be met in relation to the displaced threshold.

11. Approach lighting systems

11.1 Types and characteristics

11.1.1 The specifications in this volume provide for the basic characteristics for simple and precision approach lighting systems. For certain aspects of these systems, some latitude is permitted, for example, in the spacing between centre line lights and crossbars. The approach lighting patterns that have been generally adopted are shown in Figures A-5 and A-6. A diagram of the inner 300 m of the precision approach category II and III lighting system is shown in Figure 5-10.

11.1.2 The approach lighting configuration is to be provided irrespective of the location of the threshold, i.e. whether the threshold is at the extremity of the runway or displaced from the runway extremity. In both cases, the approach lighting system should extend up to the threshold. However, in the case of a displaced threshold, inset lights are used from the runway extremity up to the threshold to obtain the specified configuration. These inset lights are designed to...
satisfy the structural requirements specified in Chapter 5, 5.3.1.8, and the photometric requirements specified in Appendix 2, Figure 2.1 or 2.2.

11.1.3 Flight path envelopes to be used in designing the lighting are shown in Figure A-4.

11.2 Installation tolerances

**Horizontal**

11.2.1 The dimensional tolerances are shown in Figure A-6.

11.2.2 The centre line of an approach lighting system should be as coincident as possible with the extended centre line of the runway with a maximum tolerance of ±15 ′.

11.2.3 The longitudinal spacing of the centre line lights should be such that one light (or group of lights) is located in the centre of each crossbar, and the intervening centre line lights are spaced as evenly as practicable between two crossbars or a crossbar and a threshold.

11.2.4 The crossbars and barrettes should be at right angles to the centre line of the approach lighting system with a tolerance of ±30 ′, if the pattern in Figure A-6 (A) is adopted or ±2 ′, if Figure A-6 (B) is adopted.

11.2.5 When a crossbar has to be displaced from its standard position, any adjacent crossbar should, where possible, be displaced by appropriate amounts in order to reduce the differences in the crossbar spacing.

11.2.6 When a crossbar in the system shown in Figure A-6 (A) is displaced from its standard position, its overall length should be adjusted so that it remains one-twentieth of the actual distance of the crossbar from the point of origin. It is not necessary, however, to adjust the standard 2.7 m spacing between the crossbar lights, but the crossbars should be kept symmetrical about the centre line of the approach lighting.

**Vertical**

11.2.7 The ideal arrangement is to mount all the approach lights in the horizontal plane passing through the threshold (see Figure A-7), and this should be the general aim as far as local conditions permit. However, buildings, trees, etc., should not obscure the lights from the view of a pilot who is assumed to be 1° below the electronic glide path in the vicinity of the outer marker.

11.2.8 Within a stopway or clearway, and within 150 m of the end of a runway, the lights should be mounted as near to the ground as local conditions permit in order to minimize risk of damage to aeroplanes in the event of an overrun or undershoot. Beyond the stopway and clearway, it is not so necessary for the lights to be mounted close to the ground and therefore undulations in the ground contours can be compensated for by mounting the lights on poles of appropriate height.

11.2.9 It is desirable that the lights be mounted so that, as far as possible, no object within a distance of 60 m on each side of the centre line protrudes through the plane of the approach lighting system. Where a tall object exists within 60 m of the centre line and within 1350 m from the threshold for a precision approach lighting system, or 900 m for a simple approach lighting system, it may be advisable to install the lights so that the plane of the outer half of the pattern clears the top of the object.

11.2.10 In order to avoid giving a misleading impression of the plane of the ground, the lights should not be mounted below a gradient of 1 in 66 downwards from the threshold to a point 300 m out, and below a gradient of 1 in 40 beyond the 300 m point. For a precision approach category II and III lighting system, more stringent criteria may be necessary, e.g. negative slopes not permitted within 450 m of the threshold.

11.2.11 **Centre line.** The gradients of the centre line in any section (including a stopway or clearway) should be as small as practicable, and the changes in gradients should be as few and small as can be arranged and should not exceed 1 in 60. Experience has shown that as one proceeds outwards from the runway, rising gradients in any section of up to 1 in 66, and falling gradients of down to 1 in 40, are acceptable.

11.2.12 **Crossbars.** The crossbar lights should be so arranged as to lie on a straight line passing through the associated centre line lights, and wherever possible this line should be horizontal. It is permissible, however, to mount the lights on a transverse gradient not more than 1 in 80, if this enables crossbar lights within a stopway or clearway to be mounted nearer to the ground on sites where there is a crossfall.

11.3 Clearance of obstacles

11.3.1 An area, hereinafter referred to as the light plane, has been established for obstacle clearance purposes, and all lights of the system are in this plane. This plane is rectangular in shape and symmetrically located about the approach lighting system’s centre line. It starts at the threshold and extends 60 m beyond the approach end of the system, and is 120 m wide.

11.3.2 No objects are permitted to exist within the boundaries of the light plane which are higher than the light plane except as designated herein. All roads and highways are considered as obstacles extending 4.8 m above the crown of the road, except aerodrome service roads where all vehicular traffic is under control of the aerodrome authorities and coordinated with the aerodrome traffic control tower. Railroads, regardless of the amount of traffic, are considered as obstacles extending 5.4 m above the top of the rails.
Figure A-4. Flight path envelopes to be used for lighting design for category I, II and III operations
Figure A-5. Simple approach lighting systems
Figure A-7  Vertical installation tolerances
11.3.3 It is recognized that some components of electronic landing aids systems, such as reflectors, antennas, monitors, etc., must be installed above the light plane. Every effort should be made to relocate such components outside the boundaries of the light plane. In the case of reflectors and monitors, this can be done in many instances.

11.3.4 Where an ILS localizer is installed within the light plane boundaries, it is recognized that the localizer, or screen if used, must extend above the light plane. In such cases the height of these structures should be held to a minimum and they should be located as far from the threshold as possible. In general the rule regarding permissible heights is 15 cm for each 30 m the structure is located from the threshold. As an example, if the localizer is located 300 m from the threshold, the screen will be permitted to extend above the plane of the approach lighting system by $10 \times 15 = 150$ cm maximum, but preferably should be kept as low as possible consistent with proper operation of the ILS.

11.3.5 In locating an MLS azimuth antenna the guidance contained in Annex 10, Volume I, Attachment G to Part I should be followed. This material, which also provides guidance on collocating an MLS azimuth antenna with an ILS localizer antenna, suggests that the MLS azimuth antenna may be sited within the light plane boundaries where it is not possible or practical to locate it beyond the outer end of the approach lighting for the opposite direction of approach. If the MLS azimuth antenna is located on the extended centre line of the runway, it should be as far as possible from the closest light position to the MLS azimuth antenna in the direction of the runway end. Furthermore, the MLS azimuth antenna phase centre should be at least 0.3 m above the light centre of the light position closest to the MLS azimuth antenna in the direction of the runway end. (This could be relaxed to 0.15 m if the site is otherwise free of significant multipath problems.) Compliance with this requirement, which is intended to ensure that the MLS signal quality is not affected by the approach lighting system, could result in the partial obstruction of the lighting system by the MLS azimuth antenna. To ensure that the resulting obstruction does not degrade visual guidance beyond an acceptable level, the MLS azimuth antenna should not be located closer to the runway end than 300 m and the preferred location is 25 m beyond the 300 m crossbar (this would place the antenna 5 m behind the light position 330 m from the runway end). Where an MLS azimuth antenna is so located, a central part of the 300 m crossbar of the approach lighting system would alone be partially obstructed. Nevertheless, it is important to ensure that the unobstructed lights of the crossbar remain serviceable all the time.

11.3.6 Objects existing within the boundaries of the light plane, requiring the light plane to be raised in order to meet the criteria contained herein, should be removed, lowered or relocated where this can be accomplished more economically than raising the light plane.

11.3.7 In some instances objects may exist which cannot be removed, lowered or relocated economically. These objects may be located so close to the threshold that they cannot be cleared by the 2 per cent slope. Where such conditions exist and no alternative is possible, the 2 per cent slope may be exceeded or a “stair step” resorted to in order to keep the approach lights above the objects. Such “step” or increased gradients should be resorted to only when it is impracticable to follow standard slope criteria, and they should be held to the absolute minimum. Under this criterion no negative slope is permitted in the outermost portion of the system.

11.4 Consideration of the effects of reduced lengths

11.4.1 The need for an adequate approach lighting system to support precision approaches where the pilot is required to acquire visual references prior to landing, cannot be stressed too strongly. The safety and regularity of such operations is dependent on this visual acquisition. The height above runway threshold at which the pilot decides there are sufficient visual cues to continue the precision approach and land will vary, depending on the type of approach being conducted and other factors such as meteorological conditions, ground and airborne equipment, etc. The required length of approach lighting system which will support all the variations of such approaches is 900 m, and this shall always be provided whenever possible.

11.4.2 However, there are some runway locations where it is impossible to provide the 900 m length of approach lighting system to support precision approaches.

11.4.3 In such cases, every effort should be made to provide as much approach lighting system as possible. The appropriate authority may impose restrictions on operations to runways equipped with reduced lengths of lighting. There are many factors which determine at what height the pilot must have decided to continue the approach to land or execute a missed approach. It must be understood that the pilot does not make an instantaneous judgement upon reaching a specified height. The actual decision to continue the approach and landing sequence is an accumulative process which is only concluded at the specified height. Unless lights are available prior to reaching the decision point, the visual assessment process is impaired and the likelihood of missed approaches will increase substantially. There are many operational considerations which must be taken into account by the appropriate authorities in deciding if any restrictions are necessary to any precision approach and these are detailed in Annex 6.

12. Priority of installation of visual approach slope indicator systems

12.1 It has been found impracticable to develop guidance material that will permit a completely objective analysis to be made of which runway on an aerodrome should receive first
priority for the installation of a visual approach slope indicator system. However, factors that must be considered when making such a decision are:

a) frequency of use;

b) seriousness of the hazard;

c) presence of other visual and non-visual aids;

d) type of aeroplanes using the runway; and

e) frequency and type of adverse weather conditions under which the runway will be used.

12.2 With respect to the seriousness of the hazard, the order given in the application specifications for a visual approach slope indicator system, 5.3.5.1 b) to e) of Chapter 5 may be used as a general guide. These may be summarized as:

a) inadequate visual guidance because of:

1) approaches over water or featureless terrain, or absence of sufficient extraneous light in the approach area by night;

2) deceptive surrounding terrain;

b) serious hazard in approach;

c) serious hazard if aeroplanes undershoot or overrun; and

d) unusual turbulence.

12.3 The presence of other visual or non-visual aids is a very important factor. Runways equipped with ILS or MLS would generally receive the lowest priority for a visual approach slope indicator system installation. It must be remembered, though, that visual approach slope indicator systems are visual approach aids in their own right and can supplement electronic aids. When serious hazards exist and/or a substantial number of aeroplanes not equipped for ILS or MLS use a runway, priority might be given to installing a visual approach slope indicator on this runway.

12.4 Priority should be given to runways used by turbojet aeroplanes.

13. Lighting of unserviceable areas

13.1 Where a temporarily unserviceable area exists, it may be marked with fixed-red lights. These lights should mark the most potentially dangerous extremities of the area. A minimum of four such lights should be used, except where the area is triangular in shape where a minimum of three lights may be employed. The number of lights should be increased when the area is large or of unusual configuration. At least one light should be installed for each 7.5 m of peripheral distance of the area. If the lights are directional, they should be orientated so that as far as possible their beams are aligned in the direction from which aircraft or vehicles will approach. Where aircraft or vehicles will normally approach from several directions, consideration should be given to adding extra lights or using omnidirectional lights to show the area from these directions. Uns usable area lights should be frangible. Their height should be sufficiently low to preserve clearance for propellers and for engine pods of jet aircraft.

14. Intensity control of approach and runway lights

14.1 The conspicuity of a light depends on the impression received of contrast between the light and its background. If a light is to be useful to a pilot by day when on approach, it must have an intensity of at least 2,000 or 3,000 cd, and in the case of approach lights an intensity of the order of 20,000 cd is desirable. In conditions of very bright daylight fog it may not be possible to provide lights of sufficient intensity to be effective. On the other hand, in clear weather on a dark night, an intensity of the order of 100 cd for approach lights and 50 cd for the runway edge lights may be found suitable. Even then, owing to the closer range at which they are viewed, pilots have sometimes complained that the runway edge lights seemed unduly bright.

14.2 In fog the amount of light scattered is high. At night this scattered light increases the brightness of the fog over the approach area and runway to the extent that little increase in the visual range of the lights can be obtained by increasing their intensity beyond 2,000 or 3,000 cd. In an endeavour to increase the range at which lights would first be sighted at night, their intensity must not be raised to an extent that a pilot might find excessively dazzling at diminished range.

14.3 From the foregoing will be evident the importance of adjusting the intensity of the lights of an aerodrome lighting system according to the prevailing conditions, so as to obtain the best results without excessive dazzle that would disconcert the pilot. The appropriate intensity setting on any particular occasion will depend both on the conditions of background brightness and the visibility. Detailed guidance material on selecting intensity setting for different conditions is given in the Aerodrome Design Manual, Part 4.

15. Signal area

A signal area need be provided only when it is intended to use visual ground signals to communicate with aircraft in flight. Such signals may be needed when the aerodrome does not have an aerodrome control tower or an aerodrome flight
information service unit, or when the aerodrome is used by aeroplanes not equipped with radio. Visual ground signals may also be useful in the case of failure of two-way radio communication with aircraft. It should be recognized, however, that the type of information which may be conveyed by visual ground signals should normally be available in AIPs or NOTAM. The potential need for visual ground signals should therefore be evaluated before deciding to provide a signal area.

16. Rescue and fire fighting services

16.1 Administration

16.1.1 The rescue and fire fighting service at an aerodrome should be under the administrative control of the aerodrome management, which should also be responsible for ensuring that the service provided is organized, equipped, staffed, trained and operated in such a manner as to fulfil its proper functions.

16.1.2 In drawing up the detailed plan for the conduct of search and rescue operations in accordance with 4.2.1 of Annex 12, the aerodrome management should co-ordinate its plans with the relevant rescue co-ordination centres to ensure that the respective limits of their responsibilities for an aircraft accident within the vicinity of an aerodrome are clearly delineated.

16.1.3 Co-ordination between the rescue and fire fighting service at an aerodrome and public protective agencies, such as local fire brigade, police force, coast guard and hospitals, should be achieved by prior agreement for assistance in dealing with an aircraft accident.

16.1.4 A grid map of the aerodrome and its immediate vicinity should be provided for the use of the aerodrome services concerned. Information concerning topography, access roads and location of water supplies should be indicated. This map should be conspicuously posted in the control tower and fire station, and available on the rescue and fire fighting vehicles and such other supporting vehicles required to respond to an aircraft accident or incident. Copies should also be distributed to public protective agencies as desirable.

16.1.5 Co-ordinated instructions should be drawn up detailing the responsibilities of all concerned and the action to be taken in dealing with emergencies. The appropriate authority should ensure that such instructions are promulgated and observed.

16.2 Training

The training curriculum should include initial and recurrent instruction in at least the following areas:

- airport familiarization;
- aircraft familiarization;
- rescue and fire fighting personnel safety;
- emergency communications systems on the aerodrome, including aircraft fire related alarms;
- use of the fire hoses, nozzles, turrets and other appliances required for compliance with Chapter 9, 9.2;
- application of the types of extinguishing agents required for compliance with Chapter 9, 9.2;
- emergency aircraft evacuation assistance;
- fire fighting operations;
- adaptation and use of structural rescue and fire fighting equipment for aircraft rescue and fire fighting;
- dangerous goods;
- familiarization with fire fighters’ duties under the aerodrome emergency plan; and
- protective clothing and respiratory protection.

16.3 Level of protection to be provided

16.3.1 In accordance with Chapter 9, 9.2 aerodromes should be categorized for rescue and fire fighting purposes and the level of protection provided should be appropriate to the aerodrome category.

16.3.2 However, Chapter 9, 9.2.2 permits a lower level of protection to be provided for a limited period where the number of movements of the aeroplanes in the highest category normally using the aerodrome is less than 700 in the busiest consecutive three months. It is important to note that the concession included in 9.2.2 a) is applicable only where there is a wide range of difference between the dimensions of the aeroplanes included in reaching 700 movements.

16.4 Rescue equipment for difficult environments

16.4.1 Suitable rescue equipment and services should be available at an aerodrome where the area to be covered by the service includes water, swampy areas or other difficult environment that cannot be fully served by conventional wheeled vehicles. This is particularly important where a significant portion of approach/departure operations takes place over these areas.

16.4.2 The rescue equipment should be carried on boats or other vehicles such as helicopters and amphibious or air
cushion vehicles, capable of operating in the area concerned. The vehicles should be so located that they can be brought into action quickly to respond to the areas covered by the service.

16.4.3 At an aerodrome bordering the water, the boats or other vehicles should preferably be located on the aerodrome, and convenient launching or docking sites provided. If these vehicles are located off the aerodrome, they should preferably be under the control of the aerodrome rescue and fire fighting service or, if this is not practicable, under the control of another competent public or private organization working in close co-ordination with the aerodrome rescue and fire fighting service (such as police, military services, harbour patrol or coast guard).

16.4.4 Boats or other vehicles should have as high a speed as practicable so as to reach an accident site in minimum time. To reduce the possibility of injury during rescue operations, water jet-driven boats are preferred to water propeller-driven boats unless the propellers of the latter boats are ducted. Should the water areas to be covered by the service be frozen for a significant period of the year, the equipment should be selected accordingly. Vehicles used in this service should be equipped with life rafts and life preservers related to the requirements of the larger aircraft normally using the aerodrome, with two-way radio communication, and with floodlights for night operations. If aircraft operations during periods of low visibility are expected, it may be necessary to provide guidance for the responding emergency vehicles.

16.4.5 The personnel designated to operate the equipment should be adequately trained and drilled for rescue services in the appropriate environment.

16.5 Facilities

16.5.1 The provision of special telephone, two-way radio communication and general alarm systems for the rescue and fire fighting service is desirable to ensure the dependable transmission of essential emergency and routine information. Consistent with the individual requirements of each aerodrome, these facilities serve the following purposes:

a) direct communication between the activating authority and the aerodrome fire station in order to ensure the prompt alerting and dispatch of rescue and fire fighting vehicles and personnel in the event of an aircraft accident or incident;

b) emergency signals to ensure the immediate summoning of designated personnel not on standby duty;

c) as necessary, summoning essential related services on or off the aerodrome; and

d) maintaining communication by means of two-way radio with the rescue and fire fighting vehicles in attendance at an aircraft accident or incident.

16.5.2 The availability of ambulance and medical facilities for the removal and after-care of casualties arising from an aircraft accident should receive the careful consideration of the appropriate authority and should form part of the over-all emergency plan established to deal with such emergencies.

17. Operators of vehicles

17.1 The authorities responsible for the operation of vehicles on the movement area should ensure that the operators are properly qualified. This may include, as appropriate to the driver’s function, knowledge of:

a) the geography of the aerodrome;

b) aerodrome signs, markings and lights;

c) radiotelephone operating procedures;

d) terms and phrases used in aerodrome control including the ICAO spelling alphabet;

e) rules of air traffic services as they relate to ground operations;

f) airport rules and procedures; and

g) specialist functions as required, for example, in rescue and fire fighting.

17.2 The operator should be able to demonstrate competency, as appropriate, in:

a) the operation or use of vehicle transmit/receive equipment;

b) understanding and complying with air traffic control and local procedures;

c) vehicle navigation on the aerodrome; and

d) special skills required for the particular function.

In addition, as required for any specialist function, the operator should be the holder of a State driver’s licence, a State radio operator’s licence or other licences.

17.3 The above should be applied as is appropriate to the function to be performed by the operator and it is not necessary that all operators be trained to the same level, for example, operators whose functions are restricted to the apron.

17.4 If special procedures apply for operations in low visibility conditions, it is desirable to verify an operator’s knowledge of the procedures through periodic checks.
18. The ACN-PCN method of reporting pavement strength

18.1 Overload operations

18.1.1 Overloading of pavements can result either from loads too large, or from a substantially increased application rate, or both. Loads larger than the defined (design or evaluation) load shorten the design life, whilst smaller loads extend it. With the exception of massive overloading, pavements in their structural behaviour are not subject to a particular limiting load above which they suddenly or catastrophically fail. Behaviour is such that a pavement can sustain a definable load for an expected number of repetitions during its design life. As a result, occasional minor overloading is acceptable, when expedient, with only limited loss in pavement life expectancy and relatively small acceleration of pavement deterioration. For those operations in which magnitude of overload and/or the frequency of use do not justify a detailed analysis, the following criteria are suggested:

a) for flexible pavements, occasional movements by aircraft with ACN not exceeding 10 per cent above the reported PCN should not adversely affect the pavement;

b) for rigid or composite pavements, in which a rigid pavement layer provides a primary element of the structure, occasional movements by aircraft with ACN not exceeding 5 per cent above the reported PCN should not adversely affect the pavement;

c) if the pavement structure is unknown, the 5 per cent limitation should apply; and

d) the annual number of overload movements should not exceed approximately 5 per cent of the total annual aircraft movements.

18.1.2 Such overload movements should not normally be permitted on pavements exhibiting signs of distress or failure. Furthermore, overloading should be avoided during any periods of thaw following frost penetration, or when the strength of the pavement or its subgrade could be weakened by water. Where overload operations are conducted, the appropriate authority should review the relevant pavement condition regularly, and should also review the criteria for overload operations periodically since excessive repetition of overloads can cause severe shortening of pavement life or require major rehabilitation of pavement.

18.2 ACNs for several aircraft types

For convenience, several aircraft types currently in use have been evaluated on rigid and flexible pavements founded on the four subgrade strength categories in Chapter 2, 2.6.6 b) and the results tabulated in the Aerodrome Design Manual, Part 3.
ATTACHMENT B. OBSTACLE LIMITATION SURFACES

Figure B-1

Note: The figure shows the obstacle limitation surfaces as per ICAO Annex 14. It illustrates the area surrounding an airport runway and a non-instrument runway. Both are taken off runway.
# LIMITED INDEX OF SIGNIFICANT SUBJECTS INCLUDED IN ANNEX 14, VOLUME I

- **AERODROME OPERATION***
  - apron management service 9.6
  - bird hazard reduction 9.5
  - denoting closed areas 7.1
  - denoting unserviceable areas 7.4
  - disabled aircraft removal 9.3
  - emergency planning 9.1
  - ground servicing of aircraft 9.7
  - light intensity control A-14
  - lighting of unserviceable areas A-13
  - maintenance 9.4
  - marking of vehicles or mobile objects 6.1.6; 6.2.2; 6.2.14
  - measuring runway braking action/friction A-6; A-7
  - mobile obstacles on runway strips 3.3.7
  - monitoring visual aids 8.3
  - overload operations A-18.1
  - reporting aerodrome data 2
  - rescue and fire fighting 9.2; A-16
  - secondary power supply 8.1

- **APRON**
  - clearance of debris 9.4.3
  - clearance of snow, ice, etc. 9.4.11; 9.4.12
  - definition 1.1
  - physical characteristics 3.12
  - isolated aircraft parking position 3.13
  - lighting 5.3.21
  - reporting requirements 2.5.1 d)
  - safety lines 5.2.13

- **APRON MANAGEMENT SERVICE**
  - definition 1.1
  - provision 9.6

- **CLEARWAY**
  - accountability as runway length 3.1.8
  - definition 1.1
  - frangibility 8.7.1 b); 8.7.2 c)
  - general A-2
  - physical characteristics 3.5
  - reporting requirements 2.5.1 f)

- **DE-ICING/ANTI-ICING FACILITY**
  - definition 1.1
  - lighting 5.3.19
  - location 3.14.2
  - marking 5.2.10.2

- **DECLARED DISTANCES**
  - calculation A-3
  - definition 1.1
  - reporting requirements 2.8

- **DISABLED AIRCRAFT REMOVAL**
  - capability 9.3
  - reporting requirements 2.10

- **DISPLACED THRESHOLD**
  - definition 1.1
  - lights 5.3.10.1; 5.3.10.3
  - location A-10.2
  - marking 5.2.4.9; 5.2.4.10

- **FRANGIBILITY**
  - definition of frangible object 1.1
  - elevated approach lights 5.3.1.3.; 5.3.1.4
  - markers 5.5.1
  - objects on operational areas 8.7
  - objects on runway strips 3.3.7
  - other elevated lights 5.3.1.6
  - PAPI and APAPI 5.3.5.27
  - signs 5.4.1.3
  - T-VASIS and AT-VASIS 5.3.5.16

- **GRADING**
  - radio altimeter operating area 3.7.4
  - runway end safety areas 3.4.5
  - runway strips 3.3.8-3.3.11
  - strip for precision approach runways A-8.3
  - taxiway strips 3.10.4

- **HELIPORT**
  - definition 1.1
  - specifications See Annex 14, Volume II

---

*These specifications which relate to the daily operation of an aerodrome as compared with those which relate to its design or facilities to be provided.
HOLDING BAY

definition 1.1
physical characteristics 3.11

INTERMEDIATE HOLDING POSITION

definition 1.1
lighting 5.3.18
location 3.11.4
marking 5.2.10
signs 5.4.3.8A

LIGHTING

approach lighting systems 5.3.4; A-11; Appendix 2
colour specifications Appendix 1
definitions for lights, etc. 1.1
electrical systems 8.2
intensity control 5.3.1.10; 5.3.1.11; A-14
lights 5.3
lighting of unserviceable areas A-13
maintenance 9.4
monitoring 8.3
obstacle lighting 6.3; Appendix 6
photometric characteristics Appendix 2
priority of installation of visual approach slope
indicator systems A-12
reporting requirements 2.9.2 h); 2.12
secondary power supply 8.1
security lighting 8.5

MAINTENANCE

clearance of debris 9.4.3; 9.4.9
clearance of snow, ice, etc. 9.4.10-9.4.13
genral 9.4.1
pavement overlays 9.4.16-9.4.19
runway evenness 9.4.4; A-5.4
visual aids 9.4.20-9.4.32

MARKER

definition 1.1
marker aids 5.5

MARKING

colour specifications 5.2; Appendix 1
definition 1.1
marking of objects 6.2
surface marking patterns 5.2

MONITORING

condition of the movement area and related
facilities 2.9.1-2.9.3
visual aids 8.3

NON-INSTRUMENT RUNWAY

approach lighting system 5.3.4.1-5.3.4.9
definition 1.1
holding bays 3.11.6
obstacle limitation requirements 4.2.1-4.2.6
runway-holding position marking 5.2.9.2
secondary power supply Table 8-1
threshold lights 5.3.10.1; 5.3.10.4 a)

NON-PRECISION APPROACH RUNWAY

approach lighting system 5.3.4.1-5.3.4.9
definition 1.1
holding bays 3.11.6
obstacle limitation requirements 4.2.7-4.2.12
runway threshold identification lights 5.3.8
runway-holding position marking 5.2.9.2
secondary power supply Table 8-1
threshold lights 5.3.10.1; 5.3.10.4 a)

OBSTACLE/OBJECT

clearance of obstacles A-11.3
definition of obstacle and obstacle free zone 1.1
lighting 6.3; Appendix 6
limitation requirements 4.2
limitation surfaces 4.1
marking 6.2
objects to be marked and/or lighted 6.1
on clearways 3.5.6
on runway end safety areas 3.4.6
on runway strips 3.3.6; 3.3.7
on taxiway strips 3.10.3; 8.7
other objects 4.4
outside the obstacle limitation surfaces 4.3
protection surface 5.3.5.41-5.3.5.45
reporting of obstacles and obstacle free zone 2.5
secondary power supply 8.1

PAVEMENT STRENGTH

ACNs for aircraft A-18.2
aprons 3.12.3
overload operations A-18.1
reporting requirements 2.6
runways 3.1.20
shoulders A-8.1
stopways 3.6.3; A-2.9
taxiways 3.8.12

PRECISION APPROACH RUNWAY CATEGORY I

approach lighting system 5.3.4.10-5.3.4.21
centre line lights 5.3.12.2; 5.3.12.5
definition 1.1
flight path envelope Figure A-4
frangibility 8.7
holding bays 3.11.6-3.11.9
holding position signs 5.4.2.3; 5.4.2.4; 5.4.2.5; 5.4.2.7; 5.4.2.8;
5.4.2.10; 5.4.2.12; 5.4.2.13; 5.4.2.15; 5.4.2.16
Index

maintenance of visual aids 9.4.20; 9.4.29
objects on strips 3.3.7
obstacle limitation requirements 4.2.13; 4.2.14; 4.2.16-4.2.21
runway light characteristics Appendix 2
runway-holding position marking 5.2.9.3
secondary power supply Table 8-1
threshold lights 5.3.10.4 b)

PRECISION APPROACH RUNWAYS
CATEGORIES II AND III

approach lighting system 5.3.4.22-5.3.4.39
definition 1.1
flight path envelopes Figure A-4
frangibility 8.7
holding bays 3.11.6-3.11.9
holding position signs 5.4.2.3; 5.4.2.4; 5.4.2.5; 5.4.2.7; 5.4.2.8; 5.4.2.10; 5.4.2.12; 5.4.2.13; 5.4.2.15; 5.4.2.16
maintenance of visual aids 9.4.20-9.4.26
objects on strips 3.3.7
obstacle limitation requirements 4.2.15-4.2.21
runway centre line lights 5.3.12.1; 5.3.12.5
runway end lights 5.3.11.3
runway light characteristics Appendix 2
runway-holding position marking 5.2.9.3
secondary power supply Table 8-1
stop bars 5.3.17
taxiway centre line lights 5.3.15
taxiway light characteristics Appendix 2
threshold lights 5.3.10.4 c)
touchdown zone lights 5.3.13.1

RESCUE AND FIRE FIGHTING

communication and alerting system 9.2.27; 9.2.28
emergency access roads 9.2.22-9.2.24
extinguishing agents 9.2.7-9.2.17
fire stations 9.2.25; 9.2.26
general 9.2 (Introductory Note)
level of protection 9.2.2-9.2.6; A-16.3
personnel 9.2.31; 9.2.32
reporting requirements 2.11
rescue equipment 9.2.18; 9.2.29
response time 9.2.19-9.2.21
vehicles 9.2.29

RUNWAY

clearance of debris 9.4.3
clearance of snow, ice, etc. 9.4.10; A-6
closed runway marking 7.1
definition 1.1
lights 5.3.7-5.3.13; Appendix 2
markers 5.5.2; 5.5.4
marking 5.2.2-5.2.7
number, siting and orientation A-1
pavement overlays 9.4.16-9.4.19
physical characteristics 3.1
reporting requirements 2.3.2; 2.5.1 a); 2.8; 2.9.2; 2.9.4-2.9.11; A-6; A-7
runway surface evenness A-5
shoulders 3.2
slopes 3.1.12-3.1.19; A-4
strips 2.5.1 b); 3.3; 8.7.1 a); 8.7.4; 8.7.6

RUNWAY END SAFETY AREA

definition 1.1
frangibility 8.7.1 a); 8.7.2 b)
general A-9
physical characteristics 3.4
reporting requirements 2.5.1 b)

RUNWAY-HOLDING POSITION

definition 1.1
location 3.11.2; 3.11.3; 3.11.9
marking 5.2.9
runway guard lights 5.3.19
signs 5.4.2.2-5.4.2.5; 5.4.2.7; 5.4.2.8; 5.4.2.10; 5.4.2.12; 5.4.2.13; 5.4.2.15; 5.4.2.16
stop bars 5.3.17

RUNWAY MEANT FOR TAKE-OFF

climb surface 4.1.25-4.1.29
frangibility 8.7
maintenance of visual aids 9.4.20; 9.4.30; 9.4.31
obstacle limitation requirements 4.2.22-4.2.27
runway lighting 5.3.9.2; 5.3.12.3; 5.3.12.4
secondary power supply Table 8-1
taxiway lighting 5.3.15; 5.3.16

RUNWAY SURFACE FRICTION CHARACTERISTICS

maintenance 9.4.3-9.4.7; 9.4.9; 9.4.12
reporting requirements 2.9
runway design 3.1.22
runway surface friction 2.9.6; 2.9.9
snow- and ice-covered paved surfaces – general A-6
wet runways – general A-7

SECURITY

aerodrome emergency planning 9.1.2 (Note)
airport design 8.6
fencing 8.4
isolated aircraft parking position 3.13
lighting 8.5

STOPWAY

accountability as runway length 3.1.8
definition 1.1
general A-2
lights 5.3.14; Appendix 2
markers 5.5.3
physical characteristics 3.6
reporting requirements 2.5.1. b)
Annex 14 — Aerodromes

Volume I

TAXIWAY

closed taxiway marking 7.1
definition 1.1
lights 5.3.15; 5.3.16; Appendix 2
markers 5.5.5; 5.5.6; 5.5.7
marking 5.2.8; 5.2.10; 7.2
physical characteristics 3.8
removal of contaminants 9.4.9; 9.4.11; 9.4.13
reporting requirements 2.5.1 c)

shoulders 3.9
strips 3.10; 8.7.1 a); 8.7.4

VISUAL APPROACH SLOPE INDICATOR SYSTEMS

characteristics 5.3.5
priority of installation A-12
reporting requirements 2.12
secondary power supply 8.1

— END —
SUPPLEMENT TO

ANNEX 14, VOLUME I —
AERODROME DESIGN AND OPERATIONS

(Third Edition)

1. The attached Supplement supersedes all previous Supplements to Annex 14, Volume I, and includes differences notified by Contracting States up to 11 August 2000.

2. This Supplement should be inserted at the end of Annex 14, Volume I, Third Edition. Additional differences and revised comments received from Contracting States will be issued at intervals as amendments to this Supplement.
SUPPLEMENT TO ANNEX 14, VOLUME I — THIRD EDITION

AERODROME DESIGN AND OPERATIONS

Differences between the national regulations and practices of States and the corresponding International Standards contained in Annex 14, Volume I, as notified to ICAO in accordance with Article 38 of the Convention on International Civil Aviation and the Council’s resolution of 21 November 1950.
**RECORD OF AMENDMENTS TO SUPPLEMENT**

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1. Contracting States which have notified ICAO of differences

The Contracting States listed below have notified ICAO of differences which exist between their national regulations and practices and the International Standards and Recommended Practices of Annex 14, Volume I, Third Edition, or have commented on implementation.

The page numbers shown for each State and the dates of publication of those pages correspond to the actual pages in this Supplement.

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2. Contracting States which have notified ICAO that no differences exist

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CHAPTER 1

1.1 The Austrian regulations distinguish between aerodrome elevation according to ICAO Annex 14 for operational purposes and aerodrome reference elevation for determining obstruction restrictions.

The aerodrome reference elevation is the elevation of the aerodrome reference point above mean sea level in terms of the nearest metre.

CHAPTER 3

3.4.2 Although the regulations of Austria provide a runway end safety area of at least 300 m where the code number is 3 or 4, at some aerodromes the surrounding terrain does not allow to provide the full length as stated in 3.4.2.

CHAPTER 4

4.1.25 The term “take-off climb surface” is not used and is therefore not defined.

4.2 The specifications concerning approach surfaces are applicable for take-off areas and take-off surfaces as well.

CHAPTER 5

5.2.1.5 Additional markings are on aprons, i.e. white for service roads (passenger and service vehicles) and red for parking areas of service vehicles and ramp equipment, orange for limit of apron control competence.

5.3.17.10 Exemption may be provided in cases where switched-off taxiway centre lights lead to deterioration of taxi guidance.

5.3.24 For the present, road-holding position lights will not be provided. Reason: on the way to runways most vehicles use taxiways with ATC clearances.

5.4.3.6 The indication “TORA (RWY)” is added to intersection take-off signs.

CHAPTER 6

6.2.12 Flags used to mark obstructions are 0.5 m square and are coloured in either red or yellow.

Comment on implementation:
Chapter 6 will be adopted in accordance with Annex 14 as soon as possible.
CHAPTER 1

1.3 Code letter F is not applicable to Brussels/National Airport.

Remark: The present general infrastructure is not suited to code letter F.

CHAPTER 3

3.1.9 Runways 07L/25R and 07R/25L are 45 m wide. Runway 02/20 is 50 m wide. The runway width required for code letter F is 60 m.

3.2.3 The width of shoulders for runways 07L/25R, 07R/25L and 02/20 is 60 m and not 75 m, as required by code letter F.

3.3.7 Obstacle clearance limits are 60 m, not 77.5 m as required for code letter F.

3.8.3 The clearance distance between the outer main wheel of the aeroplane and the edge of the taxiway of 4.5 m was planned for code letter E aircraft and not code letter F.

3.8.4 Taxiways serving runway 02/20 are 20 m wide and not 25 m, as required by code letter F.

3.8.7 The separation distances between: a runway centre line and a taxiway centre line, a taxiway centre line and another taxiway centre line, a taxiway centre line and an object, and a taxilane and an object, throughout the airport, meet code letter E specifications and not those of code F. Taxilane to taxilane distances are 40 m and not 42.5 m as recommended by Annex 14, Volume I, for code letter E.

Remark: The design of the taxiways serving runway 25R/07L allows no further widening. The taxiways serving runways 07R/25L and 02/20 could, if necessary, be adapted through considerable investment which is not planned at this time.

3.9 Taxiway shoulders meet code letter E specifications and not those of code letter F, with, however, 50 m clearance instead of 44 m for code letter E and 60 m for code letter F.

3.11.6 Cat. I stop bars are located at 90 m (code letter E) and they cannot be used by code letter F aircraft (minimum distance 107.5 m). However, the existing Cat. II stop bars meet this requirement.

CHAPTER 5

5.3.4.10 The precision approach category I lighting system for runway 20 extends over a distance of 540 m rather than 900 m.

Remark: The length will be increased to 630 m in September 2000.

5.3.4.22 The precision approach category II lighting system for runway 25R extends over a distance of 600 m rather than 900 m.

5.3.4.23

5.3.4.24 The precision approach category II lighting system for runway 02 has two fewer centre line barrettes because of the presence of a building at 330 m and a railway line at 570 m.

*Recommended Practice
5.3.17.14 The present lighting circuit does not allow selective switching of stop bars. In addition, when a stop bar is illuminated, the taxiway centre line lights installed beyond the stop bar are not extinguished for a distance of at least 90 m. The present level of switching does not allow this.

Comment on implementation: Compliance is planned progressively, and it should be achieved completely by the end of 2001.

5.4 The present mandatory instruction and information signs are no longer compliant with respect to colour, location and information.

Comment on implementation: Compliance is planned for 1 January 2001.

CHAPTER 6

6.3.4 The low intensity obstacle lights, Type C (mobile obstacle), displayed on vehicles are presently orange/blue instead of yellow/blue.

Comment on implementation: They will be replaced in the course of 2000.

6.3.5 The low intensity obstacle lights, Type D, displayed on follow-me vehicles are presently orange instead of yellow.

Comment on implementation: They will be replaced in the course of 2000.

CHAPTER 9

9.1.6 The present emergency plan does not contain Human Factors information. It is to be provided henceforth in the emergency operations.

Comment on implementation: Must be developed in the course of 2000.

9.2.31 The present personnel training programme does not, at this time, contain information on Human Factors performance.

Comment on implementation: Must be developed in the course of 2000.
CHAPTER 2

2.6 Canada prepares and distributes pavement evaluation charts for specific aircraft relative to specific airports upon receipt of a request from an operator. Information required from the operator is: the type of aircraft, the maximum gross weight to be used and the main wheel tire pressure.

2.7 Canada does not provide pre-flight altimeter check locations.

CHAPTER 3

3.1.12* Canada computes runway longitudinal slopes by dividing the difference between the maximum and minimum runway end elevation by the runway length.

3.4.1 Canada does not provide runway end safety areas but does provide a 60 m graded strip beyond the runway end.

CHAPTER 4

4.1.1 Canada does not establish a conical surface per se but takes into consideration areas outside the horizontal surface for obstructions that require marking or removal in order to protect aircraft in the vicinity of an aerodrome.

4.1.22 Table 4-1 Comment on implementation: Length of inner edge of balked landing surface. Canada cannot always meet this requirement during the winter months because of snow accumulation.

CHAPTER 5

5.1.1.4* Canada does not require that the location of wind direction indicators be marked by a circular band 15 m in diameter but does require that their location be indicated in appropriate aeronautical publications.

5.1.2 Landing direction indicators are not used at Canadian aerodromes.

5.1.4 Signal areas are not used at Canadian aerodromes.

5.2.2.4 Runways within the Canadian Northern Domestic Airspace are designated with reference to true azimuth rather than magnetic azimuth as used in other areas.

5.2.7.2* Canada does not provide runway side stripe markings on precision approach runways except where there is a lack of contrast between runway edges and the shoulder or surrounding terrain.

5.2.8.2* Canada does not provide taxiway centre line markings on runways where the code number is 1 or 2.

5.2.8.7 Canada permits taxiway centre line markings on straight portions of taxiways to be marked with broken longitudinal stripes 3 m in length with 3 m gaps between stripes.

5.2.9 Canada does not provide a specific taxiway holding position marking for category II.

5.2.10 Canada does not provide taxiway-to-taxiway intersection markings.

*Recommended Practice
5.2.11 VOR aerodrome check-point markings are not used at Canadian aerodromes.

5.3.3 Identification beacons are not used in Canada.

5.3.4.2 Canada does not provide a crossbar in simple approach lighting systems.

5.3.4.15 Canada does not use capacitor discharge lights to supplement high-intensity approach lighting systems.

5.3.5.1 It is not Canada’s policy to provide a visual approach slope indicator system where a runway is served by an electronic precision approach system.

Remark: The Canadian standard for visual approach slope indicator systems consists of four light units in the configuration shown for AVASIS for Figure 5-9(D), Annex 14, Volume I, First Edition.

Figure 5-12 a) Canada provides 9 m wheel clearance for aircraft with eye-to-wheel distance of 13.5 m; 4.5 m wheel clearance for aircraft with eye-to-wheel distance of 7.5 m; and 3 m for aircraft with eye-to-wheel distance of 3 m.

5.3.6 Circling guidance lights are not used in Canada.

5.3.7 Runway lead-in lighting systems are not used in Canada.

5.3.8 Canada refers to this type of lighting as runway identification lights and sites the units 30 m in front of the threshold and 12 m from the edge.

5.3.9.7 a) Canada uses blue lights to delineate this area.

5.3.10.3 In the case of a displaced threshold, Canada requires wing bar lights consisting of bidirectional threshold and runway end lights for both precision and non-precision approach runways.

5.3.14.3 Canada uses blue edge lights in stopways.

5.4.2.9 Canada provides only one HOLD sign for category I runway unless the taxiway is more than 45 m in width.

5.4.5 Canada does not use aerodrome identification signs.

5.5.3.1* Canada does not provide stopway edge markers.

CHAPTER 6

6.2.6* In Canada emergency vehicles may be painted either yellow or red.

CHAPTER 7

7.2.2* Canada does not use taxiway side stripe markings.

*Recommended Practice
CHAPTER 1

1.1 “Aerodrome” means any area of land or water designed, equipped, set apart or commonly used for affording facilities for the landing and taking-off of aircraft and includes:

a) any area or space, whether on the ground, on the roof of a building, or elsewhere, which is designed, equipped or set apart for affording facilities for the landing and taking-off of aircraft capable of descending or climbing vertically; and

b) any such area of land or water or any such area or space, the management of which is vested in the Government or in the Chief Executive,

but does not include any area the use of which for affording facilities for the landing and taking-off of aircraft has been abandoned and has not been resumed.

Comment on implementation: This definition is in line with local legislation. Therefore, withdrawal of the difference is not considered for the time being.

CHAPTER 3

3.8.21* Straight sections at both ends of the bridges on Taxiway W, Taxiway V and Taxiway V4 are not available at the following locations:

Taxiway W to H and vice versa
Taxiway W to V4 and vice versa
Taxiway W to W2 and vice versa
Taxiway V to V4 and vice versa
Taxiway V to H and vice versa

Comment on implementation: This is compensated by the widened bridges (from the required 44 m to 60 m) to overcome possible oversteering. Therefore, withdrawal of the difference is not considered for the time being.

CHAPTER 5

5.2.10 Upon the commissioning of the second runway of the Hong Kong International Airport, taxi-holding position markings at the following locations consist of one solid line and one broken line:

Junction of Taxiway A and Taxiway N
Junction of Taxiway B and Taxiway N
Junction of Taxiway A and Taxiway A4
Junction of Taxiway A and Taxiway A5
Junction of Taxiway A and Taxiway A6
Junction of Taxiway A and Taxiway V
Junction of Taxiway A and Taxiway W
Junction of Taxiway B and Taxiway V
Junction of Taxiway B and Taxiway W
Junction of Taxiway A8 and Taxiway A
Junction of Taxiway H and Taxiway V
Junction of Taxiway H and Taxiway W

* Recommended Practice
Comment on implementation: More positive visual information at the critical taxiing intersection is used in order to provide an enhancement of the markings. Therefore, withdrawal of the difference is not considered for the time being.

5.4.3.30 The TORA sign script on TWY A3, A10, J2 and J8 does not show the arrow as specified.

Remark: The specified taxiways intersect the two runways close to the ends of each runway. It is highly unlikely for the pilots to take-off in the wrong direction because if the wrong direction is taken, only a 100 m length of runway would be available for take-off. Therefore, withdrawal of the difference is not considered for the time being.
CHAPTER 3

3.4.1 Runway end safety areas are not provided at all airports in Greenland.

Remark: Difficult due to terrain.
CHAPTER 3

3.4.4 The following text will be substituted for this Standard:

The width of a runway end safety area shall be at least twice that of the associated runway. However, a width of 90 m is considered sufficient in all cases.

CHAPTER 5

5.3.3.3 This Standard will be implemented as a recommendation only.

Comment on implementation: Standards 5.3.3.1 and 5.3.3.2 are considered to be adequate in determining the operational requirement in Finland. Operational need for an aerodrome beacon or an identification beacon, in Finland, is negligible.

5.3.4.1 C A number of precision approach category I runways in Finland are provided with a high intensity simple approach lighting system with an additional crossbar located 600 m from the threshold.

Remark: For high intensity simple approach lighting systems, see the ICAO Manual of All-Weather Operations (Doc 9365), paragraph 6.3.7.

5.3.5.1 c) Comment on implementation: Date of applicability: 1 January 2005.

5.3.18.1 Comment on implementation: This Standard will be implemented in Finland by 1 January 2005.

5.3.20.1 Finland complies with this Standard but with the following addition:

For runways intended to be used for take-offs only, in runway visual range conditions of less than 550 m, runway guard lights are not required if operational procedures exist to limit the number of:

1) aircraft on the manoeuvring area to one at a time; and
2) vehicles on the manoeuvring area to the essential minimum.

5.4.1.7 The following text will be substituted for this Standard:

1) Signs shall be illuminated, either internally or externally, when intended for use:
   a) in runway visual range conditions less than 800 m; or
   b) at night in association with instrument runways; or
   c) at night in association with runways where the code number is 3 or 4.

2) Signs shall be illuminated in accordance with the provisions of Appendix 4 when the runway is intended for landings with runway visual range less than 500 m.

5.4.1.8 The following text will be substituted for this Standard:

Signs shall be retroreflective and/or illuminated when intended for use at night in association with non-instrument runways where the code number is 1 or 2.
CHAPTER 6

6.1.3 Finland complies with this Standard with the following additions:

a) the marking may be omitted in the case of tree tops forming an obstacle;

b) the marking may be omitted in the case of terrain forming an obstacle not extending above an obstacle-free zone;

c) the marking may be omitted in the case of tree tops forming an obstacle not extending above an obstacle-free zone;

d) the marking may be omitted in the case of tree tops forming an obstacle not considered to constitute a hazard to aeroplanes.

6.3.26 Low intensity obstacle lights displayed in vehicles, including follow-me vehicles, differ from those specified in Table 6-3 as to the flash frequency, maximum allowable intensity and vertical beam spread.

Remark: The low intensity obstacle lights used on vehicles (special warning lamps for motor vehicles) are in accordance with the United Nations Agreement Concerning the Adoption of Uniform Conditions of Approval and Reciprocal Recognition of Approval for Motor Vehicle Equipment and Parts, Regulation No. 65, Uniform Provisions Concerning the Approval of Special Warning Lamps for Motor Vehicles.

CHAPTER 7

7.1.4 Finland complies with this Standard with the follow addition:

When the closed marking is used on a temporarily closed runway, it shall be of the form and proportions as detailed in Figure 7-1, illustration a) or b).

CHAPTER 8

8.1.8 For a runway meant for take-off in runway visual range conditions of less than 550 m, a secondary power supply capable of meeting the relevant requirements of Table 8-1 shall be provided.

Table 8-1 In Table 8-1, 550 m is substituted for 800 m.

8.7.5 It is not mandatory for a driver of a radio-equipped vehicle, before entering the apron, to establish radio communication with ATC or the apron management service.

CHAPTER 9

9.1.12 The intervals between full-scale aerodrome emergency exercises will exceed two years.

Remark: A full-scale exercise may cause expenditure of human resources for outside organizations, such as hospitals.

9.2.2 to 9.2.34 Rescue and fire fighting services specified in these Standards and Recommended Practices are required only at those aerodromes having a permanent air traffic service. Military aerodromes open to civil aircraft but not serving scheduled air carrier operations do not meet all of these Standards.
Aerodrome rescue and fire fighting categories 1 and 2 are not applied in Finland, i.e. the aerodrome category shall be at 3.

9.2.6 Finland complies with this Standard with the following addition:

If, during anticipated periods of reduced activity, the highest category of aeroplane planned to use the aerodrome is 1 or 2, protection shall be provided for these operations if the total annual number of operations at the aerodrome is less than 50 000.

9.2.10 Complementary agents are not required at all airports in Finland.

*Comment on implementation:* Date of applicability: 1 January 2003.

9.4.19 Provision of a temporary threshold marking and early provision of the runway centre line marking are required only when practicable. If a temporary threshold marking is not provided, a temporary threshold shall be indicated with runway threshold identification lights in addition to runway threshold lights and/or wing bar lights. All these shall be high-intensity lights, if applicable.

9.4.20 A light shall be deemed to be unserviceable when the main beam is less than 50 per cent of the value specified in the appropriate figure in Appendix 2.

*General comment on implementation:* The Standards of Amendment 3 will be implemented in Finland by 7 September 2000.
CHAPTER 1

1.1 In France, the runway-holding position is known as the runway stop position. The intermediate holding position is known as the intermediate stop position.

1.3 The aerodrome classification system adopted in France is a functional one, based on the characteristics of the aviation activities for which the aerodrome is intended, namely:

*Category A:* Aerodromes intended for long-haul services normally provided under all circumstances.

*Category B:* Aerodromes intended for medium-haul services normally provided under all circumstances and for some long-haul services provided under the same conditions but which do not involve a long stage on departure from these aerodromes.

*Category C:* Aerodromes intended for:

1. — Short-haul, some medium-haul and even long-haul services which only involve short stages on departure from these aerodromes; and

2. — International tourism.

*Category D:* Aerodromes intended for aeronautical training, recreational flying and tourism or for certain short-haul services.

There is therefore no direct correspondence between the above classification and the aerodrome reference code in Annex 14, Volume I; the following is an approximate table of correspondence.

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<td>1 A to 3 C</td>
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As a result, it is difficult to make a point-by-point notification of differences, in particular regarding the limitation and removal of obstacles; be that as it may, in this latter field, the applicable specifications guarantee “safety conditions at least equivalent to those provided for in the Standards and Recommended Practices of the International Civil Aviation Organization” (Code of Civil Aviation, Article R.241-3).

CHAPTER 2

2.11.4* The French regulations do not include special provisions to express unavailability of rescue and fire fighting service facilities for periods which are expected to be less than 12 hours.

CHAPTER 3

3.1.15* Minimum radii of curvature are 20 000 m for the runways of Category A aerodromes, 15 000 m for Category B and 5 000 m for Categories C and D.

*Recommended Practice
3.4 The application of the new Standard will be studied and it can only be implemented progressively.

*Remark*: There may be obstacles (vegetation) or differences in height after the strip, in the corresponding area in particular on code number 1 and 2 runways. Verification is therefore necessary.

3.7.1* A radio altimeter operating area is only established for Category II or III precision approach runways.

3.7.3* The width of the radio altimeter operating area for Category II or III precision approaches is 60 m (30 m on each side of the centre line).

3.7.4* The characteristics are as follows:

— average longitudinal slope of less than 2 per cent in absolute value
— no local slopes in excess of 5 per cent in absolute value
— no local differences in height of more than one metre

In some cases, when the terrain does not correspond to the average slope characteristics and slope changes, an artificial plane may be used.

3.11 The taxi-holding position is known as the runway-stop position.

3.11.5 The French Administration takes the following rules into account in determining the stop position:

3.11.6

3.11.7* a) *Runways used under VFR conditions*

The distance is determined in relation to the *runway edge* and is at least equal to:

— 30 m for a paved runway less than 1 000 m long, or an unpaved runway; and
— 50 m for a paved runway 1 000 m long or more.

b) *Runways used under IFR conditions*

— With non-precision approach: the distance is 75 m from the runway centre line.
— With precision approach: the distance is at least 150 m from the runway centre line.

However, in Category I, if markings or signs cannot be located 150 m from the runway centre line, a different distance may be adopted. It must not be less than:

— 90 m if the runway is used by Category D and E aircraft.
— 75 m if the runway is used by Category A, B or C aircraft.

**CHAPTER 4**

4.1.9 The inner edge of the approach surface is situated at the same elevation as the extended centre line of the runway at right angles to the inner edge.

4.1.13 to 4.1.16 The transitional surfaces are known as lateral surfaces.

4.1.27 The inner edge of the take-off climb surface is situated at the same elevation as the extended centre line of the runway at right angles to the inner edge.

4.2.1 Some Category D aerodromes do not have a conical surface.

* Recommended Practice

8/9/00
4.2.9 The runway approach surfaces at aerodromes in Categories A, B and C are horizontal above the higher of the following two levels:

— 150 m above the inner edge; and
— 100 m above the terrain.

4.2.17 The runway approach surfaces at aerodromes in Categories A, B and C are horizontal above the higher of the following two levels:

— 150 m above the inner edge; and
— 100 m above the terrain.

CHAPTER 5

5.2.1.5 The taxiway markings are yellow except when a taxiway centre line splits, in which case one of the split centre lines is blue and the other orange.

5.2.3.3 For blacktop runways used in VFR conditions or in IFR conditions (excluding precision approaches), and for concrete runways used solely in VFR conditions, the French Administration authorizes the use of a simplified marking.

5.2.5.2 The aiming point markings are put on all runways 1 500 m in length or longer.

5.2.5.5 For blacktop runways used in VFR conditions or in IFR conditions (excluding precision approaches), and for concrete runways used solely in VFR conditions, the French Administration authorizes the use of a simplified marking.

5.2.7.2* French regulations do not require the systematic provision of runway side stripe markings on precision approach runways.

5.2.10.1* When a precise holding limit has to be defined, an intermediate stop position marking will be used.

5.3.3.12 In France, identification beacons may show flashing-white rather than flashing-green.

5.3.4.1 B The French regulations do not require the regular provision of approach lighting systems for non-precision approach runways.

C The French regulations do not require the regular provision of approach lighting systems for Category I precision approach runways.

D The French regulations do not require the regular provision of approach lighting systems for Category III precision approach runways if they are not also used for Category II precision approaches.

5.3.5.1 a) The French regulations do not require the regular provision of visual approach slope indicators to serve a runway used by turbojet or other aeroplanes with similar approach guidance requirements.

5.3.12.3 Runway centre line lights are mandatory for take-off in low visibility when the runway visual range (RVR) is lower than 250 m for aircraft of Categories A, B and C and 300 m for aircraft of Category D.

*Recommended Practice
5.3.15.1 and 5.3.15.2* Taxiway centre line lights are systematically provided only on taxiways with an RVR of less than 150 m.

5.3.15.3* The French Administration applies this provision only in the case of taxiing on an inactive runway and with an RVR of less than 150 m.

5.3.16.2 The French Administration does not apply this provision.

5.3.17.1 The French Administration only requires the provision of stop bars for Category II and III precision approach runways except on aerodromes with light traffic (Categories II and III) where taxiing in the manoeuvring area is limited to one aircraft at a time.

Remark: In view of the lower RVR limits mentioned, 350 m and 550 m, and the take-off minima authorized by European regulations, this Standard would require the installation of stop bars on most runways. This is unfeasible. This is why only Category II and III runways have been considered.

5.3.17.14 c) The centre line lights will not be visible from the entry side.

5.3.18 The French Administration has no plans to provide taxiway intersection lights.

5.3.20.1 The runway guard lights (Configuration A, Figure 5-23) are installed in France at all the runway-stop positions on Category II and III precision approach runways in the absence of stop bars. Their installation is recommended at aerodromes with high traffic density.

Remark: In view of the RVR limits mentioned, 550 m and 1 200 m, and the take-off minima authorized by European regulations, this Standard would require the installation of runway guard lights on almost all runways. This is unfeasible.

5.4.1.6 The French Administration does not fully apply the provisions of Appendix 4. Sign inscription heights may be lower than the ones mentioned in that Appendix.

5.4.2 The French Administration has established an installation schedule that should end on 31 December 2000.

Remark: Upgrading all French airfields to the Standards requires very considerable investment and work.

5.4.2.8 In France, the signs specified in 5.4.2.8 are installed at least on the left side of the runway.

Remark: Upgrading all French airfields to the Standards requires very considerable investment and work. Consultation is necessary before adopting this Standard.

5.4.2.10 In France, it is not intended to require the installation of the runway-stop position sign (example “B2”) at runway-stop positions.

Remark: It can be understood that there is sometimes a need to use this type of sign in the case of multiple stop positions, but it should not be required systematically. Runway designation signs are already often complemented by a location sign (example “B”).

*Recommended Practice
5.4.2.14 The Cat. II or Cat. III inscription is always placed closest to the taxiway and the runway designation is on the outside.

<table>
<thead>
<tr>
<th>left-side sign</th>
<th>right-side sign</th>
</tr>
</thead>
<tbody>
<tr>
<td>20-02 Cat. III</td>
<td>Cat. III 20-02</td>
</tr>
</tbody>
</table>

5.4.3.3 Information signs also include special signs and frequency reminder signs.

Frequency reminder signs:

When passing from one area of responsibility to another, it may be useful to remind pilots or drivers of vehicles of the frequency associated with the area that they are entering; it is therefore that frequency which is displayed on the sign.

5.4.3.5 Runway vacated signs are only provided for Category II and III precision approach runways.

5.4.3.7* At some complex intersections, signs displaying a combination of different types of information may be used where required.

5.4.3.10* Signs of the same type as the runway-stop position signs (example “B2”) will be installed at the intermediate stop positions at least on the left side of the taxiway.

5.4.3.21 In France, the intersection take-off signs will be installed at least on one side of the taxiway, right or left.

5.4.3.30 In France, the inscriptions on these signs will indicate the value of the distance remaining to the end of the runway (TORA) expressed in metres, but the letter “m” of the unit will not be displayed. The runway designation may also be mentioned.

*Remark: The fact of not specifying the unit “m” does not jeopardize safety since confusion with units in feet would give a shorter distance which would alert the pilot.

CHAPTER 7

7.3.1* The French Administration establishes 100 m as the minimum length beyond which a paved pre-threshold area not suitable for normal use must be marked with a chevron marking.

CHAPTER 8

8.9.7* The French regulations do not provide for the systematic use of surface movement radar for the manoeuvring area.

CHAPTER 9

9.2.2 The French Administration will not be in a position to apply this Standard in January 2000. The French Administration makes a distinction between flights planned with sufficient time in advance and other flights.

*Recommended Practice
9.2.10 The French regulations for aerodrome Categories I and II require the following amounts of extinguishing agents:

<table>
<thead>
<tr>
<th>Category</th>
<th>Amount</th>
</tr>
</thead>
<tbody>
<tr>
<td>Category I</td>
<td>50 kg powder or halons</td>
</tr>
<tr>
<td>Category II</td>
<td>250 kg powder or halons</td>
</tr>
</tbody>
</table>

9.2.34 The French regulations provide for the use of respiratory equipment only at aerodrome Category IV or higher.

9.3 The French Administration does not apply the provisions of this paragraph.
CHAPTER 2

2.1.5 Publishing of the aerodrome reference points in WGS-84 will be an ongoing task for some time after the date of applicability of Amendment 2.

2.1.6

Appendix 5

Table 1 In Germany the description of obstacles differs as follows from what is given in these tables.

Table 2 a) Obstacles in the circling area for non-precision and turning departures and at the aerodrome.

b) Significant obstacles in the precision approach and straight departure area.

Table 2 The WGS-84 geoid undulation at aerodrome elevation position will not be published in Germany.

Remark: This item need not be published because for non-precision approaches the MDH is referred to the threshold position at all German IFR aerodromes.

Table 1 Threshold, runway end: for a few IFR aerodromes these positions cannot be published as from the date of applicability.

Table 5 2.1.5

2.1.6 Remark: Some more time is needed because these positions must be brought into conformity with the declared distances specified in the aerodrome licensing document (a formal act of approval by the National administration).

Comment on implementation:

Annex provision Proposed date of implementation

Appendix 5 It is not intended to comply with this provision.

Table 1

Table 2
CHAPTER 3

3.4.2 A runway end safety area is 40 m.

Comment on implementation: No amendment to the current regulation is intended.

CHAPTER 5

5.3.5.42 The length of the obstacle protection surface is 7 400 m.

Comment on implementation: It is difficult to establish the obstacle protection surface of 15 000 m in length because of mountainous terrain. No amendment to the current regulation is intended.

5.3.14 No provision.

CHAPTER 6

6.3.25 The colour of low intensity obstacle lights displayed on vehicles associated with emergency or security is flashing red.

Date of implementation: March 2005 (provisional).
CHAPTER 3

3.3.2 In the regulations applied in Kyrgyzstan, this segment is reinforced (with pavement) and is 50 m in size with a reduction down to 2/3 of its width at the end of the reinforced segment where the code number is 4.

*Remark:* The implementation of reinforced segments of this size was based on the previous regulatory document “Standards of operational serviceability of the civil aerodromes in the USSR.”
CHAPTER 2

2.1.3 The Netherlands is not making use of CRC to protect the integrity level of critical and essential electronic aeronautical data.

2.1.7 In the Netherlands the geoid undulation at the aerodrome elevation position and the geoid undulation of the threshold will not be published.

2.3.1
2.3.2
2.3.3

2.5.3 In the Netherlands it is not yet considered necessary to determine the geographical coordinates of the taxiway centre line points in terms of the World Geodetic System — 1984 (WGS-84) geodetic reference datum.

2.5.4 In the Netherlands it is not yet considered necessary to determine the geographical coordinates of the aircraft stands in terms of the World Geodetic System – 1984 (WGS-84) geodetic reference datum.

2.6.2 a) For pavements intended for aircraft of apron (ramp) mass greater than 6000 kg, the pavement classification number (PCN) shall be reported.

2.6.5 The behaviour of a pavement may be classified as rigid (R), flexible (F), special-rigid (SR) or special-flexible (SF).

2.6.6 a) When classified as special, the standard procedures for determining the ACN-value of an aircraft as given in the Aerodrome Design Manual, Part 3, are not applicable, and therefore, the reported PCN-value is only of indicative value for the present fleet of aircraft.

2.6.8 For pavements intended for aircraft of apron (ramp) mass equal to or less than 6000 kg and for unpaved airfields in general, the bearing strength shall be reported by the method described in this paragraph.
General Comment: New Zealand’s Civil Aviation Rules (CAR) Part 139: Aerodromes — Certification, Operation and Use, is currently being reviewed. More comprehensive advice regarding New Zealand’s differences to Annex 14, Volume I, will be provided following issuance of CAR Part 139, which is expected late in 2000.

CHAPTER 1

General Table 1-1 New code letter F for aeroplanes with a wing span of 65 m up to but not including 80 m and an outer main gear span of 14 m up to but not including 16 m.

Remark: This requirement will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.

CHAPTER 3

General Provisions related to code letter F aeroplanes.

3.3.2 Length of runway strips.
3.3.7 a) Objects on runway strips.

Remark: These requirements will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.

3.4.1 General
3.4.2 Dimensions of runway end safety area (RESA).
3.4.3*  

Remark: The requirement for RESA is under consideration, and it will be some months before some resolution is finalized.

Comment on implementation: Under consideration.

3.14 New specifications on de/anti-icing facilities.

Remark: These requirements will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.

CHAPTER 4

Table 4-1 Requirements for code letter F aeroplanes.

Remark: This requirement will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.

CHAPTER 5

Revised specifications on runway-holding position marking, intermediate holding position marking, runway centre line lights, taxiway centre line lights, stop bars, intermediate holding position lights and runway guard lights.

* Recommended Practice
Specifications on mandatory instruction markings, marking and lighting of de/anti-icing facilities, alternative approach and runway lighting for precision approach runways where the serviceability levels of lights can be demonstrated and intersection take-off signs.

Remark: These requirements will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.

CHAPTER 6

Table 6-2 Revised specifications on lighting of obstacles, characteristics of obstacle lights and installation setting angles for high-intensity obstacle lights.

Remark: These requirements will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.

CHAPTER 8

Requirements for switch-over times for precision approach runways, Category I, where approaches are over hazardous or precipitous terrain and revised requirements for switch-over times for runways meant for take-off in runway visual range conditions between 550 m and 800 m.

Revised provisions concerning frangibility of non-visual aids on operational areas.

Remark: These requirements will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.

CHAPTER 9

Provisions concerning a system of preventive maintenance to be employed for precision approach runways, Categories II and III, as well as provisions concerning Human Factors issues.

Remark: These requirements will be included in the review of the Civil Aviation Rules Part 139.

Comment on implementation: On issuance of the revised Civil Aviation Rules Part 139, due late 2000.
CHAPTER 2

2.9.6 Norway does not use the method described for continuous friction measuring devices.

Remark: We regard the measured information concerning slippery runway conditions in Norway to be too inaccurate to be published for official use.

CHAPTER 3

3.4.1 A runway end safety area is not provided at the end of a runway strip where the code number is 1 or 2 and the runway is an instrument one. It is physically difficult, in some cases impossible, to establish a runway end safety area at the end of the runway strip at Norwegian aerodromes where the code number is 1 or 2.

CHAPTER 5

5.2.1.4 Runway markings shall be yellow.

Remark: The reason for using yellow markings is due to the need for improved visual references during the winter season when runways are covered with ice and snow.

5.3.10.1 Runway threshold lights are provided with use of wing bars only on precision approach runways, Cat I.

Remark: In Norway, most precision approach runways have displaced thresholds. When thresholds are displaced, the threshold lights must be inset into the pavement. During the winter season, difficulties have been experienced in keeping inset lights clear of ice and snow. Norway regards the use of wing bars as threshold lights sufficient for satisfying the needs of Cat I operations.

CHAPTER 6

6.1.5 When the terrain is regarded as an obstacle, above an obstacle protection surface, obstacles of this type are normally not marked and lighted, even if the runway is used at night.

6.3.9* When the terrain is regarded as an obstacle, above an obstacle protection surface, obstacles of this type are usually not displayed with top lights.

CHAPTER 8

8.7.2 b) Some of the equipment required for air navigation purposes, penetrating the inner approach surface and the inner transitional surface, are not frangible.

Remark: The reason for using non-frangible equipment is due to climactic conditions in Norway.

CHAPTER 9

9.4.4* Norway does not regard the method described, for using a continuous friction measuring device, as satisfactory in order to be able to publish necessary information concerning slippery conditions.

9.4.5

* Recommended Practice
CHAPTER 3

3.4.1 Runway end safety areas, as specified in Amendment 3 to Annex 14, Volume I, cannot be implemented at
3.4.2 the following international airports:
3.4.3*

Lisboa (LPPT) — Runway 17
Funchal (LPFU)
Faro (LPFR)
João Paulo II (LPPD)
Horta (LPHR)
Flores (LPFL)

Remark: The differences reported herein result from physical or environmental constraints. The individual
constraints are as follows:

Lisboa, Runway 17: A road is located at the end of runway 17.
Funchal: A complete renewal project is in progress.
João Paulo II and Horta: The runways are located in the periphery of islands. The compliance process
is dependent on a change in the declared distances and a corresponding runway lighting system
modification.
Faro: The airport is located in an environmentally protected area. Any modification of the land use is
dependent on an environmental impact evaluation.

Comment on implementation:

Existing infrastructures:

Porto (LPPR): will comply by 4 November 1999
Santa Maria (LPAZ): will comply by 4 November 1999
Porto Santo (LPPS): will comply by 4 November 1999
Lisboa (LPPT) runways, 03, 21, 35: will comply by 4 November 1999
Funchal (LPFU): will comply after 31 December 2000
Flores (LPFL): will comply after 31 December 2000

The compliance date after 4 November 1999 will be notified for each airport through the specific
Aeronautical Information Publication (AIP, NOTAM CIA). The precise compliance date is subordinated
to the planning of the corresponding correction project and the result of an in-depth evaluation of the
operational and safety consequences.

New infrastructures:

Will comply with the provisions of Annex 14, Volume I, including all amendments up to and including
Amendment 3.

* Recommended Practice
CHAPTER 1

1.3 The basic aerodrome classification is the length of the paved runway. Aerodromes are subdivided into six classes. In standardizing the taxiway requirements, aeroplanes are subdivided into six groups (index numbers) depending on wing span and wheel span. For the largest (index number 6) aeroplanes, a wing span of up to 65 m and a wheel span of up to 14 m are used. The requirements for index number 6 aeroplanes also apply to aeroplanes with a wing span of up to 75 m and a wheel span of up to 10.5 m.

Remark: The national use of a wing span value of 75 m is due to the operation of An-124 type aeroplanes, with a wing span of 73.3 m, at aerodromes. Larger aeroplanes are not currently being used at the country’s aerodromes.

CHAPTER 3

3.1.2* The value of the maximum permissible cross-wind component varies from 6 to 12 m/s (from 21.6 to 43.2 km/h) as a function of the aerodrome category.

3.1.16* The requirement to provide a line of sight over the entire length of the runway, where a parallel (main) taxiway is not available, is not used. There is a general requirement for a line of sight of not less than half the runway length from a height of 3 m, regardless of the availability of a main taxiway.

3.2.3* Shoulders are not provided with a runway width of 60 m or more.

3.3.7 The width of the obstacle-free part of the runway strip is taken to be 60 m from the runway centre line for all aerodrome categories.

3.4 Runway end safety areas are not used. Their functions are performed by sections of the runway strip located beyond the runway ends.

3.6.4* The requirement to provide a stopway coefficient of friction equal to that of the adjacent runway is not used.

3.7 The requirements for the radio altimeter operating area are not used.

3.8.4* The taxiway width is from 7 to 22.5 m as a function of the aeroplane code (group).

3.8.7* As a function of the aeroplane code (group), the distance between parallel taxiway centre lines is from 38 to 95 m; the distance between a taxiway centre line and an object is from 22 to 55 m; the distance between an aircraft stand taxilane centre line and an object is from 16 to 45 m.

3.8.20* The requirement to provide access for rescue and fire fighting vehicles to aeroplanes on a taxiway bridge is not used.

3.9.1* The overall width of the taxiway with two shoulders is from 27 to 40.5 m as a function of the aeroplane code.

3.10 The requirements for taxiway strips are not used.

3.11.4* The requirement for an intermediate holding position on a taxiway is not used.

3.14 The requirements for de/anti-icing facilities are not used.

* Recommended Practice
CHAPTER 4

4.2 Insignificant differences relate to the dimensions and slopes of the obstacle limitation surfaces. The obstacle limitation surfaces are differentiated according to the runway (aerodrome) categories and to the extent to which the runway is equipped with landing aids.

CHAPTER 5

5.2.5 The aiming point is not used. Its functions are performed by the fixed distance area (300 m from the threshold).

5.2.7.1 The requirement for a runway side stripe marking, where there is a lack of contrast between the pavement and the adjacent area of the runway strip, is not used. There is a requirement for a side stripe marking for a precision approach runway.

5.2.9.3 Marking of additional runway-holding positions is not used.

5.2.9.7 The runway-holding position marking at a runway/runway intersection is not used.

5.2.10 The intermediate holding position marking at taxiway intersections is not used. The taxiway intersection positions are designated by taxiway signs.

5.2.11 The VOR aerodrome check in marking is not used.

5.2.14 The road-holding position marking is not used.

CHAPTER 6

6.2.4* Alternating horizontal bands 0.5 to 6 m wide are used to mark an object up to 100 m high. Not less than 1/3 of the height of the object is marked.

CHAPTER 7

7.1 Marking of closed runways and taxiways or parts thereof is not used.

7.3 Different markings in terms of shape, dimensions and colour are used to designate the runway displaced threshold.

* Recommended Practice

8/9/00
CHAPTER 1

1.1 South Africa will retain the use of the wording “Taxi-Holding Position” and “Taxi-Holding Position Marking”.

*Remark:* To avoid confusion when such markings are used on taxiway crossings.

CHAPTER 5

5.2.9 South Africa will retain the use of the wording “Taxi-Holding Position” and “Taxi-Holding Position Marking”.

*Remark:* To avoid confusion when such markings are used on taxiway crossings.
CHAPTER 2

2.9.9* Given the climate of our country, the proposed Recommendation is not applied.

CHAPTER 3

3.1.11* The minimum distance between parallel runways intended for independent parallel approaches is 1 310 m.

3.3.2 It is considered that not all of the airports will have been adapted to the Standard concerning the length of runway strips by 1 January 2009.

3.4.3* This Recommended Practice is not complied with in Spain.

3.7 The Recommendation to establish radio altimeter operating areas is applied at those airports requiring category II/III precision approaches.

CHAPTER 5

5.2.15 In Spain, the character height for information markings is normally 2 m.

5.3.5 On 31 December 1999, the seven thresholds of four runways at which VASIS systems still operate at the present time will have operational PAPI systems.

5.3.17.10 The current facilities are not adapted to this Standard.

5.3.22 By 31 December 1999, all visual docking guidance systems will comply with the specifications.

5.4 Adaptation to the corresponding provisions will be completed during 2001.

CHAPTER 6

6.3.16 This provision is not applied, as ICAO requirements differ from those in our Traffic Code.

* Recommended Practice
CHAPTER 5

5.2.5 1 January 2005.

*Remark:* Swedish regulations contain requirement only for code 4.

5.2.6 1 January 2005.

*Remark:* Swedish regulations contain requirement only for code 4.

5.2.16.1 Swedish regulations contain no requirement.

*Remark:* Information markings are deemed to have limited use, given the winter conditions. Signs are required to the extent necessary.

CHAPTER 6

*General comment:* Swedish regulations do not fully correspond with Annex 14. National requirements will be reviewed within the next three years.
CHAPTER 5

5.3.5 In general, we do not install a visual approach slope indicator system on a precision approach runway.

CHAPTER 6

6.2.9* The spacing between two markers, where the marker diameter is 60 cm, may be increased to a 40 m maximum.

Remark: For reasons of uniformity, it is not possible to change given the large number of marked installations in our country.

6.3.3* 3 × 80 cd occulting lights may also be used instead of medium-intensity lights.

Remark: Proven practice, high reliability of the lights and good visibility from all angles.

* Recommended Practice
CHAPTER 2

2.6 PCN for pavements has not been determined. LCN method still in use.
CHAPTER 1

1.1 Aerodrome traffic density; Runway holding point; Intermediate holding point — currently not adopted.

Remark: Pending internal assessment and external consultation.

CHAPTER 3

3.1.13* UK uses 2.5 per cent where the code is 2; 3 per cent where the code is 1.

Remark: These criteria have been in use for many years; no evidence to suggest they are inadequate.

3.1.14* UK uses 3 per cent where the code is 1.

Remark: These criteria have been in use for many years; no evidence to suggest they are inadequate.

3.3.12* UK uses 2 per cent where the code is 3; 2.5 per cent where the code is 2; 3 per cent where the code is 1.

Remark: These criteria have been in use for many years; no evidence to suggest they are inadequate.

3.4.3 UK uses 30 m for code 1 and 2 instrument runway.
No provision for code 1 and 2 visual runway.

Remark: These criteria have been in use for many years; no evidence to suggest they are inadequate.

3.4.9* UK allows up to 10 per cent down slope.

Remark: Only permitted after aeronautical study determines acceptability.

3.5.4* UK allows 2 per cent up slope where the codes are 1 and 2.

Remark: Only permitted after aeronautical study determines acceptability.

3.8.7* Column (10) (11)
Table 3-1 Code A 21.0 13.5
Code B 31.5 19.5

Remark: The distances used are adequate for the operation involved.

Table 3-2 Code 4 precision approach categories I, II, III, note ‘c’ — currently not adopted.

Remark: Pending internal assessment and external consultation.

CHAPTER 4

Table 4-1 Balked landing surface, inner edge, note ‘e’ — currently not adopted.

Remark: Pending internal assessment and external consultation.

* Recommended Practice
CHAPTER 5

5.2.4.3* Not required in UK.

*Remark:* The use of grass runways for public transport operations in the UK is limited and the use of non-paved runway markings is an over provision.

5.2.5.5 The UK uses a different style of aiming point marking, as illustrated below.

*Remark:* The UK considers that the ICAO standard aiming point marking is insufficiently conspicuous.

![Touchdown Zone Markings](image)

<table>
<thead>
<tr>
<th>Runway width (m)</th>
<th>Distance A runway C/L to marker (m)</th>
<th>Marker width B (m)</th>
<th>Marker width C (m)</th>
</tr>
</thead>
<tbody>
<tr>
<td>45</td>
<td>9</td>
<td>5.5</td>
<td>3</td>
</tr>
<tr>
<td>30</td>
<td>3</td>
<td>5</td>
<td>3</td>
</tr>
<tr>
<td>23</td>
<td>5</td>
<td>2.5</td>
<td>1.5</td>
</tr>
<tr>
<td>18</td>
<td>3</td>
<td>2.5</td>
<td>1.5</td>
</tr>
</tbody>
</table>

5.2.8.1 Taxiway centre line marking from de/anti-icing facility and apron between runway centre line and aircraft stands — currently not adopted.

*Remark:* Pending internal assessment and external consultation.

5.2.15 Mandatory instruction marking — currently not adopted.

*Remark:* Pending internal assessment and external consultation.

5.3.4.15 Composition of approach centre line lights where serviceability level of lights can meet maintenance objective in 9.4.29 — currently not adopted.

*Remark:* Pending internal assessment and external consultation.

* Recommended Practice

8/9/00
5.3.4.22 Composition of approach lights where serviceability level of lights can meet maintenance objective in 9.4.26 — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.3.4.24 Composition of approach lights (side rows) where serviceability of lights can meet maintenance objective in 9.4.26 — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.3.4.30 Composition of Cat II/III centre line approach lighting where serviceability of lights can meet maintenance objective in 9.4.26 — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.3.4.32 Composition of approach lights where serviceability can meet maintenance objective in 9.4.26 — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.3.5.42 UK uses a plane 1° below lower boundary of on slope signal originating 90 m from units where LDA 1 200 m or greater, 60 m where LDA 800 m – 1 199 m and 30 m where LDA <800 m, diverging at 15° from runway edge out to 15 NM.

Remark: This practise has been in existence since PAPI was originally designed; no safety related incidents to justify introducing more penalizing criteria.

5.3.12.2* Not required in UK.

Remark: No demonstrated safety benefit. Markings and HI edge lights considered sufficient.

5.3.12.4* Not required in UK.

Remark: No demonstrated safety benefit. Markings and HI edge lights considered sufficient.

5.3.15.1 Provision of continuous taxiway lights between runway and stands in RVR less than 350 m — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.3.15.7 UK uses amber/green both ways within OFZ.

Remark: The pattern is intended to remind pilots when they are within OFZ/ILS protected areas.

5.3.15.11 c)* UK uses 30 m spacing for taxiways used in RVR 200 m and above, 15 m spacing when used in RVR <200 m.

Remark: Tests and practical experience confirm that spacing is sufficient.

5.3.15.13* UK uses 15 m spacing for taxiways used in RVR 200 m and above, 7.5 m where used in RVR <200 m.

Remark: Tests and practical experience confirm that spacing is sufficient.

*Recommended Practice
5.3.15.17* UK uses 15 m spacing for exits used in RVR 200 m and above, 7.5 m where used in RVR < 200 m.

Remark: Tests and practical experience confirm that spacing is sufficient.

5.3.15.18* UK uses 30 m spacing for taxiways used in RVR 200 m and above, 15 m spacing where used in RVR < 200 m.

Remark: Tests and practical experience confirm that spacing is sufficient.

5.3.16.6 UK also uses omni-directional taxiway edge lights. Use of omni-directional edge lights is gradually being phased out.

5.3.19 De-anti-icing facility exit lights — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.3.22.11* UK uses systems not aligned for use by both pilots.

Remark: Many of these systems were in use before guidance material was produced. They are gradually being replaced with compliant units.

5.3.22.14 UK uses systems where pilot has to turn head.

Remark: Pending internal assessment and external consultation.

5.3.22.16* UK uses systems not aligned for use by both pilots.

Remark: Pending internal assessment and external consultation.

5.4.1.9 Blank face on variable message sign — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.4.2.8 Runway designation sign location — currently not adopted.

Remark: Pending internal assessment and external consultation.

5.4.3.5 Not used in UK.

Remark: UK uses a location sign.

5.4.3.14* UK uses taxiway ending sign but not just at ‘T’ junctions.

Remark: UK uses a taxiway ending sign which is a location sign with a yellow diagonal.

5.4.3.17 Not all runway exit signs on same side of runway as exit.

Remark: Location is on same side where possible.

5.4.3.21 Location of intersection take-off sign — currently not adopted.

Remark: Pending internal assessment and external consultation.

* Recommended Practice

8/9/00
5.4.3.25* Taxiway ending sign at ‘T’ junctions not always located as specified.

* Remark: Taxiway ending signs should be located well before end of taxiway.

5.4.3.26 Not used in UK.

5.4.3.27 All UK location signs have yellow border.

5.4.3.30 Inscription on intersection take-off sign — currently not adopted.

* Remark: Pending internal assessment and external consultation.

5.5.3.1* Not used in UK.

* Remark: When standard marking agreed, UK will implement.

5.5.3.2 Not used in UK.

* Remark: When standard marking agreed, UK will implement.

CHAPTER 6

6.2.8* Not required in UK.

6.3.16 Provision of medium-intensity obstacle lights Type A/B/C — currently not adopted.

6.3.17

6.3.18 *Remark: Pending internal assessment and external consultation.

CHAPTER 7

7.1.4 UK uses elongated legs 0.9 m wide as illustrated:

* Recommended Practice
7.2.2* Inner edge marks outer edge of load-bearing surface.

CHAPTER 8

8.3.4* Not used in UK.

*Remark: UK is working on minimum acceptable levels of lighting.

8.6.1 Design and construction of new facilities to incorporate security measures — currently not adopted.

*Remark: Pending internal assessment and external consultation.

8.7.5 Location of equipment or installation with reference to runway strip and extended centre line — currently not adopted.

*Remark: Pending internal assessment and external consultation.

8.7.6 Frangibility and mounting of equipment or installations located on or near Cat. I/II/III precision approach runways — currently not adopted.

*Remark: Pending internal assessment and external consultation.

CHAPTER 9

9.2.10 At all aerodromes, up to 50 per cent of the complimentary media may be replaced by water for foam production to performance level B.

9.4.20 Serviceability requirements for lights in visual aids — currently not adopted.

*Remark: Pending internal assessment and external consultation.

Appendix 4

paragraph 11 Not specified in UK.
CHAPTER 1

1.1 De/anti-icing facility definition was revised to include slush as another form of contamination to be removed. Holdover time definition revised to include commencement of holdover time for one-step procedures.

1.2.1 Airports in the United States are for the most part owned and operated by local governments and quasi-governmental organizations formed to operate transportation facilities. The Federal government provides air traffic control, operates and maintains navaids, provides financial assistance for airport development, certifies major airports and issues standards and guidance for airport planning, design and operational safety.

There is general conformance with the Standards and Recommended Practices of Annex 14, Volume I. At airports with scheduled passenger service using aircraft having more than nine seats, compliance with Standards is enforced through regulation and certification. At other airports, compliance is achieved through the agreements with individual airports under which federal development funds were granted; or, through voluntary actions.

1.3.1 In the United States the Airport Reference Code is a two-component indicator relating the standards used in the airport’s design to a combination of dimensional and operating characteristics of the largest aircraft expected to use the airport. The first element, Aircraft Approach Category, corresponds to the ICAO PANS-OPS approach speed groupings. The second, Airplane Design Group (ADG), corresponds to the wingspan groupings of code element 2 of the Annex 14, Aerodrome Reference Code. See below:

<table>
<thead>
<tr>
<th>Aircraft Approach Category</th>
<th>Approximate Annex 14 Code Number</th>
<th>Corresponding Annex 14 Code Letter</th>
</tr>
</thead>
<tbody>
<tr>
<td>A</td>
<td>1</td>
<td>A</td>
</tr>
<tr>
<td>B</td>
<td>2</td>
<td>B</td>
</tr>
<tr>
<td>C</td>
<td>3</td>
<td>C</td>
</tr>
<tr>
<td>D</td>
<td>4</td>
<td>D</td>
</tr>
<tr>
<td>E</td>
<td>⎕</td>
<td>E</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>Airplane Design Group</th>
<th>Corresponding Annex 14 Code Letter</th>
</tr>
</thead>
<tbody>
<tr>
<td>I</td>
<td>A</td>
</tr>
<tr>
<td>II</td>
<td>B</td>
</tr>
<tr>
<td>III</td>
<td>C</td>
</tr>
<tr>
<td>IV</td>
<td>D</td>
</tr>
<tr>
<td>V</td>
<td>E</td>
</tr>
<tr>
<td>VI</td>
<td>F (proposed)</td>
</tr>
</tbody>
</table>

CHAPTER 2

2.2.1 Comment on implementation: The aerodrome reference point is recomputed when the ultimate planned development of the aerodrome is changed.
2.9.6 Minimum friction values have not been established to indicate that runways are “slippery when wet”.

2.9.7 However, United States guidance recommends that pavements be maintained to the same levels indicated in the ICAO Airport Services Manual.

2.11.3 If an inoperative piece of fire fighting apparatus cannot be replaced immediately, a Notice to Airmen must be issued. If the apparatus is not restored to service within 48 hours, operations shall be limited to those compatible with the lower Index corresponding to operative apparatus.

2.12 e) Where the original VASI is still installed, the threshold crossing height is reported as the centre of the on-course signal, not the top of the red signal from the downwind bar.

CHAPTER 3

3.1.2* The cross-wind component is based on the Airport Reference Code (ARC): 10.5 kt for AI and BI; 13 kt for AII and BII; 16 kt for AIII, BIII and CI through DIII; 20 kts for AIV through DVI.

3.1.9* The runway widths (metres) used in design are below:

<table>
<thead>
<tr>
<th>Aircraft Approach Category</th>
<th>Width of runway (metres)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Airplane Design Group</td>
</tr>
<tr>
<td></td>
<td>I</td>
</tr>
<tr>
<td>A1</td>
<td>18</td>
</tr>
<tr>
<td>B1</td>
<td>18</td>
</tr>
<tr>
<td>C3</td>
<td>30</td>
</tr>
<tr>
<td>D3</td>
<td>30</td>
</tr>
</tbody>
</table>

*1 The width of a precision (lower than 1 200 m approach visibility minimums) runway is 23 m for a runway which is to accommodate only small (less than 5 700 kg) aeroplanes and 30 m for runways accommodating larger aeroplanes.

*2 For aeroplanes with a maximum certificated take-off mass greater than 68 000 kg, the standard runway width is 45 m.

3.1.12* Longitudinal runway slopes of up to 1.5 per cent are permitted for aircraft approach Categories C and D, except for the first and last quarter of the runway where the maximum slope is 0.8 per cent.

3.1.16* Runway profiles permit any two points five feet (1.5 m) above the runway centre line to be mutually visible for the entire runway length. However, if the runway has a full length parallel taxiway, the runway profile may be such that an unobstructed line of sight will exist from any point five feet (1.5 m) above the runway centre line for one-half the runway length. Regarding the Note, the 1.5 m as compared to 3 m height difference is additionally required for the runway visibility zone contained in the Aerodrome Design Manual (Doc 9157), Part 1, Runways.

3.1.18* Minimum and maximum transverse runway slopes are based on aircraft approach categories as follows:

For Categories A and B: 1.0-2.0 per cent
For Categories C and D: 1.0-1.5 per cent

*Recommended Practice
3.2.3* The United States does not require that the minimum combined runway and shoulder widths equal 60 m. The widths of shoulders are determined independently.

3.2.4* The transverse slope on the innermost portion of the shoulders can be as high as 5 per cent.

3.3.3 A strip width of 120 m is used for code 3 and 4 runways for precision, non-precision and non-instrumented operations. For code 1 and 2 precision runways, the width is 120 m. For non-precision/visual runways, widths vary from 37.5 m up to 120 m.

3.3.4* The frangibility and object removal requirement are applicable only to the FAA runway safety area (RSA) and the obstacle free zone (OFZ). The RSA functions similarly to the ICAO “graded portion” of the runway strip. The difference is less than 1.3 m; that is, the FAA Standard measure 76.2 m as compared to 77.5 m under Annex 14. However, the area beyond the RSA, the runway object free area, measures 120 m and requires object removal, but not frangibility for permitted objects.

3.3.5* Airports used exclusively by small aircraft (U.S. Airplane Design Group I) may be graded to distances as little as 18 m from the runway centre line.

3.3.7 The frangibility and object removal requirement are applicable only to the FAA runway safety area (RSA) and the obstacle free zone (OFZ). The RSA functions similarly to the ICAO “graded portion” of the runway strip. The difference is less than 1.3 m; that is, the FAA Standard measure 76.2 m as compared to 77.5 m under Annex 14. However, the area beyond the RSA, the runway object free area, measures 120 m and requires object removal, but not frangibility for permitted objects.

3.3.9* The maximum transverse slope of the graded portion of the strip can be 3 per cent for Aircraft Approach Categories C and D and 5 per cent for Aircraft Approach Categories A and B.

3.3.14* The United States does not have standards for the maximum transverse grade on portions of the runway strip falling beyond the area that is normally graded.

3.3.15* Runways designed for use by smaller aircraft under non-instrument conditions may be graded to distances as little as 18 m from the runway centre line (U.S. Airplane Design Groups I and II).

3.4.2 For certain code 1 runways the runway end safety areas may be only 72 m.

3.6.4* Pavement friction measurements are taken for only the runway with full strength pavement and not for the surface of stopways.

3.7.1* The United States does not provide Standards or Recommended Practices for radio altimeter operating areas.

3.8.3* The United States specifies a 6 m clearance for Design Group VI aeroplanes.

3.8.4* The taxiway width for Design Group VI aircraft is 30 m.

3.8.5* The United States also permits designing taxiway turns and intersections using the judgmental oversteering method.

3.8.7* Minimum separations between runway and taxiway centre lines are shown in Table 1. Minimum separations between taxiways and taxilanes and between taxiway/taxilanes and fixed/moveable objects are shown in Table 2. Generally, United States separations are larger for non-instrumented runways, and smaller for instrumented runways, than in the Annex. Values are also provided for aircraft with wingspans up to 80 m. For Code F, separations in all cases equal 180 m.

3.8.8* Longitudinal grades (slopes) for taxiways are based on approach category (code number) rather than wing span (code letter). However, values are: Code 1 and 2 = 3.0 per cent; Code 3 and 4 = 1.5 per cent.

*Recommended Practice
3.8.10* Line-of-sight standards for taxiways are not provided in United States practice, but there is a requirement that the sight distance along a runway, from an intersecting taxiway, must be sufficient to allow a taxiing aircraft to safely enter or cross the runway.

3.8.11* Transverse slopes of taxiways are based on aircraft approach categories. For Categories C and D slopes are 1.0-1.5 percent; for A and B 1.0-2.0 per cent.

3.9.1* The overall width of the taxiway plus shoulders is 54 m as compared to 60 m for the Annex.

Remark: For Code F aircraft the emphasis is on wider full strength pavement rather than shoulders. The United States requires 30 m taxiway widths as compared to the 25 m width under 3.8.4.

3.10.5* Grading of runway strips are downward and are based on aircraft approach categories, that is code numbers. Furthermore, we recommend a downward slope from the horizontal for the first three metres beyond the taxiway or shoulders followed by a downward range of 1.5-3.0 per cent for the remainder of the strip.

3.11.6 Regarding Table 3-2, FAA runway centre line to taxi-holding position, etc., a separation for code 1 is 38 m for non-precision operations and 53 m for precision. Code 3 and 4 precision operations require a separation of 75 m, except for “wide bodies”, which require 85 m. There is no standard clearance for Code F aircraft at this time.

CHAPTER 4

4.1 Obstacle limitation surfaces similar to those described in 4.1-4.20 are found in Federal Aviation Regulation Part 77.

4.1.21 A balked landing surface is not used.

4.1.25 The United States does not establish take-off climb obstacle limitation areas and surfaces per se but does specify protective surfaces for each end of the runway based on the type of approach procedures available or planned. The dimensions and slopes for these surfaces and areas are listed in Table 3.

4.2 For Code C, D and F aircraft, the width of the OFZ inner approach equals 120 m.

Table 4-1

Remark: Research and development for the width measurement for Code F are ongoing. Flight Standards will issue, upon completion of the R & D simulation, an FAA document with results and possible new recommendations.

CHAPTER 5

5.2.1.8* The United States does not require unpaved taxiways to be marked.

5.2.2.2* The United States does not require a runway designator marking for unpaved runways.

* Recommended Practice

8/9/00
5.2.2.4 Zeros are not used to precede single digit runway markings. An optional configuration of the numeral 1 is available to designate a Runway 1 and to prevent confusion with the runway centre line.

5.2.4.2* Threshold markings are not required, but sometimes provided, for non-instrument runways that do not serve international operations.

5.2.4.3* The current United States standard for threshold designation is eight stripes, except that more than eight stripes may be used on runways wider than 45 m. After 1 January 2008, the United States standard will comply with Annex 14.

5.2.4.5 The width and spacing of threshold stripes will comply with Annex 14 after 1 January 2008.

5.2.4.6 When a threshold is temporarily displaced, there is no requirement that runway or taxiway edge markings, prior to the displaced threshold, be obscured. These markings are removed only if the area is unsuitable for the movement of aircraft.

5.2.5.2 Aiming point markings are required on precision instrument runways and code 3 and 4 runways used by jet aircraft.

5.2.5.3* The aiming point marking commences 306 m from the threshold at all runways.

5.2.5.4 The United States pattern for touchdown zone markings, when installed on both runway ends, is only applicable to runways longer than 4 990 feet. On shorter runways, the three pairs of markings closest to the runway midpoint are eliminated.

5.2.6.3 The United States standard places the aiming point marking 306 m from the threshold where it replaces one of the pairs of three-stripe threshold markings. The 306 m location is used regardless of runway length.

5.2.6.4 Touchdown zone markings are not required at a non-precision approach runway, though they may be provided.

5.2.7.4* Runway side stripe markings on a non-instrument runway may have an overall width of 0.3 m.

5.2.8.3 Taxiway centre line markings are never installed longitudinally on a runway even if the runway is part of a standard taxi-route.

5.2.9.6* The term “ILS” is used instead of CAT I, CAT II, CAT III.

5.2.9.7 Runway-holding position markings at runway/runway intersections are as shown in Figure 5-6, Pattern A.

5.2.11.4 Check-point markings are provided, but the circle is 3 m in diameter and the directional line may be of varying width and length. The color is the yellow used for taxiway markings.

5.2.11.5* Standards for aircraft stand markings are not provided.

5.2.11.6* Apron safety lines are not required although many aerodromes have installed them. The United States does not set marking standards for aprons, finding it more effective to let airports and airlines manage activities related to aircraft parking.

*Recommended Practice
5.2.14.1 The United States does not have standards for holding position markings on roadways that cross runways. Local traffic control practices are used.

5.3.1.1 The United States does not have regulations to prevent the establishment of non-aviation ground lights that might interfere with airport operations.

5.3.1.2* New approach lighting installations will meet the frangibility requirements. Some existing non-frangible systems may not be replaced before 1 January 2005.

5.3.2.1* There is no requirement for an airport to have emergency runway lighting available if it does not have a secondary power source. Some airports do have these systems, and there is an FAA specification for these lights.

5.3.3.1 Only airports served by aircraft having more than 30 seats are required to have a beacon, though they are available at many others.

5.3.3.3 Although the present United States standard for beacons calls for 24-30 flashes per minute, some older beacons may have flash rates as low as 12 flashes per minute.

5.3.3.6 Coded identification beacons are not required and are not commonly installed. Typically, airport beacons conforming to 5.3.3.6 are installed at locations served by aircraft having more than 30 seats.

5.3.4.1 While the United States has installed an approach light system conforming to the specifications in 5.3.4.10 to 5.3.4.19, it also provides for a lower cost system consisting of medium intensity approach lighting and sequenced flashing lights (MALS) at some locations.

5.3.4.2 In addition to the system described in 5.3.4.1, a system consisting of omnidirectional strobe lights (ODALS) located at 90 m intervals extending out to 450 m from the runway threshold is used at some locations.

5.3.4.10 to 5.3.4.19 The United States standard for a precision approach Category I lighting system is medium intensity approach lighting system with runway alignment indicator lights (MALS). This system consists of 3 m barrettes at 60 m intervals out to 420 m from the threshold and sequenced flashing lights at 60 m intervals from 480 m to 900 m. A crossbar 20 m in length is provided 300 m from the threshold. The total length of this system is dependent upon the ILS glide slope path angle. For angles 2.75° and higher, the length is 720 m.

5.3.4.18 The capacitor discharge lights can be switched on or off when the steady-burning lights of the approach lighting system are operating. However, they cannot be operated when the other lights are not in operation.

5.3.4.22 The United States standard for a precision approach Category II and III lighting system has a total length dependent upon the ILS glide path angle. For angles 2.75° and higher, the length is 720 m.

5.3.4.35 The capacitor discharge lights can be switched on or off when the steady-burning lights of the approach lighting system are operating. However, they cannot be operated when the other lights are not in operation.

5.3.5.1 Visual approach slope indicator systems are not required for all runways used by turbojets except runways involved with land and hold short operations that do not have an electronic glideslope system.

*Recommended Practice

8/9/00
5.3.5.2 In addition to PAPI and APAPI systems, VASI and AVASI type systems remain in service at United States airports with commercial service. Smaller general aviation airports may have various other approach slope indicators including tri-color and pulsating visual approach slope indicators.

5.3.5.3 Visual approach slope indicator systems are not required for all runways used by turbojets except runways involved with land and hold short operations that do not have an electronic glideslope system.

5.3.5.27 The United States standard for PAPI allows for the distance between the edge of the runway and the first light unit to be reduced to 9 m for code 1 runways used by non-jet aircraft.

5.3.5.42 The PAPI obstacle protection surface used is as follows: The surface begins 90 m in front of the PAPI system (toward to the threshold) and proceeds outward into the approach zone at an angle \( \frac{1}{8} \) less than the aiming angle of the third light unit from the runway. The surface flares \( \frac{10}{8} \) on either side of the extended runway centre line and extends four statute miles from its point of origin.

5.3.8.4 The United States permits the use of omnidirectional runway threshold identification lights.

5.3.13.2 The United States does not require the lateral spacing of touchdown zone lights to be equal to that of touchdown zone marking when runways are less than 45 m wide.

5.3.14 The United States has no provision for stopway lights.

5.3.15.1 Taxiway centre line lights are required only below 183 m runway visual range (RVR) on designated taxi routes. However, they are generally recommended whenever a taxiing problem exists.

5.3.15.2* Taxiway centre line lights are not provided on runways forming part of a standard taxi route even for low visibility operations. Under these conditions, the taxi path is coincident with the runway centre line and the runway lights are illuminated.

5.3.15.7 Comment on Implementation: The United States standard for taxiway centre line lights on exit taxiways was brought into compliance with this Annex 14 provision on 1 September 1998. However, United States airports are still in the process of implementing the Standard.

5.3.15.10* The United States permits an offset of up to 60 cm.

5.3.16.2 Taxiway edge lights are not provided on runways forming part of a standard taxi route.

5.3.17.1 Stop bars are required only for RVR conditions less than a value of 183 m at taxiway/runway intersections where the taxiway is lighted during low visibility operations. Once installed, controlled stop bars are operated at RVR conditions less than a value of 350 m.

5.3.17.2* Elevated stop bar lights are normally installed longitudinally in line with taxiway edge lights. Where edge lights are not installed, the stop bar lights are installed not more than 3 m from the taxiway edge.

5.3.17.3 The beamspread of elevated stop bar lights differs from the in-pavement lights. The inner isocandela curve for the elevated lights is \( \pm 7 \) hor. and \( \pm 4 \) vert.

5.3.17.9 The United States standard for stop bars, which are switchable in groups, does not require the taxiway centre line lights beyond the stop bars to be extinguished when the stop bars are illuminated. The taxiway centre line lights which extend beyond selectively switchable stop bars are grouped into two

*Recommended Practice
segments of approximately 45 m each. A sensor at the end of the first segment re-illuminates the stop bar and extinguishes the first segment of centre line lights. A sensor at the end of the second segment extinguishes that segment of centre line lights.

5.3.18.1 Taxiway intersection lights are also used at other hold locations on taxiways such as low visibility holding points.

5.3.18.2* Taxiway intersection lights are collocated with the taxiway intersection marking. The marking is located at the following distances from the centre line of the intersecting taxiway:

<table>
<thead>
<tr>
<th>Aeroplane Design Group</th>
<th>Distance (m)</th>
</tr>
</thead>
<tbody>
<tr>
<td>I</td>
<td>13.5</td>
</tr>
<tr>
<td>II</td>
<td>20</td>
</tr>
<tr>
<td>III</td>
<td>28.5</td>
</tr>
<tr>
<td>IV</td>
<td>39</td>
</tr>
<tr>
<td>V</td>
<td>48.5</td>
</tr>
<tr>
<td>VI</td>
<td>59</td>
</tr>
</tbody>
</table>

5.3.20.1 Runway guard lights are required only for RVR conditions less than a value of 350 m.
5.3.20.2* Runway guard lights are placed at the same distance from the runway centre line as the aircraft holding distance, or within a few feet of this location.

5.3.20.4 Runway guard lights are placed at the same distance from the runway centre line as the aircraft holding distance, or within a few feet of this location.

5.3.20.17 The new United States standard for in-pavement runway guard lights complies with Annex 14. However, there may be some existing systems that do not flash.

5.3.21 The United States does not set aviation standards for floodlighting aprons.

5.3.22 The United States does not provide standards for visual docking guidance systems.

5.3.24.1 The United States does not have a requirement for providing road-holding position lights during RVR conditions less than a value of 350 m.

5.4.1.3 Signs are often installed a few centimetres taller than specified in Annex 14, Volume 1, Table 5-4.

5.4.1.6 Sign inscriptions are slightly larger, and margins around the sign slightly smaller, than indicated in Annex 14, Volume 1, Appendix 4.

5.4.1.7 The sign luminance requirements are not as high as specified in Appendix 4. The United States does not specify a night-time color requirement in terms of chromaticity.

5.4.2.2 All signs used to denote precision approach holding positions have the legend “ILS”.
5.4.2.4

5.4.2.7 United States practice uses the NO ENTRY sign to prohibit entry by aircraft only.
5.4.2.9 The second mandatory instruction sign is usually not installed unless added guidance is necessary.

5.4.2.10 All signs used to denote precision approach holding positions have the legend “ILS”.

*Recommended Practice
5.4.2.11 The second mandatory instruction sign is usually not installed unless added guidance is necessary.

5.4.2.14 All signs used to denote precision approach holding positions have the legend “ILS”.

5.4.2.15 Signs for holding aircraft and vehicles from entering areas where they would infringe on obstacle limitation surfaces or interfere with navais are inscribed with the designator of the approach, followed by the letters “APCH”. For example: “15-APCH”.

5.4.2.18 All signs used to denote precision approach holding positions have the legend “ILS”.

5.4.3.15 United States practice is to install signs about 3 to 5 m closer to the taxiway/runway (See Annex 14, Table 5-4).

5.4.3.17 The United States does not have standards for the location of runway exit signs.

5.4.3.27 A yellow border is used on all location signs, regardless of whether they are stand-alone or collocated with other signs.

5.4.3.29 United States practice is to use Pattern A on runway vacated signs, except that Pattern B is used to indicate that an ILS critical area has been cleared.

5.4.3.34* The United States does not have standards for signs used to indicate a series of taxi-holding positions on the same taxiway.

5.4.4.4* The inscription “VOR Check Course” is placed on the sign in addition to the VOR and DME data.

5.4.5.1* The United States does not have requirements for aerodrome identification signs, though they are usually installed.

5.4.6.1* Standards are not provided for signs used to identify aircraft stands.

5.4.7.2 The distance from the edge of road to the road-holding position sign conforms to local highway practice.

5.5.2.2* Boundary markers may be used to denote the edges of an unpaved runway.

5.5.3 There is no provision for stopway edge markers.

5.5.7.1* Boundary markers may be used to denote the edges of an unpaved runway.

CHAPTER 6

6.1 The FAA does not exercise regulatory authority over the marking and lighting of obstacles. However, it reviews proposed construction of tall objects in the vicinity of airports and recommends the minimum marking and lighting consistent with aviation safety.

6.2.3* The maximum dimension of the rectangles in a chequered pattern is 6 m on a side.

6.3.21* The effective intensity, for daylight-luminance background, of Type A high-intensity obstacle lights is 270 000 cd ±25 per cent. The effective intensity, for daylight-luminance background, of Type B high-intensity obstacle lights is 140 000 cd ±25 per cent. A higher intensity improves conspicuity.

*Recommended Practice
United States standards do not require 75 per cent maximum intensity at $-1^{\circ}$, as shown in Column 9 for medium- and high-intensity obstacle lights.

CHAPTER 7

7.1.2* Closed markings are not used with partially closed runways. See 5.2.4.10 above.

7.1.4 Closed markings with shapes similar to Figure 7.1b are used to indicate closed runways and taxiways. The “X” for denoting a closed runway is yellow.

7.1.5 In the United States, when a runway is permanently closed, only the threshold marking, runway designation marking, and touchdown zone marking need be obliterated. Permanently closed taxiways need not have the markings obliterated.

7.1.7 The United States does not require unserviceability lights across the entrance to a closed runway or taxiway when it is intersected by a night-use runway or taxiway.

7.4.4 Flashing yellow lights are used as unserviceability lights. The intensity is such as to be adequate to delineate a hazardous area.

CHAPTER 8

8.1.5* 8.1.6* 8.1.7* 8.1.8

8.1.5* A secondary power supply for non-precision instrument and non-instrument approach runways is not required, nor is it required for all precision approach runways.

8.2.1 *Remark:* There is no requirement in the United States to interleave lights as described in the Aerodrome Design Manual, Part 5.

8.2.3 See 5.3.15.4 and 5.3.16.2.

8.7.2 Glide slope facilities and certain other installations located within the runway strip, or which penetrate obstacle limitation surfaces, may not be frangibly mounted.

8.7.4* 8.9.7* A Surface Movement Surveillance System is recommended from 350 m RVR down to 183 m. Below 183 m RVR, a surface movement radar or alternative technology is generally required.

CHAPTER 9

9.1.1 Emergency plans such as those specified in this section are required only at airports serving scheduled air carriers using aircraft having more than 30 seats. These airports are certificated under Part 139 of the United States Federal Aviation Regulations. In practice, other airports also prepare emergency plans.

9.1.12 Full-scale aerodrome emergency exercises are conducted at intervals, not to exceed three years, at airports with scheduled passenger service using aircraft with more than 30 seats.

*Remark:* Three years has been found to be an interval that provides for an efficient allocation of airport resources.

*Recommended Practice

8/9/00
9.2.1 Rescue and fire fighting equipment and services such as those specified in this section are required only at airports serving scheduled air carriers with aircraft having more than 30 seats. Such airports generally equate to ICAO categories 4 through 9.

9.2.3* There is no plan to eliminate, after 1 January 2005, the current practice of permitting a reduction of one category in the Index when the largest aircraft has fewer than an average of five scheduled departures a day. The reduction in category is a rudimentary cost/benefit consideration and also facilitates the introduction of large aircraft into scheduled service by not making the air carrier’s planning contingent on the airport’s immediate acquisition of additional equipment.

9.2.5 The level of protection at United States airports is derived from the length of the largest aircraft serving the airport. This is similar to the Annex 14 procedure, except that maximum fuselage width is not used.

Remark: United States indices A-E are close equivalents of the Annex’s categories 5-9. The United States does not have an equivalent to Category 10. The United States will consider the requirements of Category 10 when it adopts a new index for very large aircraft. Further harmonization with the Annex will be considered in the future.

9.2.10 The required fire fighting equipment and agents by index are shown in Table 4.

The substitution equivalencies between complementary agents and foam meeting performance level A are also used for protein and fluoroprotein foam. Equivalencies for foam meeting performance level B are used only for aqueous film forming foams.

9.2.18* 9.2.19* At least one apparatus must arrive and apply foam within 3 minutes, with all other required vehicles arriving within 4 minutes. Response time is measured from the alarm at the equipment’s customary assigned post, to the commencement of the application of foam at the mid-point of the farthest runway.

Remark: The United States values a rapid response and the presence of professional fire fighters at the earliest possible time to deal with incipient conditions.

9.2.29* For ICAO Category 6 (U.S. index B) the United States allows one vehicle.

9.4.5 At the present time, there is no requirement to perform tests using a continuous friction measuring device with self-wetting features. Some United States airports own these devices, while others use less formal methods to monitor build-up of rubber deposits and the deterioration of friction characteristics.

Remark: The United States is working toward making these devices a requirement, at least at larger airports.

9.4.16 The standard grade for temporary ramps is 15 feet longitudinal per 1 inch of height (0.56 per cent slope) maximum, regardless of overlay depth.

9.4.20 There is no United States standard for declaring a light unserviceable if its intensity is less than 50 per cent of the specified or design value of the main beam average intensity.

*Recommended Practice
Table 1. Minimum separations between runway centre line and parallel taxiway/taxilane centre line

<table>
<thead>
<tr>
<th>Operation</th>
<th>Aircraft approach category</th>
<th>I1</th>
<th>I</th>
<th>II</th>
<th>III</th>
<th>IV</th>
<th>V</th>
<th>VI</th>
</tr>
</thead>
<tbody>
<tr>
<td>Visual runways and runways with not lower than ¾-statute mile (1 200 m) approach visibility minimums</td>
<td>A&amp;B</td>
<td>150 ft</td>
<td>45 m</td>
<td></td>
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<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Runways with lower than ¾-statute mile (1 200 m) approach visibility minimums</td>
<td>A&amp;B</td>
<td>200 ft</td>
<td>60 m</td>
<td></td>
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<td></td>
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</tr>
<tr>
<td>Visual runways and runways with not lower than ¾-statute mile (1 200 m) approach visibility minimums</td>
<td>C&amp;D</td>
<td>–</td>
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<tr>
<td>Runways with lower than ¾-statute mile (1 200 m) approach visibility minimums</td>
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</tbody>
</table>

1 Pertains to small (less than 5 700 kg) aeroplanes.

2 Corrections are made for altitude: 120 m separation for airports at or below 410 m; 135 m for altitudes between 410 m and 2 000 m; and 150 m for altitudes above 2 000 m.

Table 2. Minimum taxiway and taxilane separations

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<thead>
<tr>
<th>Aeroplane design group</th>
<th>I</th>
<th>II</th>
<th>III</th>
<th>IV</th>
<th>V</th>
<th>VI</th>
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</thead>
<tbody>
<tr>
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<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
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<tr>
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<td></td>
</tr>
<tr>
<td>Fixed or movable object</td>
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<tr>
<td>Taxiway centre line to:</td>
<td></td>
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<td></td>
<td></td>
</tr>
<tr>
<td>Parallel taxilane centre line</td>
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<td></td>
</tr>
<tr>
<td>Fixed or movable object</td>
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</table>

8/9/00
### Table 3. Dimensions and slopes for protective areas and surfaces

<table>
<thead>
<tr>
<th>Type of approach procedure</th>
<th>Precision approach</th>
<th>Non-precision instrument approach</th>
<th>Visual runway</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>All runways</td>
<td>Runways other than utility</td>
<td>Runways other than utility</td>
</tr>
<tr>
<td></td>
<td>305 m</td>
<td>152 m</td>
<td>152 m</td>
</tr>
<tr>
<td>Width of inner edge</td>
<td>15 per cent</td>
<td>15 per cent</td>
<td>10 per cent</td>
</tr>
<tr>
<td>Divergency (each side)</td>
<td>15 per cent</td>
<td>15 per cent</td>
<td>15 per cent</td>
</tr>
<tr>
<td>Final width</td>
<td>4 877 m</td>
<td>1 067 m(^d)</td>
<td>610 m</td>
</tr>
<tr>
<td>Length</td>
<td>15 240 m</td>
<td>3 048 m(^c)</td>
<td>1 524 m(^c)</td>
</tr>
<tr>
<td>Slope: inner 3 049 m</td>
<td>2 per cent</td>
<td>2.94 per cent(^c)</td>
<td>5 per cent(^c)</td>
</tr>
<tr>
<td>Slope: beyond 3 048 m</td>
<td>2.5 per cent(^c)</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

\(^{a}\) With visibility minimum as low as 1.2 km.

\(^{b}\) With visibility minimum greater than 1.2 km.

\(^{c}\) Criteria less demanding than Table 4-1 dimensions and slopes.

\(^{d}\) Utility runways are intended to serve propeller driven aircraft having a maximum take-off mass of 5 670 kg.

### Table 4. Fire extinguishing agents and equipment

<table>
<thead>
<tr>
<th>Index</th>
<th>Aircraft length</th>
<th>Total minimum quantities of extinguishing agents</th>
<th>Minimum number of trucks</th>
<th>Discharge rate (^1)</th>
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<tbody>
<tr>
<td></td>
<td>More than</td>
<td>Not more than</td>
<td>Dry chemical</td>
<td>Water for protein foam</td>
</tr>
<tr>
<td>A</td>
<td>27 m</td>
<td>225 kg</td>
<td>0</td>
<td>1</td>
</tr>
<tr>
<td>B</td>
<td>27 m</td>
<td>225 kg</td>
<td>5 700 L</td>
<td>1</td>
</tr>
<tr>
<td>C</td>
<td>38 m</td>
<td>225 kg</td>
<td>5 700 L</td>
<td>2</td>
</tr>
<tr>
<td>D</td>
<td>48 m</td>
<td>225 kg</td>
<td>5 700 L</td>
<td>3</td>
</tr>
<tr>
<td>E</td>
<td>60 m</td>
<td>225 kg</td>
<td>11 400 L</td>
<td>3</td>
</tr>
</tbody>
</table>

\(^1\) Truck size

1 900 L but less than 7 600 L: Discharge rate (Litres per minute): at least 1 900 but not more than 3 800.

7 600 L or greater: at least 2 280 but not more than 4 560.
INTERNATIONAL STANDARDS AND RECOMMENDED PRACTICES

AERODROMES

ANNEX 14
TO THE CONVENTION ON INTERNATIONAL CIVIL AVIATION

VOLUME II
HELIPORTS

SECOND EDITION — JULY 1995

This edition incorporates all amendments to Annex 14, Volume II, adopted by the Council prior to 14 March 1995 and supersedes on 9 November 1995 all previous editions of Annex 14, Volume II.

For information regarding the applicability of the Standards and Recommended Practices, see Foreword and the relevant clauses in each Chapter

INTERNATIONAL CIVIL AVIATION ORGANIZATION
AMENDMENTS

The issue of amendments is announced regularly in the *ICAO Journal* and in the monthly *Supplement to the Catalogue of ICAO Publications and Audio-visual Training Aids*, which holders of this publication should consult. The space below is provided to keep a record of such amendments.

**RECORD OF AMENDMENTS AND CORRIGENDA**

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**CORRIGENDA**

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# TABLE OF CONTENTS

<table>
<thead>
<tr>
<th>Chapter</th>
<th>Title</th>
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</tr>
</thead>
<tbody>
<tr>
<td>Abbreviations and symbols; manuals</td>
<td></td>
<td>(v)</td>
</tr>
<tr>
<td>FOREWORD</td>
<td></td>
<td>(vii)</td>
</tr>
<tr>
<td>CHAPTER 1. General</td>
<td></td>
<td>1</td>
</tr>
<tr>
<td>1.1 Definitions</td>
<td></td>
<td>1</td>
</tr>
<tr>
<td>1.2 Applicability</td>
<td></td>
<td>2</td>
</tr>
<tr>
<td>CHAPTER 2. Heliport data</td>
<td></td>
<td>3</td>
</tr>
<tr>
<td>2.1 Aeronautical data</td>
<td></td>
<td>3</td>
</tr>
<tr>
<td>2.2 Heliport reference point</td>
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<tr>
<td>2.3 Heliport elevation</td>
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<td>4</td>
</tr>
<tr>
<td>2.4 Heliport dimensions and related information</td>
<td></td>
<td>4</td>
</tr>
<tr>
<td>2.5 Declared distances</td>
<td></td>
<td>4</td>
</tr>
<tr>
<td>2.6 Co-ordination between aeronautical information services and heliport authorities</td>
<td></td>
<td>4</td>
</tr>
<tr>
<td>CHAPTER 3. Physical characteristics</td>
<td></td>
<td>5</td>
</tr>
<tr>
<td>3.1 Surface-level heliports</td>
<td></td>
<td>5</td>
</tr>
<tr>
<td>— Final approach and take-off areas</td>
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<td>5</td>
</tr>
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<td>— Touchdown and lift-off areas</td>
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<tr>
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<td>8</td>
</tr>
<tr>
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<td></td>
<td>8</td>
</tr>
<tr>
<td>3.3 Helidecks</td>
<td></td>
<td>9</td>
</tr>
<tr>
<td>— Final approach and take-off area and touchdown and lift-off area</td>
<td></td>
<td>9</td>
</tr>
<tr>
<td>3.4 Shipboard heliports</td>
<td></td>
<td>9</td>
</tr>
<tr>
<td>— Final approach and take-off area and touchdown and lift-off area</td>
<td></td>
<td>9</td>
</tr>
<tr>
<td>CHAPTER 4. Obstacle restriction and removal</td>
<td></td>
<td>10</td>
</tr>
<tr>
<td>4.1 Obstacle limitation surfaces and sectors</td>
<td></td>
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</tr>
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<td>13</td>
</tr>
<tr>
<td>CHAPTER 5. Visual aids</td>
<td></td>
<td>30</td>
</tr>
<tr>
<td>5.1 Indicators</td>
<td></td>
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</tr>
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<td></td>
<td>30</td>
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<tr>
<td>5.2 Markings and markers</td>
<td></td>
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</tr>
<tr>
<td>5.2.1 Winching area marking</td>
<td></td>
<td>30</td>
</tr>
<tr>
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<td></td>
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<tr>
<td>5.2.3 Maximum allowable mass marking</td>
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<td>5.2.4 Final approach and take-off area marking or marker</td>
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</tr>
<tr>
<td>5.2.6 Aiming point marking</td>
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<td>32</td>
</tr>
<tr>
<td>5.2.7 Touchdown and lift-off area marking</td>
<td></td>
<td>34</td>
</tr>
<tr>
<td>5.2.8 Touchdown marking</td>
<td></td>
<td>35</td>
</tr>
<tr>
<td>5.2.9 Heliport name marking</td>
<td></td>
<td>35</td>
</tr>
<tr>
<td>5.2.10 Helideck obstacle-free sector marking</td>
<td></td>
<td>35</td>
</tr>
<tr>
<td>5.2.11 Marking for taxiways</td>
<td></td>
<td>35</td>
</tr>
<tr>
<td>5.2.12 Air taxiway markers</td>
<td></td>
<td>36</td>
</tr>
<tr>
<td>5.2.13 Air transit route markers</td>
<td></td>
<td>36</td>
</tr>
<tr>
<td>5.3 Lights</td>
<td></td>
<td>38</td>
</tr>
<tr>
<td>5.3.1 General</td>
<td></td>
<td>38</td>
</tr>
<tr>
<td>5.3.2 Heliport beacon</td>
<td></td>
<td>38</td>
</tr>
<tr>
<td>5.3.3 Approach lighting system</td>
<td></td>
<td>39</td>
</tr>
<tr>
<td>5.3.4 Visual alignment guidance system</td>
<td></td>
<td>39</td>
</tr>
<tr>
<td>5.3.5 Visual approach slope indicator</td>
<td></td>
<td>42</td>
</tr>
<tr>
<td>5.3.6 Final approach and take-off area lights</td>
<td></td>
<td>45</td>
</tr>
<tr>
<td>5.3.7 Aiming point lights</td>
<td></td>
<td>46</td>
</tr>
<tr>
<td>5.3.8 Touchdown and lift-off area lighting system</td>
<td></td>
<td>46</td>
</tr>
<tr>
<td>5.3.9 Winching area floodlighting</td>
<td></td>
<td>47</td>
</tr>
<tr>
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<td>Page</td>
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</tr>
<tr>
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<tr>
<td>5.3.10 Taxiway lights</td>
<td>47</td>
<td></td>
</tr>
<tr>
<td>5.3.11 Visual aids for denoting obstacles</td>
<td>47</td>
<td></td>
</tr>
<tr>
<td>5.3.12 Floodlighting of obstacles</td>
<td>47</td>
<td></td>
</tr>
<tr>
<td>6.1 Rescue and fire fighting</td>
<td>49</td>
<td></td>
</tr>
<tr>
<td>— General</td>
<td>49</td>
<td></td>
</tr>
<tr>
<td>CHAPTER 6. Heliport services</td>
<td>49</td>
<td></td>
</tr>
<tr>
<td>APPENDIX 1. Aeronautical data quality requirements</td>
<td>51</td>
<td></td>
</tr>
</tbody>
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ABBREVIATIONS AND SYMBOLS
(used in Annex 14, Volume II)

Abbreviations

Abbreviations

<table>
<thead>
<tr>
<th>Abbreviation</th>
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<tbody>
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<td>cd</td>
<td>Candela</td>
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<tr>
<td>cm</td>
<td>Centimeter</td>
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<tr>
<td>D</td>
<td>Helicopter largest over-all dimension</td>
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<td>FATO</td>
<td>Final approach and take-off area</td>
</tr>
<tr>
<td>ft</td>
<td>Foot</td>
</tr>
<tr>
<td>HAPI</td>
<td>Helicopter approach path indicator</td>
</tr>
<tr>
<td>Hz</td>
<td>Hertz</td>
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<td>IMC</td>
<td>Instrument meteorological conditions</td>
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<tr>
<td>kg</td>
<td>Kilogram</td>
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<tr>
<td>km/h</td>
<td>Kilometre per hour</td>
</tr>
<tr>
<td>kt</td>
<td>Knot</td>
</tr>
<tr>
<td>L</td>
<td>Litre</td>
</tr>
<tr>
<td>LDAH</td>
<td>Landing distance available</td>
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Symbols

Symbols

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<td>°</td>
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<td>=</td>
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<td>%</td>
<td>Percentage</td>
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<tr>
<td>±</td>
<td>Plus or minus</td>
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MANUALS
(related to the specifications of this Annex)

Aerodrome Design Manual (Doc 9157)
Part 1 — Runways
Part 2 — Taxiways, Aprons and Holding Bays
Part 3 — Pavements
Part 4 — Visual Aids
Part 5 — Electrical Systems

Airport Planning Manual (Doc 9184)
Part 1 — Master Planning
Part 2 — Land Use and Environmental Control
Part 3 — Guidelines for Consultant/Construction Services

Airport Services Manual (Doc 9137)
Part 1 — Rescue and Fire Fighting
Part 2 — Pavement Surface Conditions
Part 3 — Bird Control and Reduction

Heliport Manual (Doc 9261)

Stolport Manual (Doc 9150)

Manual on the ICAO Bird Strike Information System (IBIS)
(Doc 9332)

Manual of Surface Movement Guidance and Control Systems (SMGCS) (Doc 9476)
Historical background

Standards and Recommended Practices for aerodromes were first adopted by the Council on 29 May 1951 pursuant to the provisions of Article 37 of the Convention on International Civil Aviation (Chicago 1944) and designated as Annex 14 to the Convention. The document containing these Standards and Recommended Practices is now designated as Annex 14, Volume I to the Convention. In general, Volume I addresses planning, design and operations of aerodromes but is not specifically applicable to heliports.

Therefore, Volume II is being introduced as a means of including provisions for heliports. Proposals for comprehensive Standards and Recommended Practices covering all aspects of heliport planning, design and operations have been developed with the assistance of the ANC Visual Aids Panel and the ANC Helicopter Operations Panel.

Table A shows the origin of the provisions in this volume, together with a list of the principal subjects involved and the dates on which the Annex was adopted by the Council, when it became effective and when it became applicable.

Action by Contracting States

Notification of differences. The attention of Contracting States is drawn to the obligation imposed by Article 38 of the Convention by which Contracting States are required to notify the Organization of any differences between their national regulations and practices and the International Standards contained in this Annex and any amendments thereto. Contracting States are invited to extend such notification to any differences from Recommended Practices contained in this Annex and any amendments thereto, when the notification of such differences is important for the safety of air navigation. Further, Contracting States are invited to keep the Organization currently informed of any differences which may subsequently occur, or of the withdrawal of any differences previously notified. A specified request for notification of differences will be sent to Contracting States immediately after the adoption of each amendment to this Annex.

The attention of States is also drawn to the provisions of Annex 15 related to the publication of differences between their national regulations and practices and the related ICAO Standards and Recommended Practices through the Aeronautical Information Service, in addition to the obligation of States under Article 38 of the Convention.

Promulgation of information. The establishment and withdrawal of and changes to facilities, services and procedures affecting aircraft operations provided in accordance with the Standards and Recommended Practices specified in this Annex should be notified and take effect in accordance with the provisions of Annex 15.

Status of Annex components

An Annex is made up of the following component parts, not all of which, however, are necessarily found in every Annex; they have the status indicated:

1. Material comprising the Annex proper:
   a) Standards and Recommended Practices adopted by the Council under the provisions of the Convention. They are defined as follows:

   Standard: Any specification for physical characteristics, configuration, matériel, performance, personnel or procedure, the uniform application of which is recognized as necessary for the safety or regularity of international air navigation and to which Contracting States will conform in accordance with the Convention; in the event of impossibility of compliance, notification to the Council is compulsory under Article 38.

   Recommended Practice: Any specification for physical characteristics, configuration, matériel, performance, personnel or procedure, the uniform application of which is recognized as desirable in the interest of safety, regularity or efficiency of international air navigation, and to which Contracting States will endeavour to conform in accordance with the Convention.

   b) Appendices comprising material grouped separately for convenience but forming part of the Standards and Recommended Practices adopted by the Council.

   c) Definitions of terms used in the Standards and Recommended Practices which are not self-
explanatory in that they do not have accepted dictionary meanings. A definition does not have independent status but is an essential part of each Standard and Recommended Practice in which the term is used, since a change in the meaning of the term would affect the specifications.

d) **Tables and Figures** which add to or illustrate a Standard or Recommended Practice and which are referred to therein, form part of the associated Standard or Recommended Practice and have the same status.

2. — **Material approved by the Council for publication in association with the Standards and Recommended Practices:**

a) **Forewords** comprising historical and explanatory material based on the action of the Council and including an explanation of the obligations of States with regard to the application of the Standards and Recommended Practices ensuing from the Convention and the Resolution of Adoption.

b) **Introductions** comprising explanatory material introduced at the beginning of parts, chapters or sections of the Annex to assist in the understanding of the application of the text.

c) **Notes** included in the text, where appropriate, to give factual information or references bearing on the Standards or Recommended Practices in question, but not constituting part of the Standards or Recommended Practices.

d) **Attachments** comprising material supplementary to the Standards and Recommended Practices, or included as a guide to their application.

### Selection of language

This Annex has been adopted in four languages — English, French, Russian and Spanish. Each Contracting State is requested to select one of those texts for the purpose of national implementation and for other effects provided for in the Convention, either through direct use or through translation into its own national language, and to notify the Organization accordingly.

### Editorial practices

The following practice has been adhered to in order to indicate at a glance the status of each statement: **Standards** have been printed in light face roman; **Recommended Practices** have been printed in light face italics, the status being indicated by the prefix **Recommendation**; **Notes** have been printed in light face italics, the status being indicated by the prefix **Note**.

The following editorial practice has been followed in the writing of specifications: for Standards the operative verb “shall” is used, and for Recommended Practices the operative verb “should” is used.

The units of measurement used in this document are in accordance with the International System of Units (SI) as specified in Annex 5 to the Convention on International Civil Aviation. Where Annex 5 permits the use of non-SI alternative units these are shown in parentheses following the basic units. Where two sets of units are quoted it must not be assumed that the pairs of values are equal and interchangeable. It may, however, be inferred that an equivalent level of safety is achieved when either set of units is used exclusively.

Any reference to a portion of this document, which is identified by a number and/or title, includes all subdivisions of that portion.
Table A. Amendments to Annex 14, Volume II

<table>
<thead>
<tr>
<th>Amendment</th>
<th>Source(s)</th>
<th>Subject(s)</th>
<th>Adopted</th>
<th>Effective</th>
<th>Applicable</th>
</tr>
</thead>
<tbody>
<tr>
<td>1st Edition</td>
<td>Fourth Meeting of the ANC Helicopter Operations Panel; Eleventh meeting of the ANC Visual Aids Panel and Secretariat</td>
<td>Physical characteristics; obstacle limitation surfaces; visual aids for visual meteorological conditions; rescue and fire fighting services.</td>
<td>9 March 1990</td>
<td>30 July 1990</td>
<td>15 November 1990</td>
</tr>
<tr>
<td>1</td>
<td>Twelfth Meeting of the ANC Visual Aids Panel and Secretariat</td>
<td>Standard geodetic reference system (WGS-84); frangibility; visual aids for helicopter non-precision approaches; and visual alignment guidance system.</td>
<td>13 March 1995</td>
<td>24 July 1995</td>
<td>9 November 1995</td>
</tr>
</tbody>
</table>
INTernational Standards
and Recommended Practices

Chapter 1. General

Introductory Note.— Volume II of this Annex contains Standards and Recommended Practices (specifications) that describe the physical characteristics and obstacle limitation surfaces to be provided for at heliports, and certain facilities and technical services normally provided at a heliport. It is not intended that these specifications limit or regulate the operation of an aircraft.

The specifications in this volume modify or complement those in Volume I which, where appropriate, are also applicable to heliports. In other words, where a particular issue is a subject of a specification in this volume that specification will supersede any other specification on that particular issue in Volume I. Throughout this volume the term “heliport” is used; however, it is intended that these specifications also apply to areas for the exclusive use of helicopters at an aerodrome primarily meant for the use of aeroplanes.

It is to be noted that provisions for helicopter flight operations are contained in Annex 6, Part III.

1.1 Definitions

When the following terms are used in this volume they have the meanings given below. Annex 14, Volume I contains definitions for those terms which are used in both volumes.

Accuracy. A degree of conformance between the estimated or measured value and the true value.

Note.— For measured positional data the accuracy is normally expressed in terms of a distance from a stated position within which there is a defined confidence of the true position falling.

Air taxiway. A defined path on the surface established for the air taxiing of helicopters.

Air transit route. A defined path on the surface established for the air transitting of helicopters.

Cyclic redundancy check (CRC). A mathematical algorithm applied to the digital expression of data that provides a level of assurance against loss or alteration of data.

Data quality. A degree or level of confidence that the data provided meets the requirements of the data user in terms of accuracy, resolution and integrity.

Declared distances — heliports.

a) Take-off distance available (TODAH). The length of the final approach and take-off area plus the length of helicopter clearway (if provided) declared available and suitable for helicopters to complete the take-off.

b) Rejected take-off distance available (RTODAH). The length of the final approach and take-off area declared available and suitable for performance class 1 helicopters to complete a rejected take-off.

c) Landing distance available (LDAH). The length of the final approach and take-off area plus any additional area declared available and suitable for helicopters to complete the landing manoeuvre from a defined height.

Elevated heliport. A heliport located on a raised structure on land.

Ellipsoid height (Geodetic height). The height related to the reference ellipsoid, measured along the ellipsoidal outer normal through the point in question.

Final approach and take-off area (FATO). A defined area over which the final phase of the approach manoeuvre to hover or landing is completed and from which the take-off manoeuvre is commenced. Where the FATO is to be used by performance Class 1 helicopters, the defined area includes the rejected take-off area available.

Geodetic datum. A minimum set of parameters required to define location and orientation of the local reference system with respect to the global reference system/frame.

Geoid. The equipotential surface in the gravity field of the Earth which coincides with the undisturbed mean sea level (MSL) extended continuously through the continents.

Note.— The geoid is irregular in shape because of local gravitational disturbances (wind tides, salinity, current, etc.) and the direction of gravity is perpendicular to the geoid at every point.
Annex 14 — Aerodromes

Geoid undulation. The distance of the geoid above (positive) or below (negative) the mathematical reference ellipsoid.

Note.— In respect to the World Geodetic System — 1984 (WGS-84) defined ellipsoid, the difference between the WGS-84 ellipsoidal height and orthometric height represents WGS-84 geoid undulation.

Helicopter ground taxiway. A ground taxiway for use by helicopters only.

Helicopter clearway. A defined area on the ground or water under the control of the appropriate authority, selected and/or prepared as a suitable area over which a performance class 1 helicopter may accelerate and achieve a specific height.

Helicopter stand. An aircraft stand which provides for parking a helicopter and, where air taxiing operations are contemplated, the helicopter touchdown and lift-off.

Helideck. A heliport located on a floating or fixed off-shore structure.

Heliport. An aerodrome or a defined area on a structure intended to be used wholly or in part for the arrival, departure and surface movement of helicopters.

Integrity (aeronautical data). A degree of assurance that an aeronautical data and its value has not been lost nor altered since the data origination or authorized amendment.

Orthometric height. Height of a point related to the geoid, generally presented as an MSL elevation.

Safety area. A defined area on a heliport surrounding the FATO which is free of obstacles, other than those required for air navigation purposes, and intended to reduce the risk of damage to helicopters accidentallydiverging from the FATO.

Station declination. An alignment variation between the zero degree radial of a VOR and true north, determined at the time the VOR station is calibrated.

Surface level heliport. A heliport located on the ground or on the water.

Touchdown and lift-off area (TLOF). A load bearing area on which a helicopter may touch down or lift off.

1.2 Applicability

1.2.1 The interpretation of some of the specifications in the Annex expressly requires the exercising of discretion, the taking of a decision or the performance of a function by the appropriate authority. In other specifications, the expression appropriate authority does not actually appear although its inclusion is implied. In both cases, the responsibility for whatever determination or action is necessary shall rest with the State having jurisdiction over the heliport.

1.2.2 The specifications in Annex 14, Volume II shall apply to all heliports intended to be used by helicopters in international civil aviation. The specifications of Annex 14, Volume I shall apply, where appropriate, to these heliports as well.

1.2.3 Wherever a colour is referred to in this volume, the specifications for that colour given in Appendix 1 to Annex 14, Volume I shall apply.
CHAPTER 2. HELIPORT DATA

2.1 Aeronautical data

2.1.1 Determination and reporting of heliport related aeronautical data shall be in accordance with the accuracy and integrity requirements set forth in Tables 1 to 5 contained in Appendix 1 while taking into account the established quality system procedures. Accuracy requirements for aeronautical data are based upon a 95 per cent confidence level and in that respect, three types of positional data shall be identified: surveyed points (e.g. FATO threshold), calculated points (mathematical calculations from the known surveyed points of points in space, fixes) and declared points (e.g. flight information region boundary points).

Note. — Specifications governing the quality system are given in Annex 15, Chapter 3.

2.1.2 Contracting States shall ensure that integrity of aeronautical data is maintained throughout the data process from survey/origin to the next intended user. Aeronautical data integrity requirements shall be based upon the potential risk resulting from the corruption of data and upon the use to which the data item is put. Consequently, the following classification and data integrity level shall apply:

a) critical data, integrity level $1 \times 10^{-8}$: there is a high probability when using corrupted critical data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe;

b) essential data, integrity level $1 \times 10^{-5}$: there is a low probability when using corrupted essential data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe; and

c) routine data, integrity level $1 \times 10^{-3}$: there is a very low probability when using corrupted routine data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe.

2.1.3 Protection of electronic aeronautical data while stored or in transit shall be totally monitored by the cyclic redundancy check (CRC). To achieve protection of the integrity level of critical and essential aeronautical data as classified in 2.1.2 above, a 32 or 24 bit CRC algorithm shall apply respectively.

2.1.4 Recommendation. — To achieve protection of the integrity level of routine aeronautical data as classified in 2.1.2 above, a 16 bit CRC algorithm should apply.

Note.— Guidance material on the aeronautical data quality requirements (accuracy, resolution, integrity, protection and traceability) is contained in the World Geodetic System — 1984 (WGS-84) Manual (Doc 9674). Supporting material in respect of the provisions of Appendix 1 related to accuracy and integrity of aeronautical data, is contained in RTCA Document DO-201A and European Organization for Civil Aviation Equipment (EUROCAE) Document ED-77, entitled “Industry Requirements for Aeronautical Information”.

2.1.5 Geographical coordinates indicating latitude and longitude shall be determined and reported to the aeronautical information services authority in terms of the World Geodetic System — 1984 (WGS-84) geodetic reference datum, identifying those geographical coordinates which have been transformed into WGS-84 coordinates by mathematical means and whose accuracy of original field work does not meet the requirements in Appendix 1, Table 1.

2.1.6 The order of accuracy of the field work shall be such that the resulting operational navigation data for the phases of flight will be within the maximum deviations, with respect to an appropriate reference frame, as indicated in tables contained in Appendix 1.

2.1.7 In addition to the elevation (referenced to mean sea level) of the specific surveyed ground positions at heliports, geoid undulation (referenced to the WGS-84 ellipsoid) for those positions as indicated in Appendix 1, shall be determined and reported to the aeronautical information services authority.

Note 1.— An appropriate reference frame is that which enables WGS-84 to be realized on a given heliport and with respect to which all coordinate data are related.

Note 2.— Specifications governing the publication of WGS-84 coordinates are given in Annex 4, Chapter 2 and Annex 15, Chapter 3.

2.2 Heliport reference point

2.2.1 A heliport reference point shall be established for a heliport not co-located with an aerodrome.

Note.— When the heliport is co-located with an aerodrome, the established aerodrome reference point serves both aerodrome and heliport.

2.2.2 The heliport reference point shall be located near the initial or planned geometric centre of the heliport and shall normally remain where first established.
2.2.3 The position of the heliport reference point shall be measured and reported to the aeronautical information services authority in degrees, minutes and seconds.

2.3 Heliport elevation

2.3.1 The heliport elevation and geoid undulation at the heliport elevation position shall be measured and reported to the aeronautical information services authority to the accuracy of one-half metre or foot.

2.3.2 For a heliport used by international civil aviation, the elevation of the touchdown and lift-off area and/or the elevation and geoid undulation of each threshold of the final approach and take-off area (where appropriate) shall be measured and reported to the aeronautical information services authority to the accuracy of:

— one-half metre or foot for non-precision approaches; and

— one-quarter metre or foot for precision approaches.

Note.— Geoid undulation must be measured in accordance with the appropriate system of coordinates.

2.4 Heliport dimensions and related information

2.4.1 The following data shall be measured or described, as appropriate, for each facility provided on a heliport:

a) heliport type — surface-level, elevated or helideck;

b) touchdown and lift-off area — dimensions to the nearest metre or foot, slope, surface type, bearing strength in tonnes (1 000 kg);

c) final approach and take-off area — type of FATO, true bearing to one-hundredth of a degree, designation number (where appropriate), length, width to the nearest metre or foot, slope, surface type;

d) safety area — length, width and surface type;

e) helicopter ground taxiway, air taxiway and air transit route — designation, width, surface type;

f) apron — surface type, helicopter stands;

g) clearway — length, ground profile; and

h) visual aids for approach procedures, marking and lighting of FATO, TLOF, taxiways and aprons.

i) distances to the nearest metre or foot of localizer and glide path elements comprising an instrument landing system (ILS) or azimuth and elevation antenna of microwave landing system (MLS) in relation to the associated TLOF or FATO extremities.

2.4.2 The geographical coordinates of the geometric centre of the touchdown and lift-off area and/or of each threshold of the final approach and take-off area (where appropriate) shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds.

2.4.3 The geographical coordinates of appropriate centre line points of helicopter ground taxiways, air taxiways and air transit routes shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds.

2.4.4 The geographical coordinates of each helicopter stand shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds.

2.4.5 The geographical coordinates of significant obstacles on and in the vicinity of a heliport shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and tenths of seconds. In addition, the top elevation rounded up to the nearest metre or foot, type, marking and lighting (if any) of the significant obstacles shall be reported to the aeronautical information services authority.

2.5 Declared distances

The following distances to the nearest metre or foot shall be declared, where relevant, for a heliport:

a) take-off distance available;

b) rejected take-off distance available; and

c) landing distance available.

2.6 Co-ordination between aeronautical information services and heliport authorities

2.6.1 To ensure that aeronautical information services units obtain information to enable them to provide up-to-date pre-flight information and to meet the need for in-flight information, arrangements shall be made between aeronautical information services and heliport authorities responsible for heliport services to report to the responsible aeronautical information services unit, with a minimum of delay:

a) information on heliport conditions;

b) the operational status of associated facilities, services and navigation aids within their area of responsibility;
c) any other information considered to be of operational
significance.

2.6.2 Before introducing changes to the air navigation
system, due account shall be taken by the services responsible
for such changes of the time needed by the aeronautical
information service for the preparation, production and issue
of relevant material for promulgation. To ensure timely
provision of the information to the aeronautical information
service, close co-ordination between those services concerned
is therefore required.

2.6.3 Of a particular importance are changes to
aeronautical information that affect charts and/or computer-
based navigation systems which qualify to be notified by the
aeronautical information regulation and control (AIRAC)
system, as specified in Annex 15, Chapter 6 and Appendix 4.
The predetermined, internationally agreed AIRAC effective
dates in addition to 14 days postage time shall be observed by
the responsible heliport services when submitting the raw
information/data to aeronautical information services.

2.6.4 The heliport services responsible for the provision of
raw aeronautical information/data to the aeronautical
information services shall do that while taking into account
accuracy and integrity requirements for aeronautical data as
specified in Appendix 1 to this Annex.

Note 1.— Specifications for the issue of a NOTAM
and SNOWTAM are contained in Annex 15, Chapter 5,
Appendices 6 and 2 respectively.

Note 2.— The AIRAC information is distributed by the AIS
at least 42 days in advance of the AIRAC effective dates with
the objective of reaching recipients at least 28 days in advance
of the effective date.

Note 3.— The schedule of the predetermined inter-
nationally agreed AIRAC common effective dates at intervals
of 28 days, including 6 November 1997 and guidance for the
AIRAC use are contained in the Aeronautical Information
Services Manual (Doc 8126, Chapter 3, 3.1.1 and Chapter 4,
4.4).
CHAPTER 3. PHYSICAL CHARACTERISTICS

3.1 Surface-level heliports

Note.— The following specifications are for surface-level land heliports (except where specified).

Final approach and take-off areas

3.1.1 A surface-level heliport shall be provided with at least one FATO.

Note.— A FATO may be located on or near a runway strip or taxiway strip.

3.1.2 The dimensions of a FATO shall be:

a) for a heliport intended to be used by performance class 1 helicopters, as prescribed in the helicopter flight manual except that, in the absence of width specifications, the width shall be not less than 1.5 times the over-all length/width, whichever is greater, of the longest/widest helicopter the heliport is intended to serve;

b) for a water heliport intended to be used by performance class 1 helicopters, as prescribed in a) above, plus 10 per cent;

c) for a heliport intended to be used by performance class 2 and 3 helicopters, of sufficient size and shape to contain an area within which can be drawn a circle of diameter not less than 1.5 times the over-all length/width, whichever is greater, of the longest/widest helicopter the heliport is intended to serve; and

d) for a water heliport intended to be used by performance class 2 and 3 helicopters, of sufficient size to contain an area within which can be drawn a circle of diameter not less than two times the over-all length/width, whichever is greater, of the longest/widest helicopter the heliport is intended to serve.

Note.— Local conditions, such as elevation and temperature, may need to be considered when determining the size of a FATO. Guidance is given in the Heliport Manual.

3.1.3 The over-all slope in any direction on the FATO shall not exceed 3 per cent. No portion of a FATO shall have a local slope exceeding:

a) 5 per cent where the heliport is intended to be used by performance class 1 helicopters; and

b) 7 per cent where the heliport is intended to be used by performance class 2 and 3 helicopters.

3.1.4 The surface of the FATO shall:

a) be resistant to the effects of rotor downwash;

b) be free of irregularities that would adversely affect the take-off or landing of helicopters; and

c) have bearing strength sufficient to accommodate a rejected take-off by performance class 1 helicopters.

3.1.5 Recommendation.— The FATO should provide ground effect.

Helicopter clearways

3.1.6 When it is necessary to provide a helicopter clearway, it shall be located beyond the upwind end of the rejected take-off area available.

3.1.7 Recommendation.— The width of a helicopter clearway should not be less than that of the associated safety area.

3.1.8 Recommendation.— The ground in a helicopter clearway should not project above a plane having an upward slope of 3 per cent, the lower limit of this plane being a horizontal line which is located on the periphery of the FATO.

3.1.9 Recommendation.— An object situated on a helicopter clearway which may endanger helicopters in the air should be regarded as an obstacle and should be removed.

Touchdown and lift-off areas

3.1.10 At least one touchdown and lift-off area shall be provided at a heliport.

Note.— The touchdown and lift-off area may or may not be located within the FATO.

3.1.11 The touchdown and lift-off area (TLOF) shall be of sufficient size to contain a circle of diameter 1.5 times the length or width of the undercarriage, whichever is the greater, of the largest helicopter the area is intended to serve.

Note.— A touchdown and lift-off area may be any shape.
Annex 14 — Aerodromes

3.1.12 Slopes on a touchdown and lift-off area shall be sufficient to prevent accumulation of water on the surface of the area, but shall not exceed 2 per cent in any direction.

3.1.13 A touchdown and lift-off area shall be capable of withstanding the traffic of helicopters that the area is intended to serve.

Safety areas

3.1.14 A FATO shall be surrounded by a safety area.

3.1.15 A safety area surrounding a FATO intended to be used in visual meteorological conditions (VMC) shall extend outwards from the periphery of the FATO for a distance of at least 3 m or 0.25 times the over-all length/width, whichever is greater, of the longest/widest helicopter the area is intended to serve.

3.1.16 A safety area surrounding a FATO intended to be used by helicopter operations in instrument meteorological conditions (IMC) shall extend:

a) laterally to a distance of at least 45 m on each side of the centre line; and

b) longitudinally to a distance of at least 60 m beyond the ends of the FATO.

Note.— See Figure 3-1.

3.1.17 No fixed object shall be permitted on a safety area, except for frangible objects, which, because of their function, must be located on the area. No mobile object shall be permitted on a safety area during helicopter operations.

3.1.18 Objects whose functions require them to be located on the safety area shall not exceed a height of 25 cm when located along the edge of the FATO nor penetrate a plane originating at a height of 25 cm above the edge of the FATO and sloping upwards and outwards from the edge of the FATO at a gradient of 5 per cent.

3.1.19 The surface of the safety area shall not exceed an upward slope of 4 per cent outwards from the edge of the FATO.

3.1.20 The surface of the safety area shall be treated to prevent flying debris caused by rotor downwash.

3.1.21 The surface of the safety area abutting the FATO shall be continuous with the FATO and be capable of supporting, without structural damage, the helicopters that the heliport is intended to serve.

Helicopter ground taxiways

Note.— A helicopter ground taxiway is intended to permit the surface movement of a wheeled helicopter under its own power. The specifications for taxiways, taxiway shoulders and taxiway strips included in Annex 14, Volume I are equally applicable to heliports as modified below. When a taxiway is intended for use by aeroplanes and helicopters, the provisions for taxiways and helicopter ground taxiways will be examined and the more stringent requirements will be applied.

3.1.22 The width of a helicopter ground taxiway shall not be less than:

<table>
<thead>
<tr>
<th>Helicopter main gear span</th>
<th>Helicopter ground taxiway width</th>
</tr>
</thead>
<tbody>
<tr>
<td>Up to but not including 4.5 m</td>
<td>7.5 m</td>
</tr>
<tr>
<td>4.5 m up to but not including 6 m</td>
<td>10.5 m</td>
</tr>
<tr>
<td>6 m up to but not including 10 m</td>
<td>15 m</td>
</tr>
<tr>
<td>10 m and over</td>
<td>20 m</td>
</tr>
</tbody>
</table>

3.1.23 The separation distance between a helicopter ground taxiway and another helicopter ground taxiway, an air taxiway, an object or helicopter stand shall not be less than the appropriate dimension specified in Table 3-1.

3.1.24 The longitudinal slope of a helicopter ground taxiway shall not exceed 3 per cent.

3.1.25 Recommendation.— A helicopter ground taxiway should be capable of withstanding the traffic of helicopters that the helicopter ground taxiway is intended to serve.

Figure 3-1. Safety area for instrument FATO

9/11/95
3.1.26 **Recommendation.**—A helicopter ground taxiway should be provided with shoulders which extend symmetrically on each side of the helicopter ground taxiway for at least one-half the greatest over-all width of the helicopters that the helicopter ground taxiway is intended to serve.

3.1.27 The helicopter ground taxiway and its shoulder shall provide rapid drainage but the helicopter ground taxiway transverse slope shall not exceed 2 per cent.

3.1.28 **Recommendation.**—The surface of a helicopter ground taxiway shoulder should be resistant to the effect of rotor downwash.

### Air taxiways

**Note.**—An air taxiway is intended to permit the movement of a helicopter above the surface at a height normally associated with ground effect and at groundspeed less than 37 km/h (20 kt).

3.1.29 The width of an air taxiway shall be at least two times the greatest over-all width of the helicopters that the air taxiway is intended to serve.

3.1.30 The surface of an air taxiway shall:

a) be resistant to the effects of rotor downwash; and

b) be suitable for emergency landings.

3.1.31 **Recommendation.**—The surface of an air taxiway should provide ground effect.

3.1.32 **Recommendation.**—The transverse slope of the surface of an air taxiway should not exceed 10 per cent and the longitudinal slope should not exceed 7 per cent. In any event, the slopes should not exceed the slope landing limitations of the helicopters the air taxiway is intended to serve.

### Air transit route

**Note.**—An air transit route is intended to permit the movement of a helicopter above the surface, normally at heights not above 30 m (100 ft) above ground level and at ground speeds exceeding 37 km/h (20 kt).

3.1.33 The separation distance between an air taxiway and another air taxiway, a helicopter ground taxiway, an object or a helicopter stand shall not be less than the appropriate dimension in Table 3-1.

3.1.34 The width of an air transit route shall not be less than:

a) 7.0 times RD when the air transit route is intended for use by day only; and

b) 10.0 times RD when the air transit route is intended for use at night;

when RD is the diameter of the largest rotor of the helicopters that the air transit route is intended to serve.

3.1.35 Any variation in the direction of the centre line of an air transit route shall not exceed 120° and be designed so as not to necessitate a turn of radius less than 270 m.

**Note.**—It is intended that air transit routes be selected so as to permit autorotative or one-engine-in-operative landings such that, as a minimum requirement, injury to persons on the ground or water, or damage to property are minimized.

### Aprons

**Note.**—The specifications for aprons included in Chapter 3 of Annex 14, Volume I are equally applicable to heliports as modified below.

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### Table 3-1. Helicopter ground taxiway and air taxiway separation distances

(expressed in multiples of greatest over-all width of helicopter with rotor turning)

<table>
<thead>
<tr>
<th>Facility</th>
<th>Helicopter ground taxiway</th>
<th>Air taxiway</th>
<th>Object</th>
<th>Helicopter stand</th>
</tr>
</thead>
<tbody>
<tr>
<td>Helicopter ground taxiway</td>
<td>2 (between edges)</td>
<td>4 (between centre lines)</td>
<td>1 (edge to object)</td>
<td>2 (between edges)</td>
</tr>
<tr>
<td>Air taxiway</td>
<td>4 (between centre lines)</td>
<td>4 (between centre lines)</td>
<td>1/2 (centre line to object)</td>
<td>4 (centre line to edge)</td>
</tr>
</tbody>
</table>
Annex 14 — Aerodromes

3.1.36 The slope in any direction on a helicopter stand shall not exceed 2 per cent.

3.1.37 The minimum clearance between a helicopter using a helicopter stand and an object or any aircraft on another stand shall not be less than half the greatest over-all width of the helicopters that the stand is intended to serve.

Note.— Where simultaneous hover operations are to be provided for, the separation distances specified in Table 3-1 between two air taxiways are to be applied.

3.1.38 A helicopter stand shall be of sufficient size to contain a circle of diameter of at least the largest over-all dimension of the largest helicopter the stand is expected to serve.

Location of a final approach and take-off area in relation to a runway or taxiway

3.1.39 Where a FATO is located near a runway or taxiway, and simultaneous VMC operations are planned, the separation distance between the edge of a runway or taxiway and the edge of a FATO shall not be less than the appropriate dimension in Table 3-2.

3.1.40 Recommendation.— A FATO should not be located:

a) near taxiway intersections or holding points where jet engine efflux is likely to cause high turbulence; or

b) near areas where aeroplane vortex wake generation is likely to exist.

Table 3-2. FATO minimum separation distance

<table>
<thead>
<tr>
<th>If aeroplane mass and/or helicopter mass are</th>
<th>Distance between FATO edge and runway edge or taxiway edge</th>
</tr>
</thead>
<tbody>
<tr>
<td>up to but not including 2 720 kg</td>
<td>60 m</td>
</tr>
<tr>
<td>2 720 kg up to but not including 5 760 kg</td>
<td>120 m</td>
</tr>
<tr>
<td>5 760 kg up to but not including 100 000 kg</td>
<td>180 m</td>
</tr>
<tr>
<td>100 000 kg and over</td>
<td>250 m</td>
</tr>
</tbody>
</table>

3.2 Elevated heliports

Final approach and take-off area and touchdown and lift-off area

Note.— On elevated heliports it is presumed that the FATO and the touchdown and lift-off area will be coincidental.

3.2.1 An elevated heliport shall be provided with at least one FATO.

3.2.2 The dimensions of the FATO shall be:

a) for a heliport intended to be used by performance class 1 helicopters, as prescribed in the helicopter flight manual except that, in the absence of width specifications, the width shall be not less than 1.5 times the over-all length/width, whichever is greater, of the longest/widest helicopter the heliport is intended to serve; and

b) for a heliport intended to be used by performance class 2 helicopters, of sufficient size and shape to contain an area within which can be drawn a circle of diameter not less than 1.5 times the over-all length/width, whichever is greater, of the longest/widest helicopter the heliport is intended to serve.

3.2.3 Recommendation.— The slope requirements for elevated heliports should conform to the requirements for surface level heliports specified in 3.1.3.

3.2.4 The FATO shall be capable of withstanding the traffic of helicopters the heliport is intended to serve. Design considerations shall take into account additional loading resulting from the presence of personnel, snow, freight, refuelling, fire fighting equipment, etc.

Note.— Guidance on structural design for elevated heliports is given in the Heliport Manual.

Safety area

3.2.5 The FATO shall be surrounded by a safety area.

3.2.6 The safety area shall extend outwards from the periphery of the FATO for a distance of at least 3 m or 0.25 times the over-all length/width, whichever is greater, of the longest/widest helicopter intended to use the elevated heliport.

3.2.7 No fixed object shall be permitted on a safety area, except for frangible objects, which, because of their function, must be located on the area. No mobile object shall be permitted on a safety area during helicopter operations.
Chapter 3

3.2.8 Objects whose function require them to be located on the safety area shall not exceed a height of 25 cm when located along the edge of the FATO nor penetrate a plane originating at a height of 25 cm above the edge of the FATO and sloping upwards and outwards from the edge of the FATO at a gradient of 5 per cent.

3.2.9 The surface of the safety area shall not exceed an upward slope of 4 per cent outwards from the edge of the FATO.

3.2.10 The surface of the safety area abutting the FATO shall be continuous with the FATO and be capable of supporting, without structural damage, the helicopters that the heliport is intended to serve.

3.3 Helidecks

Note.— The following specifications are for helidecks located on structures engaged in such activities as mineral exploitation, research or construction. See 3.4 for shipboard heliport provisions.

Final approach and take-off area and touchdown and lift-off area

Note.— On helidecks it is presumed that the FATO and the touchdown and lift-off area will be coincidental. Guidance on the effects of airflow direction and turbulence, prevailing wind velocity and high temperatures from gas turbine exhausts or flare radiated heat on the location of the FATO is given in the Heliport Manual.

3.3.1 A helideck shall be provided with at least one FATO.

3.3.2 A FATO may be any shape but shall, for a single main rotor helicopter or side-by-side twin main rotor helicopter, be of sufficient size to contain an area within which can be drawn a circle of diameter not less than 1.0 times D of the largest helicopter the helideck is intended to serve, where D is the largest dimension of the helicopter when the rotors are turning.

3.3.3 Where omnidirectional landings by helicopters having tandem main rotors are intended, the FATO shall be of sufficient size to contain an area within which can be drawn a circle of diameter not less than 0.9 times the distance across the rotors in a fore and aft line. Where these provisions cannot be met, the FATO may be in the form of a rectangle with a small side not less than 0.75 D and a long side not less than 0.9 D but within this rectangle, bi-directional landings only will be permitted in the direction of the 0.9 D dimension.

3.3.4 No fixed object shall be permitted around the edge of the FATO except for frangible objects, which, because of their function, must be located thereon.

3.3.5 Objects whose function require them to be located on the edge of the FATO shall not exceed a height of 25 cm.

3.3.6 The surface of the FATO shall be skid-resistant to both helicopters and persons and be sloped to prevent pooling of liquids. Where the helideck is constructed in the form of a grating, the underdeck design shall be such that ground effect is not reduced.

Note.— Guidance on rendering the surface of the FATO skid-resistant is contained in the Heliport Manual.

3.4 Shipboard heliports

3.4.1 When helicopter operating areas are provided in the bow or stern of a ship or are purpose-built above the ship’s structure, they shall be regarded as helidecks and the criteria given in 3.3 shall apply.

Final approach and take-off area and touchdown and lift-off area

Note.— On heliports located in other areas of ships it is presumed that the FATO and the touchdown and lift-off area will be coincidental. Guidance on the effects of airflow direction and turbulence, prevailing wind velocity and high temperature from gas turbine exhausts or flare radiated heat on the location of the FATO is given in the Heliport Manual.

3.4.2 Shipboard heliports shall be provided with at least one FATO.

3.4.3 A FATO on a shipboard heliport shall be circular and shall be of sufficient size to contain a diameter not less than 1.0 times D of the largest helicopter the heliport is intended to serve where D is the largest dimension of the helicopter when the rotors are turning.

3.4.4 The surface of the FATO shall be skid-resistant to both helicopters and persons.
CHAPTER 4. OBSTACLE RESTRICTION AND REMOVAL

Note.— The objectives of the specifications in this chapter are to define the airspace around heliports to be maintained free from obstacles so as to permit the intended helicopter operations at the heliports to be conducted safely and to prevent the heliports becoming unusable by the growth of obstacles around them. This is achieved by establishing a series of obstacle limitation surfaces that define the limits to which objects may project into the airspace.

4.1 Obstacle limitation surfaces and sectors

Approach surface

4.1.1 Description. An inclined plane or a combination of planes sloping upwards from the end of the safety area and centred on a line passing through the centre of the FATO.

Note.— See Figure 4-1.

4.1.2 Characteristics. The limits of an approach surface shall comprise:

a) an inner edge horizontal and equal in length to the minimum specified width of the FATO plus the safety area, perpendicular to the centre line of the approach surface and located at the outer edge of the safety area;

b) two side edges originating at the ends of the inner edge and:

1) for other than a precision approach FATO, diverging uniformly at a specified rate from the vertical plane containing the centre line of the FATO;

2) for a precision approach FATO, diverging uniformly at a specified rate from the vertical plane containing the centre line of the FATO, to a specified height above FATO, and then diverging uniformly at a specified rate to a specified final width and continuing thereafter at that width for the remaining length of the approach surface; and

c) an outer edge horizontal and perpendicular to the centre line of the approach surface and at a specified height above the elevation of the FATO.

4.1.3 The elevation of the inner edge shall be the elevation of the safety area at the point on the inner edge that is intersected by the centre line of the approach surface.

4.1.4 The slope(s) of the approach surface shall be measured in the vertical plane containing the centre line of the surface.

Note.— For heliports used by performance class 2 and 3 helicopters, it is intended that approach paths be selected so as to permit safe forced landing or one-engine-inoperative landings such that, as a minimum requirement, injury to persons on the ground or water or damage to property are minimized. Provisions for forced landing areas are expected to minimize risk of injury to the occupants of the helicopter. The most critical helicopter type for which the heliport is intended and the ambient conditions will be factors in determining the suitability of such areas.

Transitional surface

4.1.5 Description. A complex surface along the side of the safety area and part of the side of the approach surface, that slopes upwards and outwards to the inner horizontal surface or a predetermined height.

Note.— See Figure 4-1.

4.1.6 Characteristics. The limits of a transitional surface shall comprise:

a) a lower edge beginning at the intersection of the side of the approach surface with the inner horizontal surface, or beginning at a specified height above the lower edge when an inner horizontal surface is not provided, and extending down the side of the approach surface to the inner edge of the approach surface and from there along the length of the side of the safety area parallel to the centre line of the FATO; and

b) an upper edge located in the plane of the inner horizontal surface, or at a specified height above the lower edge when an inner horizontal surface is not provided.

4.1.7 The elevation of a point on the lower edge shall be:

a) along the side of the approach surface — equal to the elevation of the approach surface at that point; and

b) along the safety area — equal to the elevation of the centre line of the FATO opposite that point.

Note.— As a result of b) the transitional surface along the safety area will be curved if the profile of the FATO is curved.
or a plane if the profile is a straight line. The intersection of the transitional surface with the inner horizontal surface, or upper edge when an inner horizontal surface is not provided, will also be a curved or a straight line depending on the profile of the FATO.

4.1.8 The slope of the transitional surface shall be measured in a vertical plane at right angles to the centre line of the FATO.

Inner horizontal surface

Note.— The intent of the inner horizontal surface is to allow safe visual manoeuvring.

4.1.9 Description. A circular surface located in a horizontal plane above a FATO and its environs.

Note.— See Figure 4-1.

4.1.10 Characteristics. The radius of the inner horizontal surface shall be measured from the mid-point of the FATO.

4.1.11 The height of the inner horizontal surface shall be measured above an elevation datum established for such purpose.

Note.— Guidance on determining the elevation datum is contained in the Heliport Manual.

Conical surface

4.1.12 Description. A surface sloping upwards and outwards from the periphery of the inner horizontal surface, or from the outer limit of the transitional surface if an inner horizontal surface is not provided.

Note.— See Figure 4-1.

4.1.13 Characteristics. The limits of the conical surface shall comprise:

a) a lower edge coincident with the periphery of the inner horizontal surface, or outer limit of the transitional surface if an inner horizontal surface is not provided; and

b) an upper edge located at a specified height above the inner horizontal surface, or above the elevation of the lowest end of the FATO if an inner horizontal surface is not provided.

4.1.14 The slope of the conical surface shall be measured above the horizontal.

Take-off climb surface

4.1.15 Description. An inclined plane, a combination of planes or, when a turn is involved, a complex surface sloping upwards from the end of the safety area and centred on a line passing through the centre of the FATO.

Note.— See Figure 4-1.

4.1.16 Characteristics. The limits of a take-off climb surface shall comprise:

a) an inner edge horizontal and equal in length to the minimum specified width of the FATO plus the safety area, perpendicular to the centre line of the take-off climb surface and located at the outer edge of the safety area or clearway;

b) two side edges originating at the ends of the inner edge and diverging uniformly at a specified rate from the vertical plane containing the centre line of the FATO; and

c) an outer edge horizontal and perpendicular to the centre line of the take-off climb surface and at a specified height above the elevation of the FATO.

4.1.17 The elevation of the inner edge shall be the elevation of the safety area at the point on the inner edge that is intersected by the centre line of the take-off climb surface, except that when a clearway is provided, the elevation shall be equal to the highest point on the ground on the centre line of the clearway.

4.1.18 In the case of a straight take-off climb surface, the slope shall be measured in the vertical plane containing the centre line of the surface.

4.1.19 In the case of a take-off climb surface involving a turn, the surface shall be a complex surface containing the horizontal normals to its centre line and the slope of the centre line shall be the same as that for a straight take-off climb surface. That portion of the surface between the inner edge and 30 m above the inner edge shall be straight.

4.1.20 Any variation in the direction of the centre line of a take-off climb surface shall be designed so as not to necessitate a turn of radius less than 270 m.

Note.— For heliports used by performance class 2 and 3 helicopters, it is intended that departure paths be selected so as to permit safe forced landings or one-engine-inoperative landings such that, as a minimum requirement, injury to persons on the ground or water or damage to property are minimized. Provisions for forced landing areas are expected to minimize risk of injury to the occupants of the helicopter. The most critical helicopter type for which the heliport is intended and the ambient conditions will be factors in determining the suitability of such areas.
Annex 14 — Aerodromes

Obstacle-free sector/surface — helidecks

4.1.21 Description. A complex surface originating at a reference point on the edge of the FATO of a helideck and extending to a specified distance.

4.1.22 Characteristics. An obstacle-free sector/surface shall subtend an arc of specified angle.

4.1.23 For helidecks the obstacle-free sector shall subtend an arc of 210° and extend outwards to a distance compatible with the one-engine inoperative capability of the most critical helicopter the helideck is intended to serve. The surface shall be a horizontal plane level with the elevation of the helideck except that, over an arc of 180° passing through the centre of the FATO, the surface shall be at water level, extending outwards for a distance compatible with the take-off space required for the most critical helicopter the helideck is intended to serve (see Figure 4-2).

Limited obstacle surface — helidecks

4.1.24 Description. A complex surface originating at the reference point for the obstacle-free sector and extending over the arc not covered by the obstacle-free sector as shown in Figures 4-3, 4-4 and 4-5 and within which the height of obstacles above the level of the FATO will be prescribed.

4.1.25 Characteristics. The limited obstacle surface shall not subtend an arc greater than a specified angle and shall be sufficient to include that area not covered by the obstacle-free sector.

4.2 Obstacle limitation requirements

Note.— The requirements for obstacle limitation surfaces are specified on the basis of the intended use of a FATO, i.e. approach manoeuvre to hover or landing, or take-off manoeuvre and type of approach, and are intended to be applied when such use is made of the FATO. In cases where operations are conducted to or from both directions of a FATO, the function of certain surfaces may be nullified because of more stringent requirements of another lower surface.

Surface level heliports

4.2.1 The following obstacle limitation surfaces shall be established for a precision approach FATO:

a) take-off climb surface;

b) approach surface;

c) transitional surfaces; and

d) conical surface.

4.2.2 The following obstacle limitation surfaces shall be established for a non-precision approach FATO:

a) take-off climb surface;

b) approach surface;

c) transitional surfaces; and

d) conical surface if an inner horizontal surface is not provided.

4.2.3 The following obstacle limitation surfaces shall be established for a non-instrument FATO:

a) take-off climb surface; and

b) approach surface.

4.2.4 Recommendation.— The following obstacle limitation surfaces should be established for a non-precision approach FATO:

a) inner horizontal surface; and

b) conical surface.

Note.— An inner horizontal surface may not be required if a straight-in non-precision approach is provided at both ends.

4.2.5 The slopes of the surfaces shall not be greater than, and their other dimensions not less than those specified in Tables 4-1 to 4-4 and shall be located as shown in Figures 4-6 to 4-10.

4.2.6 New objects or extensions of existing objects shall not be permitted above any of the surfaces in 4.2.1 to 4.2.4 above except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object.

Note.— Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6.

4.2.7 Recommendation.— Existing objects above any of the surfaces in 4.2.1 to 4.2.4 above should, as far as practicable, be removed except when, in the opinion of the appropriate authority, the object is shielded by an existing immovable object or after aeronautical study it is determined that the object would not adversely affect the safety or significantly affect the regularity of operations of helicopters.

Note.— The application of curved take-off climb surfaces as specified in 4.1.19 may alleviate the problems created by objects infringing these surfaces.
Chapter 4

4.2.8 A surface level heliport shall have at least two take-off climb and approach surfaces, separated by not less than 150°.

4.2.9 Recommendation.— The number and orientation of take-off climb and approach surfaces should be such that the usability factor of a heliport is not less than 95 per cent for the helicopters the heliport is intended to serve.

Elevated heliports

4.2.10 The obstacle limitation requirements for elevated heliports shall conform to the requirements for surface level heliports specified in 4.2.1 to 4.2.7.

4.2.11 An elevated heliport shall have at least two take-off climb and approach surfaces separated by not less than 150°.

Helidecks

Note.— The following specifications are for helidecks located on a structure and engaged in such activities as mineral exploitation, research, or construction, but excluding heliports on ships.

4.2.12 A helideck shall have an obstacle-free sector and, where necessary, a limited obstacle sector.

4.2.13 There shall be no fixed obstacles within the obstacle-free sector above the obstacle-free surface.

4.2.14 In the immediate vicinity of the helideck, obstacle protection for helicopters shall be provided below the heliport level. This protection shall extend over an arc of at least 180° with the origin at the centre of the FATO, with a descending gradient having a ratio of one unit horizontally to five units vertically from the edges of the FATO within the 180° sector.

4.2.15 Where a mobile obstacle or combination of obstacles within the obstacle-free sector is essential for the operation of the installation, the obstacle(s) shall not subtend an arc exceeding 30°, as measured from the centre of the FATO.

4.2.16 For single-main-rotor and side-by-side twin rotor helicopters, within the 150° limited obstacle surface/sector out to a distance of 0.62 D, measured from the centre of the FATO, objects shall not exceed a height of 0.05 D above the FATO. Beyond that arc, out to an over-all distance of 0.83 D the limited obstacle surface rises at a rate of one unit vertically for each two units horizontally (see Figure 4-3).

4.2.17 For omnidirectional operations by tandem-main-rotor helicopters within the 150° limited obstacle surface/sector out to a distance of 0.62 D, measured from the centre of the FATO, there shall be no fixed obstacles. Beyond that arc, out to an over-all distance of 0.83 D, objects shall not penetrate a level surface which has a height equivalent to 0.05 D above the FATO (see Figure 4-4).

4.2.18 For bi-directional operations by tandem-main-rotor helicopters, within the 0.62 D arc in the 150° limited obstacle surface/sector, objects shall not penetrate a level surface which has a height equivalent to 1.1 m above the FATO (see Figure 4-5).

Shipboard heliports

Amidships location

4.2.19 Forward and aft of the FATO shall be two symmetrically located sectors, each covering an arc of 150°, with their apexes on the periphery of the FATO D reference circle. Within the area enclosed by these two sectors, there shall be no objects rising above the level of the FATO, except those aids essential for the safe operation of a helicopter and then only up to a maximum height of 25 cm.

4.2.20 To provide further protection from obstacles fore and aft of the FATO, rising surfaces with gradients of one unit vertically to five units horizontally shall extend from the entire length of the edges of the two 150° sectors. These surfaces shall extend for a horizontal distance equal to at least the diameter of the FATO and shall not be penetrated by any obstacle (see Figure 4-11).

Ship’s side location

4.2.21 From the fore and aft mid-points of the D reference circle, an area shall extend to the ship’s rail to a fore and aft distance of 1.5 times the diameter of the FATO, located symmetrically about the athwartships bisector of the reference circle. Within this sector there shall be no objects rising above the level of the FATO, except those aids essential to the safe operation of the helicopter and then only up to a maximum height of 25 cm (see Figure 4-12).

4.2.22 A horizontal surface shall be provided, at least 0.25 times the diameter of the D reference circle, which shall surround the FATO and the obstacle-free sector, at a height of 0.05 times the diameter of the reference circle, which no object shall penetrate.
Annex 14 — Aerodromes

Note.—The figure shows the obstacle limitation surfaces at a heliport with a non-precision approach FATO and a clearway.

Figure 4-1. Obstacle limitation surfaces
210° sector

Alternative positions on the periphery and swinging the whole sector ±15° from that shown may be used in satisfying requirements

PLAN

Limited obstacle surface

±15°

PROFILE

Landing area

Area in which rig structure is permitted in 180° sector

±15°

No fixed obstacle between these lines in 180° sector

Water level

No fixed obstacle between these lines in 180° sector

5:1 Falling gradient

Figure 4-2. Helideck obstacle-free sector
Figure 4-3. Helideck obstacle limitation sectors 
Single-main-rotor and side-by-side twin rotor helicopters
150° sector (Alternative positions on the periphery and swinging the whole sector ± 15° from that shown may be used in satisfying requirements)

D = Helicopter largest over-all dimension

Obstacles limited to 0.05 D

Section AA

Figure 4-4. Helideck obstacle limitation sectors
Tandem-main-rotor helicopters — Omnidirectional operations
Final approach and take-off area

Landing direction

Landing direction

150° sector
(No alternative position is allowed)

Obstacles limited to 1.1 m

D = Helicopter largest over-all dimension

0.9 D

0.75 D

0.62 D

Section A-A

Figure 4-5. Helideck obstacle limitation sectors
Tandem-main-rotor helicopters — Bi-directional operations
Chapter 4

Annex 14 — Aerodromes

Final approach and take-off area (FATO)

Take-off climb/approach surface

Shaded area to have same characteristics as safety area

A. Circular final approach and take-off area (straight approach-departure)

B. Squared final approach and take-off area (straight approach-departure)

C. Squared final approach and take-off area (curved approach-departure)

Figure 4-6. Take-off climb/approach surface (non-instrument FATO)
Figure 4-7. Take-off climb surface for instrument FATO
Figure 4-8. Approach surface for precision approach FATO
Figure 4-9. Approach surface for non-precision approach FATO
Chapter 4

Annex 14 — Aerodromes

Conical surface 5% (1:20)

Transitional surface 14.3% (1:7)

Inner horizontal surface

Safety area

315 m

1 100 m

4 000 m

100 m

Non-precision approach (end profiles)

Conical surface 5% (1:20)

Transitional surface 14.3% (1:7)

Safety area

315 m

1 100 m

Alternative when no inner horizontal surface is provided

Precision approach (end profiles)

Figure 4-10. Transitional, inner horizontal and conical obstacle limitation surfaces
PLAN VIEW

D = Helicopter largest over-all dimension

Figure 4-11. Midship non-purpose built heliport obstacle limitation surfaces
Figure 4-12. Ships-side non-purpose built heliport obstacle limitation surfaces

D = Helicopter largest over-all dimension
### Table 4-1. Dimensions and slopes of obstacle limitation surfaces

#### NON-INSTRUMENT AND NON-PRECISION FATO

<table>
<thead>
<tr>
<th>Surface and dimensions</th>
<th>Non-instrument (visual) FATO</th>
<th>Non-precision (instrument approach) FATO</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Helicopter performance class 1</td>
<td>2</td>
</tr>
<tr>
<td>APPROACH SURFACE</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Width of inner edge</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Location of inner edge</td>
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<td></td>
</tr>
<tr>
<td>First section</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Divergence</td>
<td>— day</td>
<td>10%</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>15%</td>
</tr>
<tr>
<td>Length</td>
<td>— day</td>
<td>245 m^a</td>
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<tr>
<td></td>
<td>— night</td>
<td>245 m^a</td>
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<tr>
<td>Outer width</td>
<td>— day</td>
<td>49 m^b</td>
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<tr>
<td></td>
<td>— night</td>
<td>73.5 m^b</td>
</tr>
<tr>
<td>Slope (maximum)</td>
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<td>Second section</td>
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<tr>
<td>Divergence</td>
<td>— day</td>
<td>10%</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>15%</td>
</tr>
<tr>
<td>Length</td>
<td>— day</td>
<td>c</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>c</td>
</tr>
<tr>
<td>Outer width</td>
<td>— day</td>
<td>d</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>d</td>
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<tr>
<td>Slope (maximum)</td>
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<td>12.5%</td>
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<td>Divergence</td>
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<tr>
<td>Length</td>
<td>— day</td>
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<tr>
<td></td>
<td>— night</td>
<td>e</td>
</tr>
<tr>
<td>Outer width</td>
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<td>d</td>
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<td></td>
<td>— night</td>
<td>d</td>
</tr>
<tr>
<td>Slope (maximum)</td>
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<tr>
<td>Radius</td>
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<td>—</td>
</tr>
<tr>
<td>Height</td>
<td></td>
<td>—</td>
</tr>
<tr>
<td>TRANSITIONAL</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Slope</td>
<td></td>
<td>—</td>
</tr>
<tr>
<td>Height</td>
<td></td>
<td>—</td>
</tr>
</tbody>
</table>

a. Slope and length enables helicopters to decelerate for landing while observing “avoid” areas.
b. The width of the inner edge shall be added to this dimension.
c. Determined by the distance from the inner edge to the point where the divergence produces a width of 7 rotor diameters for day operations or 10 rotor diameters for night operations.
d. Seven rotor diameters over-all width for day operations or 10 rotor diameters over-all width for night operations.
e. Determined by the distance from inner edge to where the approach surface reaches a height of 150 m above the elevation of the inner edge.
### Table 4-2. Dimensions and slopes of obstacle limitation surfaces

#### Instrument (Precision Approach) FATO

<table>
<thead>
<tr>
<th>Surface and dimensions</th>
<th>Height above FATO</th>
<th>3° approach</th>
<th>6° approach</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>90 m (300 ft)</td>
<td>60 m (200 ft)</td>
<td>45 m (150 ft)</td>
</tr>
<tr>
<td>APPROACH SURFACE</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Length of inner edge</td>
<td>90 m</td>
<td>90 m</td>
<td>90 m</td>
</tr>
<tr>
<td>Distance from end of FATO</td>
<td>60 m</td>
<td>60 m</td>
<td>60 m</td>
</tr>
<tr>
<td>Divergence each side to height above FATO</td>
<td>25%</td>
<td>25%</td>
<td>25%</td>
</tr>
<tr>
<td>Distance to height above FATO</td>
<td>1 745 m</td>
<td>1 163 m</td>
<td>872 m</td>
</tr>
<tr>
<td>Width at height above FATO</td>
<td>962 m</td>
<td>671 m</td>
<td>526 m</td>
</tr>
<tr>
<td>Divergence to parallel section</td>
<td>15%</td>
<td>15%</td>
<td>15%</td>
</tr>
<tr>
<td>Distance to parallel section</td>
<td>2 793 m</td>
<td>3 763 m</td>
<td>4 246 m</td>
</tr>
<tr>
<td>Width of parallel section</td>
<td>1 800 m</td>
<td>1 800 m</td>
<td>1 800 m</td>
</tr>
<tr>
<td>Distance to outer edge</td>
<td>5 462 m</td>
<td>5 074 m</td>
<td>4 882 m</td>
</tr>
<tr>
<td>Width at outer edge</td>
<td>1 800 m</td>
<td>1 800 m</td>
<td>1 800 m</td>
</tr>
<tr>
<td>Slope of first section</td>
<td>2.5% (1:40)</td>
<td>2.5% (1:40)</td>
<td>2.5% (1:40)</td>
</tr>
<tr>
<td>Length of first section</td>
<td>3 000 m</td>
<td>3 000 m</td>
<td>3 000 m</td>
</tr>
<tr>
<td>Slope of second section</td>
<td>3% (1:33.3)</td>
<td>3% (1:33.3)</td>
<td>3% (1:33.3)</td>
</tr>
<tr>
<td>Length of second section</td>
<td>2 500 m</td>
<td>2 500 m</td>
<td>2 500 m</td>
</tr>
<tr>
<td>Total length of surface</td>
<td>10 000 m</td>
<td>10 000 m</td>
<td>10 000 m</td>
</tr>
<tr>
<td>CONICAL</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Slope</td>
<td>5%</td>
<td>5%</td>
<td>5%</td>
</tr>
<tr>
<td>Height</td>
<td>55 m</td>
<td>55 m</td>
<td>55 m</td>
</tr>
<tr>
<td>TRANSITIONAL</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Slope</td>
<td>14.3%</td>
<td>14.3%</td>
<td>14.3%</td>
</tr>
<tr>
<td>Height</td>
<td>45 m</td>
<td>45 m</td>
<td>45 m</td>
</tr>
</tbody>
</table>
### Table 4-3. Dimensions and slopes of obstacle limitation surfaces

#### STRAIGHT TAKE-OFF

<table>
<thead>
<tr>
<th>Surface and dimensions</th>
<th>Helicopter performance class</th>
<th>Non-instrument (visual)</th>
<th>Instrument</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>1</td>
<td>2</td>
<td>3</td>
</tr>
<tr>
<td><strong>TAKE-OFF CLIMB</strong></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Width of inner edge</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Location of inner edge</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td><strong>First section</strong></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Divergence</td>
<td>— day</td>
<td>10%</td>
<td>10%</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>15%</td>
<td>15%</td>
</tr>
<tr>
<td>Length</td>
<td>— day</td>
<td>a</td>
<td>245 m&lt;sup&gt;b&lt;/sup&gt;</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>a</td>
<td>245 m&lt;sup&gt;b&lt;/sup&gt;</td>
</tr>
<tr>
<td>Outer width</td>
<td>— day</td>
<td>c</td>
<td>49 m&lt;sup&gt;d&lt;/sup&gt;</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>c</td>
<td>73.5 m&lt;sup&gt;d&lt;/sup&gt;</td>
</tr>
<tr>
<td>Slope (maximum)</td>
<td></td>
<td>4.5%*</td>
<td>8%&lt;sup&gt;b&lt;/sup&gt;</td>
</tr>
<tr>
<td><strong>Second section</strong></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Divergence</td>
<td>— day</td>
<td>parallel</td>
<td>10%</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>parallel</td>
<td>15%</td>
</tr>
<tr>
<td>Length</td>
<td>— day</td>
<td>e</td>
<td>a</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>e</td>
<td>a</td>
</tr>
<tr>
<td>Outer width</td>
<td>— day</td>
<td>c</td>
<td>c</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>c</td>
<td>c</td>
</tr>
<tr>
<td>Slope (maximum)</td>
<td></td>
<td>4.5%*</td>
<td>15%</td>
</tr>
<tr>
<td><strong>Third section</strong></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Divergence</td>
<td>— day</td>
<td>parallel</td>
<td>parallel</td>
</tr>
<tr>
<td>Length</td>
<td>— day</td>
<td>—</td>
<td>e</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>—</td>
<td>e</td>
</tr>
<tr>
<td>Outer width</td>
<td>— day</td>
<td>—</td>
<td>c</td>
</tr>
<tr>
<td></td>
<td>— night</td>
<td>—</td>
<td>c</td>
</tr>
<tr>
<td>Slope (maximum)</td>
<td></td>
<td>—</td>
<td>15%</td>
</tr>
</tbody>
</table>

a. Determined by the distance from the inner edge to the point where the divergence produces a width of 7 rotor diameters for day operations or 10 rotor diameters for night operations.
b. Slope and length provides helicopters with an area to accelerate and climb while observing “avoid” areas.
c. Seven rotor diameters over-all width for day operations or 10 rotor diameters over-all width for night operations.
d. The width of the inner edge shall be added to this dimension.
e. Determined by the distance from the inner edge to where the surface reaches a height of 150 m above the elevation of the inner edge.

* This slope exceeds the maximum mass one-engine-inoperative climb gradient of many helicopters which are currently operating.

9/11/95
### Chapter 4  
### Annex 14 — Aerodromes

**Table 4-4. Criteria for curved take-off climb/approach area**

**NON-INSTRUMENT FINAL APPROACH AND TAKE-OFF**

<table>
<thead>
<tr>
<th>Facility</th>
<th>Requirement</th>
</tr>
</thead>
<tbody>
<tr>
<td>Directional change</td>
<td>As required (120° max).</td>
</tr>
<tr>
<td>Radius of turn on centre line</td>
<td>Not less than 270 m.</td>
</tr>
<tr>
<td>Distance to inner gate*</td>
<td>(a) For performance class 1 helicopters — not less than 305 m from the end of the safety area or helicopter clearway.</td>
</tr>
<tr>
<td></td>
<td>(b) For performance class 2 and 3 helicopters — not less than 370 m from the end of the FATO.</td>
</tr>
<tr>
<td>Width of inner gate — day</td>
<td>Width of the inner edge plus 20% of distance to inner gate.</td>
</tr>
<tr>
<td></td>
<td>— night Width of the inner edge plus 30% of distance to inner gate.</td>
</tr>
<tr>
<td>Width of outer gate — day</td>
<td>Width of inner edge plus 20% of distance to inner gate out to minimum width of 7 rotor diameters.</td>
</tr>
<tr>
<td></td>
<td>— night Width of inner edge plus 30% of distance to inner gate out to a minimum width of 10 rotor diameters.</td>
</tr>
<tr>
<td>Elevation of inner and outer gates</td>
<td>Determined by the distance from the inner edge and the designated gradient(s).</td>
</tr>
<tr>
<td>Slopes</td>
<td>As given in Tables 4-1 and 4-3.</td>
</tr>
<tr>
<td>Divergence</td>
<td>As given in Tables 4-1 and 4-3.</td>
</tr>
<tr>
<td>Total length of area</td>
<td>As given in Tables 4-1 and 4-3.</td>
</tr>
</tbody>
</table>

* This is the minimum distance required prior to initiating a turn after take-off or completing a turn in the final phase.

*Note.* More than one turn may be necessary in the total length of the take-off climb/approach area. The same criteria will apply for each subsequent turn except that the widths of the inner and outer gates will normally be the maximum width of the area.
CHAPTER 5. VISUAL AIDS

5.1 Indicators

5.1.1 Wind direction indicators

Application

5.1.1.1 A heliport shall be equipped with at least one wind direction indicator.

Location

5.1.1.2 A wind direction indicator shall be located so as to indicate the wind conditions over the final approach and take-off area and in such a way as to be free from the effects of airflow disturbances caused by nearby objects or rotor downwash. It shall be visible from a helicopter in flight, in a hover or on the movement area.

5.1.1.3 Recommendation.— Where a touchdown and lift-off area may be subject to a disturbed air flow, then additional wind direction indicators located close to the area should be provided to indicate the surface wind on the area.

Note.— Guidance on the location of wind direction indicators is given in the Heliport Manual.

Characteristics

5.1.1.4 A wind direction indicator shall be constructed so that it gives a clear indication of the direction of the wind and a general indication of the wind speed.

5.1.1.5 Recommendation.— An indicator should be a truncated cone made of lightweight fabric and should have the following minimum dimensions:

<table>
<thead>
<tr>
<th></th>
<th>Surface level heliports</th>
<th>Elevated heliports and helidecks</th>
</tr>
</thead>
<tbody>
<tr>
<td>Length</td>
<td>2.4 m</td>
<td>1.2 m</td>
</tr>
<tr>
<td>Diameter (larger end)</td>
<td>0.6 m</td>
<td>0.3 m</td>
</tr>
<tr>
<td>Diameter (smaller end)</td>
<td>0.3 m</td>
<td>0.15 m</td>
</tr>
</tbody>
</table>

5.1.1.6 Recommendation.— The colour of the wind direction indicator should be so selected as to make it clearly visible and understandable from a height of at least 200 m (650 ft) above the heliport, having regard to background. Where practicable, a single colour, preferably white or orange, should be used. Where a combination of two colours is required to give adequate conspicuity against changing backgrounds, they should preferably be orange and white, red and white, or black and white, and should be arranged in five alternate bands the first and last band being the darker colour.

5.1.1.7 A wind direction indicator at a heliport intended for use at night shall be illuminated.

5.2 Markings and markers

Note.— See Annex 14, Volume I, 5.2.1.4, Note I, concerning improving conspicuity of markings.

5.2.1 Winching area marking

Application

5.2.1.1 Recommendation.— A winching area marking should be provided at a winching area.

Location

5.2.1.2 A winching area marking shall be located so that its centre coincides with the centre of the clear zone of the winching area.

Characteristics

5.2.1.3 A winching area marking shall consist of a solid circle of not less than 5 m in diameter and painted yellow.

5.2.2 Heliport identification marking

Application

5.2.2.1 A heliport identification marking shall be provided at a heliport.
Chapter 5

Annex 14 — Aerodromes

Location

5.2.2.2 A heliport identification marking shall be located within the final approach and take-off area, at or near the centre of the area or when used in conjunction with runway designation markings at each end of the area.

Characteristics

5.2.2.3 A heliport identification marking, except for a heliport at a hospital, shall consist of a letter H, white in colour. The dimensions of the marking shall be no less than those shown in Figure 5-1 and where the marking is used in conjunction with the final approach and take-off area designation marking specified in 5.2.5 its dimensions shall be increased by a factor of 3.

Note.—On a helideck covered with a rope netting, it may be advantageous to increase the height of the marking to 4 m and the other dimensions proportionally.

5.2.2.4 A heliport identification marking for a heliport at a hospital shall consist of a letter H, red in colour, on a white cross made of squares adjacent to each of the sides of a square containing the H as shown in Figure 5-1.

5.2.2.5 A heliport identification marking shall be oriented with the cross arm of the H at right angles to the preferred final approach direction. For a helideck the cross arm shall be on or parallel to the bisector of the obstacle-free sector as shown in Figure 5-1.

5.2.3 Maximum allowable mass marking

Application

5.2.3.1 Recommendation.—A maximum allowable mass marking should be displayed at an elevated heliport and at a helideck.
Annex 14 — Aerodromes

Location

5.2.3.2 Recommendation.— A maximum allowable mass marking should be located within the touchdown and lift-off area and so arranged as to be readable from the preferred final approach direction.

Characteristics

5.2.3.3 A maximum allowable mass marking shall consist of a two digit number followed by a letter “t” to indicate the allowable helicopter mass in tonnes (1 000 kg).

5.2.3.4 Recommendation.— The numbers and the letter of the marking should have a colour contrasting with the background and should be in the form and proportion shown in Figure 5-2.

5.2.4 Final approach and take-off area marking or marker

Application

5.2.4.1 Final approach and take-off area marking or markers shall be provided at a surface level heliport on ground where the extent of the final approach and take-off area is not self-evident.

Location

5.2.4.2 Final approach and take-off area marking or markers shall be located on the boundary of the final approach and take-off area.

Characteristics

5.2.4.3 Final approach and take-off area marking or markers shall be spaced:

a) for a square or rectangular area at equal intervals of not more than 50 m with at least three markings or markers on each side including a marking or marker at each corner; and

b) for any other shaped area, including a circular area, at equal intervals of not more than 10 m with a minimum number of five markings or markers.

5.2.4.4 A final approach and take-off area marking shall be a rectangular stripe with a length of 9 m or one-fifth of the side of the final approach and take-off area which it defines and a width of 1 m. Where a marker is used its characteristics shall conform to those specified in Annex 14, Volume I, 5.5.8.3 except that the height of the marker shall not exceed 25 cm above ground or snow level.

5.2.5 Final approach and take-off area designation marking

Application

5.2.5.1 Recommendation.— A final approach and take-off area designation marking should be provided where it is necessary to designate the final approach and take-off area to the pilot.

Location

5.2.5.2 A final approach and take-off area designation marking shall be located at the beginning of the final approach and take-off area as shown in Figure 5-3.

Characteristics

5.2.5.3 A final approach and take-off area designation marking shall consist of a runway designation marking described in Annex 14, Volume I, 5.2.2.4 and 5.2.2.5 supplemented by an H, specified in 5.2.2 above, and as shown in Figure 5-3.

5.2.6 Aiming point marking

Application

5.2.6.1 Recommendation.— An aiming point marking should be provided at a heliport where it is necessary for a pilot to make an approach to a particular point before proceeding to the touchdown and lift-off area.

Location

5.2.6.2 The aiming point marking shall be located within the final approach and take-off area.

Characteristics

5.2.6.3 The aiming point marking shall be an equilateral triangle with the bisector of one of the angles aligned with the preferred approach direction. The marking shall consist of continuous white lines and the dimensions of the marking shall conform to those shown in Figure 5-4.
Figure 5-2. Form and proportions of numbers and letter for maximum allowable mass marking

Note.—All units are expressed in centimetres.
5.2.7 Touchdown and lift-off area marking

**Application**

5.2.7.1 A touchdown and lift-off area marking shall be provided on a helideck.

5.2.7.2 **Recommendation.**—A touchdown and lift-off area marking should be provided on a heliport other than a helideck if the perimeter of the touchdown and lift-off area is not self-evident.

**Location**

5.2.7.3 The touchdown and lift-off area marking shall be located along the perimeter of the touchdown and lift-off area.

**Characteristics**

5.2.7.4 A touchdown and lift-off area marking shall consist of a continuous white line with a width of at least 30 cm.
Chapter 5

5.2.8 Touchdown marking

Application

5.2.8.1 Recommendation.— A touchdown marking should be provided where it is necessary for a helicopter to touch down in a specific position.

Location

5.2.8.2 A touchdown marking shall be located so that when a helicopter for which the marking is intended is positioned, with the main undercarriage inside the marking and the pilot situated over the marking, all parts of the helicopter will be clear of any obstacle by a safe margin.

5.2.8.3 On a helideck or on an elevated heliport the centre of the touchdown marking shall be located at the centre of the touchdown and lift-off area except that the marking may be offset away from the origin of the obstacle-free sector by no more than 0.1 D where an aeronautical study indicates such offsetting to be necessary and that a marking so offset would not adversely affect the safety.

Characteristics

5.2.8.4 A touchdown marking shall be a yellow circle and have a line width of at least 0.5 m. For a helideck the line width shall be at least 1 m.

5.2.8.5 On helidecks the inner diameter of the circle shall be half the D value of the helideck or 6 m whichever is the greater.

5.2.9 Heliport name marking

Application

5.2.9.1 Recommendation.— A heliport name marking should be provided at a heliport where there is insufficient alternative means of visual identification.

Location

5.2.9.2 Recommendation.— The heliport name marking should be placed on the heliport so as to be visible, as far as practicable, at all angles above the horizontal. Where an obstacle sector exists the marking should be located on the obstacle side of the H identification marking.

Characteristics

5.2.9.3 A heliport name marking shall consist of the name or the alphanumeric designator of the heliport as used in the R/T communications.

5.2.9.4 Recommendation.— The characters of the marking should be not less than 3 m in height at surface level heliports and not less than 1.2 m on elevated heliports and helidecks. The colour of the marking should contrast with the background.

5.2.9.5 A heliport name marking intended for use at night or during conditions of poor visibility shall be illuminated, either internally or externally.

5.2.10 Helideck obstacle-free sector marking

Application

5.2.10.1 Recommendation.— A helideck obstacle-free sector marking should be provided at a helideck.

Location

5.2.10.2 A helideck obstacle-free sector marking shall be located on the touchdown and lift-off area marking.

Characteristics

5.2.10.3 The helideck obstacle-free sector marking shall indicate the origin of the obstacle free sector, the directions of the limits of the sector and the D value of the helideck as shown in Figure 5-5 for a hexagonal-shaped helideck.

Note.— D is the largest dimension of the helicopter when the rotors are turning.

5.2.10.4 The height of the chevron shall equal the width of the touchdown and lift-off area marking.

5.2.10.5 The chevron shall be black.

5.2.11 Marking for taxiways

Note.— The specifications for taxiway centre line marking and taxi-holding position markings in Annex 14, Volume I, 5.2.8 and 5.2.9 are equally applicable to taxiways intended for ground taxiing of helicopters.
Annex 14 — Aerodromes

5.2.12 Air taxiway markers

Application

5.2.12.1 Recommendation. — An air taxiway should be marked with air taxiway markers.

Note. — These markers are not meant to be used on helicopter ground taxiways.

Location

5.2.12.2 Air taxiway markers shall be located along the centre line of the air taxiway and shall be spaced at intervals of not more than 30 m on straight sections and 15 m on curves.

Characteristics

5.2.12.3 An air taxiway marker shall be frangible and when installed shall not exceed 35 cm above ground or snow level. The surface of the marker as viewed by the pilot shall be a rectangle with a height to width ratio of approximately 3 to 1 and shall have a minimum area of 150 cm² as shown in Figure 5-6.

5.2.12.4 An air taxiway marker shall be divided into three equal, horizontal bands coloured yellow, green and yellow, respectively. If the air taxiway is to be used at night, the markers shall be internally illuminated or retro-reflective.

5.2.13 Air transit route markers

Application

5.2.13.1 Recommendation. — When established an air transit route should be marked with air transit route markers.

Location

5.2.13.2 Air transit route markers shall be located along the centre line of the air transit route and shall be spaced at intervals of not more than 60 m on straight sections and 15 m on curves.

Figure 5-5. Helideck obstacle-free sector marking

9/11/95
Chapter 5

Annex 14 — Aerodromes

Figure 5-6. Air taxiway marker

Figure 5-7. Air transit route marker
Annex 14 — Aerodromes

Characteristics

5.2.13.3 An air transit route marker shall be frangible and when installed shall not exceed 1 m above ground or snow level. The surface of the marker as viewed by the pilot shall be a rectangle with a height to width ratio of approximately 1 to 3 and shall have a minimum area of 1 500 cm² as shown in the examples in Figure 5-7.

5.2.13.4 An air transit route marker shall be divided into three equal, vertical bands coloured yellow, green and yellow, respectively. If the air transit route is to be used by night, the marker shall be internally illuminated or retro-reflective.

5.3 Lights

5.3.1 General

Note 1.— See Annex 14, Volume I, 5.3.1 concerning specifications on screening of non-aeronautical ground lights and design of elevated and inset lights.

Note 2.— In the case of helidecks and heliports located near navigable waters, consideration needs to be given to ensuring that aeronautical ground lights do not cause confusion to mariners.

Note 3.— As helicopters will generally come very close to extraneous light sources, it is particularly important to ensure that, unless such lights are navigation lights exhibited in accordance with international regulations, they are screened or located so as to avoid direct and reflected glare.

5.3.2 Heliport beacon

Application

5.3.2.1 Recommendation.— A heliport beacon should be provided at a heliport where:

a) long-range visual guidance is considered necessary and is not provided by other visual means; or

b) identification of the heliport is difficult due to surrounding lights.

Location

5.3.2.2 The heliport beacon shall be located on or adjacent to the heliport preferably at an elevated position and so that it does not dazzle a pilot at short range.

Note.— Where a heliport beacon is likely to dazzle pilots at short range it may be switched off during the final stages of the approach and landing.

Characteristics

5.3.2.3 The heliport beacon shall emit repeated series of equispaced short duration white flashes in the format in Figure 5-8.

Figure 5-8. Heliport beacon flash characteristics
Chapter 5

5.3.2.4 The light from the beacon shall show at all angles of azimuth.

5.3.2.5 **Recommendation.**— The effective light intensity distribution of each flash should be as shown in Figure 5-9, Illustration 1.

**Note.**— Where brilliancy control is desired, settings of 10 per cent and 3 per cent have been found to be satisfactory. In addition, shielding may be necessary to ensure that pilots are not dazzled during the final stages of the approach and landing.

5.3.3 Approach lighting system

**Application**

5.3.3.1 **Recommendation.**— An approach lighting system should be provided at a heliport where it is desirable and practicable to indicate a preferred approach direction.

**Location**

5.3.3.2 The approach lighting system shall be located in a straight line along the preferred direction of approach.

**Characteristics**

5.3.3.3 **Recommendation.**— An approach lighting system should consist of a row of three lights spaced uniformly at 30 m intervals and of a crossbar 18 m in length at a distance of 90 m from the perimeter of the final approach and take-off area as shown in Figure 5-10. The lights forming the crossbar should be as nearly as practicable in a horizontal straight line at right angles to, and bisected by, the line of the centre line lights and spaced at 4.5 m intervals. Where there is the need to make the final approach course more conspicuous additional lights spaced uniformly at 30 m intervals should be added beyond the crossbar. The lights beyond the crossbar may be steady or sequenced flashing, depending upon the environment.

**Note.**— Sequenced flashing lights may be useful where identification of the approach lighting system is difficult due to surrounding lights.

5.3.3.4 **Recommendation.**— Where an approach lighting system is provided for a non-precision final approach and take-off area, the system should not be less than 210 m in length.

5.3.3.5 The steady lights shall be omnidirectional white lights.

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Annex 14 — Aerodromes

5.3.3.6 **Recommendation.**— The light distribution of steady lights should be as indicated in Figure 5-9, Illustration 2 except that the intensity should be increased by a factor of 3 for a non-precision final approach and take-off area.

5.3.3.7 Sequenced flashing lights shall be omnidirectional white lights.

5.3.3.8 **Recommendation.**— The flashing lights should have a flash frequency of one per second and their light distribution should be as shown in Figure 5-9, Illustration 3. The flash sequence should commence from the outermost light and progress towards the crossbar.

5.3.3.9 **Recommendation.**— A suitable brilliancy control should be incorporated to allow for adjustment of light intensity to meet the prevailing conditions.

**Note.**— The following intensity settings have been found suitable:

a) steady lights — 100 per cent, 30 per cent and 10 per cent; and

b) flashing lights — 100 per cent, 10 per cent and 3 per cent.

5.3.4 Visual alignment guidance system

**Application**

5.3.4.1 **Recommendation.**— A visual alignment guidance system should be provided to serve the approach to a heliport where one or more of the following conditions exist especially at night:

a) obstacle clearance, noise abatement or traffic control procedures require a particular direction to be flown;

b) the environment of the heliport provides few visual surface cues; and

c) it is physically impracticable to install an approach lighting system.

**Location**

5.3.4.2 The visual alignment guidance system shall be located such that a helicopter is guided along the prescribed track towards the final approach and take-off area.

5.3.4.3 **Recommendation.**— The system should be located at the downwind edge of the final approach and take-off area and aligned along the preferred approach direction.

9/11/95
### Figure 5.9: Iso-candela diagrams of lights meant for helicopter non-instrument and non-precision approaches

#### Illustration 1 — Heliport beacon

<table>
<thead>
<tr>
<th>Elevation</th>
<th>10°</th>
<th>250 cd*</th>
</tr>
</thead>
<tbody>
<tr>
<td>7°</td>
<td>750 cd*</td>
<td></td>
</tr>
<tr>
<td>4°</td>
<td>1700 cd*</td>
<td></td>
</tr>
<tr>
<td>2½°</td>
<td>2500 cd*</td>
<td></td>
</tr>
<tr>
<td>1½°</td>
<td>2500 cd*</td>
<td></td>
</tr>
<tr>
<td>0°</td>
<td>1700 cd*</td>
<td></td>
</tr>
</tbody>
</table>

#### Illustration 2 — Approach light steady burning

<table>
<thead>
<tr>
<th>Elevation</th>
<th>15°</th>
<th>25 cd</th>
</tr>
</thead>
<tbody>
<tr>
<td>9°</td>
<td>250 cd*</td>
<td></td>
</tr>
<tr>
<td>6°</td>
<td>350 cd</td>
<td></td>
</tr>
<tr>
<td>5°</td>
<td>350 cd</td>
<td></td>
</tr>
<tr>
<td>2°</td>
<td>250 cd*</td>
<td></td>
</tr>
</tbody>
</table>

#### Illustration 3 — Approach light flashing

<table>
<thead>
<tr>
<th>Elevation</th>
<th>2°</th>
<th>2500 cd*</th>
</tr>
</thead>
<tbody>
<tr>
<td>0°</td>
<td>250 cd*</td>
<td></td>
</tr>
</tbody>
</table>

#### Illustration 4 — HAPI system

<table>
<thead>
<tr>
<th>Elevation</th>
<th>30°</th>
<th>10 cd</th>
</tr>
</thead>
<tbody>
<tr>
<td>25°</td>
<td>50 cd</td>
<td></td>
</tr>
<tr>
<td>20°</td>
<td>100 cd</td>
<td></td>
</tr>
<tr>
<td>10°</td>
<td>100 cd</td>
<td></td>
</tr>
<tr>
<td>0°</td>
<td>100 cd</td>
<td></td>
</tr>
</tbody>
</table>

#### Illustration 5 — Final approach and take-off area lights and aiming point lights

<table>
<thead>
<tr>
<th>Elevation</th>
<th>30°</th>
<th>3 cd</th>
</tr>
</thead>
<tbody>
<tr>
<td>25°</td>
<td>15 cd</td>
<td></td>
</tr>
<tr>
<td>20°</td>
<td>25 cd</td>
<td></td>
</tr>
<tr>
<td>0°</td>
<td>3 cd</td>
<td></td>
</tr>
</tbody>
</table>

#### Illustration 6 — Touchdown and lift-off area perimeter lights

<table>
<thead>
<tr>
<th>Elevation</th>
<th>90°</th>
<th>55 cd/m²</th>
</tr>
</thead>
<tbody>
<tr>
<td>60°</td>
<td>55 cd/m²</td>
<td></td>
</tr>
<tr>
<td>40°</td>
<td>50 cd/m²</td>
<td></td>
</tr>
<tr>
<td>30°</td>
<td>45 cd/m²</td>
<td></td>
</tr>
<tr>
<td>20°</td>
<td>45 cd/m²</td>
<td></td>
</tr>
<tr>
<td>10°</td>
<td>5 cd/m²</td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>Elevation</th>
<th>(yellow light)</th>
</tr>
</thead>
<tbody>
<tr>
<td>-180°</td>
<td>5 cd/m²</td>
</tr>
<tr>
<td>180°</td>
<td></td>
</tr>
</tbody>
</table>

#### Illustration 7 — Touchdown and lift-off area electro-luminescent panels
Chapter 5

5.3.4.4 The light units shall be frangible and mounted as low as possible.

5.3.4.5 Where the lights of the system need to be seen as discrete sources, light units shall be located such that at the extremes of system coverage the angle subtended between units as seen by the pilot shall not be less than 3 minutes of arc.

5.3.4.6 The angles subtended between light units of the system and other units of comparable or greater intensities shall also be not less than 3 minutes of arc.

Note.—Requirements of 5.3.4.5 and 5.3.4.6 can be met for lights on a line normal to the line of sight if the light units are separated by 1 metre for every kilometre of viewing range.

Signal format

5.3.4.7 The signal format of the alignment guidance system shall include a minimum of three discrete signal sectors providing "offset to the right", "on track" and "offset to the left" signals.

5.3.4.8 The divergence of the "on track" sector of the system shall be as shown in Figure 5-11.

5.3.4.9 The signal format shall be such that there is no possibility of confusion between the system and any associated visual approach slope indicator or other visual aids.

5.3.4.10 The system shall avoid the use of the same coding as any associated visual approach slope indicator.

Figure 5-10. Approach lighting system

Figure 5-11. Divergence of the "on track" sector
Annex 14 — Aerodromes

5.3.4.11 The signal format shall be such that the system is unique and conspicuous in all operational environments.

5.3.4.12 The system shall not significantly increase the pilot workload.

Light distribution

5.3.4.13 The useable coverage of the visual alignment guidance system shall be equal to or better than that of the visual approach slope indicator system, with which it is associated.

5.3.4.14 A suitable intensity control shall be provided so as to allow adjustment to meet the prevailing conditions and to avoid dazzling the pilot during approach and landing.

Approach track and azimuth setting

5.3.4.15 A visual alignment guidance system shall be capable of adjustment in azimuth to within ± 5 minutes of arc of the desired approach path.

5.3.4.16 The angle of azimuth guidance system shall be such that during an approach the pilot of a helicopter at the boundary of the “on track” signal will clear all objects in the approach area by a safe margin.

5.3.4.17 The characteristics of the obstacle protection surface specified in 5.3.5.23, Table 5-1 and Figure 5-13 shall equally apply to the system.

Characteristics of the visual alignment guidance system

5.3.4.18 In the event of the failure of any component affecting the signal format the system shall be automatically switched off.

5.3.4.19 The light units shall be so designed that deposits of condensation, ice, dirt, etc. on optically transmitting or reflecting surfaces will interfere to the least possible extent with the light signal and will not cause spurious or false signals to be generated.

5.3.5 Visual approach slope indicator

Application

5.3.5.1 Recommendation.— A visual approach slope indicator should be provided to serve the approach to a heliport, whether or not the heliport is served by other visual approach aids or by non-visual aids, where one or more of the following conditions exist especially at night:

a) obstacle clearance, noise abatement or traffic control procedures require a particular slope to be flown;

b) the environment of the heliport provides few visual surface cues; and

c) the characteristics of the helicopter require a stabilized approach.

5.3.5.2 The standard visual approach slope indicator systems for helicopter operations shall consist of the following:

a) PAPI and APAPI systems conforming to the specifications contained in Annex 14, Volume I, 5.3.5.23 to 5.3.5.40 inclusive except that the angular size of the on-slope sector of the systems shall be increased to 45 minutes; or

b) helicopter approach path indicator (HAPI) system conforming to the specifications in 5.3.5.6 to 5.3.5.21 inclusive.

Location

5.3.5.3 A visual approach slope indicator shall be located such that a helicopter is guided to the desired position within the final approach and take-off area and so as to avoid dazzling the pilot during final approach and landing.

5.3.5.4 Recommendation.— A visual approach slope indicator should be located adjacent to the nominal aiming point and aligned in azimuth with the preferred approach direction.

5.3.5.5 The light unit(s) shall be frangible and mounted as low as possible.

HAPI signal format

5.3.5.6 The signal format of the HAPI shall include four discrete signal sectors, providing an “above slope”, an “on slope”, a “slightly below” and a “below slope” signal.

5.3.5.7 The signal format of the HAPI shall be as shown in Figure 5-12, Illustrations A and B.

Note.— Care is required in the design of the unit to minimize spurious signals between the signal sectors and at the azimuth coverage limits.

5.3.5.8 The signal repetition rate of the flashing sector of the HAPI shall be at least 2 Hz.

5.3.5.9 Recommendation.— The on-to-off ratio of pulsing signals of the HAPI should be 1 to 1 and the modulation depth should be at least 80 per cent.
Chapter 5

5.3.5.10 The angular size of the “on-slope” sector of the HAPI shall be 45 minutes.

5.3.5.11 The angular size of the “slightly below” sector of the HAPI shall be 15 minutes.

Light distribution

5.3.5.12 **Recommendation.**— The light intensity distribution of the HAPI in red and green colours should be as shown in Figure 5-9, Illustration 4.

Note.— A larger azimuth coverage can be obtained by installing the HAPI system on a turntable.

5.3.5.13 Colour transition of the HAPI in the vertical plane shall be such as to appear to an observer at a distance of not less than 300 m to occur within a vertical angle of not more than three minutes.

5.3.5.14 The transmission factor of a red or green filter shall be not less than 15 per cent at the maximum intensity setting.

5.3.5.15 At full intensity the red light of the HAPI shall have a Y-coordinate not exceeding 0.320 and the green light shall be within the boundaries specified in Annex 14, Volume I, Appendix 1, 2.1.3.

5.3.5.16 A suitable intensity control shall be provided so as to allow adjustment to meet the prevailing conditions and to avoid dazzling the pilot during approach and landing.

Approach slope and elevation setting

5.3.5.17 A HAPI system shall be capable of adjustment in elevation at any desired angle between 1 degree and 12 degrees above the horizontal with an accuracy of ±5 minutes of arc.

5.3.5.18 The angle of elevation setting of HAPI shall be such that during an approach, the pilot of a helicopter observing the upper boundary of the “below slope” signal will clear all objects in the approach area by a safe margin.

Characteristics of the light unit

5.3.5.19 The system shall be so designed that:

a) in the event the vertical misalignment of a unit exceeds ±0.5" (±30 minutes), the system will switch off automatically; and

b) if the flashing mechanism fails, no light will be emitted in the failed flashing sector(s).

<table>
<thead>
<tr>
<th>Surface and Dimensions</th>
<th>Non-Instrument FATO</th>
<th>Non-Precision FATO</th>
</tr>
</thead>
<tbody>
<tr>
<td>Length of inner edge</td>
<td>Width of safety area</td>
<td>Width of safety area</td>
</tr>
<tr>
<td>Distance from end of FATO</td>
<td>3 m minimum</td>
<td>60 m</td>
</tr>
<tr>
<td>Divergence</td>
<td>10%</td>
<td>15%</td>
</tr>
<tr>
<td>Total length</td>
<td>2 500 m</td>
<td>2 500 m</td>
</tr>
<tr>
<td>Slope</td>
<td>PAPI A° - 0.57&quot;</td>
<td>A° - 0.57&quot;</td>
</tr>
<tr>
<td></td>
<td>HAPI A° - 0.65&quot;</td>
<td>A° - 0.65&quot;</td>
</tr>
<tr>
<td></td>
<td>APAPI A° - 0.9&quot;</td>
<td>A° - 0.9&quot;</td>
</tr>
</tbody>
</table>

a. As indicated in Annex 14, Volume I, Figure 5-13.

b. The angle of the upper boundary of the “below slope” signal.
## Annex 14 — Aerodromes

### Sector Format

<table>
<thead>
<tr>
<th>Sector</th>
<th>Format</th>
</tr>
</thead>
<tbody>
<tr>
<td>Above</td>
<td>Flashing green</td>
</tr>
<tr>
<td>On slope</td>
<td>Green</td>
</tr>
<tr>
<td>Slightly below</td>
<td>Red</td>
</tr>
<tr>
<td>Below</td>
<td>Flashing red</td>
</tr>
</tbody>
</table>

#### Figure 5-12. HAPI signal format

![Figure 5-12. HAPI signal format](image)

#### Figure 5-13. Obstacle protection surface for visual approach slope indicator systems

![Figure 5-13. Obstacle protection surface for visual approach slope indicator systems](image)

9/11/95
Chapter 5

5.3.5.20 The light unit of the HAPI shall be so designed that deposits of condensation, ice, dirt, etc. on optically transmitting or reflecting surfaces will interfere to the least possible extent with the light signal and will not cause spurious or false signals to be generated.

5.3.5.21 **Recommendation.**— A HAPI system intended for installation on a floating helideck should afford a stabilization of the beam to an accuracy of ± 1/4° within ± 3° pitch and roll movement of the heliport.

Obstacle protection surface

**Note.**— The following specifications apply to PAPI, APAPI and HAPI.

5.3.5.22 An obstacle protection surface shall be established when it is intended to provide a visual approach slope indicator system.

5.3.5.23 The characteristics of the obstacle protection surface, i.e. origin, divergence, length and slope shall correspond to those specified in the relevant column of Table 5-1 and in Figure 5-13.

5.3.5.24 New objects or extensions of existing objects shall not be permitted above an obstacle protection surface except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object.

**Note.**— Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6.

5.3.5.25 Existing objects above an obstacle protection surface shall be removed except when, in the opinion of the appropriate authority, the object is shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the safety of operations of helicopters.

5.3.5.26 Where an aeronautical study indicates that an existing object extending above an obstacle protection surface could adversely affect the safety of operations of helicopters one or more of the following measures shall be taken:

a) suitably raise the approach slope of the system;

b) reduce the azimuth spread of the system so that the object is outside the confines of the beam;

c) displace the axis of the system and its associated obstacle protection surface by no more than 5°;

d) suitably displace the final approach and take-off area; and

e) install a visual alignment guidance system specified in 5.3.4.

**Note.**— Guidance on this issue is contained in the Heliport Manual.

5.3.6 Final approach and take-off area lights

**Application**

5.3.6.1 Where a final approach and take-off area is established at a surface level heliport on ground intended for use at night, final approach and take-off area lights shall be provided except that they may be omitted where the final approach and take-off area and the touchdown and lift-off area are nearly coincidental or the extent of the final approach and take-off area is self-evident.

**Location**

5.3.6.2 Final approach and take-off area lights shall be placed along the edges of the final approach and take-off area. The lights shall be uniformly spaced as follows:

a) for an area in the form of a square or rectangle, at intervals of not more than 50 m with a minimum of four lights on each side including a light at each corner; and

b) for any other shaped area, including a circular area, at intervals of not more than 5 m with a minimum of ten lights.

**Characteristics**

5.3.6.3 Final approach and take-off area lights shall be fixed omnidirectional lights showing white. Where the intensity of the lights is to be varied the lights shall show variable white.

5.3.6.4 **Recommendation.**— The light distribution of final approach and take-off area lights should be as shown in Figure 5-9, Illustration 5.

5.3.6.5 **Recommendation.**— The lights should not exceed a height of 25 cm and should be inset when a light extending above the surface would endanger helicopter operations. Where a final approach and take-off area is not meant for lift-off or touchdown, the lights should not exceed a height of 25 cm above ground or snow level.
Annex 14 — Aerodromes

5.3.7 Aiming point lights

Application

5.3.7.1 Recommendation.— Where an aiming point marking is provided at a heliport intended for use at night, aiming point lights should be provided.

Location

5.3.7.2 Aiming point lights shall be collocated with the aiming point marking.

Characteristics

5.3.7.3 Aiming point lights shall form a pattern of at least six omnidirectional white lights as shown in Figure 5-4. The lights shall be inset when a light extending above the surface could endanger helicopter operations.

5.3.7.4 Recommendation.— The light distribution of aiming point lights should be as shown in Figure 5-9, Illustration 5.

5.3.8 Touchdown and lift-off area lighting system

Application

5.3.8.1 A touchdown and lift-off area lighting system shall be provided at a heliport intended for use at night.

5.3.8.2 The touchdown and lift-off area lighting system for a surface level heliport shall consist of one or more of the following:

a) perimeter lights; or

b) floodlighting; or

c) luminescent panel lighting when a) and b) are not practicable and final approach and take-off area lights are available.

5.3.8.3 The touchdown and lift-off area lighting system for an elevated heliport or helideck shall consist of:

a) perimeter lights; and

b) floodlighting and/or luminescent panel lighting.

Note.— At elevated heliports and helidecks, surface texture cues within the touchdown and lift-off area are essential for helicopter positioning during the final approach and landing.

Such cues are provided by using floodlighting or luminescent panel lighting or a combination of these two forms of lighting, in addition to perimeter lights.

5.3.8.4 Recommendation.— Touchdown and lift-off area floodlighting or luminescent panel lighting should be provided at a surface-level heliport intended for use at night when enhanced surface texture cues are required.

Location

5.3.8.5 Touchdown and lift-off area perimeter lights shall be placed along the edge of the area designated for use as the touchdown and lift-off area or within a distance of 1.5 m from the edge. Where the touchdown and lift-off area is a circle the lights shall be:

a) located on straight lines in a pattern which will provide information to pilots on drift displacement; and

b) where a) is not practicable, evenly spaced around the perimeter of the touchdown and lift-off area at the appropriate interval except that over a sector of 45° the lights shall be spaced at half spacing.

5.3.8.6 Touchdown and lift-off area perimeter lights shall be uniformly spaced at intervals of not more than 3 m for elevated heliports and helidecks and not more than 5 m for surface level heliports. There shall be a minimum number of four lights on each side including a light at each corner. For a circular touchdown and lift-off area, where lights are installed in accordance with 5.3.8.5 b) there shall be a minimum of fourteen lights.

Note.— Guidance on this issue is contained in the Heliport Manual.

5.3.8.7 The touchdown and lift-off area perimeter lights shall be installed at an elevated heliport or fixed helideck such that the pattern cannot be seen by the pilot from below the elevation of the touchdown and lift-off area.

5.3.8.8 The touchdown and lift-off area perimeter lights shall be installed at a floating helideck, such that the pattern cannot be seen by the pilot from below the elevation of the touchdown and lift-off area.

5.3.8.9 On surface level heliports, luminescent panel lights shall be placed along the marking designating the edge of the touchdown and lift-off area. Where the touchdown and lift-off area is a circle the luminescent panels shall be located on straight lines circumscribing the area.

5.3.8.10 On surface level heliports the minimum number of panels on a touchdown and lift-off area shall be nine. The total length of luminescent panels in a pattern shall not be less than 50 per cent of the length of the pattern. There shall be an odd number with a minimum number of three panels on each
Chapter 5

side of the touchdown and lift-off area including a panel at each corner. Luminescent panels shall be uniformly spaced with a distance between adjacent panel ends of not more than 5 m on each side of the touchdown and lift-off area.

5.3.8.11 **Recommendation.**—When luminescent panels are used on an elevated heliport or helideck to enhance surface texture cues the panels should not be placed adjacent to the perimeter lights. They should be placed around a touchdown marking where it is provided or coincident with heliport identification marking.

5.3.8.12 Touchdown and lift-off area floodlights shall be located so as to avoid glare to pilots in flight or to personnel working on the area. The arrangement and aiming of floodlights shall be such that shadows are kept to a minimum.

**Characteristics**

5.3.8.13 The touchdown and lift-off area perimeter lights shall be fixed omnidirectional lights showing yellow.

5.3.8.14 At a surface level heliport the luminescent panels shall emit yellow light when used to define the boundary of the touchdown and lift-off area.

**Note.**—In other circumstances, luminescent panels may emit light of other colours.

5.3.8.15 **Recommendation.**—The chromaticity and luminance of colours of luminescent panels should conform to Annex 14, Volume I, Appendix 1, 3.4.

5.3.8.16 A luminescent panel shall have a minimum width of 6 cm. The panel housing shall be the same colour as the marking it defines.

5.3.8.17 **Recommendation.**—The perimeter lights should not exceed a height of 25 cm and should be inset when a light extending above the surface could endanger helicopter operations.

5.3.8.18 **Recommendation.**—The touchdown and lift-off area floodlights should not exceed a height of 25 cm.

5.3.8.19 The luminescent panels shall not extend above the surface by more than 2.5 cm.

5.3.8.20 **Recommendation.**—The light distribution of the perimeter lights should be as shown in Figure 5-9, Illustration 6.

5.3.8.21 **Recommendation.**—The light distribution of the luminescent panels should be as shown in Figure 5-9, Illustration 7.

Annex 14 — Aerodromes

5.3.8.22 The spectral distribution of touchdown and lift-off area floodlights shall be such that the surface and obstacle marking can be correctly identified.

5.3.8.23 **Recommendation.**—The average horizontal illuminance of the floodlighting should be at least 10 lux, with a uniformity ratio (average to minimum) of not more than 8:1 measured on the surface of the touchdown and lift-off area.

5.3.9 Winching area floodlighting

**Application**

5.3.9.1 Winching area floodlighting shall be provided at a winching area intended for use at night.

**Location**

5.3.9.2 Winching area floodlights shall be located so as to avoid glare to pilots in flight or to personnel working on the area. The arrangement and aiming of floodlights shall be such that shadows are kept to a minimum.

**Characteristics**

5.3.9.3 The spectral distribution of winching area floodlights shall be such that the surface and obstacle markings can be correctly identified.

5.3.9.4 **Recommendation.**—The average horizontal illuminance should be at least 10 lux, measured on the surface of the winching area.

5.3.10 Taxiway lights

**Note.**—The specifications for taxiway centre line lights and taxiway edge lights in Annex 14, Volume I, 5.3.15 and 5.3.16 are equally applicable to taxiways intended for ground taxiing of helicopters.

5.3.11 Visual aids for denoting obstacles

**Note.**—The specifications for marking and lighting of obstacles included in Annex 14, Volume I, Chapter 6, are equally applicable to heliports and winching areas.

5.3.12 Floodlighting of obstacles

**Application**

5.3.12.1 At a heliport intended for use at night, obstacles shall be floodlighted if it is not possible to display obstacle lights on them.

47

9/11/95
Annex 14 — Aerodromes

Location

5.3.12.2 Obstacle floodlights shall be arranged so as to illuminate the entire obstacle and as far as practicable in a manner so as not to dazzle the helicopter pilots.

Characteristics

5.3.12.3 **Recommendation.** Obstacle floodlighting should be such as to produce a luminance of at least 10 cd/m².
CHAPTER 6. HELIPORT SERVICES

6.1 Rescue and fire fighting

General

Introductory Note.— These specifications apply to surface level heliports and elevated heliports only. The specifications complement those in Annex 14, Volume I, 9.2 concerning rescue and fire fighting requirements at aerodromes.

The principal objective of a rescue and fire fighting service is to save lives. For this reason, the provision of means of dealing with a helicopter accident or incident occurring at or in the immediate vicinity of a heliport assumes primary importance because it is within this area that there are the greatest opportunities of saving lives. This must assume at all times the possibility of, and need for, extinguishing a fire which may occur either immediately following a helicopter accident or incident or at any time during rescue operations.

The most important factors bearing on effective rescue in a survivable helicopter accident are the training received, the effectiveness of the equipment and the speed with which personnel and equipment designated for rescue and fire fighting purposes can be put into use.

For an elevated heliport, requirements to protect any building or structure on which the heliport is located are not taken into account.

Rescue and fire fighting requirements for helidecks may be found in the Heliport Manual.

Level of protection to be provided

6.1.1 Recommendation.— The level of protection to be provided for rescue and fire fighting should be based on the over-all length of the longest helicopter normally using the heliport and in accordance with the heliport fire fighting category determined from Table 6-1, except at an unattended heliport with a low movement rate.

Note.— Guidance to assist the appropriate authority in providing rescue and fire fighting equipment and services at surface-level and elevated heliports is given in the Heliport Manual.

6.1.2 Recommendation.— During anticipated periods of operations by smaller helicopters, the heliport fire fighting category may be reduced to that of the highest category of helicopter planned to use the heliport during that time.

Extinguishing agents

6.1.3 Recommendation.— The principal extinguishing agent should be a foam meeting the minimum performance level B.

Note.— Information on the required physical properties and fire extinguishing performance criteria needed for a foam to achieve an acceptable performance level B rating is given in the Airport Services Manual, Part 1.

6.1.4 Recommendation.— The amounts of water for foam production and the complementary agents to be provided should be in accordance with the heliport fire fighting category determined under 6.1.1 and Table 6-2 or Table 6-3 as appropriate.

Note.— The amounts of water specified for elevated heliports do not have to be stored on or adjacent to the heliport if there is a suitable adjacent pressurized water main system capable of sustaining the required discharge rate.

6.1.5 Recommendation.— At a surface-level heliport it is permissible to replace all or part of the amount of water for foam production by complementary agents.

6.1.6 Recommendation.— The discharge rate of the foam solution should not be less than the rates shown in Table 6-2 or Table 6-3 as appropriate. The discharge rate of complementary agents should be selected for optimum effectiveness of the agent used.

6.1.7 Recommendation.— At an elevated heliport, at least one hose spray line capable of delivering foam in a jet

Table 6-1. Heliport fire fighting category

<table>
<thead>
<tr>
<th>Category</th>
<th>Helicopter over-all lengtha</th>
</tr>
</thead>
<tbody>
<tr>
<td>H1</td>
<td>up to but not including 15 m</td>
</tr>
<tr>
<td>H2</td>
<td>from 15 m up to but not including 24 m</td>
</tr>
<tr>
<td>H3</td>
<td>from 24 m up to but not including 35 m</td>
</tr>
</tbody>
</table>

Note. a. Helicopter length, including the tail boom and the rotors.
spray pattern at 250 L/min should be provided. Additionally at elevated heliports in categories 2 and 3, at least two monitors should be provided each having a capability of achieving the required discharge rate and positioned at different locations around the heliports so as to ensure the application of foam to any part of the heliport under any weather condition and to minimize the possibility of both monitors being impaired by a helicopter accident.

**Rescue equipment**

6.1.8 **Recommendation.**—At an elevated heliport rescue equipment should be stored adjacent to the heliport.

Note.— Guidance on the rescue equipment to be provided at a heliport is given in the Heliport Manual.

**Response time**

6.1.9 **Recommendation.**—At a surface-level heliport, the operational objective of the rescue and fire fighting service should be to achieve response times not exceeding two minutes in optimum conditions of visibility and surface conditions.

Note.—Response time is considered to be the time between the initial call to the rescue and fire fighting service and the time when the first responding vehicle(s) (the service) is (are) in position to apply foam at a rate of at least 50 per cent of the discharge rate specified in Table 6-2.

6.1.10 **Recommendation.**—At an elevated heliport, the rescue and fire fighting service should be immediately available on or in the vicinity of the heliport while helicopter movements are taking place.

### Table 6-2. Minimum usable amounts of extinguishing agents for surface level heliports

<table>
<thead>
<tr>
<th>Category</th>
<th>Water (L)</th>
<th>Discharge rate foam solution (L/min)</th>
<th>Dry chemical powders (kg)</th>
<th>Complementary agents</th>
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<tr>
<td>H1</td>
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<td>250</td>
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<td>H2</td>
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<tr>
<td>H3</td>
<td>1 600</td>
<td>800</td>
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### Table 6-3. Minimum usable amounts of extinguishing agents for elevated heliports

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<th>Discharge rate foam solution (L/min)</th>
<th>Dry chemical powders (kg)</th>
<th>Complementary agents</th>
</tr>
</thead>
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<tr>
<td>H1</td>
<td>2 500</td>
<td>250</td>
<td>45</td>
<td>45</td>
</tr>
<tr>
<td>H2</td>
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<tr>
<td>H3</td>
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<td>800</td>
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## APPENDIX 1. AERONAUTICAL DATA QUALITY REQUIREMENTS

Table 1. Latitude and longitude

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<td>30 m</td>
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<td>NAVAIDS located at the heliport</td>
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</tr>
<tr>
<td>Obstacles in the circling area and at the heliport</td>
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<td>essential</td>
<td>$1 \times 10^{-5}$</td>
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<tr>
<td>Significant obstacles in the approach and take-off area</td>
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<td>essential</td>
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</tr>
<tr>
<td>Geometric centre of TLOF or FATO thresholds</td>
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<td>critical</td>
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<tr>
<td>Ground taxiway centre line points, air taxiways and transit routes points</td>
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<tr>
<td>Helicopter stand-points/INS check-points</td>
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<td>essential</td>
<td>$1 \times 10^{-5}$</td>
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<tr>
<td>Obstacles in the approach and take-off areas</td>
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<td>Obstacles in the circling areas and at the heliport</td>
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<td>Distance measuring equipment/precision (DME/P)</td>
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### Table 3. Declination and magnetic variation

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<tr>
<td>MLS azimuth antenna magnetic variation</td>
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### Appendix Annex 14 — Aerodromes

#### Table 4. Bearing

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<td>surveyed</td>
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<td>MLS zero azimuth alignment</td>
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<td></td>
<td>surveyed</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>FATO bearing</td>
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#### Table 5. Length/distance/dimension

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<td>ILS markers-threshold distance</td>
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<td>essential</td>
<td>$1 \times 10^{-5}$</td>
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</tr>
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<td>ILS DME antenna-threshold,</td>
<td>3 m (10 ft)</td>
<td>calculated</td>
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<td>distance along centre line</td>
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<tr>
<td>MLS azimuth antenna-FATO end, distance</td>
<td>3 m (10 ft)</td>
<td>calculated</td>
<td>routine</td>
<td>$1 \times 10^{-3}$</td>
</tr>
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<tr>
<td>MLS elevation antenna-threshold,</td>
<td>3 m (10 ft)</td>
<td>calculated</td>
<td>routine</td>
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<td>3 m (10 ft)</td>
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<td>distance along centre line</td>
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Supplement Note

Supplement to

Annex 14 — Aerodromes

Volume II — Heliports

(Second Edition)

1. The attached Supplement supersedes all previous Supplements to Annex 14, Volume II, and includes differences notified by Contracting States up to 21 January 1997.

2. This Supplement should be inserted at the end of Annex 14, Volume II, Second Edition. Additional differences and revised comments received from Contracting States will be issued at intervals as amendments to this Supplement.
SUPPLEMENT TO
ANNEX 14 — AERODROMES
VOLUME II — HELIPORTS
(Second Edition)

Differences between the national regulations and practices of States and the corresponding International Standards contained in Annex 14, Volume II, as notified to ICAO in accordance with Article 38 of the Convention on International Civil Aviation and the Council’s resolution of 21 November 1950.

JANUARY 1997

INTERNATIONAL CIVIL AVIATION ORGANIZATION
### RECORD OF AMENDMENTS TO SUPPLEMENT

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### RECORD OF AMENDMENTS TO ANNEX 14, VOLUME II SUBSEQUENT TO SECOND EDITION ISSUED JULY 1995

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1. Contracting States which have notified ICAO of differences

The Contracting States listed below have notified ICAO of differences which exist between their national regulations and practices and the International Standards and Recommended Practices of Annex 14, Volume II, Second Edition, or have commented on implementation.

The page numbers shown for each State and the dates of publication of those pages correspond to the actual pages in this Supplement.

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2. Contracting States which have notified ICAO that no differences exist

Bahrain  Kyrgyzstan  Portugal
Barbados  Namibia  Switzerland
China (Hong Kong SAR)  Niger  Tunisia
Finland  Oman  United Kingdom
Ireland  Pakistan  United Republic of Tanzania
Jordan  Peru  Uruguay

3. Contracting States from which no information has been received

Afghanistan  Bhutan  Chad
Albania  Bolivia  Colombia
Algeria  Bosnia and Herzegovina  Comoros
Angola  Botswana  Congo
Antigua and Barbuda  Brazil  Cook Islands
Armenia  Brunei Darussalam  Costa Rica
Azerbaijan  Bulgaria  Côte d’Ivoire
Bahamas  Burkina Faso  Croatia
Bangladesh  Burundi  Cuba
Belarus  Cambodia  Cyprus
Belgium  Cameroon  Czech Republic
Belize  Cape Verde  Democratic People’s Republic
Benin  Central African Republic  of Korea
Democratic Republic of the Congo | Latvia | Saint Lucia  
---|---|---  
Denmark | Lebanon | Saint Vincent and the Grenadines  
Djibouti | Lesotho | Samoa  
Dominican Republic | Liberia | San Marino  
Ecuador | Libyan Arab Jamahiriya | Sao Tome and Principe  
Egypt | Lithuania | Saudi Arabia  
El Salvador | Luxembourg | Senegal  
Equatorial Guinea | Madagascar | Seychelles  
Eritrea | Malawi | Sierra Leone  
Estonia | Malaysia | Singapore  
Ethiopia | Maldives | Slovakia  
Fiji | Mali | Slovenia  
Gabon | Malta | Solomon Islands  
Gambia | Marshall Islands | Somalia  
Georgia | Mauritania | South Africa  
Ghana | Mauritius | Sri Lanka  
Grenada | Mexico | Sudan  
Guatemala | Micronesia (Federated States of) | Suriname  
Guinea | Monaco | Swaziland  
Guinea-Bissau | Mongolia | Syrian Arab Republic  
Guyana | Morocco | Tajikistan  
Haiti | Mozambique | Thailand  
Honduras | Myanmar | The former Yugoslav Republic  
        of Macedonia  
Hungary | Nauru | Togo  
Iceland | Nepal | Tonga  
India | Nicaragua | Trinidad and Tobago  
Indonesia | Nigeria | Turkey  
Iran (Islamic Republic of) | Palau | Turkmenistan  
Iraq | Panama | Uganda  
Israel | Paraguay | Ukraine  
Italy | Philippines | United States  
Jamaica | Poland | Uzbekistan  
Japan | Qatar | Vanuatu  
Kazakhstan | Republic of Korea | Venezuela  
Kenya | Republic of Moldova | Viet Nam  
Kiribati | Romania | Yemen  
Kuwait | Russian Federation | Zambia  
Lao People’s Democratic Republic | Rwanda | Zimbabwe
4. Paragraphs with respect to which differences have been notified

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Appendix 1
CHAPTER 2

2.1.7 The geoid undulation is not provided.
2.3.1
2.3.2 Remark: It is not possible to provide it with the required precision.
2.4.1 g) The ground profile in clearways is not always provided.

Remark: It is not possible to provide such information until topographic measurements are made.

APPENDIX 1

Table 1 The geographical coordinates of the obstacles in the circling area and at the heliport are not provided, nor are those of the significant obstacles in the approach and take-off area.

Remark: It is not possible at present to provide such information.

Table 2 The geoid undulation is not provided at the heliport elevation position, at the FATO threshold and at the geometric centre of the TLOF for non-precision approaches; at the FATO threshold and at the geometric centre of the TLOF for precision approaches.

Remark: It is not possible to provide it with the required precision.

The elevation/altitude/height of the distance measuring equipment/precision (DME/P) is not provided.

Remark: It is not possible to provide such information until topographic measurements are made.

The integrity and classification of the aeronautical data are not provided.

Remark: There is no electronic database available.

Comment on implementation:

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CHAPTER 3

3.1.3 The overall slope is not to exceed 7.5 degrees.

3.1.11 The landing and lift-off area (LLA equivalent to TLOF) should have an area equal in size to the undercarriage contact points plus one metre on all sides.

3.1.12 The overall slope of the LLA, in any direction, should not exceed the slope landing capability of the helicopter.

3.1.14 Australian guidelines do not require a safety area.

Remark: Australian FATO is 2 times the overall length/width in lieu of 1.5 times the overall length/width as required by Annex 14, Volume II.

3.1.34 Australian guidelines do not specify the dimensions of an air transit route.

CHAPTER 5

5.3.3.2 The direction of approach should be indicated by at least two omnidirectional green lights or by one white lead-in light.

5.3.6.2 The edge of the FATO should be defined by omnidirectional white lights spaced not more than eight metres apart or by a combination of markings and floodlighting. Where this is not practicable, the GEA should be so defined.

*Recommended Practice
CHAPTER 2

2.1.1 The geographical coordinates are reported in Austria in degrees, minutes, seconds. Geographical coordinates with accuracy as defined in Annex 14 will be reported with application of the WGS-84 latest at 1 January 1998.

CHAPTER 3

3.1.2 For designing and classifying heliports no distinction is made between performance classes of helicopters. FATOs are divided into three classes by minimum length (diameter) requirement.

3.1.3 For designing and classifying heliports no distinction is made between performance classes of helicopters. FATOs are divided into three classes by minimum length (diameter) requirement.

3.2.2 For designing and classifying heliports no distinction is made between performance classes of helicopters. FATOs are divided into three classes by minimum length (diameter) requirement.
CHAPTER 3

3.1.37 The minimum clearance between a helicopter stand and an object or other aircraft stand may be reduced to 3 metres in Canada.

3.2.1 Canada makes provision for performance class 3 helicopter operations at elevated heliports. The dimensions of the FATO at elevated heliports for performance class 3 helicopters is the same as that used for performance class 2 helicopters.

3.2.2 b) Canada makes provision for performance class 3 helicopter operations at elevated heliports. The dimensions of the FATO at elevated heliports for performance class 3 helicopters is the same as that used for performance class 2 helicopters.

3.2.5 Canada does not require a safety area around the FATO of elevated heliports.

3.3.2 The FATO for single main rotor helicopters operating on helidecks within the inland waters of Canada shall be of sufficient size to contain a circle of a diameter not less than the main rotor diameter of the design helicopter.

3.4.1 The Canadian standards for the size of a FATO located at the bow or stern of a vessel for a single main rotor helicopter requires the FATO to be of sufficient size to contain a circle with a diameter not less than the main rotor diameter of the design helicopter.

CHAPTER 5

5.2.3.3 Canada indicates the maximum allowable mass markings in thousands of pounds on the touchdown and lift-off area of elevated heliports and helidecks.

5.3.6.2 b) Canada requires a minimum of five lights to mark a circular FATO.

5.3.8.13 Canada permits the use of retro-reflective markers as the minimum lighting requirements at remote heliports where it is impractical to provide lighting.
CHAPTER 2

2.3.1 The heliport elevation shall be measured and reported to the aeronautical information services authority to the accuracy of one-half metre or foot.

Remark: Chilean authorities responsible for providing geodetic data have not yet determined WGS-84 vertical reference data sufficient to obtain the geoid undulation. Therefore, once these authorities have provided that information, we shall be in a position to make the corresponding publications.

Note.— In view of the above, we are still waiting to adopt the Standards in which subjects are mentioned relating to geoid elevation data, for example, paragraphs 2.1.7, 2.3.2, etc.
CHAPTER 6

6.1 In France the minimum level of protection is assured at surface level heliports by a quantity of 50 kg of powder or equivalent and at elevated heliports by a quantity of 250 kg of powder or equivalent.
CHAPTER 2

2.1.2 Full implementation of the quality system can only be achieved by the date indicated.

2.1.3

2.1.4*

2.1.5 Publishing of the heliport reference points in WGS-84 will be an ongoing task for some time after the date of applicability of Amendment 2.

2.1.6

APPENDIX 1

Table 1 In Germany the description of obstacles differs as follows from what is given in these tables.

Table 2

a) Obstacles in the circling area for non-precision and turning departures and at the heliport.

b) Significant obstacles in the precision approach and straight departure area.

Table 2 The WGS-84 geoid undulation at heliport elevation position will not be published in Germany.

Remark: This item is not considered to be required for VFR heliports. As concerns potential German IFR heliports, the WGS-84 geoid undulation would always be published for the FATO threshold, TLOF geometric centre. This would even apply for non-precision approaches because the MDH is also referred to this position.

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*Recommended Practice
CHAPTER 5

5.3.3.4* The approach lighting system provided for a non-precision final approach and take-off area is 90 metres in length.

5.3.3.6* The installation of steady lights of the heliports regarding their light distribution and intensity has been realized in compliance with the previous editions of Annex 14, Volume II.
CHAPTER 2

2.1.2 b) and c) In the Netherlands it is not yet considered necessary to determine the geographical coordinates of the geometric centre of the touchdown and lift-off area, thresholds of the final approach and take-off area, centre line points of the helicopter ground taxiways, air taxiways and air transit routes and helicopter stands in terms of the World Geodetic System — 1984 (WGS-84) geodetic reference datum.

CHAPTER 3

3.1.18 The height limitation of 25 cm applies for helidecks and elevated heliports only. For ground level heliports the maximum height of objects permitted in the safety area shall not exceed 35 cm.

3.1.23 The separation distances between:
— a helicopter-ground-taxiway and object shall not be less than 1x the greatest overall width of a helicopter with rotor turning (centre line to object);
— a helicopter-ground-taxiway and a helicopter stand shall not be less than 1x the greatest overall width of a helicopter with rotor turning (centre line to edge).

3.1.33 The separation distance between:
— an air-taxiway and another air-taxiway shall not be less than 3x the greatest width of a helicopter with rotor turning (between centre lines);
— an air-taxiway and a helicopter-ground-taxiway shall not be less than 3x the greatest width of a helicopter with rotor turning (between centre lines);
— an air-taxiway and a helicopter stand shall not be less than 1.5x the greatest width of a helicopter with rotor turning (centre line to edge).

3.1.39 The separation distance between the edge of a runway strip and the edge of a FATO shall not be less than 2x the overall length of the largest helicopter the FATO is intended to serve.

CHAPTER 4

4.2.5 The slopes for visual approach and visual take-off for Class 2 and Class 3 helicopters for the first section shall be 12.5 per cent.

CHAPTER 5

5.2.4.4 The height of the marker shall not exceed 35 cm above ground or snow level.
CHAPTER 1

1.2 New Zealand has no heliports intended to be used by helicopters in international civil aviation.

New Zealand Civil Aviation Rules Part 139 prescribe rules governing the certification and operation of aerodromes and rules for operators of aircraft using aerodromes.

No person shall operate an aerodrome serving any aeroplane having a certified seating capacity of more than 30 passengers that is engaged in regular air transport operations except under the authority of, and in accordance with the provisions of, an aerodrome operating certificate issued for that aerodrome under New Zealand Civil Aviation Rules Part 139.

Remark: An aerodrome operator who is not required to hold an aerodrome operating certificate may apply for an aerodrome operating certificate.

New Zealand heliports are not required to be certificated.

Except for a person operating a helicopter on an external load operation, no person operating a helicopter shall use any place within a populous area as a heliport unless the heliport has physical characteristics, obstacle limitation surfaces and visual aids commensurate with the characteristics of the helicopter being operated and the ambient light conditions during operations, and the heliport is clear of all persons, animals, vehicles or other obstructions during the hover, touchdown or lift-off other than persons and vehicles essential to the operation, and the selected approach and take-off paths are such that, if the helicopter is not a performance Class 1 helicopter, an autorotative landing can be conducted without any undue risk to any person on the ground, and the helicopter can be manoeuvred in the aerodrome traffic circuit clear of any obstructions, and not in conflict with the aerodrome traffic circuit or instrument approach of any other aerodrome.

Remark: New Zealand CAA Advisory Circular AC139-08 contains heliport design standards for heliports in populous areas that are acceptable to the Director.
CHAPTER 5

5.2.2.3 Markings shall be yellow in colour instead of white in colour.

5.2.2.4

5.2.7.4

*Remark:* Yellow markings are used due to the need for improved visual references during the winter season when the heliport identifications are covered with ice and snow.

5.2.12.2 Air transit route markings are located along the edge of the air transit route only.

5.2.13.2

*Remark:* The reason for this is due to the possible damage which markers along the centre line may cause if the helicopter is forced to a sudden touchdown.
CHAPTER 2

2.1 At the present time, geographical coordinates indicating latitude and longitude are not expressed in terms of the WGS-84 system.
CHAPTER 4

4.2.8 The separation between the take-off climb surface and the approach surface is required to be 90 degrees or more, instead of 150 degrees.

APPENDIX 1

Table 1 to Table 5

Comment on implementation:

For heliports with instrument approach procedures, Appendix 1, Tables 1-5 will be implemented on 23 April 1997, except that obstacles in the circling area and in the outer parts (>3 km) of the approach and take-off areas will have an accuracy of 30 m in latitude/longitude (Table 1) and 5 m in elevation (Table 2). Obstacle data not meeting the requirements of Appendix 1 will be identified with effect from 23 April 1998. For other heliports, the implementation of Appendix 1 is yet to be determined.
CHAPTER 2

2.1 CRC not yet implemented.

2.1.7 Geoid undulation information not available.

2.3

Comment on implementation:

2.1 November 2000.

2.1.7

CHAPTER 3

3.2.7 Frangible objects on some older structures do not meet new standards.

3.3.2 Some helidecks on older structures are of non-standard size.

CHAPTER 4

4.2.15 Some helidecks on older structures do not meet requirements for mobile obstacles.

CHAPTER 5

5.1.1.2 Some helidecks are not equipped with standard wind direction indicators.

5.3 Non-standard lighting systems are installed on some older helidecks.